

**Report 10805A
July 1996**

CR-203621

**GENCORP
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**Meteorological Satellites (METSAT) and
Earth Observing System (EOS)
Advanced Microwave Sounding Unit-A (AMSU-A)
Stress Analysis Report**

**Contract No. NAS 5-32314
CDRL: 113**

Submitted to:

**National Aeronautics and Space Administration
Goddard Space Flight Center
Greenbelt, Maryland 20771**

Submitted by:

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170:8611 INTAMSU-A 96-101
29 August 1996

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Subject: Contract NAS5-32314 (INTEGRATED AMSU-A)
CDRL 113B METSAT Stress Analysis Report, A1 Module
Report No. 10805A July 1996 Final

Attachment: (1) Aerojet Letter RHJ:metop96.doc, dated 31 July 1996

Dear Mrs. Farnan Lockard:

Aerojet submits the subject METSAT Stress Analysis Report, A1 in accordance with contract negotiations.

Should any questions arise concerning this submittal, please contact Mr. R. Hauerwaas at (818) 812-1163 or me at (818) 812-2010.

Sincerely yours,



Cathy Woodington
Contract Manager

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Electronic Systems Plant

TO: B. Kelly DATE: 31 July 1996
FROM: R. Heffner 170-1:8611:x2398
SUBJECT: Action Item Responses to April 1996 Critical Design Review (CDR)
Comments - METSAT/AMSU-A1 Mechanical/Structural Subsystem
COPIES TO: R. Burnes, R. Hauerwaas, B. Morris, P. Patel, File
ENCLOSURE: (1) Revision A to AESP CDRL 113, "Meteorological Satellites (METSAT) and Earth Observing System (EOS) Advanced Microwave Sounding Unit-A (AMSU-A) Stress Analysis Report", A1 Module, Aerojet Report No. 10805, July 1996.
REFERENCE: (1) AESP CDRL 113, "Meteorological Satellites (METSAT) and Earth Observing System (EOS) Advanced Microwave Sounding Unit-A (AMSU-A) Stress Analysis Report", A1 Module, Aerojet Report No. 10805, 1 April 1996.
(2) NASA/GSFC Comments to AMSU-A1 METSAT Stress Analysis Report, (CDRL 113), W. Case, April 11, 1996.
(3) "Critical Design Review for The Advanced Microwave Sounding Unit-A (AMSU-A)", Report 10808, Book 2 of 2, April 1996.

PURPOSE

The purposes of this writing is to respond to the action items of Reference 2 resulting from the Reference 1 CDRL document and the Reference 3 CDR presentation. Additional stress analysis required per Reference 2, on the Aerojet CDRL 113 AMSU-A1 METSAT Stress Analysis Report (Reference 1), is presented in the A revision to the 10805 report. The Report 10805 A revision is a complete document, superseding the 1 April 1996 submittal. The Revision A analysis in Report 10805 is the result of an updated finite element model, therefore all analysis in the Report 10805 A revision is new.

RESPONSES

The action items of Reference 2 pertaining to the METSAT/AMSU-A1 Stress Analysis are responded to in this writing. The action items along with their responses follow.

Action Item: Add to the A1 stress analysis for METSAT (CDRL 113B - A1): (1) Margin of Safety (MS) calculations for the panel flanges (as was done for the EOS loads), (2) A review of the MS's calculated for the sidemount in the areas of the mounting bolts where the calculated stress gradients are shown to be high (in report 10805). One area shows a MS of 0.16 in an area where the model appears that it might not be fine enough to accurately predict the stress gradients shown in the report.

Response: The Reference 1 CDRL report was composed from analysis performed using a NASTRAN finite element model with an error in the sidemount structure in the vicinity of the upper aft spacecraft mounting bolt (in the vicinity of the above mentioned +0.16 MS). This anomaly invalidates the Reference 1 analysis. An entirely new stress analysis is presented in the A revision to Report 10805.

The A revision to Report 10805 also responds to the Reference 2 comments. Analysis on the panel flanges to the METSAT random vibration loads is included in the revised CDRL report. Correction of the modeling anomaly at the sidemount spacecraft upper aft mounting bolt significantly reduces the sidemount large stress gradient reported in Reference 1. The minimum MS of +0.16, reported in Reference 1 at the sidemount (and referred to in the action item), per random vibration 8.8 Grms Z-loading, increases to +1.58 in the corrected Revision A analysis. The finite element model of Revision A is also refined in the sidemount spacecraft upper aft mounting bolt region.

Concerning the panel flanges, without design changes, the minimum MS in the METSAT Revision A analysis would be at the upper aft panel side flanges in the vicinities of the upper aft panel reinforced DC/DC converter upper mounting regions, where negative margins would be seen (MS = -.84, see Figure 1). Thus, a design modification is warranted. Taking into consideration the slightly more severe METOP random vibration 9.66 Grms spectrum (versus the 8.8 Grms METSAT spectrum), and the knowledge that a potential DC/DC converter weight increase is expected (from the present 3.50 maximum specification weight used in the finite element model to as high as 4.50 lb), the flange thicknesses are increased as follows (see Figure 2):

1331642-3 Upper Aft Panel

	Was	Is
Lower flange	.095/.105	.110/.120
Upper flange	.050/.060	.060/.070
Left Side Flange	.050/.060	.145/.155 @ DC/DC Conv Upper Mounting Region .075/.085 Below DC/DC Converter Upper Mounting Region .060/.070 All other locations
Right Side Flange	.050/.060	.145/.155 @ DC/DC Conv Upper Mounting Region .075/.085 Below DC/DC Converter Upper Mounting Region .060/.070 All other locations

With the above flange thickness increases, the minimum MS in the upper aft panel becomes +.36 at the right side flange DC/DC Converter upper mounting region with METSAT 3 σ random vibration loads and factors of safety of 1.25 yield/1.4 ultimate applied (see Figure 2). Projected weight increase with flange thickening is +.05 lb.

The upper front panel lower flange requires a thickness of $\geq .045$ inches to demonstrate, analytically, a positive MS with 3 σ random vibration loads and factors of safety of 1.25 yield/1.4 ultimate applied. Lower flange thickness per drawing 1331352-3 is .050 +/- .010. Therefore, the as-built hardware was inspected and found to have minimum $t = .050$ thick lower flanges in the area of the maximum load. With minimum $t = .050$ lower flanges, the minimum MS is +.34 at the upper aft panel lower flange.

After considering the above upper aft panel design change and the lower front panel as-built dimensions, the remaining flange minimum MS is then seen in the lower motor mount panel upper flange, where a MS of +.06 is noted with METSAT 3 σ random vibration loads and factors of safety of 1.25 yield/1.4 ultimate applied.

The Report 10805 A revision is written similarly to Reference 1. Static design load results are reported first (Tables 1 through 9), followed by the modal effective weight summary (Table 10) random vibration loads (Tables 11 through 14), natural frequencies, transmissibility's, rattlespace (Table 15), mounting bolt hardware stresses (Tables 16 through 26), panel flange bending stresses (Tables 27 through 36), and panel mounting screw stresses due to flange bending loads (Table 37 through 46).

A comparison of margins of safety of the A revision report to the Ref 1 report is as follows:

Margins of Safety Summary

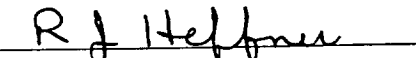
Subject	A Revision Report	Reference 1 Report
Design Loads Cases 1-8 (Table 1)	+1.61	+0.26
Random Vibration 8.8 Grms (Table 11 Rev A, Table 10 Ref 1)	+0.62	+0.16
Rattlespace - Lower Cage Cards (Table 15 Rev A, Table 14 Ref 1)	.037-.005=.032 clearance	.037-.025=.012 clearance
Mounting Screws Tension/Shear (Table 19 Rev A, Table 18 Ref 1)	+0.25	+0.14
Shear Pins Sidemount Shear Tearout (Table 20 Rev A, Table 19 Ref 1)	+0.23	+0.23
Shear Pins Sidemount Bearing (Table 20 Rev A, Table 19 Ref 1)	+0.72	+0.72
Mounting Screws Tension W/O Shear (Table 24 Rev A, Table 21 Ref 1)	+0.13	+0.11
Mounting Screws Member Compression (Table 25 Rev A, Table 24 Ref 1)	+0.76	+0.76
Mounting Screws Sidemount Thread Shear (Table 25 Rev A, Table 24 Ref 1)	+0.71	+0.69

Panel Flange Bending

Upper Front Panel	+0.34	-----
Upper Right Front Support Panel	+0.76	-----
Lower Right Panel	+4.84	-----
Lower Aft Panel	+1.05	-----
Upper Motor Mount Panel	+1.75	-----
Lower Motor Mount Panel	+0.06	-----
Lower Front Panel	+0.33	-----
Upper Aft Panel	+0.36	-----
Upper Right Panel	+0.48	-----
Lower Right Front Support Panel	+0.94	-----

Panel Mounting Screws

Upper Front Panel	+5.6	-----
Upper Right Front Support Panel	+6.1	-----
Lower Right Panel	+13	-----
Lower Aft Panel	+5.3	-----
Upper Motor Mount Panel	+11	-----
Lower Motor Mount Panel	+5.5	-----
Lower Front Panel	+6.0	-----
Upper Aft Panel	+0.32	-----
Upper Right Panel	+7.7	-----
Lower Right Front Support Panel	+22	-----


R. J. Heffner
Mechanical/Structural Analysis


B. T. Kelly, Team Leader
Mechanical/Structural/Thermal Team

Was Configuration

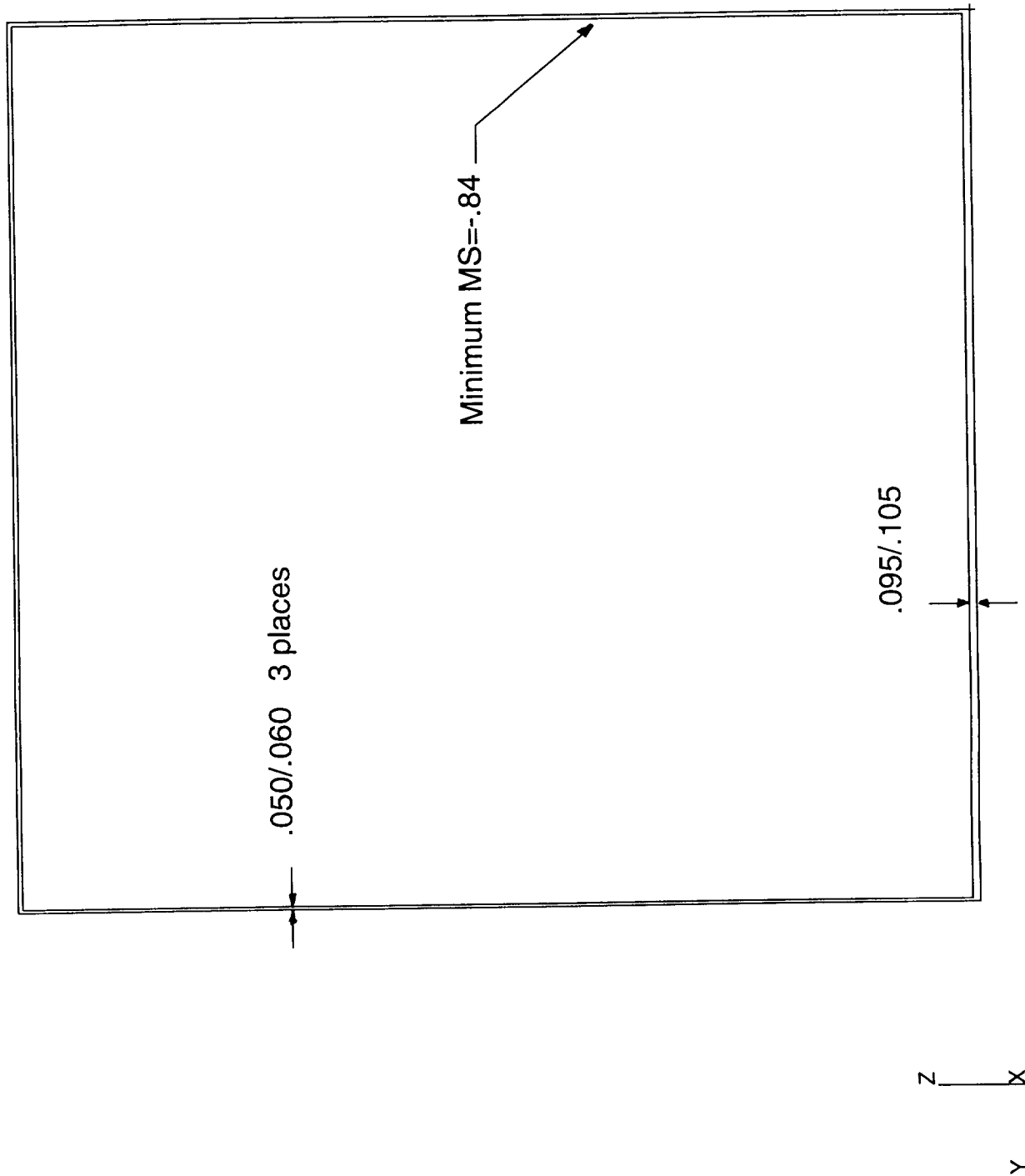


Figure 1 1331642-3 Upper Aft Panel Flanges

Is Configuration

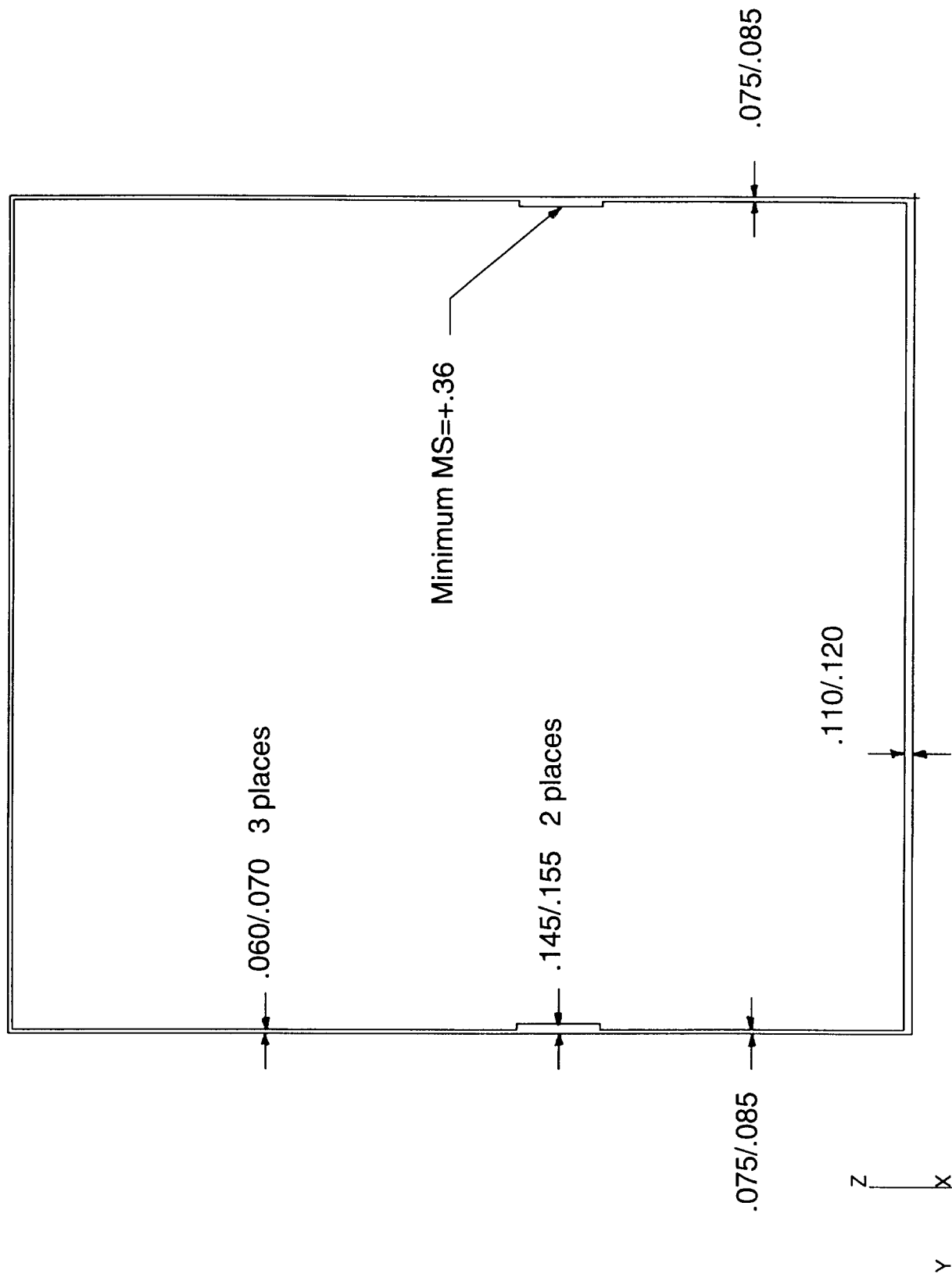


Figure 2 1331642-3 Upper Aft Panel Flanges

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Section 1

INTRODUCTION

This Stress Analysis Report for the Meteorological Satellites Project (METSAT) Advanced Microwave Sounding Units-A (AMSU-A), A1 Module, reports the structural margins of safety and natural frequency predictions for the METSAT design at Critical Design Review (CDR). The report has been prepared in accordance with GSFC S-480-79, Performance Assurance Requirements (PAR) for the Earth Observing System (EOS) & Meteorological Satellites Project (METSAT) Advanced Microwave Sounding Units-A.

1.1 Identification

This is the Stress Analysis Report for the Meteorological Satellites Project (METSAT) Advanced Microwave Sounding Units-A (AMSU-A), Module A1. This report is submitted to fulfill the requirements of Contract NAS 5-32314 CDRL 113, Stress Analysis Report, for the METSAT AMSU-A1 Module. The Stress Analysis Report for the A2 module has been submitted under separate cover.

1.2 Purpose and Objectives

The purpose of this analysis is to show that the METSAT AMSU-A1 Module exhibits positive structural margins of safety when subjected to the METSAT environmental conditions given in Appendix E of the GSFC-S-79 PAR document.

In addition, the dynamic analysis results are used to show that the METSAT AMSU-A1 Module natural frequencies are above 100 hertz (Hz) and that the simplified test and analytical model requirements of GSFC-S-79 PAR for METSAT General Instruments can be applied to the METSAT AMSU-A1 module.

1.3 Document Status and Schedule

This is the submittal of the Stress Analysis Report for the METSAT AMSU-A1 Module after CDR, in July 1996.

Section 2

SUMMARY OF RESULTS AND CONCLUSIONS

A stress analysis of the primary structure of the METSAT AMSU-A1 Module has been performed using the static loads given in Appendix E, Section 1.2, of GSFC 422-S-79 PAR. Combined with factors of safety (1.4 ultimate, 1.25 yield), the static loads minimum margins of safety summary is given in Table I. The METSAT AMSU-A1 Module exhibits positive margins of safety (MS) in all parts of the structure for static design loads. Minimum MS is +1.61 and is in the sidemount at the -X upper mounting bolt region for case 4 loads of $G_y = -20.6$ $G_z = 19.6$ (in Aerojet METSAT coordinate system). Figures 5 and 6 show the stress contour plot of the sidemount critical region, with maximum *von Mises* stress of 10,727 psi. Figures 7 through 14 show the full model contour plots, for load cases 1 through 4, shell surfaces 1 and 2. Figures 15 through 22 show full model beam stress maximums and minimums, for load cases 1 through 4. Maximum beam stress is 11,596 psi, in the power control/monitor standoffs for load case 2 ($G_y = -20.6$ $G_z = -19.6$), producing a MS of +2.04. Tables II through IX present summarized MS and stresses for individual component hardware for load cases 1 through 4.

The first natural frequency of the METSAT AMSU-A1 Module is predicted at 101.8 Hz. As shown in Figure 23, the top panel canopy over the sidemount flexes to produce the first mode. This mode shape is local to the top panel, with no mass participation elsewhere in the model, as demonstrated in Table X, the modal effective weight and participation factor summary for the first 16 modes.

Figures 23 through 34 identify the first twelve non-rigid body modes. Modes 2 (109.0 Hz), 3 (109.5 Hz), 4 (109.7 Hz), and 5 (109.7 Hz) are all the result of the lower card cage cards flexing in the lower card cage (see Figures 24 through 27). Of these four modes, only mode 4 possesses any significant modal effective weight (X-axis 1.2 percent), with the card cage and power control/monitor moving along with the lower cards.

Mode 6 (120.4 Hz) shows significant power control/monitor circuit board movement, along with the upper aft, top, and radiator panels distortions. Both reflectors, the lower baseplate, and lower shelf are also agitated (Figure 28). The significant modal effective weight for mode 6 is along the X-axis, with 8.5 percent of the model reacting. Mode 6 is a rocking of the entire structure about the Aerojet METSAT yz-plane. Mode 7 (121.4 Hz) shows pronounced rotation along the X-axis by the reflectors, with more movement in the upper reflector (Figure 29). Mode 8 (121.9 Hz) is similar to mode 7, but with the predominant movement in lower reflector (Figure 30). Mode 9 (122.0 Hz) movement in also an X-axis rotation in the lower reflector (Figure 31).

The 10th and 11th modes at 130.9 Hz and 131.8 Hz, respectively, are the first extensive modal mass participation modes with 27.7 percent and 28.8 percent Z-axis participation (Figures 32 and 33). Both modes exhibit an overall rocking of the structure about the X-axis with overall Z-axis translation. The 12th mode (138.4 Hz) with 6.2 percent X-axis modal mass participation shows considerable Z-axis rotation at the lower reflector (Figure 34).

The analysis demonstrates a slight margin over the 100 Hz threshold first natural frequency.

A stress analysis of the primary structure of the METSAT AMSU-A1 Module has been performed using the random vibration spectrum given in Appendix E, Section 1.3, of the GSFC 422-S-79 PAR. An extensive random vibration stress spreadsheet is included in Appendix D of this report. Q levels resulting from application of the qualification level random vibration spectrum (8.8 grms) with critical damping of 7 percent are tabulated for each component for each loading condition. Combined with factors of safety of 1.4 ultimate, 1.25 yield, the random vibration minimum margins of safety summary is given in Table XI. RMS stress levels are taken to the 3σ level ($3.0 \times$ stress at 8.8 grms load), multiplied by factors of safety (1.25 yield, 1.4 ultimate), and compared to material yield and ultimate strengths. All positive margins

result from this task, with the highest loaded region being the power control/monitor standoffs with an MS of +0.62. Minimum margins of safety are given for each of the three individual loading directions in Tables XII through XIV.

Transmissibility's of various regions of the NASTRAN model are identified in the Transmissibility Summary Table of Appendix E. Q levels resulting from application of the qualification level random vibration spectrum (8.8 grms) with critical damping of 7 percent are tabulated for each component for each loading condition. Large Q-values (greater than 4) are found at the power control/monitor bracket (Z-load, Y-response $Q = 8.6$, X-load, Y-response $Q = 7.8$, and Y-load, Y-response $Q = 4.3$), the upper right front support panel (Y-load, Y-response $Q = 8.0$), the upper reflector (X-load, Z-response $Q = 8.0$), the lower reflector (X-load, X-response $Q = 7.0$), the left panel (Y-load, Y-response $Q = 5.4$), the lower warmload (Y-load, Y-response $Q = 4.4$), the top panel (X-load, X-response $Q = 4.0$), and the radiator panel (X-load, X-response $Q = 4.0$). This information agrees fairly well with the NOAA AMSU-A1 test data used to predict the applicable damping. Based on NOAA testing, a maximum Q of 7 was measured at the reflectors. Note that the power control/monitor, with maximum Q of 8.6, is hardware new to the AMSU design in METSAT/EOS. Plots of PSD input spectrum versus frequency and response versus frequency are included for the higher loaded locations (see Figures 35 through 46 for X-axis loading, Figures 47 through 55 for Y-axis loading, and Figures 56 through 60 for Z-axis loading).

The rattlespace of the circuit cards of the upper and lower card cages is investigated based on the qualification level random vibration spectrum (8.8 grms). As shown in Table XV, the more severe condition occurs at the lower card cage where minimum card spacing is 0.037 inch and calculated maximum displacement is 0.005 inch at the 3σ level (3.0 x displacement at 8.8 grms load). At the upper card cage, minimum spacing is 0.137 inch, with maximum 3σ displacement of 0.002 inch. There will be no contact of adjacent circuit cards in either card cage. Appendix E contains rattlespace deflections in the summary sheets.

Mounting bolt load results for the A-286 #10-32UNRF-3A screws identify an overall minimum margin of safety of +0.13 (Table XXIV). The condition of bolt tensile load only (with all shear reacted at the shear pins) is the higher loaded case, for each of design load case considered. To the preload (1315 lb for $T = 50$ in-lb), the external load is added, considering the bolted joint stiffness. Factors of safety of 1.4 ultimate, 1.25 yield are applied to the external load but not the preload. Additionally, interactive tension plus shear stress (with shear reacted at both mounting bolts and shear pins), bearing and shear tearout in the sidemount at the 2 shear pins (all shear at shear pins), joint member compression, thread shear, and sidemount bearing and shear tearout at the 16 mounting bolts, all identify no margins less than +0.23 (see Tables XVI through XXVI).

The panel flange bending stresses, summarized in Tables XXVII through XXXVI, show all positive margins of safety. Loading condition is the 8.8 grms random vibration 3σ loading with factors of safety 1.25 yield, 1.4 ultimate applied. The upper aft panel side flanges, when thickened locally at the stiffened panel DC/DC converter upper mount, show a minimum MS = +0.36 with minimum $t = 0.145$ local flanges. The upper front panel lower flange with minimum as-built lower flange thickness of $t = 0.050$ for random vibration loads, shows a minimum MS of +0.34. Minimum calculated MS, at the lower motor mount panel upper flange, is +0.06 for random vibration 3σ loading with factors of safety 1.25 yield, 1.4 ultimate applied.

Stresses in the panel flanges' attachment screws are highlighted in Tables XXVIII through XXXVII, with lowest margin at the upper aft panel left (+Y) side flange at the locally stiffened region where the stiffened panel DC/DC converter upper mount intersects. Minimum MS = +0.32 for the NAS1352N06 screws.

Based on the large margins of safety for static and random vibration loads application, the upper aft panel side flanges redesign, and the upper front panel lower flange inspected thicknesses, the AMSU-A1 Module will maintain structural integrity in the METSAT launch environment.

Section 3

REFERENCE DOCUMENTS

The following documents were used in the preparation of this Report:

SPECIFICATIONS

GSFS-S-480-79
October 10, 1994

Performance Assurance Requirements
for the Earth Observing System (EOS)
& Meteorological Satellites Project (METSAT)
Advanced Microwave Sounding Units-A

GSFC 422-11-12-01A
January 1994

General Interface Requirements
Document (GIRD) for EOS Common
Spacecraft/Instruments EOS PM Project

Aerojet

Report 10381
June 1995

Earth Observing System (EOS)/
Advanced Microwave Sounding Unit-A
(AMSU-A) Stress Analysis Report,
A1 Module

Report 10381A
Addendum 1
July 1996

Earth Observing System (EOS)/
Advanced Microwave Sounding Unit-A
(AMSU-A) Stress Analysis Report,
A1 Module

Report 10738A
July 1996

METSAT Advanced Microwave
Sounding Unit -A1 (AMSU-A1)
Structural Mathematical Model

Section 4

METHOD OF ANALYSIS

4.1 Finite Element Model

Figures 1 through 4 display the NASTRAN finite element model of the METSAT AMSU-A1 structure. The basic coordinate system used in the METSAT analysis is identical to that used in the EOS analysis (Ref. Aerojet Report 10381). This orientation differs from the METSAT PAR document that utilizes the spacecraft coordinate system identified on GE ATN Spacecraft Assembly Drawing 3278200. A comparison of systems follows:

Aerojet METSAT Coordinate Sys.	GSFC-S-79 PAR Coordinate Sys.
+X Axis	-Y Axis
+Y Axis	-Z Axis
+Z Axis	+X Axis

The METSAT AMSU-A1 finite element model is a modification of the EOS AMSU-A1 finite element model. The METSAT AMSU-A1 Module is mounted to the spacecraft through a sidemount bracket. The EOS AMSU-A1 Module is mounted from the lower baseplate. The design of the primary structure is essentially the same for the two units except for the following:

- a. **Lower Baseplate** Different designs and drawings.
- b. **Sidemount** Only used on METSAT.
- c. **Top Panel** Different designs and drawings.
- d. **Left Panel** Different designs and drawings.
- e. **Radiator Panel** Only used on METSAT.
- f. **Signal Processor** METSAT has two Cards in Upper Card Cage. EOS Upper Card Cage is empty. METSAT has four different Cards in Lower Card Cage.
- g. **Power Control/Monitor** Different designed enclosure and connectors.
- h. **Insulation and Mirror Tile representations** Modified.

The NASTRAN finite element model prepared for EOS AMSU-A1 and shown in Aerojet Report 10381 is modified for the METSAT AMSU-A1 design. In the current METSAT effort, a more refined NASTRAN model has been developed to more accurately represent the METSAT structure. The model statistics are:

9307	Grids
1018	Bar/Beam Elements
8288	Rectangular Plate Elements
688	Triangular Plate Elements
385	Point Mass Elements

The total mass of the model is 119.4 pounds. Mass properties, in the Aerojet METSAT coordinate system axes, at the center of gravity (cg), are shown below. The cg is measured relative to the sidemount lower corner on the spacecraft side by the lower reflector (GRID 7508).

Mass	119.4 lb.
Center of Gravity	
x	15.218 in
y	8.831
z	9.926
Moment of Inertia (@ cg)	
Ixx	7351 lb-in ²
Ixy	32
Ixz	1803
Iyx	32
Iyy	12587
Iyz	145
Izx	1803
Izy	145
Izz	9114

Appendix A, in Figures A1 through A4, shows the model assembly. Figures A1, A2, and A3 show different views of the assembled structure, while Figure A4 is a section view of the model, showing the components modeled in the interior (i.e. shelves, card cages, warmload structures). A detailed description of the METSAT NASTRAN finite element model is found in Report 10738A, where elements and grids are highlighted in the piece part models.

4.2 Boundary Conditions

The model is constrained along the three orthogonal axes on the sidemount at each of the 16 spacecraft mounting bolt locations and along the two transverse axes (X and Z in Figure 61) at the two dowel pin locations. The constraints are imposed using NASTRAN single point constraint cards.

4.3 Load Application

Eight combinations of design loads are evaluated. Referencing Appendix E, Section 1.2, PAR individual loads, Load Cases 1 through 8 in the Aerojet METSAT Coordinate System are:

<u>Load Case</u>	<u>Loading</u>
1	Gy= -20.6, Gx=-21.2
2	Gy= -20.6, Gx= 21.2
3	Gy= -20.6, Gz= -19.6
4	Gy= -20.6, Gz= 19.6
5	Gy= 20.6, Gx=-21.2
6	Gy= 20.6, Gx= 21.2
7	Gy= 20.6, Gz= -19.6
8	Gy= 20.6, Gz= 19.6

With the exception of algebraic sign, Load Cases 1, 2, 3, and 4 are identical to Load Cases 6, 5, 8, and 7, respectively. Thus only Cases 1 through 4 are evaluated.

For the random vibration loadings (Ref. Appendix E, Section 1.3, PAR), the qualification level (8.8 grms) PSD spectrum is applied along the three orthogonal axes. Three load cases are run, each load case representing the random spectrum in one of the axes shown in Figure 1.

The random vibration analysis is performed using the NASTRAN code with a type G structural damping value of 0.14. The critical damping percentage is 1/2 of this value, or 7 percent. For large Q , the amplification or quality factor, Q , is approximated by

$$Q = 1/[2(c/c_c)]$$

where: c/c_c = fraction of critical damping

With c/c_c equal to 0.07, an amplification (Q), of 7.14 is determined. A $Q = 7.14$ corresponds to the largest Q found in test data of the NOAA AMSU-A1 unit. To the stresses derived from the NASTRAN random vibration stress output, a 3.0 factor is applied to produce 3σ values, plus an additional factor of safety (FS) of 1.25 on yield or 1.4 on ultimate is also employed.

Section 5

RESULTS

The finite element model is primarily made up of beam (NASTRAN CBAR and CBEAM) elements, shell (NASTRAN CQUAD4 and CTRIA3) elements, and point (NASTRAN CONM2) elements. The maximum *von Mises* stresses for plate elements and the maximum and minimum principal stresses for beam elements are used for margins of safety calculations. For the static loads analyses, the output from each evaluated load case (cases 1 through 4) is reviewed, with maximum stress contour plots drawn up based on the structural component for the shell elements, and maximum/minimum stress plots constructed for the beam elements. The static load cases maximum contour and maximum stress plots are included in Appendix C.

The stresses shown in Tables II through IX are summarized results of the structural component contour and stress plots. The stresses are multiplied by factor of safety 1.25 to calculate the yield margins of safety and by factor of safety 1.4 to calculate the ultimate margins of safety. Table I shows the overall highest stresses and lowest margins of Tables II through VIII for the static design loads.

The first 12 non-rigid body natural frequency mode shapes are depicted in Figures 23 through 34, from analysis using the large mass method in the finite element model. To demonstrate the mathematical soundness of the NASTRAN model, the model is subjected to a stiffness-equilibrium check similar to the EOS requirement of GSFC 422-11-12-01, Paragraph 11.1.4.i, Deliverable Model Validity Check. The METSAT model is shown to satisfy this check, with the normalized KFFRN matrix empty (see Appendix B).

For the random vibration loading, the output from each orthogonal axis load case is reviewed. The maximum principal stresses for plate elements and beam elements are used for margins of safety calculations. For plate elements, the normal and shear stresses are assumed in phase in calculating the principal stresses. Likewise, for bar elements, the axial and bending components are also considered to be in phase, and are thus algebraically added.

The stresses shown in Tables XII through XIV are statistical 3σ stress levels, and are the results of the random vibration spectrum (qualification level, 8.8 grms) orthogonal loading cases, with 3.0 multiplication factors employed. These 3σ stresses are then multiplied by 1.25 to calculate the yield margins of safety and by 1.4 to calculate the ultimate margins of safety. Tables XII through XIV present the worst case stresses for the groups of elements representing the structural components of the primary structure. The tables give the results for each of the three load cases. Table XI summarizes the minimum margins for the three cases. The worst axis of load application depends on the orientation of the structural item. Appendix D contains an extensive random vibration stress table, identifying the stress levels for the elements evaluated, for each orthogonal load direction.

The transmissibility's (Q levels) shown in Appendix E are the result of the output level grms acceleration divided by the input level grms acceleration (8.8 grms for each qualification level), for critical damping at 7 percent. Output shown in Appendix E consists of the root mean square (rms) response (rms in units of in/sec^2 , the grms response (grms in units of g), and the transmissibility's Q (response grms divided by input grms). Plots of the input spectra are shown in Figures 35, 47, and 56 for the X, Y, and Z loadings, respectively, identifying 8.84 grms input levels. Plots response (in units of $(\text{in/sec}^2)^2/\text{Hz}$) versus frequency are included for the higher loaded locations (see Figures 36 through 46 for X-axis loading, Figures 48 through 55 for Y-axis loading, and Figures 57 through 60 for Z-axis loading).

The rattlespace calculations are tabulated in Appendix E for both upper and lower card cage circuit cards. Circuit card displacements based on 3σ random vibration loadings for each orthogonal axis are

listed. The use of MPC elements between adjacent modeled circuit cards in the NASTRAN finite element model is made in determining the remaining gaps between cards.

Mounting bolt hand calculated stress analysis is included for the sixteen #10-32UNF mounting screws (and two 0.375 diameter shear pins). Interactive tension plus shear stress checks are performed on the 16 screws for the 8 design load cases, summarized in Tables XVI through XIX, with resulting minimum margin +0.25 (Table XIX). Bearing and shear tearout calculations in the sidemount at the 2 shear pins, assuming all shear is taken out only at the 2 pins, shows a minimum margin of +0.23 (Table XX). Mounting bolt tensile stresses, based on the same shear reaction assumption, show a +0.13 minimum margin of safety (see Table XXIV of Tables XXI through XXIV). Additional stress calculations for joint member compression, thread shear, and sidemount bearing and shear tearout at the 16 mounting bolts, identify no margins less than +0.71 (Tables XXV and XXVI).

The flange bending stresses, summarized in Tables XXVII through XXXVI, show all positive margins of safety. The upper aft panel side flanges are thickened locally to a minimum $t = 0.145$ at the locations where the panel stiffened DC/DC converter $t = 0.250$ upper mount region intersects. With this action, positive MS (+0.36) for the random vibration 3σ loading with factors of safety 1.25 yield 1.4 ultimate applied are found. The upper front panel lower flange has a required minimum $t = 0.045$ for the random vibration 3σ loading with factors of safety 1.25 yield, 1.4 ultimate applied. With lower flange specified drawing thickness of 0.050 ± 0.010 , the as-built hardware has been inspected and shows a minimum $t = 0.050$ lower flange thickness. Thus a positive MS (+0.15) is determined. Minimum calculated MS in the panel flanges is at the lower motor mount panel upper flange with a MS = +0.06 for the random vibration 3σ loading with factors of safety 1.25 yield, 1.4 ultimate applied.

Stresses in the panel flanges' attachment screws are highlighted in Tables XXVII through XLVI, with lowest margin at the upper aft panel left (+Y) side flange at the locally stiffened region where the stiffened panel DC/DC converter upper mount intersects. Minimum MS = +0.32 for the NAS1352N06 screws.

5.1 AMSU-A1 METSAT 1331552 Sidemount Mounting Bolt/Pin Stresses per Static Loads

<u>Aerojet METSAT</u> <u>Coordinate Sys.</u>	<u>GSFC-S-79 PAR</u> <u>Coordinate Sys.</u>
+X Axis	-Y Axis
+Y Axis	-Z Axis
+Z Axis	+X Axis

Eight combinations of design loads are evaluated. Referencing Appendix E, Section 1.2, PAR individual loads, Load Cases 1 through 8 in the Aerojet METSAT Coordinate System are:

<u>Load Case</u>	<u>Loading</u>
1	Gy= -20.6, Gx= -21.2
2	Gy= -20.6, Gx= 21.2
3	Gy= -20.6, Gz= -19.6
4	Gy= -20.6, Gz= -19.6
5	Gy= 20.6, Gx= -21.2
6	Gy= 20.6, Gx= 21.2
7	Gy= 20.6, Gz= -19.6
8	Gy= 20.6, Gz= 19.6

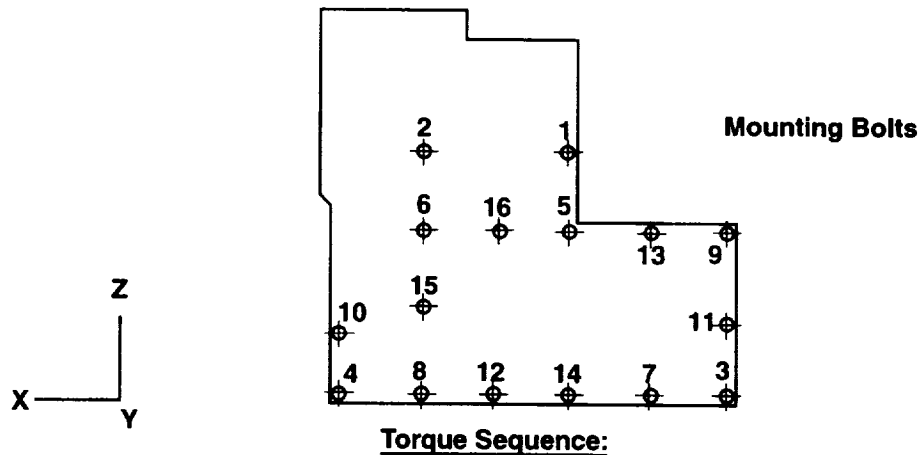
With the exception of algebraic sign, load Cases 1,2,3, and 4 are identical to Load Cases 6, 5, 8, and 7, respectively. Thus only Cases 1 through 4 are evaluated.

FACTOR OF SAFETY

1.4 ULTIMATE

ASSUMPTIONS

1. Stresses in bolts derived from NASTRAN SPC forces based on above load cases plus bolt preload (50 in-lb as shown in drawing 1333964).
2. There are 16 bolts plus two dowel pins shown in the NASTRAN sketch. A list of NASTRAN grid numbers is included. Grids 7855 and 10724 are the shear pins.
3. Bolt grids react tension/compression (NASTRAN T2) plus shear (NASTRAN T1, T3).



Torque screws to 20 ± 2 in lbs in the order shown
 Torque screws to 30 ± 2 in lbs in the order shown
 Torque screws to 40 ± 2 in lbs in the order shown
 Torque screws to 50 ± 2 in lbs in the order shown

396-3116M

4. Pin grids react only shear (NASTRAN T1, T3).
5. Interactive tension plus shear considered in accordance with MIL-HDBK-5, Figure 1.5.3.5, $R_s^3 + R_t^2 = 1$ curve.
6. Allowables for A286 screws and pins based on NAS1351N3-32 with $F_{TU} = 160000$ psi and tensile area of $A_s = 0.01999$ in² (.190-32UNRF-3A). Shear load allowable based on $F_{su} = 0.6 \times F_{TU}$ and area per minor diameter, 0.1517 in (0.01807 in²).

Allowable

Tension	3200 lb	.190-32UNRF-3A
Shear	1735 lb	

7. The factor of safety, 1.4, is applied only to applied loads, not to preload.
8. Preload at 50 in-lb torque, F_i

$$F_i = \frac{T}{.2d} = \frac{50}{(.2)(.190)} = 1315 \text{ lb}$$

Bolt loads based on the four evaluated load cases (NASTRAN) follow.

Detailed calculations are performed for the most severe case, Grid 8313, Case 4.

$$G_y = -20.6, G_z = 19.6$$

$$F_t = T_2 = 1415 \text{ lb}$$

$$F_s = \sqrt{T_1^2 + T_3^2} = \sqrt{61^2 + 66^2} = 90 \text{ lb}$$

$$F_i = 1315 \text{ lb}$$

Total bolt tensile load, F_b , is a function of preload, applied tension, and joint stiffness.

$$F_b = F_i + \frac{k_b F_t}{k_b + k_m} \quad (\text{Shigley, "Mechanical Engineering Design," 4th Edition})$$

$$k_b = \frac{E_b A}{\ell} \quad E_b = 28 \times 10^6 \text{ psi}$$

$$A = \pi/4 (.190)^2 = 0.2835 \text{ in}^2$$

$$\ell = \text{Grip of screw}$$

$$= 983743 \text{ lb/in}$$

$$0.437 + 0.31 + 0.06 = 0.807 \text{ in}$$

$$k_m = \frac{\pi E_m d}{2L_n \frac{5(\ell + d/2)}{\ell + 2.5d}} \quad \text{for G-10 spacer only with fibers } \perp \text{ thickness use only properties}$$

$$E_m = 2,500,000 \text{ psi}$$

$$\ell = 0.807 \text{ in}$$

$$d = 0.190 \text{ in}$$

$$= 593,160 \text{ lb/in}$$

Therefore,

$$\frac{k_b}{k_b + k_m} = 0.623845$$

Apply the applied load directly to the preload for GRID 8313 Case 4, applying FS = 1.4, and stiffness factor

$$f_t = 1315 + 1.4(0.624)(1415) = 2551 \text{ lb}$$

$$f_s = 1.4 (90) = 126 \text{ lb}$$

$$r_t = \frac{f_t}{F_{tu}} = \frac{2551}{3200} = 0.797$$

$$r_s = \frac{f_s}{F_{su}} = \frac{126}{1735} = 0.0726$$

POINT ID.	TYPE	FORCES OF SINGLE-POINT CONSTRAINT									
		T1	T2	T3	R1	R2	R3				
7240	G	-1.150475E+01	-2.308063E+02	-1.932467E+02	0.0	0.0	0.0				
7255	G	1.704103E+02	-6.667104E+01	-4.122255E+01	0.0	0.0	0.0				
7446	G	7.653661E+01	4.904039E+00	-1.094048E+01	0.0	0.0	0.0				
7462	G	3.402306E+01	2.066600E+02	1.057857E+02	0.0	0.0	0.0				
7497	G	8.737622E+01	5.608583E+01	3.483112E+01	0.0	0.0	0.0				
7507	G	1.515906E+02	3.129030E+02	1.229639E+02	0.0	0.0	0.0				
7555	G	3.094059E+01	-7.092370E+01	-4.221728E+01	0.0	0.0	0.0				
7745	G	4.916122E+01	2.460296E+01	-3.874358E+01	0.0	0.0	0.0				
7840	G	1.711216E+02	6.325173E+01	-4.153853E+01	0.0	0.0	0.0				
7855	G	2.407013E+02	0.0	-8.143425E+00	0.0	0.0	0.0				
7933	G	6.933346E+01	4.503124E+01	-1.269742E+01	0.0	0.0	0.0				
8313	G	4.291951E+02	4.078265E+02	1.436252E+02	0.0	0.0	0.0				
9576	G	2.573442E+02	4.393478E+02	-5.588730E+01	0.0	0.0	0.0				
9834	G	4.878310E+02	4.620385E+02	2.127497E+01	0.0	0.0	0.0				
10271	G	5.440712E+01	4.240285E+02	1.898773E+02	0.0	0.0	0.0				
10703	G	1.815681E+02	3.277552E+02	-9.097468E+01	0.0	-2.798942E-20	0.0				
10724	G	1.480344E+01	0.0	9.045010E+00	0.0	-7.231528E-20	0.0				
10748	G	3.571492E+01	5.290018E+01	-9.179119E+01	0.0	1.730802E-20	0.0				

Forces of Single-Point Constraint (Sheet 1 of 4)

SUBCASE 2

POINT ID.	TYPE	FORCES OF SINGLE-POINT CONSTRAINT									
		T1	T2	T3	R1	R2	R3				
7240	G	2.160906E+01	4.324717E+02	2.576815E+02	0.0	0.0	0.0				
7255	G	-2.261883E+02	1.942011E+02	-1.476231E+01	0.0	0.0	0.0				
7446	G	-1.846833E+02	9.340278E+00	-2.008085E+00	0.0	0.0	0.0				
7462	G	-1.585676E+02	-5.606078E+01	-1.344829E+02	0.0	0.0	0.0				
7497	G	-1.419548E+02	-2.624223E+01	-4.006616E+01	0.0	0.0	0.0				
7507	G	-1.290168E+02	-2.449961E+02	-7.712297E+01	0.0	0.0	0.0				
7555	G	-2.378767E+01	5.094666E+02	4.958028E+01	0.0	0.0	0.0				
7745	G	-5.408296E+01	8.851603E+01	-1.069770E+02	0.0	0.0	0.0				
7840	G	7.241022E+01	6.070613E+02	1.212534E+02	0.0	0.0	0.0				
7855	G	-1.675526E+02	0.0	2.793463E+01	0.0	0.0	0.0				
7933	G	-7.426967E+01	2.251427E+01	8.110397E+00	0.0	0.0	0.0				
8313	G	-4.209350E+02	9.004758E+02	5.747405E+01	0.0	0.0	0.0				
9576	G	-1.934309E+02	1.146213E+02	3.604144E+01	0.0	0.0	0.0				
9834	G	-3.898563E+02	-1.767841E+02	9.045721E+00	0.0	0.0	0.0				
10271	G	-2.426548E+02	2.165265E+01	-1.288466E+02	0.0	0.0	0.0				
10703	G	-1.237897E+02	-1.905697E+01	-3.162248E+01	0.0	9.938755E-21	0.0				
10724	G	-8.318752E+01	0.0	-2.101788E+01	0.0	8.653684E-20	0.0				
10748	G	-1.061557E+01	8.175165E+01	-1.021499E+01	0.0	9.729974E-21	0.0				

Forces of Single-Point Constraint (Sheet 2 of 4)

SUBCASE 3

FORCES OF SINGLE-POINT CONSTRAINT

POINT ID.	TYPE	T1	T2	T3	R1	R2	R3
7240	G	1.496918E+02	4.248118E+02	3.083591E+02	0.0	0.0	0.0
7255	G	-1.714283E+02	3.691648E+02	1.301964E+02	0.0	0.0	0.0
7446	G	1.334422E+00	5.559219E+01	8.157663E+01	0.0	0.0	0.0
7462	G	-1.650544E+02	4.968589E+02	2.779029E+02	0.0	0.0	0.0
7497	G	3.208227E-01	9.325952E+01	7.492542E+01	0.0	0.0	0.0
7507	G	1.122518E+02	3.222846E+02	1.554500E+02	0.0	0.0	0.0
7555	G	1.455312E+01	2.296248E+02	9.036020E+01	0.0	0.0	0.0
7745	G	-3.447092E+01	1.439095E+02	7.911532E+01	0.0	0.0	0.0
7840	G	6.165983E+01	2.808315E+02	2.730477E+02	0.0	0.0	0.0
7855	G	9.443030E+01	0.0	2.352559E+02	0.0	0.0	0.0
7933	G	-2.566635E+01	2.428530E+01	9.705036E+00	0.0	0.0	0.0
8313	G	-5.293840E+01	-1.067254E+02	1.350590E+02	0.0	0.0	0.0
9576	G	-1.014966E+02	-1.723588E+02	1.049865E+02	0.0	0.0	0.0
9834	G	1.460581E+02	1.853117E+02	1.734016E+01	0.0	0.0	0.0
10271	G	-2.787283E+01	2.178517E+02	2.355006E+02	0.0	0.0	0.0
10703	G	-1.266849E+00	-8.053241E-01	4.574073E+01	0.0	-7.839818E-21	0.0
10724	G	2.745042E+01	0.0	2.643910E+00	0.0	-5.137628E-20	0.0
10748	G	-2.755592E+01	-1.049623E+02	8.240316E+01	0.0	-1.324761E-20	0.0

Forces of Single-Point Constraint (Sheet 3 of 4)

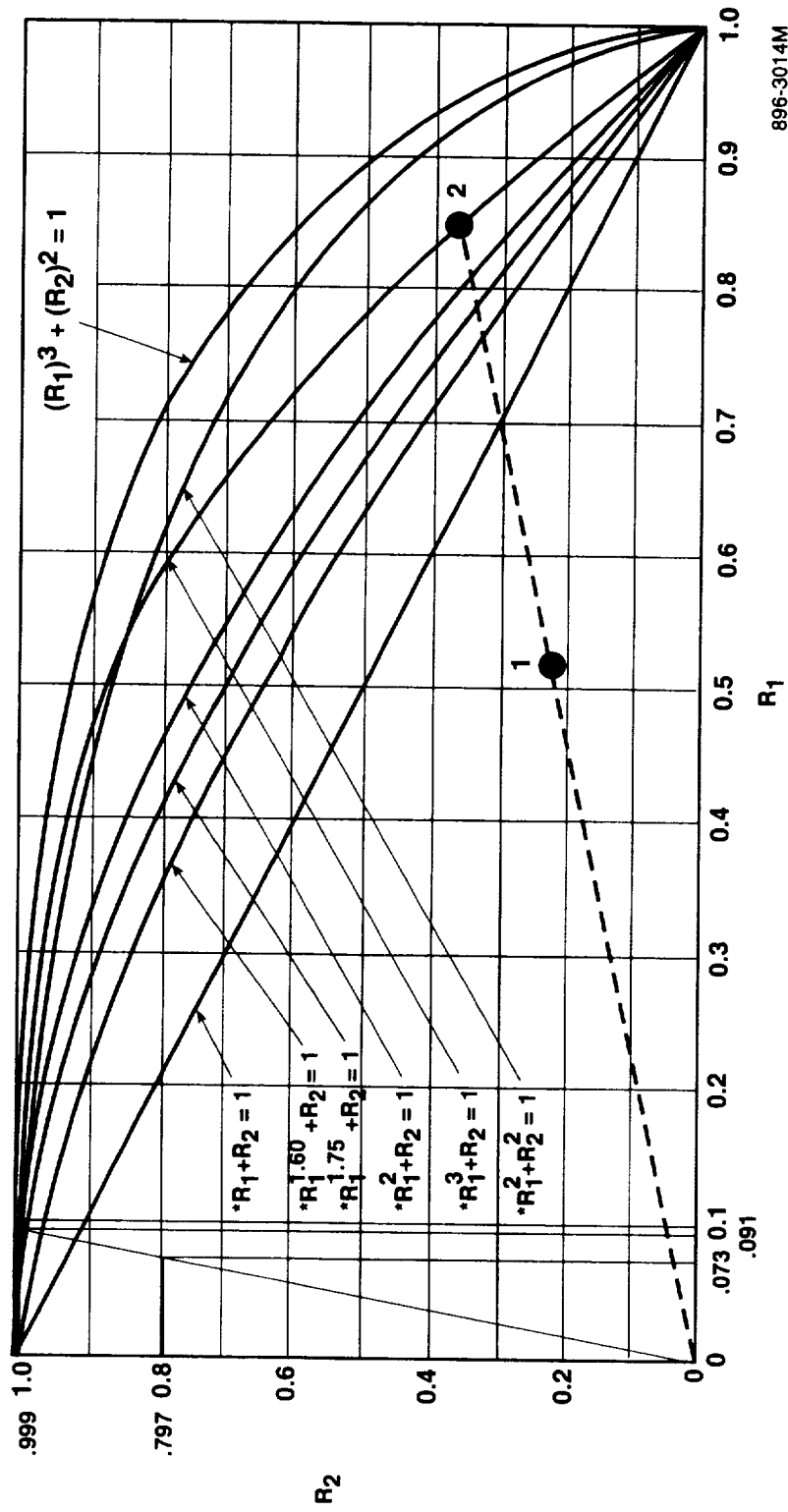
SUBCASE 4

FORCES OF SINGLE-POINT CONSTRAINT

POINT ID.	TYPE	T1	T2	T3	R1	R2	R3
7240	G	-1.395875E+02	-2.231464E+02	-2.439243E+02	0.0	0.0	0.0
7255	G	1.156502E+02	-2.416347E+02	-1.861812E+02	0.0	0.0	0.0
7446	G	-1.094811E+02	-4.134787E+01	-9.452520E+01	0.0	0.0	0.0
7462	G	4.050983E+01	-3.462596E+02	-3.066002E+02	0.0	0.0	0.0
7497	G	-5.489940E+01	-6.341592E+01	-8.016047E+01	0.0	0.0	0.0
7507	G	-8.967787E+01	-2.543777E+02	-1.096091E+02	0.0	0.0	0.0
7555	G	-7.400199E+00	2.089180E+02	-8.299719E+01	0.0	0.0	0.0
7745	G	2.954918E+01	-3.078848E+01	-2.248359E+02	0.0	0.0	0.0
7840	G	1.818719E+02	3.894816E+02	-1.933329E+02	0.0	0.0	0.0
7855	G	-2.128159E+01	0.0	-2.154647E+02	0.0	0.0	0.0
7933	G	2.073013E+01	4.326021E+01	-1.429206E+01	0.0	0.0	0.0
8313	G	6.119852E+01	1.415028E+03	6.604028E+01	0.0	0.0	0.0
9576	G	1.654100E+02	7.263279E+02	-1.248324E+02	0.0	0.0	0.0
9834	G	-4.808335E+01	9.994270E+01	1.298054E+01	0.0	0.0	0.0
10271	G	-1.603748E+02	2.278294E+02	-1.744700E+02	0.0	0.0	0.0
10703	G	5.904524E+01	3.095035E+02	-1.683379E+02	0.0	-1.021085E-20	0.0
10724	G	-9.583450E+01	0.0	-1.461678E+01	0.0	6.559783E-20	0.0
10748	G	5.265528E+01	2.396142E+02	-1.844093E+02	0.0	4.028561E-20	0.0

Forces of Single-Point Constraint (Sheet 4 of 4)

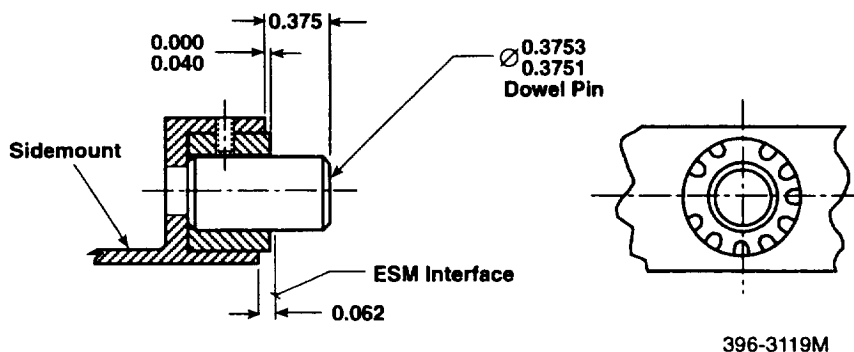
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1 June 1987



Typical Interaction Curves for Combined Loading Conditions

Shear Pins Shear Tearout

As shown in Drawing 1333964, Sheet 2, View D, and the 1331552-1 sidemount with the 1333395-2 and -3 spacers.



Shear tearout of the sidemount and bearing on the sidemount can be determined. The 133395-2 and -3 T1-6AL-4V spacers are scalloped along the OD, thus reducing the bearing area to the 6061-T6 sidemount.

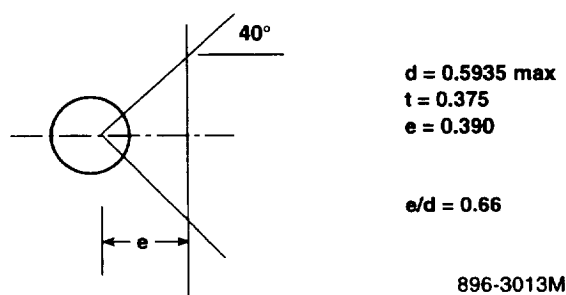
Shear Tearout of Aluminum Sidemount

Based on Case 4, GRID 7855 pin, from the NASTRAN solution with all shear reacted at the two shear pins (no SPC restraint in T1 or T3 for the 16 screws).

Max shear load

$$F = \sqrt{(111)^2 + (1913)^2} = 1916 \text{ lb}$$

Shear tearout of the aluminum sidemount for dimensions as noted, per Bruhn, "Analysis and Design of Airplane Structures," 1973, pages D.1.4-D.1.5.



$$AS = (e - \frac{d}{2} \cos 40^\circ)(t)(2) = 0.122 \text{ in}^2$$

Based on an ultimate load, with FS=1.4

$$T = \frac{1.4(1916)}{0.122} = 21,986 \text{ psi}$$

Shear allowable for sidemount, 6061-T6

$$F_{su} = 27,000 \text{ psi}$$

$$MS = \frac{27,000}{21,986} - 1 = +0.23 \text{ sidemount}$$

See Table XX for shear tearout MS summary.

Bearing of 1333395-2 Bushing onto Sidemount

With F = 1916 lb load

Bearing area without considering scallops in 1333395-2 bushing

$$A = (0.375 - 0.010 - 0.035) (0.5936) = 0.196 \text{ in}^2$$

Scallops considered for 40 percent of circumference

$$\pi d = \pi (0.5936) = 1.865$$

$$\text{Remove } 12 \times 0.062 = 0.744$$

$$(0.744/1.865) \times 100 = 40 \text{ percent}$$

Bearing area with scallops

$$A = (0.60) (0.196) = 0.118 \text{ in}^2$$

With 1.25 FS for limit load

$$\sigma = \frac{1.25 \times 1916}{0.118} = 20,300 \text{ psi Limit with 1.25 FS}$$

With $e/D = 0.66 < 1.5$ use $F_{bry} = F_{cy}$

$$F_{cy} = 35,000 \text{ psi}$$

$$MS = \frac{35,000}{20,300} - 1 = +0.72 \text{ Sidemount with 1.25 FS limit}$$

With 1.4 FS for ultimate load

$$\sigma = \frac{1.4 \times 1916}{0.118} = 22,740 \text{ psi}$$

With $e/D = 0.66 < 1.5$ Use $F_{bu} = F_{tu} = 42,000 \text{ psi}$

$$MS = \frac{42,000}{22,740} - 1 = +0.84 \text{ Sidemount with 1.4 FS ultimate}$$

The NASTRAN loads are given for load cases 1 through 4 in Table XX for the condition of all shear reacted in pins.

Mounting Bolt Stresses based on Static Loads for Condition of all Shear Reacted at Pins

Summary tables (Tables XXI through XXIV) for the four evaluated design load cases contain all data. Worst location is bolt at GRID 8313 for load case 4 (based on the assumption of $k_b/k_b + k_m = 0.624$) and FS of 1.4 applied to applied load minimum margin.

$$MS = \frac{3200}{(1.4)(0.624)(1749) + 1315} - 1 = +0.13$$

SUBCASE 1

FORCES OF SINGLE-POINT CONSTRAINT

POINT ID.	TYPE	T1	T2	T3	R1	R2	R3
7240	G	0.0	-9.874654E+01	0.0	0.0	0.0	0.0
7255	G	0.0	-3.824416E+01	0.0	0.0	0.0	0.0
7446	G	0.0	6.221199E+00	0.0	0.0	0.0	0.0
7462	G	0.0	2.256227E+02	0.0	0.0	0.0	0.0
7497	G	0.0	6.256901E+01	0.0	0.0	0.0	0.0
7507	G	0.0	2.515095E+02	0.0	0.0	0.0	0.0
7555	G	0.0	-1.666355E+02	0.0	0.0	0.0	0.0
7745	G	0.0	1.364205E+01	0.0	0.0	0.0	0.0
7840	G	0.0	-1.242564E+02	0.0	0.0	0.0	0.0
7855	G	1.795526E+03	0.0	4.336206E+01	0.0	0.0	0.0
7933	G	0.0	5.684992E+01	0.0	0.0	0.0	0.0
8313	G	0.0	3.677610E+02	0.0	0.0	0.0	0.0
9576	G	0.0	4.375062E+02	0.0	0.0	0.0	0.0
9834	G	0.0	2.937166E+02	0.0	0.0	0.0	0.0
10271	G	0.0	6.437571E+02	0.0	0.0	0.0	0.0
10703	G	0.0	4.379902E+02	0.0	0.0	0.0	0.0
10724	G	7.350280E+02	0.0	-4.336206E+01	0.0	-4.004632E-20	0.0
10748	G	0.0	8.967153E+01	0.0	0.0	-1.448398E-19	0.0
						3.437454E-20	0.0

Forces of Single-Point Constraint (Sheet 1 of 4)

SUBCASE 2

FORCES OF SINGLE-POINT CONSTRAINT

POINT ID.	TYPE	T1	T2	T3	R1	R2	R3
7240	G	0.0	2.381153E+02	0.0	0.0	0.0	0.0
7255	G	0.0	1.739431E+02	0.0	0.0	0.0	0.0
7446	G	0.0	9.301189E+00	0.0	0.0	0.0	0.0
7462	G	0.0	-2.812684E+01	0.0	0.0	0.0	0.0
7497	G	0.0	-1.764843E+01	0.0	0.0	0.0	0.0
7507	G	0.0	-1.406480E+02	0.0	0.0	0.0	0.0
7555	G	0.0	6.201996E+02	0.0	0.0	0.0	0.0
7745	G	0.0	6.905199E+01	0.0	0.0	0.0	0.0
7840	G	0.0	7.680912E+02	0.0	0.0	0.0	0.0
7855	G	-1.672131E+03	0.0	-4.336207E+01	0.0	0.0	0.0
7933	G	0.0	-1.096556E+01	0.0	0.0	0.0	0.0
8313	G	0.0	1.047597E+03	0.0	0.0	0.0	0.0
9576	G	0.0	1.194597E+02	0.0	0.0	0.0	0.0
9834	G	0.0	7.781662E+00	0.0	0.0	0.0	0.0
10271	G	0.0	-1.721214E+02	0.0	0.0	0.0	0.0
10703	G	0.0	-2.270925E+02	0.0	0.0	1.574671E-20	0.0
10724	G	-8.584225E+02	0.0	4.336208E+01	0.0	9.844039E-20	0.0
10748	G	0.0	1.996736E+00	0.0	0.0	-6.383711E-21	0.0

Forces of Single-Point Constraint (Sheet 2 of 4)

SUBCASE 3

FORCES OF SINGLE-POINT CONSTRAINT

POINT ID.	TYPE	T1	T2	T3	R1	R2	R3
7240	G	0.0	2.700318E+02	0.0	0.0	0.0	0.0
7255	G	0.0	3.912783E+02	0.0	0.0	0.0	0.0
7446	G	0.0	5.406951E+01	0.0	0.0	0.0	0.0
7462	G	0.0	4.622511E+02	0.0	0.0	0.0	0.0
7497	G	0.0	9.547428E+01	0.0	0.0	0.0	0.0
7507	G	0.0	3.011851E+02	0.0	0.0	0.0	0.0
7555	G	0.0	3.374338E+02	0.0	0.0	0.0	0.0
7745	G	0.0	2.343174E+02	0.0	0.0	0.0	0.0
7840	G	0.0	3.678293E+02	0.0	0.0	0.0	0.0
7855	G	1.273211E+01	0.0	1.913108E+03	0.0	0.0	0.0
7933	G	0.0	6.072196E+01	0.0	0.0	0.0	0.0
8313	G	0.0	-3.337893E+02	0.0	0.0	0.0	0.0
9576	G	0.0	-5.296381E+01	0.0	0.0	0.0	0.0
9834	G	0.0	2.370037E+02	0.0	0.0	0.0	0.0
10271	G	0.0	3.219608E+02	0.0	0.0	0.0	0.0
10703	G	0.0	-1.446809E+02	0.0	0.0	-7.974856E-21	0.0
10724	G	-1.273211E+01	0.0	4.264612E+02	0.0	-3.689414E-19	0.0
10748	G	0.0	-1.431885E+02	0.0	0.0	-9.468250E-21	0.0

Forces of Single-Point Constraint (Sheet 3 of 4)

SUBCASE 4

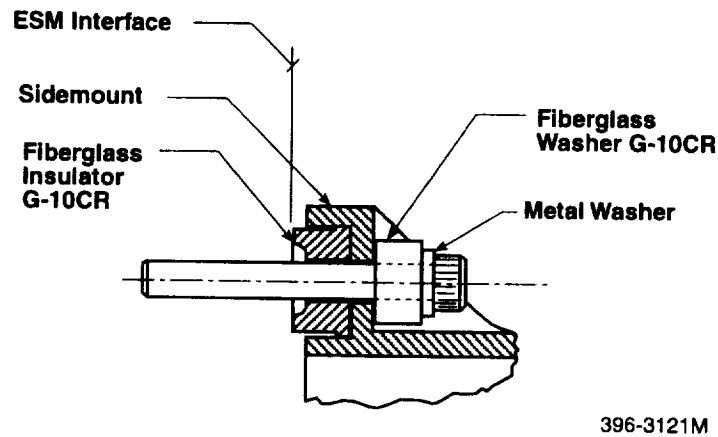
FORCES OF SINGLE-POINT CONSTRAINT

POINT ID.	TYPE	T1	T2	T3	R1	R2	R3
7240	G	0.0	-1.306630E+02	0.0	0.0	0.0	0.0
7255	G	0.0	-2.555793E+02	0.0	0.0	0.0	0.0
7446	G	0.0	-3.854712E+01	0.0	0.0	0.0	0.0
7462	G	0.0	-2.647552E+02	0.0	0.0	0.0	0.0
7497	G	0.0	-5.055370E+01	0.0	0.0	0.0	0.0
7507	G	0.0	-1.903235E+02	0.0	0.0	0.0	0.0
7555	G	0.0	1.161304E+02	0.0	0.0	0.0	0.0
7745	G	0.0	-1.516234E+02	0.0	0.0	0.0	0.0
7840	G	0.0	2.760055E+02	0.0	0.0	0.0	0.0
7855	G	1.106625E+02	0.0	-1.913108E+03	0.0	0.0	0.0
7933	G	0.0	-1.483760E+01	0.0	0.0	0.0	0.0
8313	G	0.0	1.749147E+03	0.0	0.0	0.0	0.0
9576	G	0.0	6.099297E+02	0.0	0.0	0.0	0.0
9834	G	0.0	6.449465E+01	0.0	0.0	0.0	0.0
10271	G	0.0	1.496748E+02	0.0	0.0	0.0	0.0
10703	G	0.0	3.555786E+02	0.0	0.0	-1.632476E-20	0.0
10724	G	-1.106625E+02	0.0	-4.264612E+02	0.0	3.225420E-19	0.0
10748	G	0.0	2.348568E+02	0.0	0.0	3.745908E-20	0.0

Forces of Single-Point Constraint (Sheet 4 of 4)

Member Compression Under Preload

The mounting bolt joint is shown from a view on 1333964.



Where the metal CRES washer is considered as 0.440 OD, 0.195 ID, 0.060 thick. The fiberglass washer is G-10, 0.435 $\varnothing$ x 0.250 thick. The fiberglass insulator is also G-10, 0.590 $\varnothing$ x 0.310 thick (1333395-6).

The sidemount hole dimensions are a 0.287 max dia hole with 0.5925 max dia counterbore. Thickness of 0.287 $\varnothing$ is 0.125.

Maximum load is the preload $F = 1315$ lb for 50 in-lb torque.

$FS = 1.4$ ultimate, 1.25 limit, material allowable.

Bolt	Assumed A-286 Steel	$F_{TU} = 160,000$ psi
Washer	Assumed A-286 Steel	$F_{TU} = 160,000$ psi
G-10CR Isolators	Laminates $\perp$ (centerline)	$F_{TU} = 60,000$ psi
Sidemount	6061-T6 Aluminum	$F_{TU} = 42,000$ psi
		$F_{TY} = 35,000$ psi

With summarized MS calculations in Table XXV, member analysis follows.

Bolthead to Washer

Assume NAS1351N3-32 Screw
Assume NAS1149E0363P Washer

Head OD	0.313 minimum
Washer ID	$0.203 + 0.010 = 0.213$ maximum

$$A = \pi/4 (0.303^2 - 0.213^2) = 0.0365 \text{ in}^2 \text{ minimum}$$

$$FA = \frac{1315}{0.0365} = 36,100 \text{ psi}$$

$$MS = \frac{160,000}{1.4 \times 36,100} - 1 = + 2.1 \text{ Bolt / Washer}$$

Washer to 0.435 OD x 0.250 thick Washer

Assume NAS 11440E0363P Washer

Washer OD 0.438 - 0.005 = 0.433 min ID = 0.213 max
Isolator OD 0.435 - 0.002 = 0.433 min (Assumed)

$$A = \pi/4 (0.433^2 - 0.213^2) = 0.112 \text{ in}^2$$

$$F / A = \frac{1315}{0.112} = 11,781 \text{ psi}$$

$$MS = \frac{60,000}{1.4 \times 11,781} - 1 = +2.6 \text{ Isolator}$$

Isolator to Sidemount Ledge

Isolator OD 0.433 min (Assumed)
Sidemount ID 0.287 max

$$A = \pi/4 (0.433^2 - 0.287^2) = 0.083 \text{ in}^2$$

$$F/A = \frac{1315}{0.083} = 15,928 \text{ psi}$$

$$MS = \frac{35,000}{1.25 \times 15,928} - 1 = +0.76 \text{ Sidemount}$$

$$MS = \frac{42,000}{1.4 \times 15,928} - 1 = +0.88 \text{ Sidemount}$$

Sidemount Ledge to 1333395-6 .590 Ø Isolator

Isolator OD 0.590 - 2 (0.030) = 0.530 min
Sidemount ID 0.287 max

$$A = \pi/4 (0.530^2 - 0.287^2) = 0.156 \text{ in}^2$$

$$F/A = \frac{1315}{0.156} = 8433 \text{ psi}$$

$$MS = \frac{35,000}{1.25 \times 8,433} - 1 = +2.3 \text{ Sidemount}$$

$$MS = \frac{42,000}{1.4 \times 8,433} - 1 = +2.6 \text{ Sidemount}$$

0.590 Ø Isolator to Spacecraft

Isolator OD 0.590 min
Isolator ID 0.385 max

$$A = \pi/4 (0.590^2 - 0.385^2) = 0.157 \text{ in}^2$$

$$F/A = \frac{1315}{0.157} = 8377 \text{ psi}$$

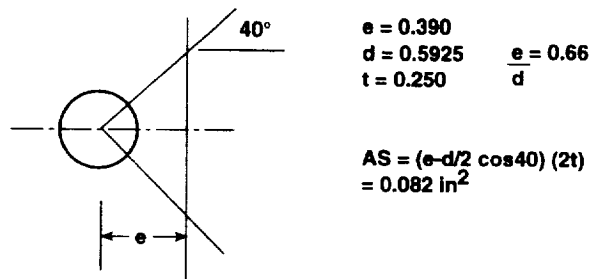
$$MS = \frac{60,000}{1.4 \times 8,377} - 1 = +4.1 \text{ Isolator}$$

Sidemount Mounting Bolts Shear Tearout

Maximum #10-32 screw shear load is from case 1, GRID 9834, from the condition of shear load reacted at both the 16 screws and the 2 pins, as shown in Table XXVI.

$$F = \sqrt{488^2 + 21^2} = 488 \text{ lb}$$

Shear tearout of the aluminum sidemount at the lower, 0.590 $\varnothing$ isolator, per Bruhn.



896-3015M

$$T = \frac{(14)(488)}{0.082} = 8387 \text{ psi}$$

$$F_{su} = 27,000 \text{ psi}$$

$$MS = \frac{27,000}{8,387} - 1 = +2.2 \text{ Sidemount}$$

Bearing of 0.590 $\varnothing$ Isolator onto Sidemount Hole for Maximum Shear Load

$$F_{smax} = 488 \text{ lb GRID 9834 Case 1}$$

$$F_t = (0.590) (0.250) = 0.148 \text{ in}^2$$

$$\sigma = \frac{1.25 \times 488}{0.148} = 4137 \text{ psi Limit with FS} = 1.25$$

With $e/D = 0.66 \ll 1.5$ Use $F_{bry} = F_{cy}$

$$F_{cy} = 35,000 \text{ psi 6061-T6}$$

$$MS = \frac{35,000}{4,137} - 1 = +7.5 \text{ Sidemount}$$

$$\text{Use } F_{bru} = F_{tu} = 42,000 \text{ psi}$$

$$MS = \frac{42,000}{\left(\frac{1.4 \times 488}{0.148} \right)} - 1 = +8.1 \quad \text{Sidemount}$$

For the isolator use $F_{bru} = F_{cu \text{ edge}} = 35,000 \text{ psi}$

$$MS = \frac{35,000}{\left(\frac{1.4 \times 488}{0.148} \right)} - 1 = +6.6 \quad 0.540\text{Ø isolator}$$

Other load cases are summarized in the summary table, Table XXVI.

Thread Shear of 0.190-32UNF Screw into Spacecraft Mounting Surface

Assumptions

1. Screw material is A286 w/ $F_{tu} = 160,000 \text{ psi}$
2. Thread type is 0.190-32UNRF-3A (NAS1351N3-32)
3. Spacecraft INT THRD is of similar material as screw
4. Length of engagement of joint $\geq 1\text{Ø} (\geq 0.190)$

Max shear load per Case 4 tensile max at GRID 8313 (all shear at shear pins)

$$F = 2843 \text{ lb} \quad \text{Preload} + \text{applied load with assumed joint stiffness } k_b/k_b + k_m = 0.624 \text{ with FS} = 1.4.$$

Shear area

$$AS = \frac{pEL_e}{2} = 0.05065 \text{ in}^2 \quad E = 0.1697 \\ Le = 0.190$$

Allowable, $F_{su} = 0.6 \times F_{tu} = 96,000 \text{ psi}$

$$MS = \frac{96,000}{\frac{2,843}{0.05065}} - 1 = +7.1$$

Other load cases are addressed in Table XXV.

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Table I METSAT AMSU-A1 Margins of Safety Summary, All Elements,
Design Loads Cases 1 through 8

ITEM NO.*	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	STRESS** (psi)	MARGINS OF SAFETY		CASE EL TYPE
							YIELD	ULTIMATE	
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	5181	4.40	4.79	3 BEAM
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	4263	11.39	11.57	1 BEAM
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	2703	18.53	18.82	1 BEAM
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	2344	10.95	11.80	1 BEAM
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	2159	20.49	20.84	4 BEAM
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	2576	9.87	10.65	2 BEAM
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	4533	10.65	10.82	1 BEAM
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	5087	8.12	8.27	1 SHELL
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR*200E	50000	70000	8745	3.57	4.72	1 SHELL
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	3804	9.52	12.14	4 SHELL
11	LOWER RIGHT FRONT SUPT PANEL	1331447-1	ALUM/6061-T651	35000	42000	1525	17.36	18.67	1 BEAM
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	1973	13.19	14.21	4 BEAM
13	UPPER RIGHT FRONT SUPT PANEL	1331390-1	ALUM/6061-T651	35000	42000	2457	10.40	11.21	2 BEAM
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	1721	15.27	16.43	1 SHELL
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	2301	11.17	12.04	2 SHELL
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	1948	13.37	14.40	4 SHELL
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	3593	6.79	7.35	1 BEAM
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	3552	6.88	7.45	1 BEAM
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	3630	6.71	7.26	1 BEAM
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	2096	12.36	13.31	1 BEAM
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	3107	8.01	8.66	4 SHELL
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	994	18.32	27.74	4 SHELL
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	3415	7.20	7.78	4 BEAM
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	3057	3.19	6.01	4 BEAM
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	529	35.29	53.01	1 SHELL
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	1984	13.11	14.12	3 SHELL
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	1904	13.71	14.76	3 SHELL
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	2576	19.50	19.80	2 BEAM
28	LOWER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	1602	16.48	17.73	1 BEAM
29	POWER CONTROL/MONITOR, BKT	1356790-1	ALUM/6061-T6	35000	42000	3211	7.72	8.34	2 SHELL
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	891	20.55	31.07	1 SHELL
31	PWR CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	11596	2.04	6.51	2 BEAM
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	2622	9.68	10.44	2 SHELL
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	10727	1.61	1.80	4 SHELL
34	UPPER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	1610	16.39	17.63	1 BEAM

DESIGN LOADS CASE 1 Gy=-20.6 Gx=-21.2, CASE 2 Gy=-20.6,Gx=21.2, CASE 3 Gy=-20.6,Gz=-19.6, CASE 4 Gy=-20.6,Gz=19.6
DESIGN LOADS CASE 5 Gy=20.6 Gx=-21.2, CASE 6 Gy=20.6,Gx=21.2, CASE 7 Gy=20.6,Gz=-19.6, CASE 8 Gy=20.6,Gz=19.6
**VON MISES STRESS FOR SHELL ELEMENTS, ABSOLUTE VALUE OF MAXIMUM PRINCIPAL STRESS OR
MINIMUM PRINCIPAL STRESS FOR BEAM ELEMENTS.

Table II METSAT AMSU-A1 Margins of Safety Summary, Shell Elements,
Design Loads Case 1 Gy=-20.6 Gx=-21.2

ITEM NO.*	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	VON MISES STRESS (psi)	MARGINS OF SAFETY	
							YIELD	ULTIMATE
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	771	35.32	37.91
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	3543	13.90	14.12
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	1972	25.77	26.17
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	1625	16.23	17.46
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	731	62.47	63.49
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	1064	25.32	27.20
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	869	59.76	60.65
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	5087	8.12	8.27
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	8745	3.57	4.72
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	991	39.36	49.45
11	LOWER RIGHT FRONT SUPPORT PANEL	1331447-1	ALUM/6061-T651	35000	42000	573	47.87	51.36
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	1221	21.93	23.57
13	UPPER RIGHT FRONT SUPPORT PANEL	1331390-1	ALUM/6061-T651	35000	42000	928	29.17	31.33
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	1721	15.27	16.43
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	2043	12.71	13.68
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	1666	15.81	17.01
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	2011	12.92	13.92
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	766	35.55	38.16
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	1245	21.49	23.10
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	397	69.53	74.57
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	2693	9.40	10.14
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	873	20.99	31.73
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	537	51.14	54.87
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	537	22.84	38.90
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	529	35.29	53.01
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	1014	26.61	28.59
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	853	31.83	34.17
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	*****		
28	LOWER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	777	35.04	37.61
29	POWER CONTROL/MONITOR, BRACKET	1356790-1	ALUM/6061-T6	35000	42000	2336	10.99	11.84
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	891	20.55	31.07
31	POWER CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	*****		
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	1773	14.79	15.92
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	6437	3.35	3.66
34	UPPER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	771	35.32	37.91

* ALSO DESIGN LOADS CASE 6 Gy=20.6 Gx=21.2

Table III METSAT AMSU-A1 Margins of Safety Summary, Beam Elements,
Design Loads Case 1 Gy=-20.6 Gx=-21.2

ITEM NO.*	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	MAXIMUM** STRESS (psi)	MARGINS OF SAFETY	
							YIELD	ULTIMATE
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	1449	18.32	19.70
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	4263	11.39	11.57
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	2703	18.53	18.82
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	2344	10.95	11.80
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	1478	30.39	30.90
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	1395	19.07	20.51
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	4533	10.65	10.82
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	2581	16.98	17.27
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	1588	24.19	30.49
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	437	90.53	113.42
11	LOWER RIGHT FRONT SUPPORT PANEL	1331447-1	ALUM/6061-T651	35000	42000	1525	17.36	18.67
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	1301	20.52	22.06
13	UPPER RIGHT FRONT SUPPORT PANEL	1331390-1	ALUM/6061-T651	35000	42000	2096	12.36	13.31
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	1254	21.33	22.92
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	500	55.00	59.00
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	*****		
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	3593	6.79	7.35
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	3552	6.88	7.45
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	3630	6.71	7.26
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	2096	12.36	13.31
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	*****		
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	*****		
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	1657	15.90	17.11
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	323	38.63	65.34
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	*****		
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	*****		
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	*****		
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	1002	51.69	52.46
28	LOWER REFLECTOR	1355777-1	ALUM/7075-T6	66000	75000	1602	31.96	32.44
29	POWER CONTROL/MONITOR, BRACKET	1356790-1	ALUM/6061-T6	35000	42000	*****		
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	*****		
31	POWER CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	10547	2.34	7.26
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	*****		
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	2437	10.49	11.31
34	UPPER REFLECTOR	1355777-1	ALUM/7075-T6	66000	75000	1610	31.80	32.27

*ALSO DESIGN LOADS CASE 6 Gy=20.6 Gx=21.2

**ABSOLUTE VALUE OF MAXIMUM PRINCIPAL STRESS OR MINIMUM PRINCIPAL STRESS.

Table IV METSAT AMSU-A1 Margins of Safety Summary, Shell Elements,
Design Loads Case 2 Gy=-20.6 Gx=21.2

ITEM NO.*	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	VON MISES STRESS (psi)	MARGINS OF SAFETY	
							YIELD	ULTIMATE
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	675	40.48	43.44
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	4101	11.87	12.06
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	1426	36.03	36.57
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	1045	25.79	27.71
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	1418	31.72	32.25
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	1632	16.16	17.38
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	1717	29.75	30.20
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	4667	8.94	9.10
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	1489	25.86	32.58
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	1110	35.04	44.05
11	LOWER RIGHT FRONT SUPPORT PANEL	1331447-1	ALUM/6061-T651	35000	42000	544	50.47	54.15
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	1212	22.10	23.75
13	UPPER RIGHT FRONT SUPPORT PANEL	1331390-1	ALUM/6061-T651	35000	42000	821	33.10	35.54
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	1345	19.82	21.30
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	2301	11.17	12.04
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	1846	14.17	15.25
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	1959	13.29	14.31
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	726	37.57	40.32
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	1466	18.10	19.46
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	327	84.63	90.74
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	2753	9.17	9.90
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	851	21.56	32.57
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	715	38.16	40.96
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	550	22.27	37.96
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	526	35.50	53.32
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	1050	25.67	27.57
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	765	35.60	38.22
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	*****		
28	LOWER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	677	40.36	43.31
29	POWER CONTROL/MONITOR, BRACKET	1356790-1	ALUM/6061-T6	35000	42000	3211	7.72	8.34
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	868	21.12	31.92
31	POWER CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	*****		
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	2622	9.68	10.44
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	10137	1.76	1.96
34	UPPER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	689	39.64	42.54

* ALSO DESIGN LOADS CASE 5 Gy=20.6 Gx=-21.2

Table V METSAT AMSU-A1 Margins of Safety Summary, Beam Elements,
Design Loads Case 2 Gy=-20.6 Gx=21.2

ITEM NO.*	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	MAXIMUM** STRESS (psi)	MARGINS OF SAFETY	
							YIELD	ULTIMATE
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	1155	23.24	24.97
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	3819	12.83	13.03
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	2457	20.49	20.80
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	1471	18.03	19.39
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	120	385.67	391.86
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	2576	9.87	10.65
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	2746	18.23	18.51
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	2311	19.08	19.40
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	956	40.84	51.30
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	784	50.02	62.78
11	LOWER RIGHT FRONT SUPPORT PANEL	1331447-1	ALUM/6061-T651	35000	42000	1332	20.02	21.52
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	842	32.25	34.63
13	UPPER RIGHT FRONT SUPPORT PANEL	1331390-1	ALUM/6061-T651	35000	42000	2457	10.40	11.21
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	1048	25.72	27.63
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	692	39.46	42.35
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	*****		
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	3588	6.80	7.36
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	3211	7.72	8.34
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	3172	7.83	8.46
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	1702	15.45	16.63
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	*****		
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	*****		
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	1816	14.42	15.52
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	336	37.10	62.78
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	*****		
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	*****		
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	*****		
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	2576	19.50	19.80
28	LOWER REFLECTOR	1355777-1	ALUM/7075-T6	66000	75000	986	52.55	53.33
29	POWER CONTROL/MONITOR, BRACKET	1356790-1	ALUM/6061-T6	35000	42000	*****		
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	*****		
31	POWER CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	11596	2.04	6.51
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	*****		
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	1865	14.01	15.09
34	UPPER REFLECTOR	1355777-1	ALUM/7075-T6	66000	75000	903	57.47	58.33

* ALSO DESIGN LOADS CASE 5 Gy=20.6 Gx=-21.2

** ABSOLUTE VALUE OF MAXIMUM PRINCIPAL STRESS OR MINIMUM PRINCIPAL STRESS.

Table VI METSAT AMSU-A1 Margins of Safety Summary, Shell Elements,
Design Loads Case 3 Gy=-20.6 Gz=-19.6

ITEM NO.	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	VON MISES STRESS (psi)	MARGINS OF SAFETY	
							YIELD	ULTIMATE
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	2065	12.56	13.53
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	1026	50.46	51.21
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	516	101.33	102.82
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	1235	21.67	23.29
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	1745	25.59	26.02
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	1056	25.52	27.41
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	1175	43.94	44.59
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	1197	37.76	38.38
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	4607	7.68	9.85
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	3603	10.10	12.88
11	LOWER RIGHT FRONT SUPPORT PANEL	1331447-1	ALUM/6061-T651	35000	42000	194	143.33	153.64
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	1554	17.02	18.31
13	UPPER RIGHT FRONT SUPPORT PANEL	1331390-1	ALUM/6061-T651	35000	42000	303	91.41	98.01
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	839	32.37	34.76
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	1621	16.27	17.51
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	1616	16.33	17.56
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	684	39.94	42.86
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	181	153.70	164.75
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	441	62.49	67.03
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	51	548.02	587.24
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	3047	8.19	8.85
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	779	23.65	35.68
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	2358	10.87	11.72
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	1312	8.76	15.33
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	264	71.73	107.23
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	1984	13.11	14.12
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	1904	13.71	14.76
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	*****		
28	LOWER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	929	29.14	31.29
29	POWER CONTROL/MONITOR, BRACKET	1356790-1	ALUM/6061-T6	35000	42000	884	30.67	32.94
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	188	101.13	150.98
31	POWER CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	*****		
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	2099	12.34	13.29
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	9509	1.94	2.15
34	UPPER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	929	29.14	31.29

*ALSO DESIGN LOADS CASE 8 Gy=20.6 Gz=19.6

Table VII METSAT AMSU-A1 Margins of Safety Summary, Beam Elements,
Design Loads Case 3 Gy=-20.6 Gz=-19.6

ITEM NO.*	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	MAXIMUM** STRESS (psi)	MARGINS OF SAFETY	
							YIELD	ULTIMATE
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	5181	4.40	4.79
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	1251	41.21	41.82
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	775	67.13	68.12
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	1375	19.36	20.82
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	1527	29.39	29.87
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	1600	16.50	17.75
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	3266	15.17	15.40
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	757	60.29	61.28
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	1581	24.30	30.63
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	1313	29.46	37.08
11	LOWER RIGHT FRONT SUPPORT PANEL	1331447-1	ALUM/6061-T651	35000	42000	480	57.33	61.50
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	1581	16.71	17.98
13	UPPER RIGHT FRONT SUPPORT PANEL	1331390-1	ALUM/6061-T651	35000	42000	1494	17.74	19.08
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	691	39.52	42.42
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	563	48.73	52.29
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	*****		
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	890	30.46	32.71
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	799	34.04	36.55
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	2154	12.00	12.93
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	515	53.37	57.25
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	*****		
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	*****		
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	2628	9.65	10.42
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	2398	4.34	7.94
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	*****		
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	*****		
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	*****		
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	1600	32.00	32.48
28	LOWER REFLECTOR	1355777-1	ALUM/7075-T6	66000	75000	950	54.58	55.39
29	POWER CONTROL/MONITOR, BRACKET	1356790-1	ALUM/6061-T6	35000	42000	*****		
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	*****		
31	POWER CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	3496	9.07	23.93
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	*****		
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	1045	25.79	27.71
34	UPPER REFLECTOR	1355777-1	ALUM/7075-T6	66000	75000	966	53.66	54.46

*ALSO DESIGN LOADS CASE 8 Gy=20.6 Gz=19.6

**ABSOLUTE VALUE OF MAXIMUM PRINCIPAL STRESS OR MINIMUM PRINCIPAL STRESS.

Table VIII METSAT AMSU-A1 Margins of Safety Summary, Shell Elements,
Design Loads Case 4 Gy=-20.6 Gz=19.6

ITEM NO.*	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	VON MISES STRESS (psi)	MARGINS OF SAFETY	
							YIELD	ULTIMATE
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	2016	12.89	13.88
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	1400	36.71	37.27
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	956	54.23	55.04
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	2017	12.88	13.87
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	1631	27.45	27.90
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	1811	14.46	15.57
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	1858	27.42	27.83
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	1499	29.95	30.45
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	4621	7.66	9.82
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	3804	9.52	12.14
11	LOWER RIGHT FRONT SUPPORT PANEL	1331447-1	ALUM/6061-T651	35000	42000	171	162.74	174.44
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	1237	21.64	23.25
13	UPPER RIGHT FRONT SUPPORT PANEL	1331390-1	ALUM/6061-T651	35000	42000	348	79.46	85.21
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	733	37.20	39.93
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	1668	15.79	16.99
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	1948	13.37	14.40
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	589	46.54	49.93
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	180	154.56	165.67
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	938	28.85	30.98
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	80	349.00	374.00
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	3107	8.01	8.66
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	994	18.32	27.74
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	3213	7.71	8.34
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	1585	7.08	12.52
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	174	109.34	163.20
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	1804	14.52	15.63
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	1857	14.08	15.16
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	*****		
28	LOWER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	785	34.67	37.22
29	POWER CONTROL/MONITOR, BRACKET	1356790-1	ALUM/6061-T6	35000	42000	945	28.63	30.75
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	200	95.00	141.86
31	POWER CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	*****		
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	1164	23.05	24.77
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	10727	1.61	1.80
34	UPPER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	785	34.67	37.22

*ALSO DESIGN LOADS CASE 7 Gy=20.6 GZ=-19.6

Table IX METSAT AMSU-A1 Margins of Safety Summary, Beam Elements,
Design Loads Case 4 Gy=-20.6 Gz=19.6

ITEM NO.*	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	MAXIMUM** STRESS (psi)	MARGINS OF SAFETY	
							YIELD	ULTIMATE
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	4816	4.81	5.23
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	1005	51.54	52.30
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	1441	35.64	36.18
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	2209	11.68	12.58
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	2159	20.49	20.84
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	2479	10.29	11.10
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	2953	16.88	17.14
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	1442	31.18	31.69
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	1973	19.27	24.34
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	1555	24.72	31.15
11	LOWER RIGHT FRONT SUPPORT PANEL	1331447-1	ALUM/6061-T651	35000	42000	416	66.31	71.12
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	1973	13.19	14.21
13	UPPER RIGHT FRONT SUPPORT PANEL	1331390-1	ALUM/6061-T651	35000	42000	1256	21.29	22.89
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	1121	23.98	25.76
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	573	47.87	51.36
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	*****		
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	1552	17.04	18.33
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	977	27.66	29.71
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	1755	14.95	16.09
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	357	77.43	83.03
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	*****		
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	*****		
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	3415	7.20	7.78
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	3057	3.19	6.01
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	*****		
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	*****		
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	*****		
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	2479	20.30	20.61
28	LOWER REFLECTOR	1355777-1	ALUM/7075-T6	66000	75000	1155	44.71	45.38
29	POWER CONTROL/MONITOR, BRACKET	1356790-1	ALUM/6061-T6	35000	42000	*****		
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	*****		
31	POWER CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	3178	10.08	26.42
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	*****		
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	3082	8.09	8.73
34	UPPER REFLECTOR	1355777-1	ALUM/7075-T6	66000	75000	1127	45.85	46.53

*ALSO DESIGN LOADS CASE 7 Gy=20.6 GZ=-19.6

**ABSOLUTE VALUE OF MAXIMUM PRINCIPAL STRESS OR MINIMUM PRINCIPAL STRESS.

Table X METSAT AMSU-A1 Modal Effective Weight and Participation Factor Summary

MODE NO.	NATURAL FREQ. Hz.	X-AXIS %	Y-AXIS %	Z-AXIS %			
1	101.8	0.003	0.000	0.129			
2	109.0	0.000	0.000	0.000			
3	109.0	0.000	0.000	0.000			
4	109.5	1.195	0.001	0.017			
5	109.7	0.000	0.000	0.000			
6	120.4	8.452	0.127	0.028			
7	121.4	0.026	0.002	0.883			
8	121.9	0.017	0.256	0.043			
9	122.0	0.064	0.150	0.147			
10	130.9	1.876	0.567	27.73			
11	131.8	0.877	0.811	28.75			
12	138.5	6.239	0.028	0.733			
13	139.6	0.000	0.000	0.000			
14	140.8	0.050	0.004	0.004			
15	149.1	1.312	0.052	0.100			
16	154.2	0.217	0.019	0.053			

Table XI METSAT AMSU-A1 Margins of Safety Summary, All Elements,
Random Vibration 8.8 grms

ITEM NO.*	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	3 SIGMA** STRESS (psi)	MARGINS OF SAFETY		DIRECTION EL TYPE
							YIELD	ULTIMATE	
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	9554	1.93	2.14	Z BEAM
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	15093	2.50	2.55	X BEAM
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	8478	5.23	5.32	X BEAM
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	5884	3.76	4.10	X BEAM
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	2323	18.97	19.29	Z BEAM
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	5384	4.20	4.57	X BEAM
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	16837	2.14	2.18	X BEAM
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	18035	1.57	1.61	X SHELL
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	7156	4.59	5.99	Y SHELL
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	4429	8.03	10.29	Z SHELL
11	LOWER RIGHT FRONT SUPT PANEL	1331447-1	ALUM/6061-T651	35000	42000	3896	6.19	6.70	X BEAM
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	4214	5.64	6.12	Y SHELL
13	UPPER RIGHT FRONT SUPT PANEL	1331390-1	ALUM/6061-T651	35000	42000	6716	3.17	3.47	X BEAM
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	3591	6.80	7.35	X SHELL
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	4858	4.76	5.18	X SHELL
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	2369	10.82	11.66	X SHELL
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	5832	3.80	4.14	X BEAM
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	6477	3.32	3.63	X BEAM
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	8969	2.12	2.34	X BEAM
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	10715	1.61	1.80	X BEAM
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	7342	2.81	3.09	Z SHELL
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	1929	8.95	13.81	Y SHELL
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	7652	2.66	2.92	Z BEAM
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	3826	2.35	4.60	Z BEAM
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	1659	10.57	16.22	X SHELL
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	6451	3.34	3.65	Y SHELL
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	7876	2.56	2.81	Y SHELL
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	4104	11.87	12.05	Y BEAM
28	LOWER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	8758	2.20	2.43	X SHELL
29	POWER CONTROL/MONITOR, BKT	1356790-1	ALUM/6061-T6	35000	42000	8050	2.48	2.73	X SHELL
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	3189	5.02	7.96	X SHELL
31	PWR CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	21709	0.62	3.01	X BEAM
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	6071	3.61	3.94	X SHELL
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	10836	1.58	1.77	X SHELL
34	UPPER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	7721	2.63	2.89	X SHELL

*RANDOM VIBRATION LOAD CASES 8.8 GRMS INDEPENDENTLY IN X, Y, AND Z

**MAXIMUM PRINCIPAL STRESS BASED ON ASSUMED IN PHASE COMPONENT STRESSES

Table XII METSAT AMSU-A1 Margins of Safety Summary, All Elements,
Random Vibration 8.8 grms, X Direction

ITEM NO.	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	3 SIGMA ^{***} STRESS (psi)	MARGINS OF SAFETY YIELD	MARGINS OF SAFETY ULTIMATE	DIRECTION EL TYPE
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	2053	12.64	13.61	X BEAM
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	15093	2.50	2.55	X BEAM
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	8478	5.23	5.32	X BEAM
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	5884	3.76	4.10	X BEAM
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	1922	23.14	23.53	X SHELL
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	5384	4.20	4.57	X BEAM
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	16837	2.14	2.18	X BEAM
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	18035	1.57	1.61	X SHELL
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	5685	6.04	7.80	X SHELL
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	3728	9.73	12.41	X SHELL
11	LOWER RIGHT FRONT SUPT PANEL	1331447-1	ALUM/6061-T651	35000	42000	3896	6.19	6.70	X BEAM
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	3045	8.20	8.85	X BEAM
13	UPPER RIGHT FRONT SUPT PANEL	1331390-1	ALUM/6061-T651	35000	42000	6716	3.17	3.47	X BEAM
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	3591	6.80	7.35	X SHELL
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	4858	4.76	5.18	X SHELL
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	2369	10.82	11.66	X SHELL
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	5832	3.80	4.14	X BEAM
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	6477	3.32	3.63	X BEAM
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	8969	2.12	2.34	X BEAM
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	10715	1.61	1.80	X BEAM
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	3665	6.64	7.19	X SHELL
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	901	20.31	30.71	X SHELL
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	6356	3.41	3.72	X BEAM
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	2602	3.92	7.24	X SHELL
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	1659	10.57	16.22	X SHELL
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	6405	3.37	3.68	X SHELL
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	3039	8.21	8.87	X SHELL
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	2649	18.93	19.22	X BEAM
28	LOWER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	8758	2.20	2.43	X SHELL
29	POWER CONTROL/MONITOR, BKT	1356790-1	ALUM/6061-T6	35000	42000	8050	2.48	2.73	X SHELL
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	3189	5.02	7.96	X SHELL
31	PWR CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	21709	0.62	3.01	X BEAM
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	6071	3.61	3.94	X SHELL
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	10836	1.58	1.77	X SHELL
34	UPPER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	7721	2.63	2.89	X SHELL

^{***} MAXIMUM PRINCIPAL STRESS BASED ON ASSUMED IN PHASE COMPONENT STRESSES

Table XIII METSAT AMSU-A1 Margins of Safety Summary, All Elements,
Random Vibration 8.8 grms, Y Direction

ITEM NO.*	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	3 SIGMA** STRESS (psi)	MARGINS OF SAFETY		DIRECTION EL TYPE
							YIELD	ULTIMATE	
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	4685	4.98	5.40	Y BEAM
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	4023	12.12	12.32	Y SHELL
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	2604	19.28	19.57	Y SHELL
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	3187	7.79	8.41	Y BEAM
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	1470	30.56	31.07	Y SHELL
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	3505	6.99	7.56	Y BEAM
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	13997	2.77	2.83	Y BEAM
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	5494	7.45	7.58	Y SHELL
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	7156	4.59	5.99	Y SHELL
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	3873	9.33	11.91	Y SHELL
11	LOWER RIGHT FRONT SUPT PANEL	1331447-1	ALUM/6061-T651	35000	42000	1038	25.97	27.90	Y BEAM
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	4214	5.64	6.12	Y SHELL
13	UPPER RIGHT FRONT SUPT PANEL	1331390-1	ALUM/6061-T651	35000	42000	4543	5.16	5.60	Y BEAM
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	3231	7.67	8.29	Y SHELL
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	1175	22.83	24.53	Y SHELL
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	2856	8.80	9.50	Y SHELL
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	2651	9.56	10.32	Y BEAM
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	2612	9.72	10.49	Y BEAM
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	2829	8.90	9.60	Y BEAM
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	3005	8.32	8.98	Y BEAM
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	6730	3.16	3.46	Y SHELL
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	1929	8.95	13.81	Y SHELL
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	6706	3.18	3.47	Y BEAM
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	3359	2.81	5.38	Y SHELL
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	330	57.18	85.58	Y SHELL
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	6451	3.34	3.65	Y SHELL
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	7876	2.56	2.81	Y SHELL
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	4104	11.87	12.05	Y BEAM
28	LOWER REFLECTOR	1355777-1	ALUM/7075-T6	66000	75000	10388	4.08	4.16	Y BEAM
29	POWER CONTROL/MONITOR, BKT	1356790-1	ALUM/6061-T6	35000	42000	1419	18.73	20.14	Y SHELL
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	1663	10.55	16.18	Y SHELL
31	PWR CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	7064	3.98	11.34	Y BEAM
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	1510	17.54	18.87	Y SHELL
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	8001	2.50	2.75	Y SHELL
34	UPPER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	4615	5.07	5.50	Y SHELL

**MAXIMUM PRINCIPAL STRESS BASED ON ASSUMED IN PHASE COMPONENT STRESSES

Table XIV METSAT AMSU-A1 Margins of Safety Summary, All Elements,
Random Vibration 8.8 grms, Z Direction

ITEM NO.¹	DESCRIPTION	PART NUMBER	MATERIAL/ ALLOY	YIELD (psi)	ULTIMATE (psi)	3 SIGMA ^² STRESS (psi)	MARGINS OF SAFETY		DIRECTION EL TYPE
							YIELD	ULTIMATE	
1	LOWER BASEPLATE ASSEMBLY	1331404-1	ALUM/6061-T651	35000	42000	9554	1.93	2.14	Z BEAM
2	LOWER MOTOR MOUNT PANEL	1331414-1	ALUM/7075-T651	66000	75000	3271	15.14	15.38	Z BEAM
3	UPPER MOTOR MOUNT PANEL	1331389-1	ALUM/7075-T651	66000	75000	1966	25.86	26.25	Z BEAM
4	LOWER FRONT PANEL	1331401-1	ALUM/6061-T651	35000	42000	3660	6.65	7.20	Z BEAM
5	LOWER AFT PANEL	1331652-1	ALUM/2024-T851	58000	66000	2323	18.97	19.29	Z BEAM
6	UPPER BASEPLATE ASSEMBLY	1331356-3	ALUM/6061-T651	35000	42000	2549	9.98	10.77	Z BEAM
7	UPPER FRONT PANEL	1331352-3	ALUM/7075-T651	66000	75000	5846	8.03	8.16	Z BEAM
8	UPPER AFT PANEL	1331642-3	ALUM/2024-T851	58000	66000	3469	12.38	12.59	Z SHELL
9	LOWER RF SHELF ASSEMBLY	1356429-1	BE/SR-200E	50000	70000	7068	4.66	6.07	Z SHELL
10	UPPER RF SHELF ASSEMBLY	1356409-1	BE/SR-200E	50000	70000	4429	8.03	10.29	Z SHELL
11	LOWER RIGHT FRONT SUPT PANEL	1331447-1	ALUM/6061-T651	35000	42000	1209	22.16	23.81	Z BEAM
12	LOWER RIGHT PANEL	1331650-1	ALUM/6061-T6	35000	42000	3567	6.85	7.41	Z SHELL
13	UPPER RIGHT FRONT SUPT PANEL	1331390-1	ALUM/6061-T651	35000	42000	3010	8.30	8.97	Z BEAM
14	UPPER RIGHT PANEL	1331651-1	ALUM/6061-T6	35000	42000	2076	12.49	13.45	Z SHELL
15	TOP PANEL	1331648-1	ALUM/6061-T6	35000	42000	2343	10.95	11.80	Z SHELL
16	LEFT PANEL	1331640-1	ALUM/6061-T6	35000	42000	2278	11.29	12.17	Z SHELL
17	LOWER LEFT WARMLOAD SHIELD	1331445-1	ALUM/6061-T6	35000	42000	3888	6.20	6.72	Z BEAM
18	LOWER RIGHT WARMLOAD SHIELD	1331405-1	ALUM/6061-T6	35000	42000	3058	8.16	8.81	Z BEAM
19	UPPER LEFT WARMLOAD SHIELD	1331647-1	ALUM/6061-T6	35000	42000	3384	7.27	7.87	Z BEAM
20	UPPER RIGHT WARMLOAD SHIELD	1331646-1	ALUM/6061-T6	35000	42000	2046	12.69	13.66	Z BEAM
21	LOWER CARD CAGE ASSEMBLY	1331600-1	ALUM/6061-T6	35000	42000	7342	2.81	3.09	Z SHELL
22	LOWER CARD	1331600-1	REINF PLASTIC	24000	40000	665	27.87	41.96	Z SHELL
23	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T6	35000	42000	7652	2.66	2.92	Z BEAM
23a	UPPER CARD CAGE ASSEMBLY	1331162-1	ALUM/6061-T4	16000	30000	3826	2.35	4.60	Z BEAM
24	UPPER CARD	1331162-1	REINF PLASTIC	24000	40000	363	51.89	77.71	Z SHELL
25	LOWER CALIBRATION SOURCE ASSY	1331380-1	ALUM/6061-T6	35000	42000	4817	4.81	5.23	Z SHELL
26	UPPER CALIBRATION SOURCE ASSY	1331380-2	ALUM/6061-T6	35000	42000	4348	5.44	5.90	Z SHELL
27	BEAM SUPPORT	1331406-1	ALUM/7075-T6	66000	75000	2793	17.90	18.18	Z BEAM
28	LOWER REFLECTOR	1355777-1	ALUM/7075-T6	66000	75000	7165	6.37	6.48	Z BEAM
29	POWER CONTROL/MONITOR, BKT	1356790-1	ALUM/6061-T6	35000	42000	2135	12.11	13.05	Z SHELL
30	POWER CNTL/MNTR ASSY, PWB	1356962-1	REINF PLASTIC	24000	40000	1233	14.57	22.17	Z SHELL
31	PWR CNTL/MNTR ASSY, STANDOFFS	1356962-1	CRES 1/4 HARD	44000	122000	7051	3.99	11.36	Z BEAM
32	RADIATOR PANEL	1331511-1	ALUM/6061-T6	35000	42000	1739	15.10	16.25	Z SHELL
33	SIDEMOUNT ASSEMBLY	1331552-1	ALUM/6061-T6	35000	42000	10670	1.62	1.81	Z SHELL
34	UPPER REFLECTOR	1355777-1	ALUM/6061-T6	35000	42000	3768	6.43	6.96	Z SHELL

¹MAXIMUM PRINCIPAL STRESS BASED ON ASSUMED IN PHASE COMPONENT STRESSES

Table XV METSAT AMSU-A1 Rattlespace Summary

SUMMARY - RATTLESPLACE OF THE LOWER CARD CAGE CIRCUIT CARDS				
CASE	GRID	GRID	LOAD	$3\Sigma \Delta X$
1	4804	4844	X	0.00102
2	4803	4843	Y	0.00528
3	4808	4848	Z	0.00170
MINIMUM GAP				0.037
SUMMARY - RATTLESPLACE OF THE UPPER CARD CAGE CIRCUIT CARDS				
CASE	GRID	GRID	LOAD	$3\Sigma \Delta X$
1	11308	11875	X	0.00235
2	11316	11883	Y	0.00059
3	11316	11883	Z	0.00113
MINIMUM GAP				0.137

Table XVI METSAT AMSU-A1 Sidemount Mounting Bolts Margins of Safety
Gy=-20.6 Gx=-21.2

LOAD CASE	BOLT GRID	APPLIED TENSION T2 (LB)*	TENSILE FS	PRELOAD TENSION (LB)	TOTAL TENSION (LB)	TENSILE ALLOW (LB)	APPLIED SHEAR T1 (LB)	APPLIED SHEAR T3 (LB)	SHEAR FS	TOTAL SHEAR (LB)	SHEAR ALLOW (LB)	MARGIN OF SAFETY
1	7240	231	1.4	1315	1517	3200	-12	-193	1.4	271	1735	1.07
	7255	67	1.4	1315	1373	3200	170	-41	1.4	245	1735	1.29
	7446	5	1.4	1315	1319	3200	77	-11	1.4	108	1735	1.42
	7462	207	1.4	1315	1495	3200	34	106	1.4	156	1735	1.13
	7497	56	1.4	1315	1364	3200	87	35	1.4	132	1735	1.34
	7507	313	1.4	1315	1588	3200	152	123	1.4	273	1735	0.98
	7555	71	1.4	1315	1377	3200	31	-42	1.4	73	1735	1.32
	7745	25	1.4	1315	1336	3200	49	-39	1.4	88	1735	1.39
	7840	63	1.4	1315	1370	3200	171	-42	1.4	247	1735	1.29
	7933	45	1.4	1315	1354	3200	69	-13	1.4	99	1735	1.36
	8313	408	1.4	1315	1671	3200	429	144	1.4	634	1735	0.68
	9576	439	1.4	1315	1699	3200	257	-56	1.4	369	1735	0.83
	9834	462	1.4	1315	1719	3200	488	21	1.4	684	1735	0.61
	10271	424	1.4	1315	1685	3200	54	190	1.4	277	1735	0.87
	10703	328	1.4	1315	1601	3200	182	-91	1.4	284	1735	0.96
	10748	53	1.4	1315	1361	3200	36	-92	1.4	138	1735	1.34
*ALL TENSILE LOADS MADE POSITIVE TO ACCOUNT FOR LOAD CASES 1 AND 6 w/ Kb/Kb+Km = .623845												

Table XVII METSAT AMSU-A1 Sidemount Mounting Bolts Margins of Safety
Gy=-20.6 Gx=21.2

LOAD CASE	BOLT GRID	APPLIED TENSION T2 (LB)*	TENSILE FS	PRELOAD TENSION (LB)	TOTAL TENSION (LB)	TENSILE ALLOW (LB)	APPLIED SHEAR T1 (LB)	APPLIED SHEAR T3 (LB)	SHEAR FS	TOTAL SHEAR (LB)	SHEAR ALLOW (LB)	MARGIN OF SAFETY
2	7240	432	1.4	1315	1693	3200		258	1.4	362	1735	0.84
	7255	194	1.4	1315	1485	3200	-226	-15	1.4	317	1735	1.09
	7446	9	1.4	1315	1323	3200	-185	-2	1.4	259	1735	1.36
	7462	56	1.4	1315	1364	3200	-159	-134	1.4	291	1735	1.28
	7497	26	1.4	1315	1338	3200	-142	-40	1.4	207	1735	1.36
	7507	245	1.4	1315	1529	3200	-129	-77	1.4	210	1735	1.08
	7555	509	1.4	1315	1760	3200	-24	50	1.4	77	1735	0.82
	7745	89	1.4	1315	1392	3200	-54	-107	1.4	168	1735	1.29
	7840	607	1.4	1315	1845	3200	72	121	1.4	198	1735	0.73
	7933	23	1.4	1315	1335	3200	-74	8	1.4	105	1735	1.39
	8313	900	1.4	1315	2101	3200	-421	57	1.4	595	1735	0.43
	9576	115	1.4	1315	1415	3200	-193	36	1.4	275	1735	1.21
	9834	177	1.4	1315	1469	3200	-390	9	1.4	546	1735	0.92
	10271	22	1.4	1315	1334	3200	-243	-129	1.4	385	1735	1.25
10703	19	1.4	1315	1332	3200	-124	-32	1.4	179	1735	1.39	
10748	82	1.4	1315	1386	3200	-11	-10	1.4	21	1735	1.31	
*ALL TENSILE LOADS MADE POSITIVE TO ACCOUNT FOR LOAD CASES 2 AND 5 w/ Kb/Kb+Km = .623845												

Table XVIII METSAT AMSU-A1 Sidemount Mounting Bolts Margins of Safety
Gy=-20.6 Gz=-19.6

LOAD CASE	BOLT GRID	APPLIED TENSION T2 (LB)*	TENSILE FS	PRELOAD TENSION (LB)	TOTAL TENSION (LB)	TENSILE ALLOW (LB)	APPLIED SHEAR T1 (LB)	APPLIED SHEAR T3 (LB)	SHEAR FS	TOTAL SHEAR (LB)	SHEAR ALLOW (LB)	MARGIN OF SAFETY
3	7240	425	1.4	1315	1686	3200	150	308	1.4	480	1735	0.78
	7255	369	1.4	1315	1637	3200	-171	130	1.4	301	1735	0.92
	7446	56	1.4	1315	1364	3200	1	82	1.4	114	1735	1.34
	7462	497	1.4	1315	1749	3200	-165	278	1.4	453	1735	0.74
	7497	93	1.4	1315	1396	3200	0	75	1.4	105	1735	1.29
	7507	322	1.4	1315	1596	3200	112	155	1.4	268	1735	0.98
	7555	230	1.4	1315	1516	3200	15	90	1.4	128	1735	1.11
	7745	144	1.4	1315	1441	3200	-34	79	1.4	121	1735	1.22
	7840	281	1.4	1315	1560	3200	62	273	1.4	392	1735	0.96
	7933	24	1.4	1315	1336	3200	-26	10	1.4	38	1735	1.39
	8313	107	1.4	1315	1408	3200	-53	135	1.4	203	1735	1.25
	9576	172	1.4	1315	1466	3200	-101	105	1.4	204	1735	1.17
	9834	185	1.4	1315	1477	3200	146	17	1.4	206	1735	1.15
	10271	218	1.4	1315	1505	3200	-28	236	1.4	332	1735	1.06
	10703	1	1.4	1315	1316	3200	-1	46	1.4	64	1735	1.43
	10748	105	1.4	1315	1407	3200	-28	82	1.4	122	1735	1.27
*ALL TENSILE LOADS MADE POSITIVE TO ACCOUNT FOR LOAD CASES 3 AND 8 w/ Kb/Kb+Km = .623845												

Table XIX METSAT AMSU-A1 Sidemount Mounting Bolts Margins of Safety
Gy=-20.6 Gz=19.6

LOAD CASE	BOLT GRID	APPLIED TENSION T2 (LB)*	TENSILE FS	PRELOAD TENSION (LB)	TOTAL TENSION (LB)	TENSILE ALLOW (LB)	APPLIED SHEAR T1 (LB)	APPLIED SHEAR T3 (LB)	SHEAR FS	TOTAL SHEAR (LB)	SHEAR ALLOW (LB)	MARGIN OF SAFETY
4	7240	223	1.4	1315	1510	3200	-140	-244	1.4	393	1735	1.02
	7255	242	1.4	1315	1526	3200	116	-186	1.4	307	1735	1.05
	7446	41	1.4	1315	1351	3200	-109	-95	1.4	202	1735	1.34
	7462	346	1.4	1315	1617	3200	41	-307	1.4	433	1735	0.87
	7497	63	1.4	1315	1370	3200	-55	-80	1.4	136	1735	1.33
	7507	254	1.4	1315	1537	3200	-90	-110	1.4	198	1735	1.07
	7555	209	1.4	1315	1497	3200	-7	-83	1.4	117	1735	1.13
	7745	31	1.4	1315	1342	3200	30	-225	1.4	317	1735	1.29
	7840	389	1.4	1315	1655	3200	182	-193	1.4	372	1735	0.87
	7933	43	1.4	1315	1353	3200	21	-14	1.4	35	1735	1.37
	8313	1415	1.4	1315	2551	3200	61	66	1.4	126	1735	0.25
	9576	726	1.4	1315	1949	3200	165	-125	1.4	290	1735	0.63
	9834	100	1.4	1315	1402	3200	-48	13	1.4	70	1735	1.28
	10271	228	1.4	1315	1514	3200	-160	-174	1.4	332	1735	1.05
	10703	310	1.4	1315	1585	3200	59	-168	1.4	250	1735	0.99
	10748	240	1.4	1315	1524	3200	53	-184	1.4	268	1735	1.06
*ALL TENSILE LOADS MADE POSITIVE TO ACCOUNT FOR LOAD CASES 4 AND 7 w/ Kb/Kb+Km = .623845												

Table XX METSAT AMSU-A1 Sidemount Shear Pins
Shear Tearout Margins of Safety

LOAD CASE	BOLT GRID	COMP	APPLIED SHEAR T1 (LB)	APPLIED SHEAR T3 (LB)	TOTAL SHEAR (LB)	SHEAR TEAROUT (PSI)	FS	SHEAR Fsu (PSI)	MS
1	7855	SIDEMT	1796	43	1796	14721	1.4	27000	0.31
2	7855	SIDEMT	-1672	-43	1673	13710	1.4	27000	0.41
3	7855	SIDEMT	13	1913	1913	15681	1.4	27000	0.23
4	7855	SIDEMT	111	-1913	1916	15707	1.4	27000	0.23
*ALL SHEAR LOAD REACTED AT SHEAR PINS									
LOAD CASE	BOLT GRID	COMP	APPLIED LOAD T1 (LB)	APPLIED LOAD T3 (LB)	TOTAL LOAD (LB)	BEARING STRESS (PSI)	FS	ALLOW Fcy/Ftu (PSI)	MS
1	7855	SIDEMT	1796	43	1796	15281	1.25	35000	0.83
	7855	SIDEMT	1796	43	1796	15281	1.4	42000	0.96
2	7855	SIDEMT	-1672	-43	1673	14232	1.25	35000	0.97
	7855	SIDEMT	-1672	-43	1673	14232	1.4	42000	1.11
3	7855	SIDEMT	13	1913	1913	16278	1.25	35000	0.72
	7855	SIDEMT	13	1913	1913	16278	1.4	42000	0.84
4	7855	SIDEMT	111	-1913	1916	16304	1.25	35000	0.72
	7855	SIDEMT	111	-1913	1916	16304	1.4	42000	0.84
*ALL SHEAR LOAD REACTED AT SHEAR PINS									

Table XXI METSAT AMSU-A1 Sidemount Mounting Bolts Margins of Safety
Gy=20.6 Gx=21.2 (Shear at Pins Only)

LOAD CASE	BOLT GRID	APPLIED TENSION T2 (LB)*	TENSILE FS	PRELOAD TENSION (LB)	TOTAL TENSION (LB)	TENSILE ALLOW (LB)	MARGIN OF SAFETY
1	7240	99	1.4	1315	1401	3200	1.28
	7255	38	1.4	1315	1348	3200	1.37
	7446	6	1.4	1315	1320	3200	1.42
	7462	226	1.4	1315	1512	3200	1.12
	7497	63	1.4	1315	1370	3200	1.34
	7507	252	1.4	1315	1535	3200	1.09
	7555	167	1.4	1315	1461	3200	1.19
	7745	14	1.4	1315	1327	3200	1.41
	7840	124	1.4	1315	1424	3200	1.25
	7933	57	1.4	1315	1365	3200	1.34
	8313	368	1.4	1315	1636	3200	0.96
	9576	438	1.4	1315	1697	3200	0.89
	9834	294	1.4	1315	1572	3200	1.04
	10271	644	1.4	1315	1877	3200	0.70
	10703	438	1.4	1315	1698	3200	0.89
	10748	90	1.4	1315	1393	3200	1.30
*ALL TENSILE LOADS MADE POSITIVE TO ACCOUNT FOR LOAD CASES 1 AND 6 w/ Kb/Kb+Km = .623845							
*ALL SHEAR LOAD REACTED AT SHEAR PINS							

Table XXII METSAT AMSU-A1 Sidemount Mounting Bolts Margins of Safety
Gy=-20.6 Gx=-21.2 (Shear at Pins Only)

LOAD CASE	BOLT GRID	APPLIED TENSION T2 (LB)*	TENSILE FS	PRELOAD TENSION (LB)	TOTAL TENSION (LB)	TENSILE ALLOW (LB)	MARGIN OF SAFETY
2	7240	238	1.4	1315	1523	3200	1.10
	7255	174	1.4	1315	1467	3200	1.18
	7446	9	1.4	1315	1323	3200	1.42
	7462	28	1.4	1315	1340	3200	1.39
	7497	18	1.4	1315	1330	3200	1.41
	7507	141	1.4	1315	1438	3200	1.23
	7555	620	1.4	1315	1857	3200	0.72
	7745	69	1.4	1315	1375	3200	1.33
	7840	768	1.4	1315	1986	3200	0.61
	7933	11	1.4	1315	1325	3200	1.42
	8313	1048	1.4	1315	2230	3200	0.44
	9576	119	1.4	1315	1419	3200	1.25
	9834	8	1.4	1315	1322	3200	1.42
	10271	172	1.4	1315	1465	3200	1.18
	10703	227	1.4	1315	1513	3200	1.11
	10748	2	1.4	1315	1317	3200	1.43
*ALL TENSILE LOADS MADE POSITIVE TO ACCOUNT FOR LOAD CASES 2 AND 5 w/ Kb/Kb+Km = .623845							
*ALL SHEAR LOAD REACTED AT SHEAR PINS							

Table XXIII METSAT AMSU-A1 Sidemount Mounting Bolts Margins of Safety
Gy=-20.6 Gz=-19.6 (Shear at Pins Only)

LOAD CASE	BOLT GRID	APPLIED TENSION T2 (LB)*	TENSILE FS	PRELOAD TENSION (LB)	TOTAL TENSION (LB)	TENSILE ALLOW (LB)	MARGIN OF SAFETY
3	7240	270	1.4	1315	1551	3200	1.06
	7255	391	1.4	1315	1657	3200	0.93
	7446	54	1.4	1315	1362	3200	1.35
	7462	462	1.4	1315	1719	3200	0.86
	7497	95	1.4	1315	1398	3200	1.29
	7507	301	1.4	1315	1578	3200	1.03
	7555	337	1.4	1315	1610	3200	0.99
	7745	234	1.4	1315	1520	3200	1.11
	7840	368	1.4	1315	1636	3200	0.96
	7933	61	1.4	1315	1368	3200	1.34
	8313	334	1.4	1315	1607	3200	0.99
	9576	53	1.4	1315	1361	3200	1.35
	9834	237	1.4	1315	1522	3200	1.10
	10271	322	1.4	1315	1596	3200	1.00
	10703	145	1.4	1315	1441	3200	1.22
	10748	143	1.4	1315	1440	3200	1.22
*ALL TENSILE LOADS MADE POSITIVE TO ACCOUNT FOR LOAD CASES 3 AND 8 w/ Kb/Kb+Km = .623845							
*ALL SHEAR LOAD REACTED AT SHEAR PINS							

Table XXIV METSAT AMSU-A1 Sidemount Mounting Bolts Margins of Safety
Gy=-20.6 Gz=-19.6 (Shear at Pins Only)

LOAD CASE	BOLT GRID	APPLIED TENSION T2 (LB)*	TENSILE FS	PRELOAD TENSION (LB)	TOTAL TENSION (LB)	TENSILE ALLOW (LB)	MARGIN OF SAFETY
4	7240	131	1.4	1315	1429	3200	1.24
	7255	256	1.4	1315	1538	3200	1.08
	7446	39	1.4	1315	1349	3200	1.37
	7462	265	1.4	1315	1546	3200	1.07
	7497	51	1.4	1315	1359	3200	1.35
	7507	190	1.4	1315	1481	3200	1.16
	7555	116	1.4	1315	1416	3200	1.26
	7745	152	1.4	1315	1447	3200	1.21
	7840	276	1.4	1315	1556	3200	1.06
	7933	15	1.4	1315	1328	3200	1.41
	8313	1749	1.4	1315	2843	3200	0.13
	9576	610	1.4	1315	1848	3200	0.73
	9834	64	1.4	1315	1371	3200	1.33
	10271	150	1.4	1315	1446	3200	1.21
	10703	356	1.4	1315	1626	3200	0.97
	10748	235	1.4	1315	1520	3200	1.11
*ALL TENSILE LOADS MADE POSITIVE TO ACCOUNT FOR LOAD CASES 4 AND 7 w/ Kb/Kb+Km = .623845							
*ALL SHEAR LOAD REACTED AT SHEAR PINS							

Table XXV METSAT AMSU-A1 Sidemount Mounting Bolts Member Compression
Margins of Safety

LOAD CASE	MEMBER 1	MEMBER 2	APPLIED LOAD (LB)	BEARING AREA (IN2)	BEARING STRESS (PSI)	FS	ALLOW Ftu (PSI)	MS			
PRELOAD	BOLT	WASHER	1315	0.036	36053	1.40	160000	2.17			
	WASHER	.435 ISO	1315	0.112	11781	1.40	60000	2.64			
	.435 ISO	BASEPL	1315	0.083	15928	1.25	35000	0.76			
	.435 ISO	BASEPL	1315	0.083	15928	1.40	42000	0.88			
	BASEPL	.590 ISO	1315	0.156	8433	1.25	35000	2.32			
	BASEPL	.590 ISO	1315	0.156	8433	1.40	42000	2.56			
	.590 ISO	SPCCR	1315	0.157	8377	1.40	60000	4.12			
METSAT AMSU-A1 SIDEMOUNT MOUNTING BOLTS THREAD SHEAR MARGINS OF SAFETY											
LOAD CASE	INT/EXT THREAD	GRID	PRELOAD	APPLIED LOAD LB	APPLIED LOAD FS	JOINT STIFF FACTOR	TOTAL LOAD LB	SHEAR AREA SQ IN	THREAD SHEAR PSI	ALLOW Fsu PSI	MARGIN OF SAFETY
1	EXTERNAL .190-32UNF	10271	1315	644	1.4	0.623845	1877.247	0.05065	37063	96000	1.59
	A-286 SCR										
2	EXTERNAL .190-32UNF	8313	1315	1048	1.4	0.623845	2229.953	0.05065	44027	96000	1.18
	A-286 SCR										
3	EXTERNAL .190-32UNF	7462	1315	462	1.4	0.623845	1718.722	0.05065	33933	96000	1.83
	A-286 SCR										
4	EXTERNAL .190-32UNF	8313	1315	1749	1.4	0.623845	2842.675	0.05065	56124	96000	0.71
	A-286 SCR										
*ALL SHEAR LOAD REACTED AT SHEAR PINS											

Table XXVI METSAT AMSU-A1 Sidemount Mounting Bolts Shear Tearout
Margins of Safety (Shear at Pins Only)

METSAT AMSU-A1 SIDEMOUNT MOUNTING BOLTS SHEAR TEAROUT MARGINS OF SAFETY*										
LOAD CASE	BOLT GRID	COMP	APPLIED SHEAR T1 (LB)	APPLIED SHEAR T3 (LB)	TOTAL SHEAR (LB)	SHEAR TEAROUT (PSI)	FS	SHEAR Fsu (PSI)	MS	
1	9834	SIDEMT	488	21	488	5991	1.4	27000	2.22	
2	8313	SIDEMT	-421	57	425	5213	1.4	27000	2.70	
3	7240	SIDEMT	150	308	343	4206	1.4	27000	3.59	
4	7462	SIDEMT	41	-307	309	3795	1.4	27000	4.08	
*W/SHEAR AT PINS AND BOLTS										
METSAT AMSU-A1 SIDEMOUNT MOUNTING BOLTS BEARING MARGINS OF SAFETY*										
LOAD CASE	BOLT GRID	COMP	APPLIED LOAD T1 (LB)	APPLIED LOAD T3 (LB)	TOTAL LOAD (LB)	BEARING STRESS (PSI)	FS	ALLOW Fcy/Ftu (PSI)	MS	
1		SIDEMT	488	21	488	3310	1.25	35000	7.46	
		SIDEMT	488	21	488	3310	1.4	42000	8.06	
		.590 ISO	488	21	488	3310	1.4	35000	6.55	
2		SIDEMT	-421	57	425	2880	1.25	35000	8.72	
		SIDEMT	-421	57	425	2880	1.4	35000	7.68	
		.590 ISO	-421	57	425	2880	1.4	35000	7.68	
3		SIDEMT	150	308	343	2324	1.25	35000	11.05	
		SIDEMT	150	308	343	2324	1.4	35000	9.76	
		.590 ISO	150	308	343	2324	1.4	35000	9.76	
4		SIDEMT	41	-307	309	2097	1.25	35000	12.35	
		SIDEMT	41	-307	309	2097	1.4	35000	10.92	
		.590 ISO	41	-307	309	2097	1.4	35000	10.92	
*W/SHEAR AT PINS AND SCREWS										

Table XXVII METSAT AMSU-A1 Panel Flange Bending Stress Summary
Random Vibration 8.8 grms, Upper Front Panel

UPPER FRONT PANEL																
LOWER FLANGE																
FEM ELEM. NO.	LOAD CASE	FORCE Fx (lb/in)	MOM Mx (in-lb/in)	SHR Vx (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)	Fty/Ftu (psi)	FS	MS
1434	Y	5.482	0.006	0.016	0.563	0.312	0.313	0.040	10.994	5.512	1.725	0.016	19412	66000	1.25	1.72
1470	Z	10.273	0.066	0.060	0.750	0.312	0.313	0.040	20.868	10.595	3.316	0.060	37311	75000	1.4	1.76
1538	Z	17.053	0.093	0.139	0.563	0.312	0.313	0.050	34.598	17.545	5.492	0.139	39549	66000	1.25	0.42
1550	Z	5.383	0.041	0.034	0.563	0.312	0.313	0.040	10.941	5.558	1.740	0.034	19573	75000	1.4	0.44
* MINIMUM MEASURED VALUE USED FOR t AT ELEMENT 1538																
RIGHT FLANGE																
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)	Fty/Ftu (psi)	FS	MS
1438	X	3.050	0.177	1.411	0.469	0.312	0.313	0.040	8.768	5.718	1.790	1.411	20239	66000	1.25	1.61
1439	X	2.421	0.186	3.114	0.360	0.312	0.313	0.040	9.005	6.584	2.061	3.114	23418	75000	1.4	1.65
1439	Y	2.788	0.159	1.887	0.360	0.312	0.313	0.040	8.242	5.454	1.707	1.887	19348	75000	1.4	1.25
1441	Y	2.857	0.022	0.214	0.219	0.312	0.313	0.040	5.925	3.068	0.960	0.214	10818	75000	1.4	1.29
UPPER FLANGE																
FEM ELEM. NO.	LOAD CASE	FORCE Fx (lb/in)	MOM Mx (in-lb/in)	SHR Vx (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)	Fty/Ftu (psi)	FS	MS
1491	Z	2.869	0.138	0.312	0.551	0.312	0.313	0.040	6.719	3.850	1.205	0.312	13580	66000	1.25	2.89
														75000	1.4	2.94

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LOWER AFT PANEL																
RIGHT FLANGE																
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)	Fly/Ftu (psi)	FS	MS
429	X	4.888	0.032	0.099	0.219	0.312	0.313	0.070	9.932	5.044	1.579	0.099	5804	58000	1.25	7.00
														66000	1.4	7.12
LOWER FLANGE																
FEM ELEM. NO.	LOAD CASE	FORCE Fx (lb/in)	MOM Mx (in-lb/in)	SHR Vx (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)	Fly/Ftu (psi)	FS	MS
340	X	6.999	0.110	0.151	0.597	0.312	0.313	0.125	14.615	7.616	2.384	0.151	2750	58000	1.25	15.87
														66000	1.4	16.14
340	Z	12.940	0.126	0.074	0.597	0.312	0.313	0.125	26.382	13.442	4.207	0.074	4849	58000	1.25	8.57
														66000	1.4	8.72
UPPER FLANGE																
FEM ELEM. NO.	LOAD CASE	FORCE Fx (lb/in)	MOM Mx (in-lb/in)	SHR Vx (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)	Fly/Ftu (psi)	FS	MS
429	X	18.607	0.132	0.452	0.338	0.312	0.313	0.070	38.064	19.457	6.090	0.452	22391	58000	1.25	1.07
														66000	1.4	1.11
339	X	19.353	0.071	0.152	0.353	0.312	0.313	0.070	39.042	19.689	6.163	0.152	22645	58000	1.25	1.05
														66000	1.4	1.08

Table XXXI METSAT AMSU-A1 Panel Flange Bending Stress Summary
Random Vibration 8.8 grms, Upper Motor Mount Panel

UPPER MOTOR MOUNT PANEL													
LOWER FLANGE													
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)
2271	X	6.416	0.030	0.104	0.638	0.312	0.438	0.050	11.206	4.790	2.098	0.104	15113
													66000
													75000
													1.25
													1.4
													2.49
													2.54
2272	X	7.184	0.031	0.055	0.584	0.312	0.438	0.050	12.445	5.261	2.305	0.055	16596
													66000
													75000
													1.25
													1.4
													2.18
													2.23
2279	Z	7.999	0.033	0.114	0.563	0.312	0.438	0.050	13.919	5.920	2.593	0.114	18675
													66000
													75000
													1.25
													1.4
													1.83
													1.87
RIGHT FLANGE													
FEM ELEM. NO.	LOAD CASE	FORCE Fx (lb/in)	MOM Mx (in-lb/in)	SHR Vx (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)
2271	X	8.301	0.036	0.087	0.424	0.312	0.438	0.050	14.380	6.079	2.663	0.087	19177
													66000
													75000
													1.25
													1.4
													1.75
													1.79
2280	X	7.031	0.054	0.095	0.298	0.312	0.438	0.050	12.227	5.196	2.276	0.095	16392
													66000
													75000
													1.25
													1.4
													2.22
													2.27
UPPER FLANGE													
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)
2364	X	5.773	0.086	0.150	0.380	0.312	0.438	0.050	10.212	4.439	1.944	0.150	14007
													66000
													75000
													1.25
													1.4
													2.77
													2.82

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Table XXXIII METSAT AMSU-A1 Panel Flange Bending Stress Summary
Random Vibration 8.8 grms, Lower Front Panel

LOWER FRONT PANEL													
LOWER FLANGE													
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)
													Fty/Ftu (psi)
													FS
													MS
1251	Z	15.174	0.193	0.145	0.497	0.313	0.312	0.090	31.246	16.072	5.015	0.145	11148
													35000
													42000
													1.25
													1.4
													1.51
1261	Y	14.007	0.131	0.130	0.754	0.313	0.312	0.090	28.793	14.786	4.613	0.130	10256
													35000
													42000
													1.25
													1.4
													1.73
1261	Z	18.308	0.236	0.140	0.754	0.313	0.312	0.090	37.769	19.461	6.072	0.140	13498
													35000
													42000
													1.25
													1.4
													1.07
1289	Z	15.906	0.149	0.116	0.754	0.313	0.312	0.090	32.621	16.715	5.215	0.116	11593
													35000
													42000
													1.25
													1.4
													1.59
1333	X	9.999	0.039	1.183	0.666	0.313	0.312	0.090	22.678	12.679	3.956	1.183	8830
													35000
													42000
													1.25
													1.4
													2.17
													2.40
RIGHT FLANGE													
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)
													Fty/Ftu (psi)
													FS
													MS
1333	X	4.373	0.073	1.409	0.296	0.313	0.312	0.040	10.331	5.958	1.859	1.409	21017
													35000
													42000
													1.25
													1.4
													0.33
1338	X	4.083	0.137	0.514	0.312	0.313	0.312	0.040	9.132	5.049	1.575	0.514	17761
													35000
													42000
													1.25
													1.4
													0.58
													0.69

Table XXXIII METSAT AMSU-A1 Panel Flange Bending Stress Summary
Random Vibration 8.8 grms, Lower Front Panel (Continued)

LOWER FRONT PANEL																
UPPER FLANGE																
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	St (psi)	Fty/Ftu (psi)	FS	MS
1240	Y	11.806	0.012	0.022	0.368	0.313	0.312	0.060	23.714	11.908	3.715	0.022	18578	35000	1.25	0.51
														42000	1.4	0.61
1240	X	12.986	0.035	0.038	0.368	0.313	0.312	0.060	26.171	13.185	4.114	0.038	20570	35000	1.25	0.36
														42000	1.4	0.46

Table XXXIV METSAT AMSU-A1 Panel Flange Bending Stress Summary
Random Vibration 8.8 grms, Upper Aft Panel

UPPER AFT PANEL													
LOWER FLANGE													
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)
5778	X	16.015	9.308	4.722	0.306	0.562	0.188	0.110	121.09	105.07	19.753	4.722	29514
													58000 1.25 0.57
													66000 1.4 0.60
5781	X	12.473	9.596	5.073	0.354	0.562	0.188	0.110	110.35	97.88	18.402	5.073	27513
													58000 1.25 0.69
													66000 1.4 0.71
5785	X	15.262	9.746	5.202	0.306	0.562	0.188	0.110	121.19	105.93	19.915	5.202	29768
													58000 1.25 0.56
													66000 1.4 0.58
UPPER FLANGE													
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)
6980	X	6.431	0.261	0.242	0.245	0.562	0.188	0.060	27.359	20.928	3.935	0.242	19685
													58000 1.25 1.36
													66000 1.4 1.39
RIGHT FLANGE													
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)
6963	X	8.303	0.505	0.086	0.655	0.562	0.188	0.060	36.109	27.806	5.228	0.086	26142
													58000 1.25 0.77
													66000 1.4 0.80
6856	X	1.406	3.552	12.510	0.362	0.562	0.188	0.145	48.591	47.185	8.871	12.510	7853
													58000 1.25 4.91
													66000 1.4 5.00
W/LOCAL THICKNESS													

Table XXXIV METSAT AMSU-A1 Panel Flange Bending Stress Summary
Random Vibration 8.8 grms, Upper Aft Panel (Continued)

UPPER AFT PANEL																
LEFT FLANGE																
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)	Fty/Ftu (psi)	FS	MS
6944	X	13.944	9.470	13.208	0.267	0.562	0.188	0.145	124.76	110.81	20.833	13.208	18109	58000	1.25	1.56
W/LOCAL THICKNESS																
6944	Z	16.833	2.997	1.658	0.267	0.562	0.188	0.145	85.449	68.616	12.900	1.658	11078	58000	1.25	3.19
W/LOCAL THICKNESS																
6945	X	26.137	6.928	4.990	0.468	0.562	0.188	0.145	153.54	127.41	23.952	4.990	20609	58000	1.25	1.25
W/LOCAL THICKNESS																
6946	X	53.813	9.059	2.067	0.312	0.562	0.188	0.145	266.30	212.48	39.947	2.067	34242	58000	1.25	0.36
W/LOCAL THICKNESS																
6925	X	17.269	0.649	0.166	0.382	0.562	0.188	0.075	72.68	55.41	10.418	0.166	33343	58000	1.25	0.39
W/LOCAL THICKNESS																
6926	Z	11.986	0.237	0.105	0.362	0.562	0.188	0.075	49.28	37.29	7.011	0.105	22440	58000	1.25	1.07
W/LOCAL THICKNESS																

Table XXXV METSAT AMSU-A1 Panel Flange Bending Stress Summary
Random Vibration 8.8 grms, Upper Right Panel

UPPER RIGHT PANEL									
LOWER FLANGE									
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	3 σ St (psi)
									V1 (lb/in)
									M1 (in-lb/in)
									F2 (lb/in)
									F1 (lb/in)
									Fty/Ftu (psi)
									FS
									MS
1648	X	11.633	0.028	0.283	0.395	0.312	0.188	0.060	18861
									35000
									42000
									1.25
									1.4
									0.48
									0.59

LOWER RIGHT FRONT SUPPORT PANEL																	
LOWER FLANGE																	
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)	Fly/Ftu (psi)	FS	MS	
2201	X	3.326	0.034	0.067	0.566	0.312	0.313	0.045	6.871	3.545	1.110	0.067	9868	35000	1.25	1.84	
														42000	1.4	2.04	
AFT FLANGE																	
FEM ELEM. NO.	LOAD CASE	FORCE Fy (lb/in)	MOM My (in-lb/in)	SHR Vy (lb/in)	h (in)	d (in)	d' (in)	t (in)	F1 (lb/in)	F2 (lb/in)	M1 (in-lb/in)	V1 (lb/in)	3σ St (psi)	Fly/Ftu (psi)	FS	MS	
2204	X	4.940	0.057	0.031	0.870	0.312	0.313	0.045	10.132	5.192	1.625	0.031	14449	35000	1.25	0.94	
														42000	1.4	1.08	

Table XXXVII METSAT AMSU-A1 Panel Flange Attachment Screws Stress Summary
Random Vibration 8.8 grms, Upper Front Panel

UPPER FRONT PANEL														
LOWER FLANGE - BOLT STRESSES (NAS1352N06-6)														
FEM ELEM. NO.	LOAD CASE	F1 (lb/in)	V1 (lb/in)	NO. BOLTS	L (in)	F _B (lb)	V _B (lb)	3σ F _B	3σ V _B	F _{tu} ' (lb)	FS	F _{su} ' (lb)	FS	MS
1538	Z	34.598	0.139	7	10.575	52.268	0.210	156.81	0.630	1458	1.4	875	1.4	5.6
RIGHT FLANGE - BOLT STRESSES (NAS1352N06-6)														
FEM ELEM. NO.	LOAD CASE	F1 (lb/in)	V1 (lb/in)	NO. BOLTS	L (in)	F _B (lb)	V _B (lb)	3σ F _B	3σ V _B	F _{tu} ' (lb)	FS	F _{su} ' (lb)	FS	MS
1439	X	9.005	3.114	8	11.500	12.945	4.476	38.83	13.4291	1458	1.4	875	1.4	24
UPPER FLANGE - BOLT STRESSES (NAS1352N06-6)														
FEM ELEM. NO.	LOAD CASE	F1 (lb/in)	V1 (lb/in)	NO. BOLTS	L (in)	F _B (lb)	V _B (lb)	3σ F _B	3σ V _B	F _{tu} ' (lb)	FS	F _{su} ' (lb)	FS	MS
1491	Z	6.719	0.312	9	10.447	7.799	0.362	23.40	1.086	1458	1.4	875	1.4	43

Table XXXVIII METSAT AMSU-A1 Panel Flange Attachment Screws Stress Summary
Random Vibration 8.8 grms, Upper Right Front Support Panel

UPPER RIGHT FRONT SUPPORT PANEL													
AFT FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM ELEM. NO.	LOAD CASE	F1 (lb/in)	V1 (lb/in)	NO. BOLTS	L (in)	F _B (lb)	V _B (lb)	3 σ F _B	3 σ V _B	F _{tu} ' (lb)	FS	F _{su} ' (lb)	FS
2241	X	23.709	0.033	3	6.200	48.998	0.068	146.99	0.205	1458	1.4	875	1.4
LOWER FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM ELEM. NO.	LOAD CASE	F1 (lb/in)	V1 (lb/in)	NO. BOLTS	L (in)	F _B (lb)	V _B (lb)	3 σ F _B	3 σ V _B	F _{tu} ' (lb)	FS	F _{su} ' (lb)	FS
2230	X	10.729	0.075	3	6.500	23.246	0.163	69.74	0.488	1458	1.4	875	1.4

Table XXXIX METSAT AMSU-A1 Panel Flange Attachment Screws Stress Summary
Random Vibration 8.8 grms, Lower Right Panel

LOWER RIGHT PANEL												
LOWER FLANGE - BOLT STRESSES (NAS1352N06-6)												
FEM	LOAD	F1	V1	NO.	L	F _B	V _B	3 σ	3 σ	F _{tu} '	FS	F _{su} '
ELEM.	CASE	(lb/in)	(lb/in)	BOLTS	(in)	(lb)	(lb)	F _B	V _B	(lb)		(lb)
NO.												
575	Z	18.769	0.091	15	19.290	24.136	0.117	72.41	0.351	1458	1.4	875
												1.4
												13

Table XL METSAT AMSU-A1 Panel Flange Attachment Screws Stress Summary
Random Vibration 8.8 grms, Lower Aft Panel

LOWER AFT PANEL													
RIGHT FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM	LOAD	F1	V1	NO.	L	F _B	V _B	3 σ	3 σ	F _{tu} '	FS	F _{su} '	MS
ELEM.	CASE	(lb/in)	(lb/in)	BOLTS	(in)	(lb)	(lb)	F _B	V _B	(lb)		(lb)	
NO.													
429	X	9.932	0.099	7	8.770	12.443	0.124	37.33	0.372	1458	1.4	875	1.4 27
LOWER FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM	LOAD	F1	V1	NO.	L	F _B	V _B	3 σ	3 σ	F _{tu} '	FS	F _{su} '	MS
ELEM.	CASE	(lb/in)	(lb/in)	BOLTS	(in)	(lb)	(lb)	F _B	V _B	(lb)		(lb)	
NO.													
340	Z	26.382	0.074	8	10.563	34.835	0.098	104.50	0.293	1458	1.4	875	1.4 9.0
UPPER FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM	LOAD	F1	V1	NO.	L	F _B	V _B	3 σ	3 σ	F _{tu} '	FS	F _{su} '	MS
ELEM.	CASE	(lb/in)	(lb/in)	BOLTS	(in)	(lb)	(lb)	F _B	V _B	(lb)		(lb)	
NO.													
429	X	38.064	0.452	6	8.498	53.911	0.640	161.73	1.921	1458	1.4	875	1.4 5.4
339	X	39.042	0.152	6	8.498	55.297	0.215	165.89	0.646	1458	1.4	875	1.4 5.3

Table XLI METSAT AMSU-A1 Panel Flange Attachment Screws Stress Summary
Random Vibration 8.8 grms, Upper Motor Mount Panel

UPPER MOTOR MOUNT PANEL														
LOWER FLANGE - BOLT STRESSES (NAS1352N06-6)														
FEM ELEM. NO.	LOAD CASE	F1 (lb/in)	V1 (lb/in)	NO. BOLTS	L (in)	F _B (lb)	V _B (lb)	3σ F _B	3σ V _B	F _{tu} ' (lb)	FS	F _{su} ' (lb)	FS	MS
2272	X	12.445	0.055	6	10.563	21.910	0.097	65.73	0.290	1458	1.4	875	1.4	15
2279	Z	13.919	0.114	6	10.563	24.504	0.201	73.51	0.602	1458	1.4	875	1.4	13
RIGHT FLANGE - BOLT STRESSES (NAS1352N06-6)														
FEM ELEM. NO.	LOAD CASE	F1 (lb/in)	V1 (lb/in)	NO. BOLTS	L (in)	F _B (lb)	V _B (lb)	3σ F _B	3σ V _B	F _{tu} ' (lb)	FS	F _{su} ' (lb)	FS	MS
2271	X	14.380	0.087	3	6.245	29.935	0.181	89.81	0.543	1458	1.4	875	1.4	11

Table XLII METSAT AMSU-A1 Panel Flange Attachment Screws Stress Summary
Random Vibration 8.8 grms, Lower Motor Mount Panel

LOWER MOTOR MOUNT PANEL														
LOWER FLANGE - BOLT STRESSES (NAS1352N06-6)														
FEM	LOAD	F1	V1	NO.	L	F _B	V _B	3σ	3σ	F _{tu} '	F _S	F _{Su} '	F _S	MS
ELEM.	CASE	(lb/in)	(lb/in)	BOLTS	(in)	(lb)	(lb)	F _B	V _B	(lb)		(lb)		
NO.														
2449	X	30.343	0.179	6	10.562	53.414	0.315	160.24	0.945	1458	1.4	875	1.4	5.5

Table XLIII METSAT AMSU-A1 Panel Flange Attachment Screws Stress Summary
Random Vibration 8.8 grms, Lower Front Panel

LOWER FRONT PANEL													
LOWER FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM ELEM. NO.	LOAD CASE	F1 (lb/in)	V1 (lb/in)	NO. BOLTS	L (in)	F _B (lb)	V _B (lb)	3 σ F _B	3 σ V _B	F _{tu} ' (lb)	FS	F _{su} ' (lb)	MS
1261	Z	37.769	0.140	8	10.563	49.870	0.185	149.61	0.555	1458	1.4	875	6.0
RIGHT FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM ELEM. NO.	LOAD CASE	F1 (lb/in)	V1 (lb/in)	NO. BOLTS	L (in)	F _B (lb)	V _B (lb)	3 σ F _B	3 σ V _B	F _{tu} ' (lb)	FS	F _{su} ' (lb)	MS
1333	Z	10.331	1.409	8	9.440	12.190	1.663	36.57	4.988	1458	1.4	875	27
UPPER FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM ELEM. NO.	LOAD CASE	F1 (lb/in)	V1 (lb/in)	NO. BOLTS	L (in)	F _B (lb)	V _B (lb)	3 σ F _B	3 σ V _B	F _{tu} ' (lb)	FS	F _{su} ' (lb)	MS
1240	X	26.171	0.038	7	10.288	38.463	0.056	115.39	0.168	1458	1.4	875	8.0

Table XLIV METSAT AMSU-A1 Panel Flange Attachment Screws Stress Summary
Random Vibration 8.8 grms, Upper Aft Panel

UPPER AFT PANEL													
LOWER FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM	LOAD	F1	V1	NO.	L	F _B	V _B	3σ	F _B	3σ	F _{tu} '	FS	F _{su} '
ELEM.	CASE	(lb/in)	(lb/in)	BOLTS	(in)	(lb)	(lb)		(lb)		(lb)		(lb)
NO.													
5778	X	121.086	4.722	9	11.700	157.412	6.139	472.24	18.416	1458	1.4	875	1.4
5781	X	110.354	5.073	9	11.700	143.460	6.595	430.38	19.785	1458	1.4	875	1.4
5785	X	121.193	5.202	9	11.700	157.551	6.763	472.65	20.288	1458	1.4	875	1.4
UPPER FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM	LOAD	F1	V1	NO.	L	F _B	V _B	3σ	F _B	3σ	F _{tu} '	FS	F _{su} '
ELEM.	CASE	(lb/in)	(lb/in)	BOLTS	(in)	(lb)	(lb)		(lb)		(lb)		(lb)
NO.													
6980	X	27.359	0.242	7	9.800	38.303	0.339	114.91	1.016	1458	1.4	875	1.4
RIGHT FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM	LOAD	F1	V1	NO.	L	F _B	V _B	3σ	F _B	3σ	F _{tu} '	FS	F _{su} '
ELEM.	CASE	(lb/in)	(lb/in)	BOLTS	(in)	(lb)	(lb)		(lb)		(lb)		(lb)
NO.													
6856	X	48.591	12.510	8	11.600	70.457	18.140	211.37	54.419	1458	1.4	875	1.4
LEFT FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM	LOAD	F1	V1	NO.	L	F _B	V _B	AVG 3σ	AVG 3σ	F _{tu} '	FS	F _{su} '	MS
ELEM.	CASE	(lb/in)	(lb/in)	BOLTS	(in)	(lb)	(lb)			(lb)		(lb)	
NO.													
6944	X	124.758	13.208	8	11.600	180.899	19.152						
6945	X	153.543	4.990	8	11.600	222.637	7.236	789.67	29.384	1458	1.4	875	1.4
6946	X	266.296	2.067	8	11.600	386.129	2.997						

Table XLV METSAT AMSU-A1 Panel Flange Attachment Screws Stress Summary
Random Vibration 8.8 grms, Upper Right Panel

UPPER RIGHT PANEL														
LOWER FLANGE - BOLT STRESSES (NAS1352N06-6)														
FEM ELEM. NO.	LOAD CASE	F1 (lb/in)	V1 (lb/in)	NO. BOLTS	L (in)	F _B (lb)	V _B (lb)	3σ F _B	3σ V _B	F _{tu} ' (lb)	FS	F _{su} ' (lb)	FS	MS
1648	X	31.682	0.283	7	8.783	39.752	0.355	119.26	1.065	1458	1.4	875	7.7302	7.7

Table XLVI METSAT AMSU-A1 Panel Flange Attachment Screws Stress Summary
Random Vibration 8.8 grms, Lower Right Front Support Panel

LOWER RIGHT FRONT SUPPORT PANEL													
LOWER FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM	LOAD	F1	V1	NO.	L	F _B	V _B	3 σ	3 σ	F _{tu} '	FS	F _{su} '	MS
ELEM.	CASE	(lb/in)	(lb/in)	BOLTS	(in)	(lb)	(lb)	F _B	V _B	(lb)		(lb)	
NO.													
2201	X	6.871	0.067	3	6.500	14.887	0.145	44.66	0.436	1458	1.4	875	22
AFT FLANGE - BOLT STRESSES (NAS1352N06-6)													
FEM	LOAD	F1	V1	NO.	L	F _B	V _B	3 σ	3 σ	F _{tu} '	FS	F _{su} '	MS
ELEM.	CASE	(lb/in)	(lb/in)	BOLTS	(in)	(lb)	(lb)	F _B	V _B	(lb)		(lb)	
NO.													
2204	X	10.132	0.031	3	3.450	11.652	0.036	34.96	0.107	1458	1.4	875	29

METSAT AMSU-A1 1331720-1 MESH ONLY

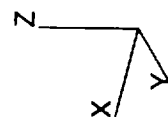
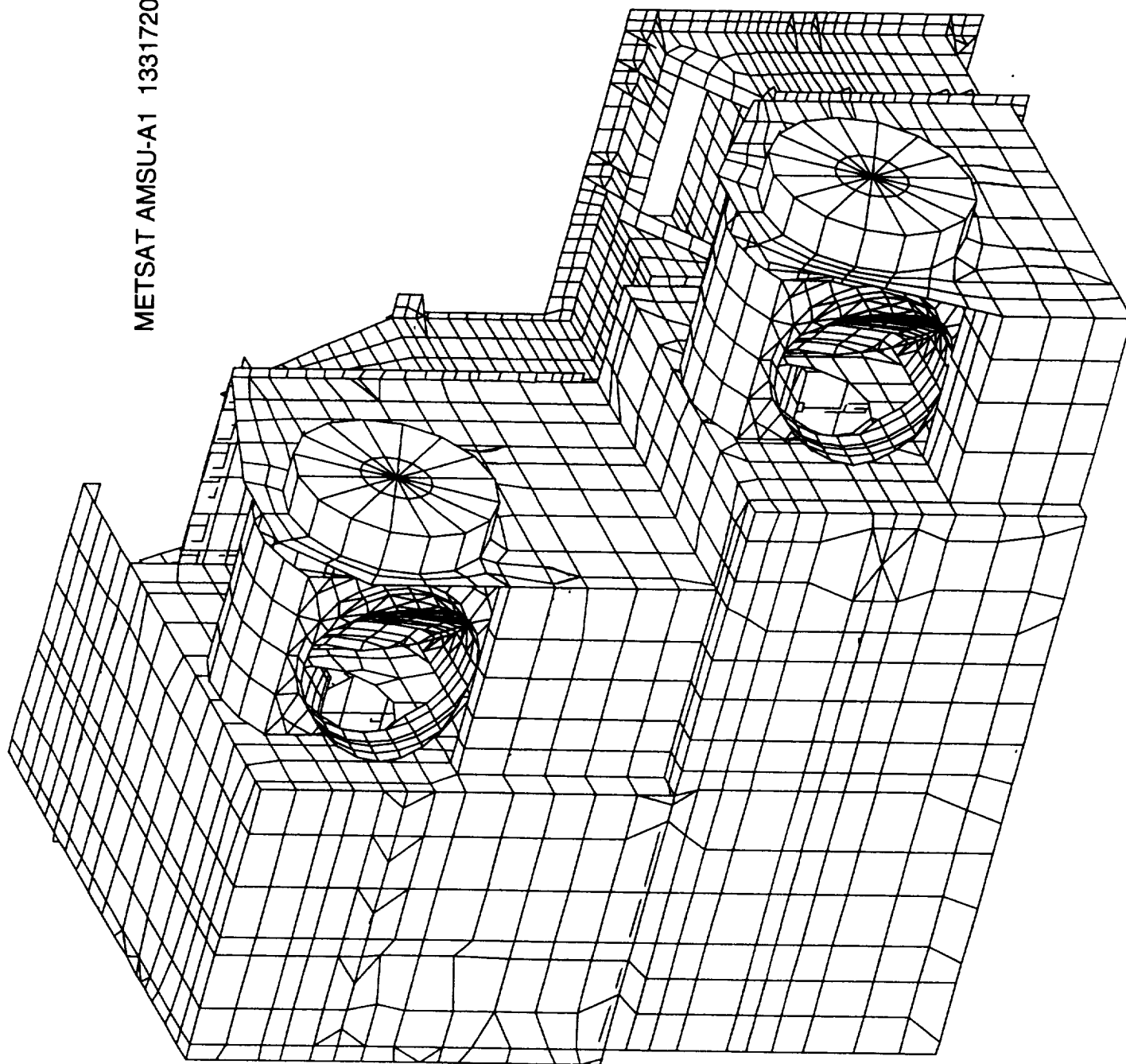
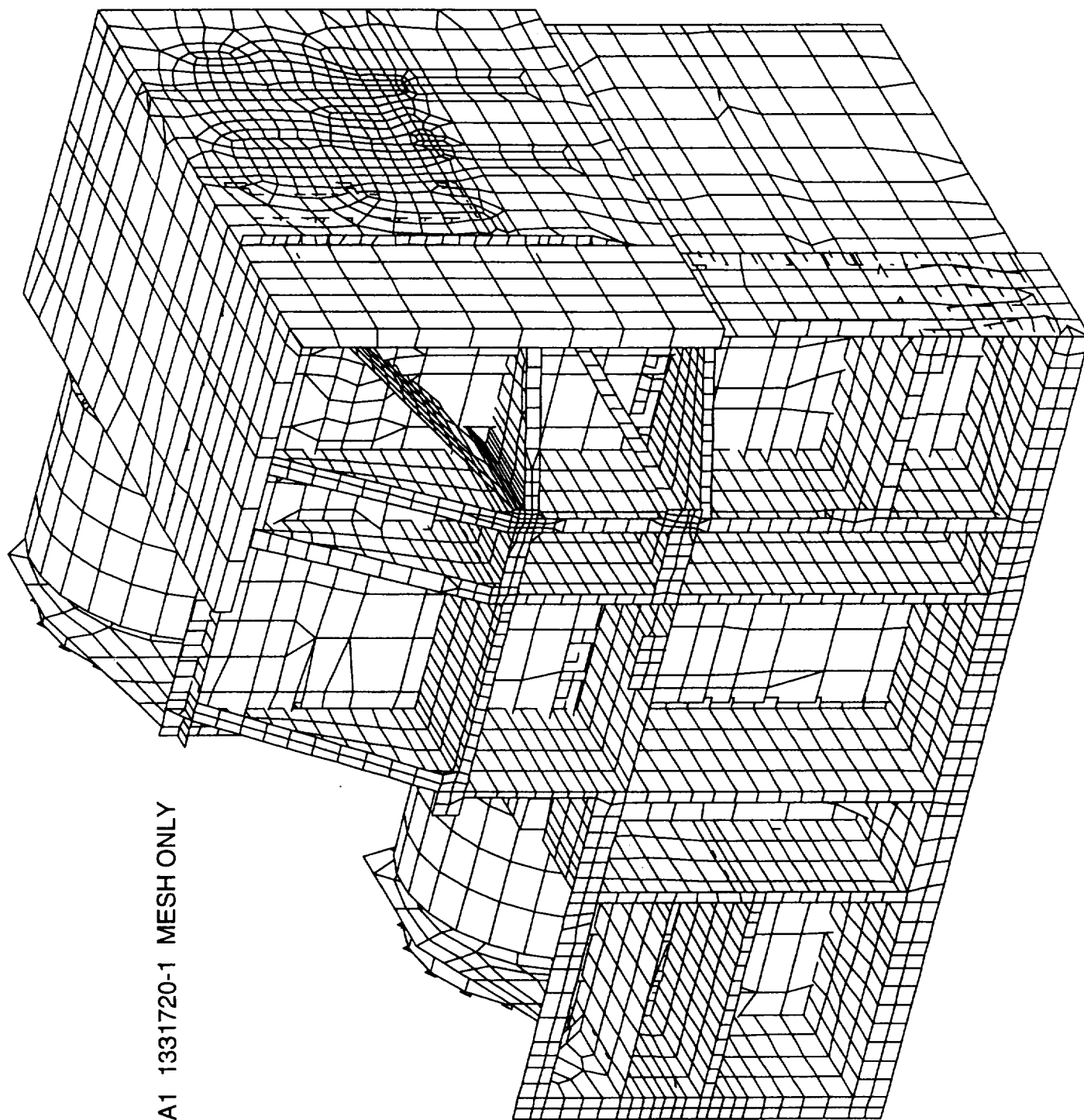


Figure 1 METSAT AMSU-A1 1331720-1 Mesh. View 1



METSAT AMSU-A1 1331720-1 MESH ONLY

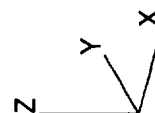


Figure 2 METSAT AMSU-A1 1331720-1 Mesh, View 2

METSAT AMSU-A1 1331720-1 MESH ONLY

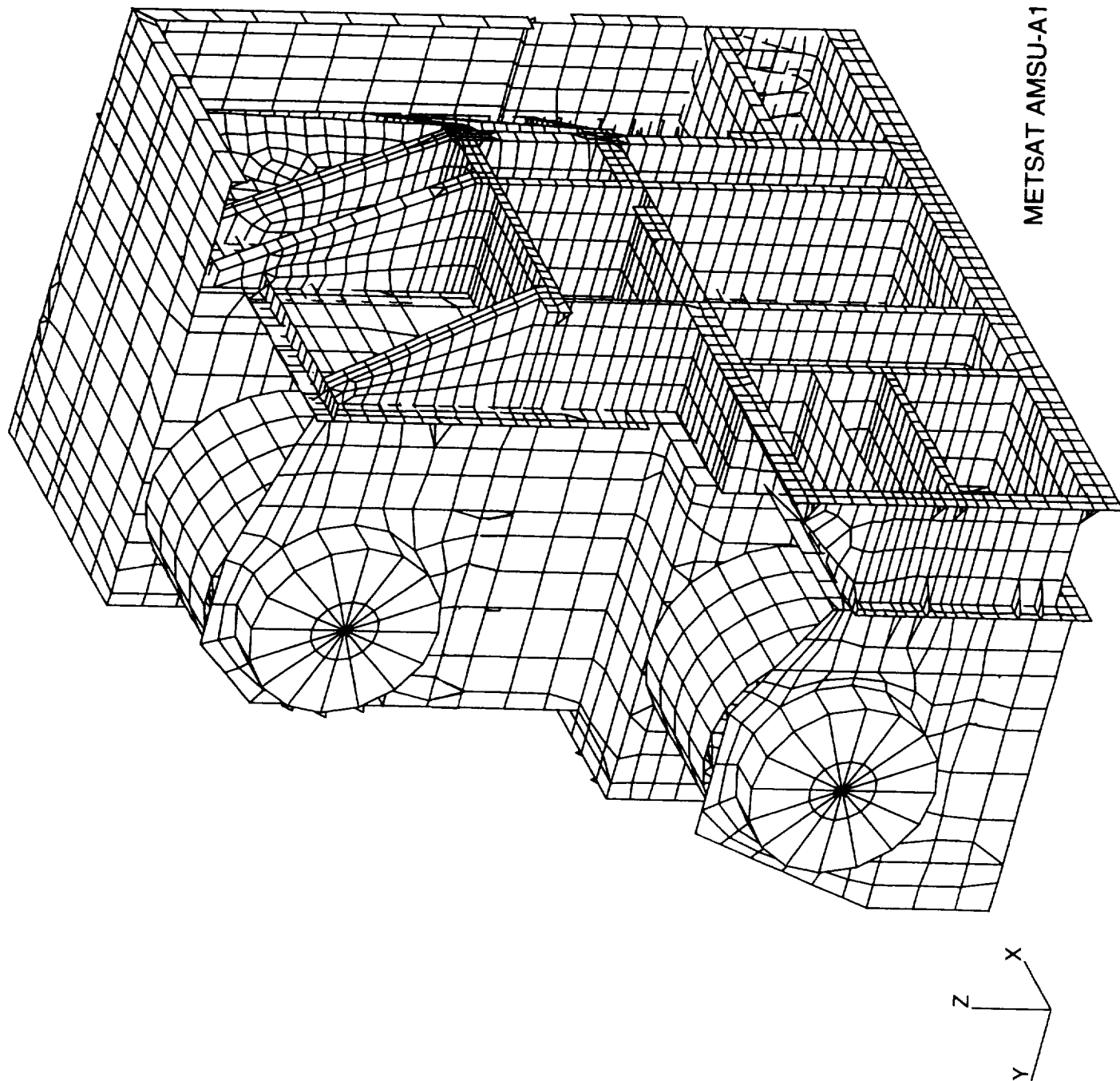


Figure 3 METSAT AMSU-A1 1331720-1 Mesh, View 3

METSAT AMSU-A1 INTERNAL VIEW - MESH ONLY

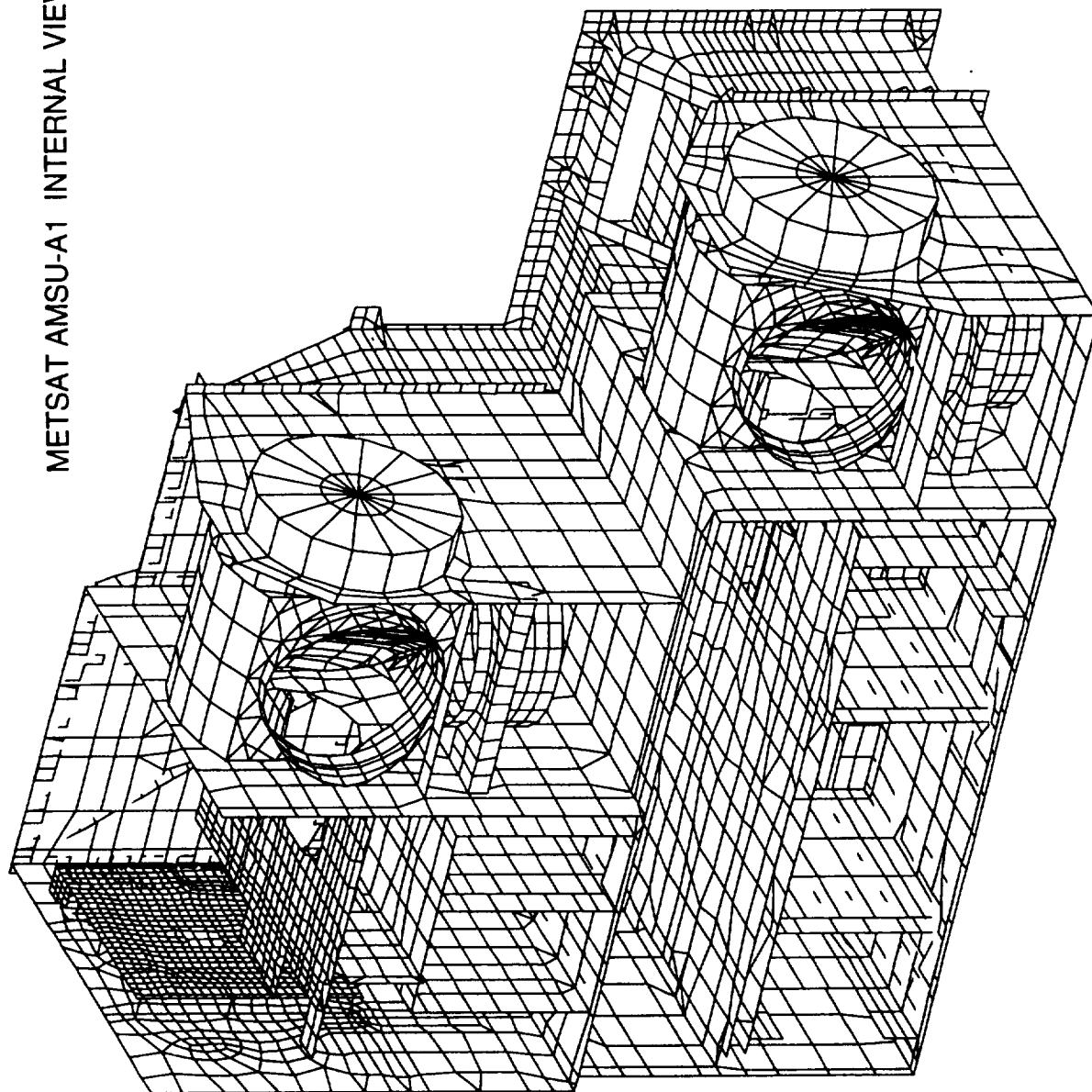
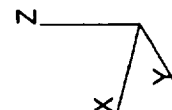


Figure 4 METSAT AMSU-A1 1331720-1 Mesh, View 4



Time: 09:43:28
Date: 06/14/96

Contour
Mode Scalar1

Color Index

0	0.966E+04
9	0.858E+04
8	0.751E+04
7	0.644E+04
6	0.537E+04
5	0.429E+04
4	0.322E+04
3	0.215E+04
2	0.108E+04
1	0.649E+01

Min = 6.487691E+00

Max = 1.072748E+04

Min ID= 7936

Max ID= 9642

Contour_8:

Stress Tensor

Von Mises

At Z2

Design_Gy=-20.6_Gz=19.6

Static Subcase

METSAT AMSU-A1

Sidemount

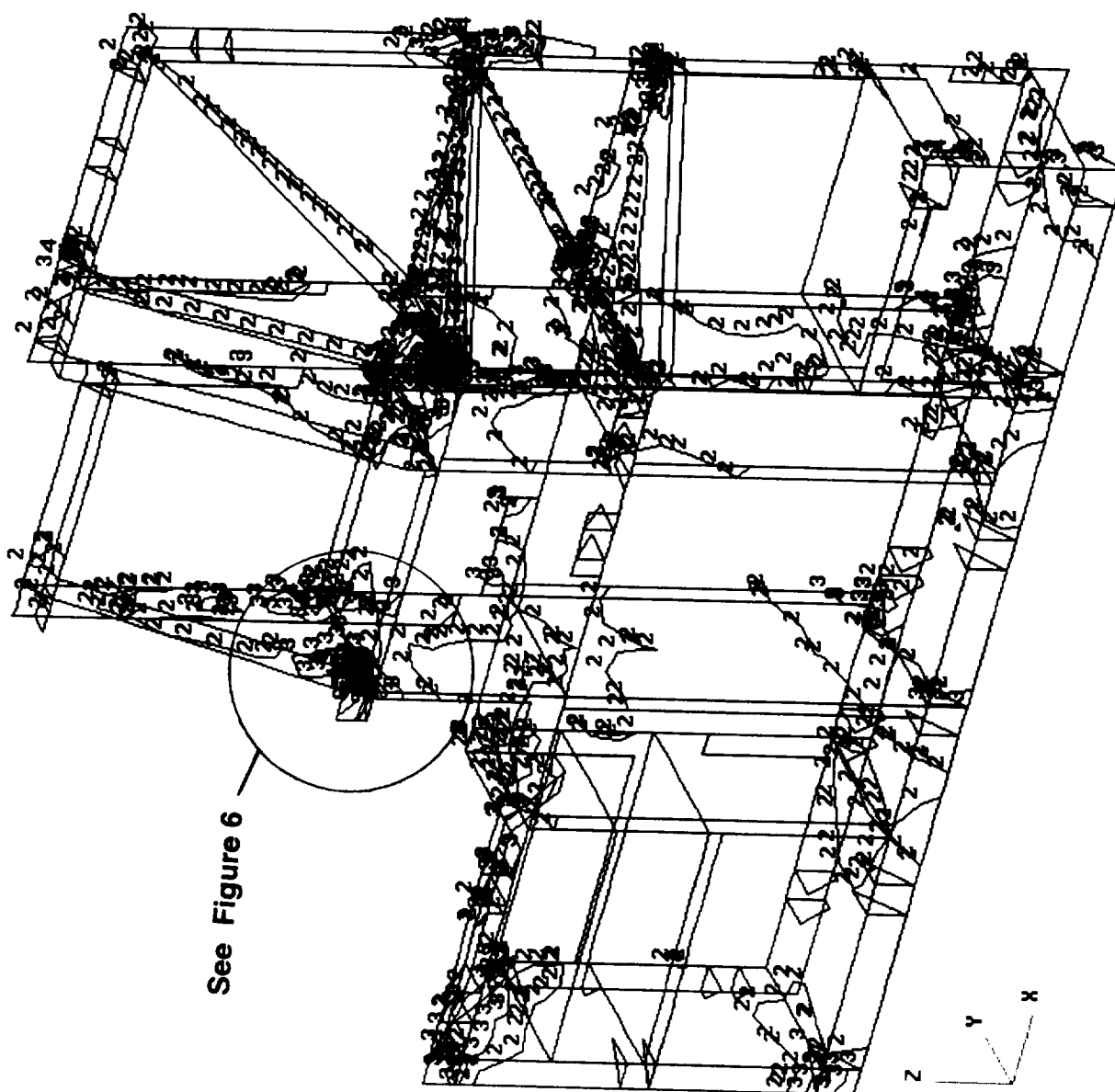


Figure 5 METSAT AMSU-A1 Sidemount, Maximum Stress

Time: 09:45:48
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.966E+04
9	0.858E+04
8	0.751E+04
7	0.644E+04
6	0.537E+04
5	0.429E+04
4	0.322E+04
3	0.215E+04
2	0.108E+04
1	0.649E+01

Min = 6.487691E+00
Max = 1.072748E+04
Min ID= 7936
Max ID= 9642
Contour_8:
Stress Tensor

Von Mises
At 22

Design_Gy=-20.6_Gz=19.6
Static Subcase

METSAT AMSU-A1
Sidemount

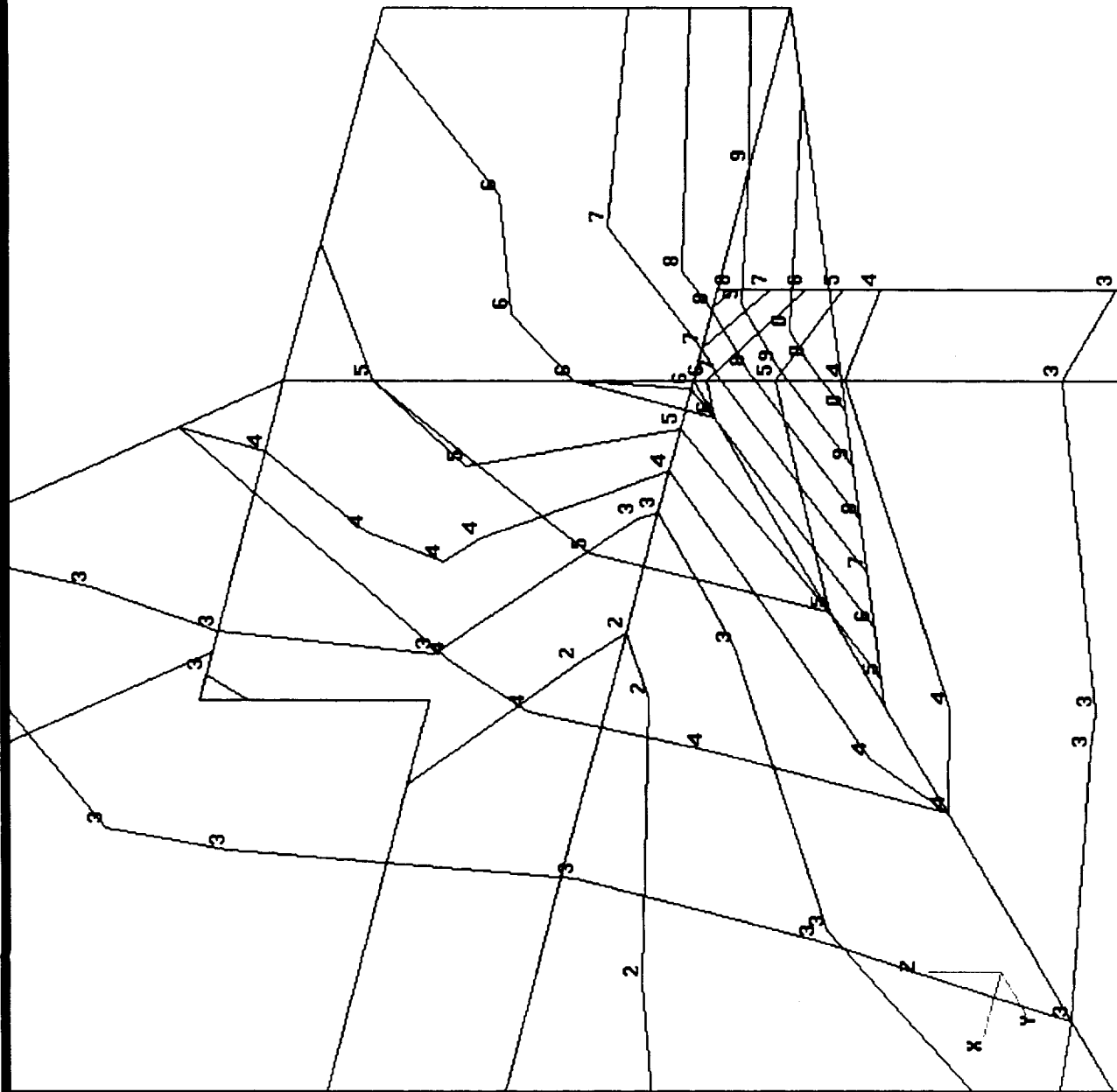


Figure 6 METSAT AMSU-A1 Sidemount, Maximum Stress, Enlarged View

Time: 14:13:41
Date: 06/09/96

Contour
Mode Scalar1

Color Index

—	0	0.787E+04
—	9	0.700E+04
—	8	0.612E+04
—	7	0.525E+04
—	6	0.437E+04
—	5	0.350E+04
—	4	0.263E+04
—	3	0.175E+04
—	2	0.878E+03
—	1	0.398E+01

Min = 3.976558E+00

Max = 8.745224E+03

Min ID= 5408

Max ID= 1772

Contour_9:

Stress Tensor

Von Mises

At Z1

Design_Gy=-20.6_Gx=-21.2

Static Subcase

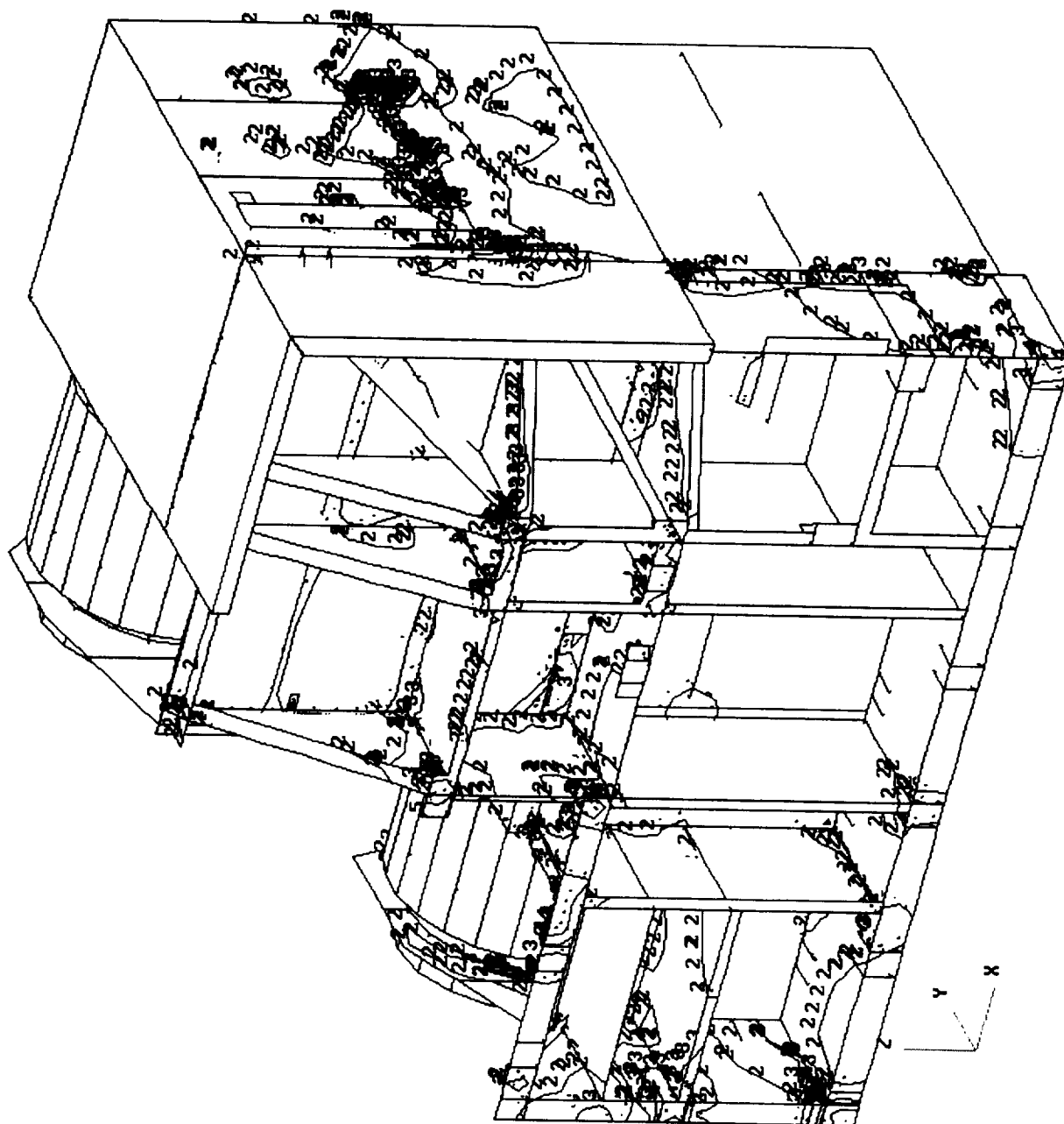


Figure 7 METSAT AMSU-A1 Shell Stresses, Case 1

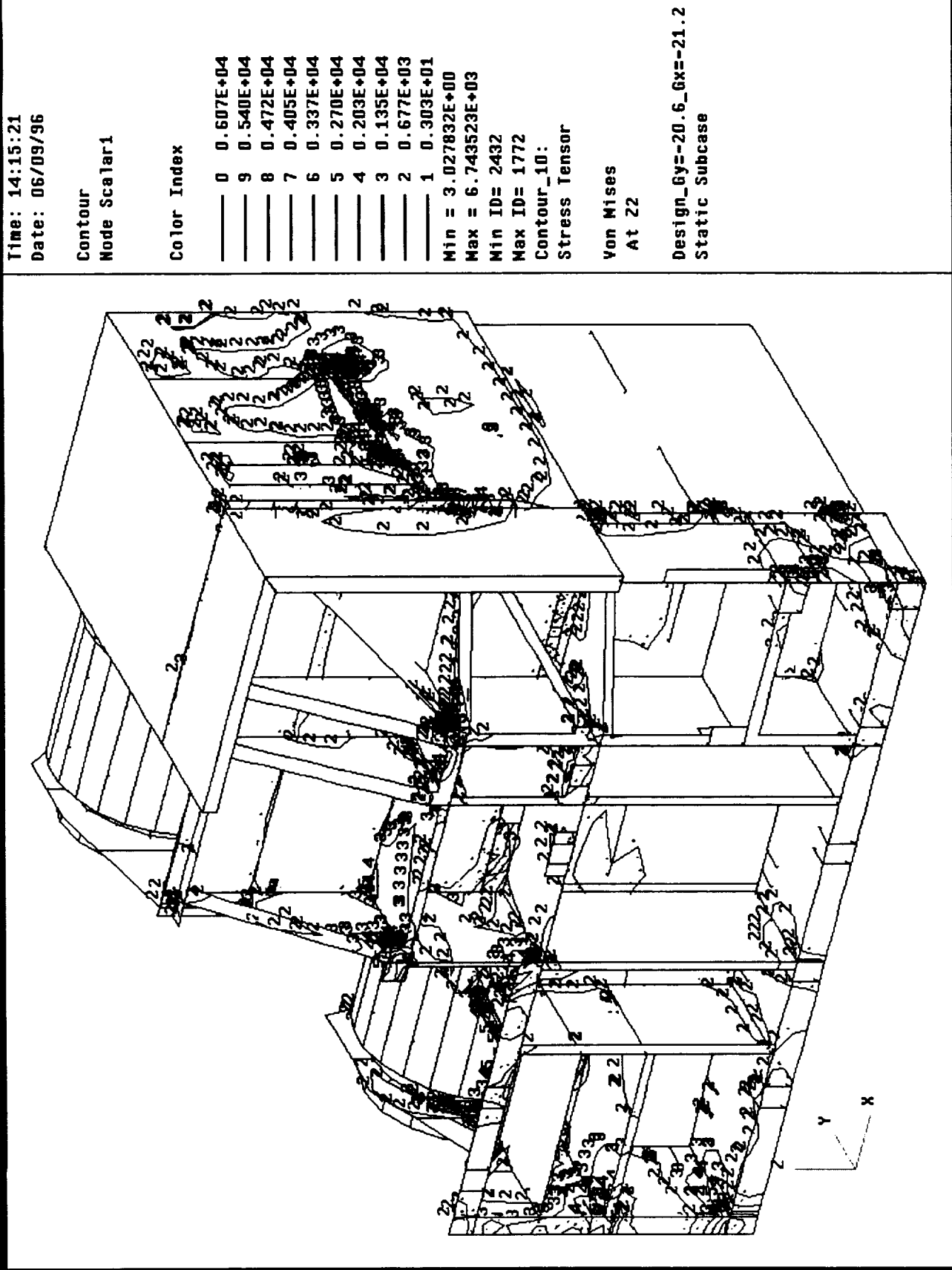


Figure 8 METSAT AMSU-A1 Shell Stresses, Case 1

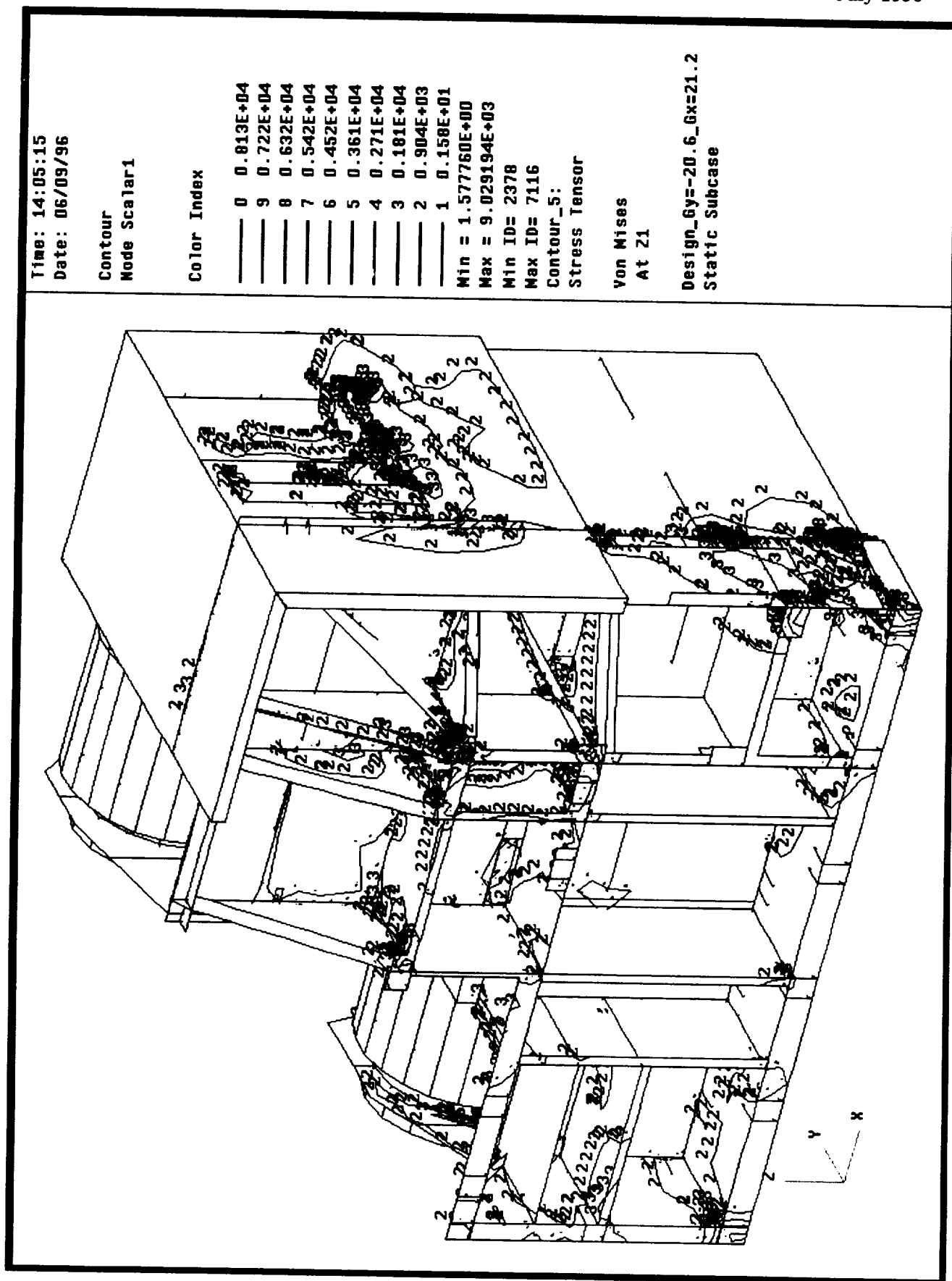


Figure 9 METSAT AMSU-A1 Shell Stresses, Case 2

Time: 14:07:09
Date: 06/09/96

Contour
Mode Scalar1

Color Index

—	0	0.827E+04
—	9	0.735E+04
—	8	0.643E+04
—	7	0.551E+04
—	6	0.459E+04
—	5	0.368E+04
—	4	0.276E+04
—	3	0.184E+04
—	2	0.921E+03
—	1	0.231E+01

Min = 2.314946E+00

Max = 9.184245E+03

Min ID= 2338

Max ID= 7116

Contour_6:

Stress Tensor

Von Mises
At 22

Design_Gy=-20.6_Gx=21.2
Static Subcase

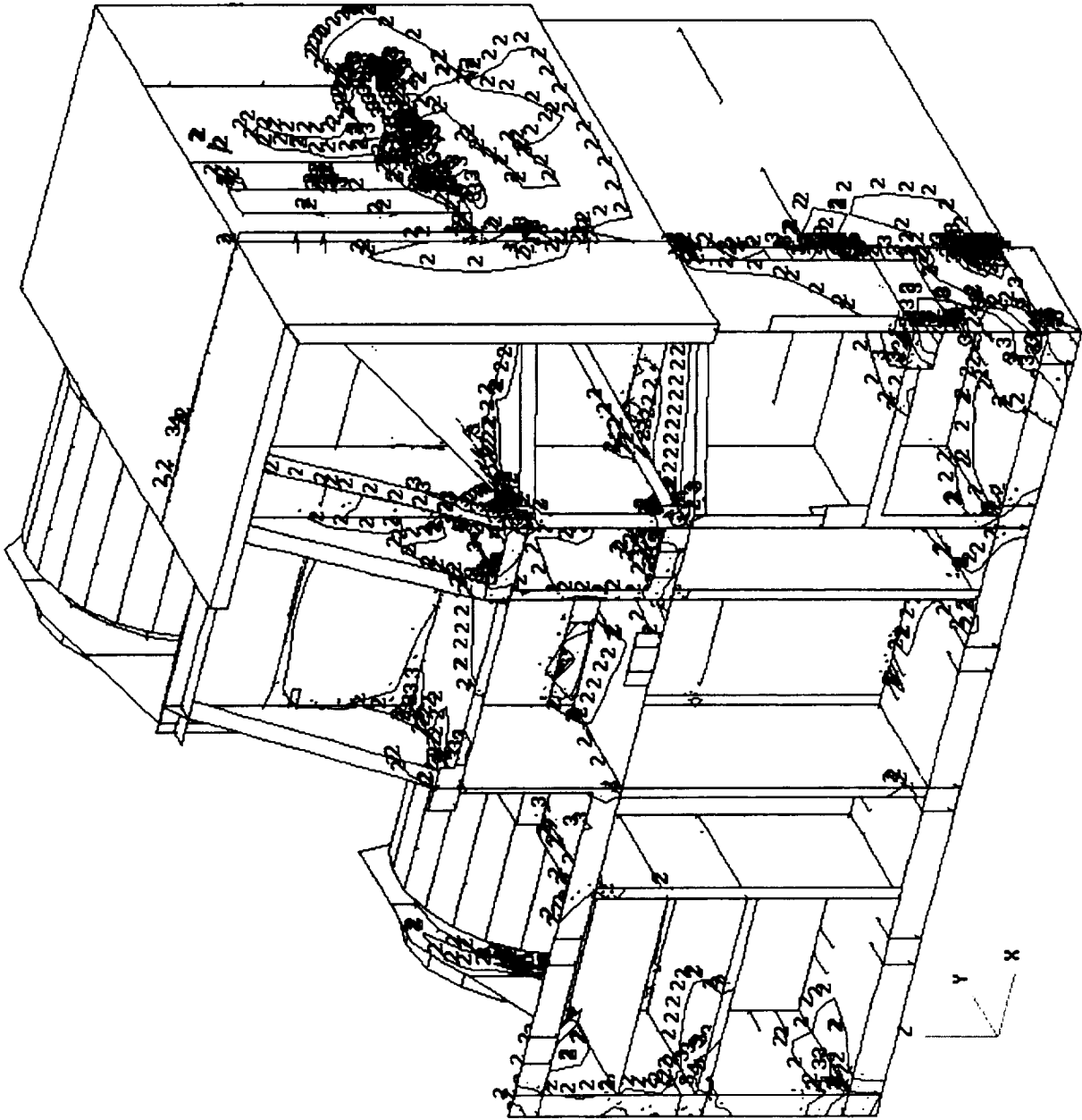


Figure 10 METSAT AMSU-A1 Shell Stresses, Case 2

Time: 14:21:12
Date: 06/09/96

Contour
Mode Scalar1

Color Index

—	0	0.787E+04
—	9	0.699E+04
—	8	0.612E+04
—	7	0.525E+04
—	6	0.437E+04
—	5	0.350E+04
—	4	0.263E+04
—	3	0.175E+04
—	2	0.878E+03
—	1	0.407E+01

Min = 4.066542E+00

Max = 8.742698E+03

Min ID= 11813

Max ID= 7116

Contour_13:

Stress Tensor

Von Mises

At 21

Design_Gy=-20.6_Gz=-19.6
Static Subcase

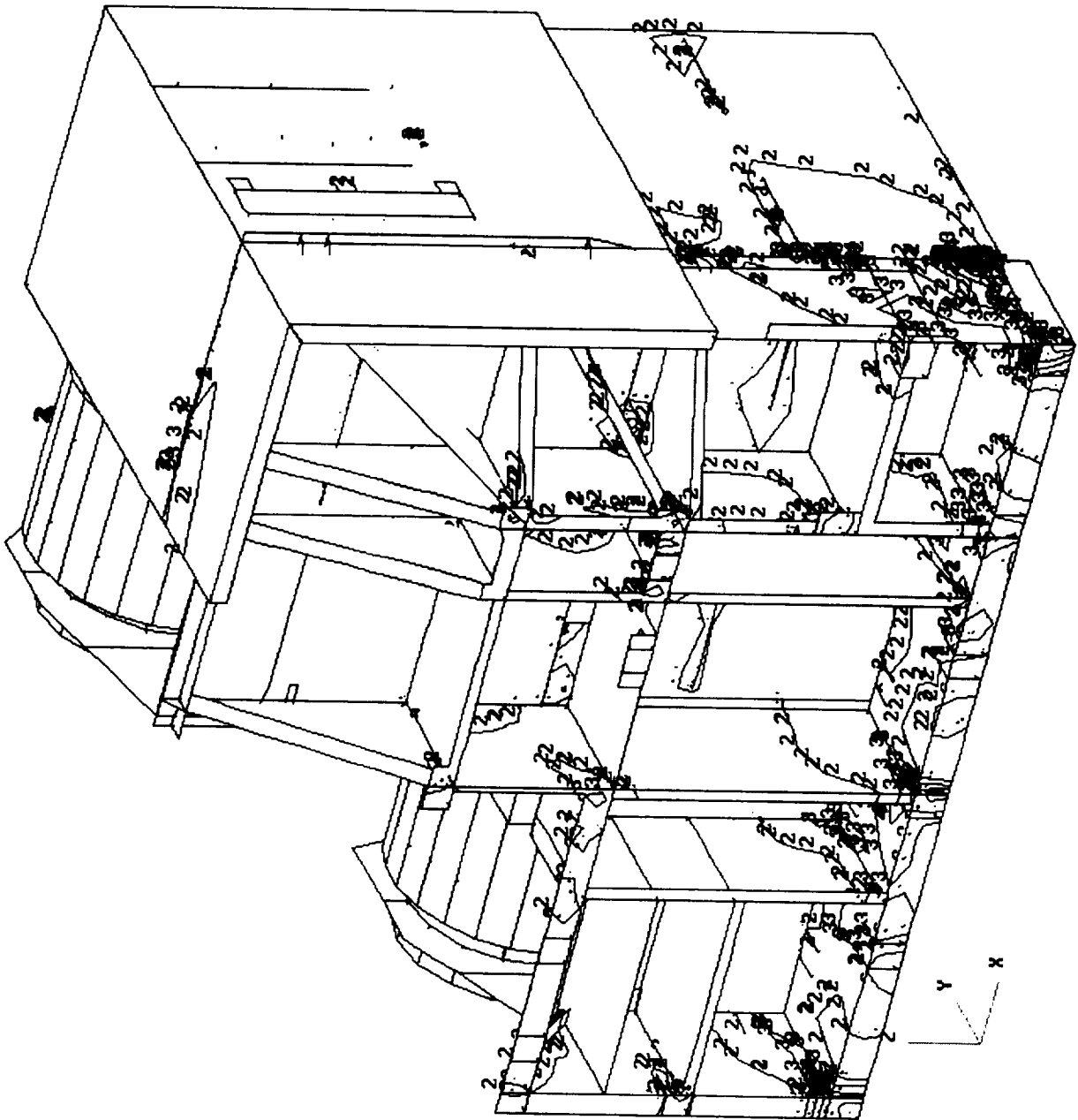


Figure 11 METSAT AMSU-A1 Shell Stresses, Case 3

Time: 14:22:28
Date: 06/09/96

Contour
Mode Scalar1

Color Index

0	0.749E+04
9	0.666E+04
8	0.582E+04
7	0.499E+04
6	0.416E+04
5	0.333E+04
4	0.250E+04
3	0.167E+04
2	0.834E+03
1	0.281E+01

Min = 2.811205E+00
Max = 8.319500E+03
Min ID= 11852
Max ID= 7116
Contour_14:
Stress Tensor

Von Mises
At 22

Design_Gy=-20.6_Gz=-19.6
Static Subcase

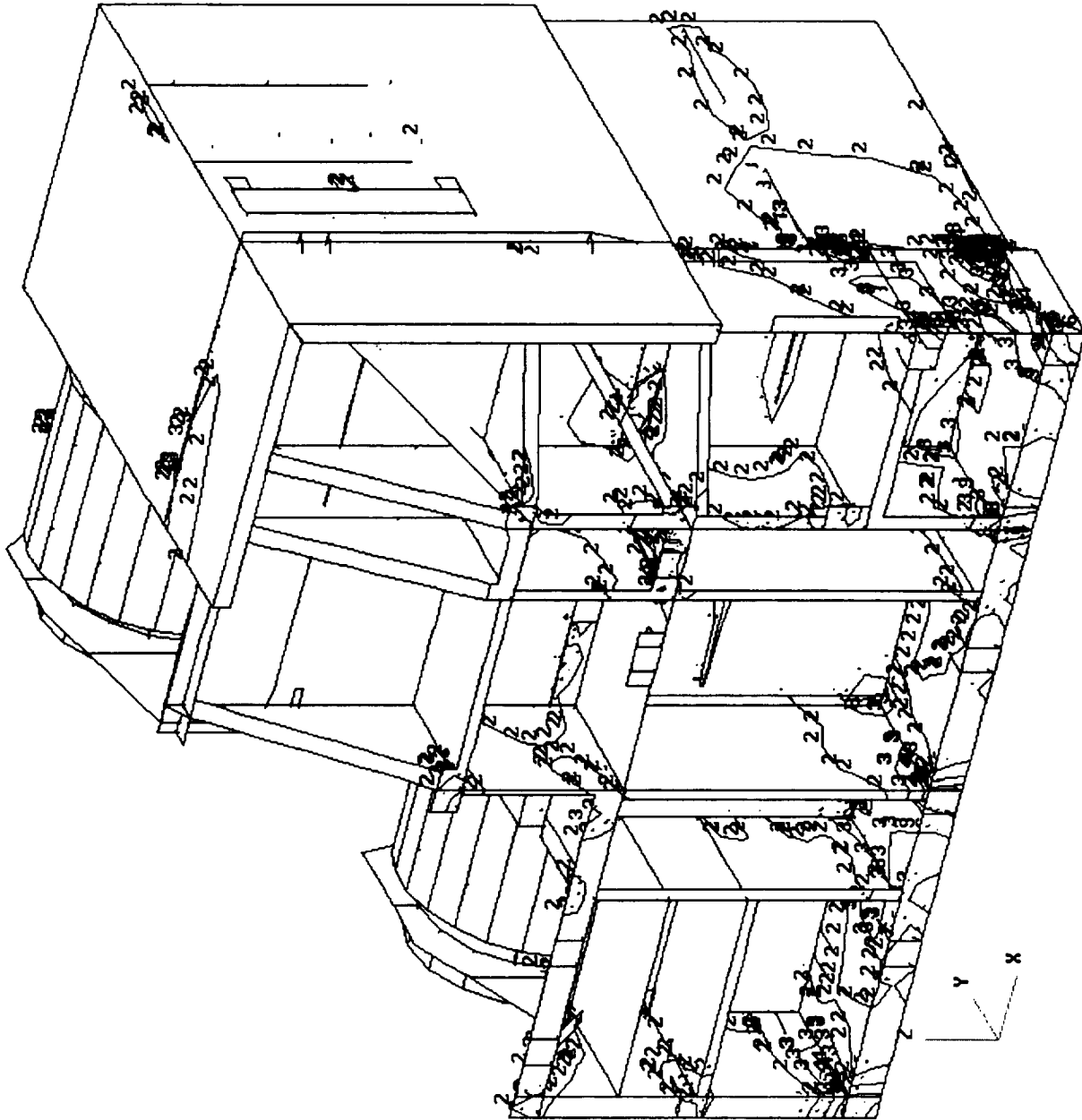


Figure 12 METSAT AMSU-A1 Shell Stresses, Case 3

Time: 12:25:59
Date: 06/08/96

Contour
Node Scalar1

Color Index

—	0	0.820E+04
—	9	0.729E+04
—	8	0.638E+04
—	7	0.546E+04
—	6	0.455E+04
—	5	0.364E+04
—	4	0.273E+04
—	3	0.182E+04
—	2	0.914E+03
—	1	0.364E+01

Min = 3.644375E+00

Max = 9.105877E+03

Min ID= 11279

Max ID= 8313

Contour_1:

Stress Tensor

Von Mises

At 21

Design_Gy=-20.6_Gz=19.6
Static Subcase

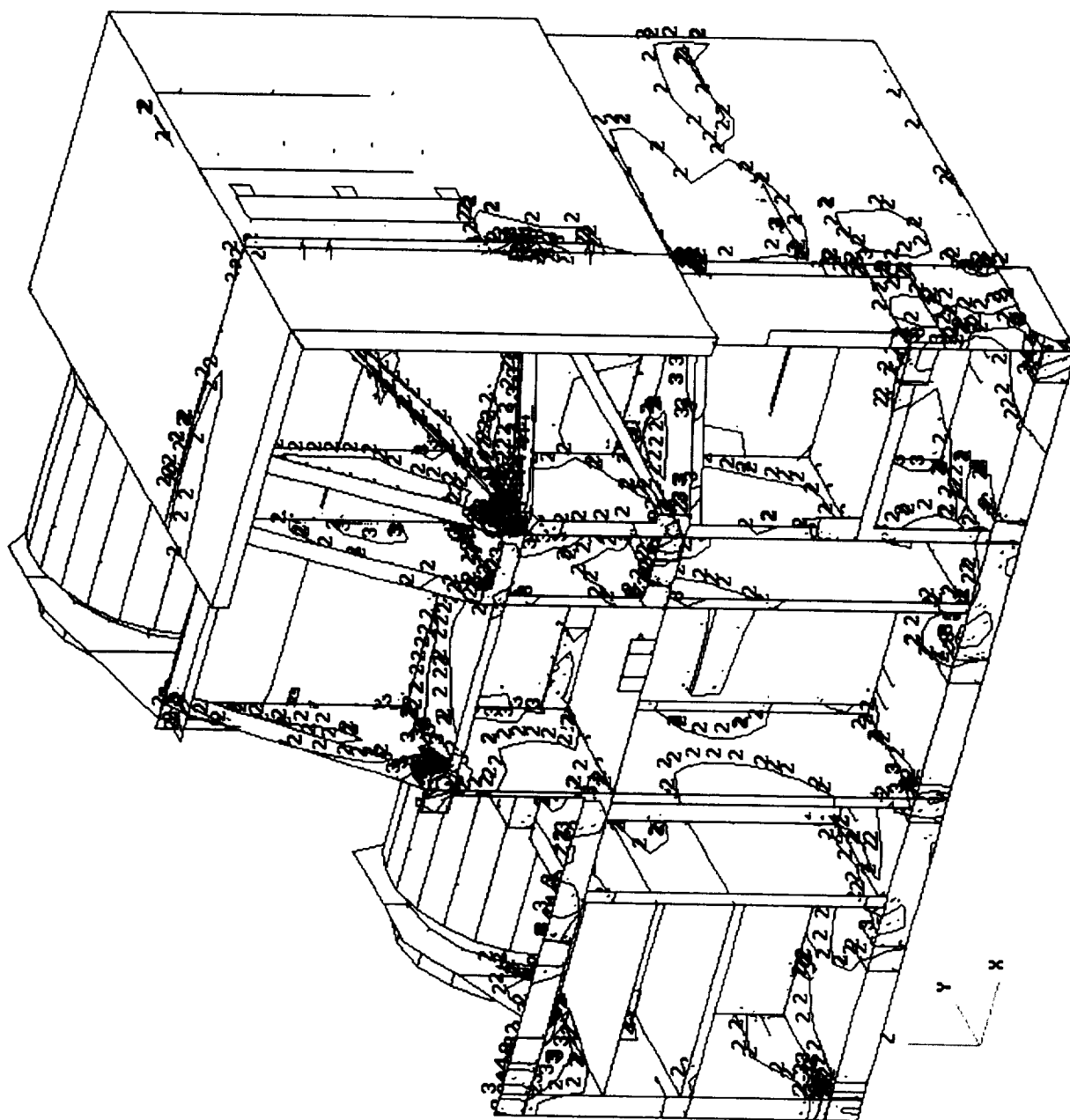


Figure 13 METSAT AMSU-A1 Shell Stresses, Case 4

Time: 12:27:29
Date: 06/08/96

Contour
Mode Scalar1

Color Index

—	0	0.965E+04
—	9	0.858E+04
—	8	0.751E+04
—	7	0.644E+04
—	6	0.537E+04
—	5	0.429E+04
—	4	0.322E+04
—	3	0.215E+04
—	2	0.108E+04
—	1	0.269E+01

Min = 2.694405E+00

Max = 1.072748E+04

Min ID= 11834

Max ID= 9642

Contour_2:

Stress Tensor

Von Mises

At 22

Design_Gy=-20.6_Gz=19.6

Static Subcase

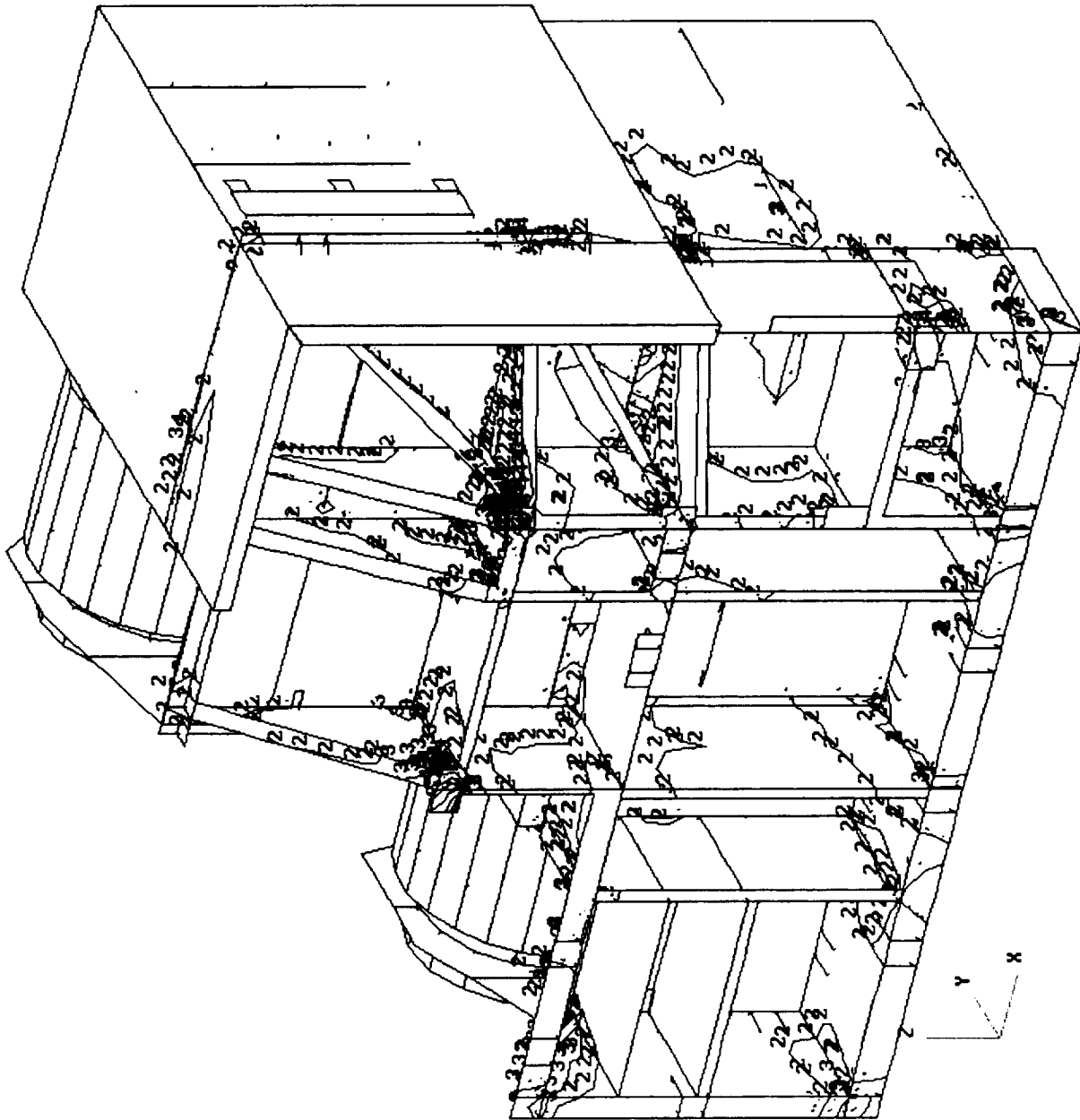


Figure 14 METSAT AMSU-A1 Shell Stresses, Case 4

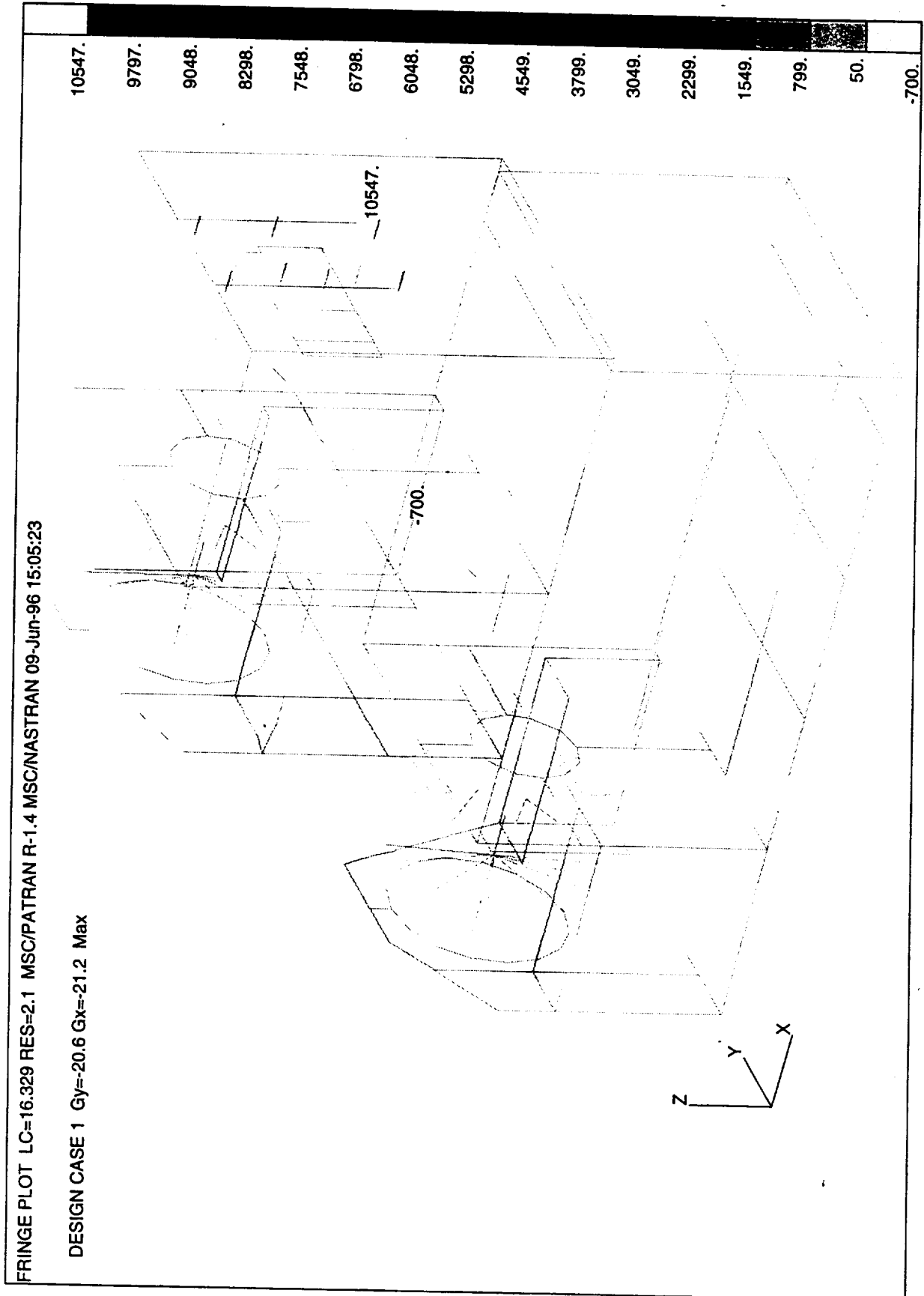


Figure 15 METSAT AMSU-A1 Beam Stresses, Case 1

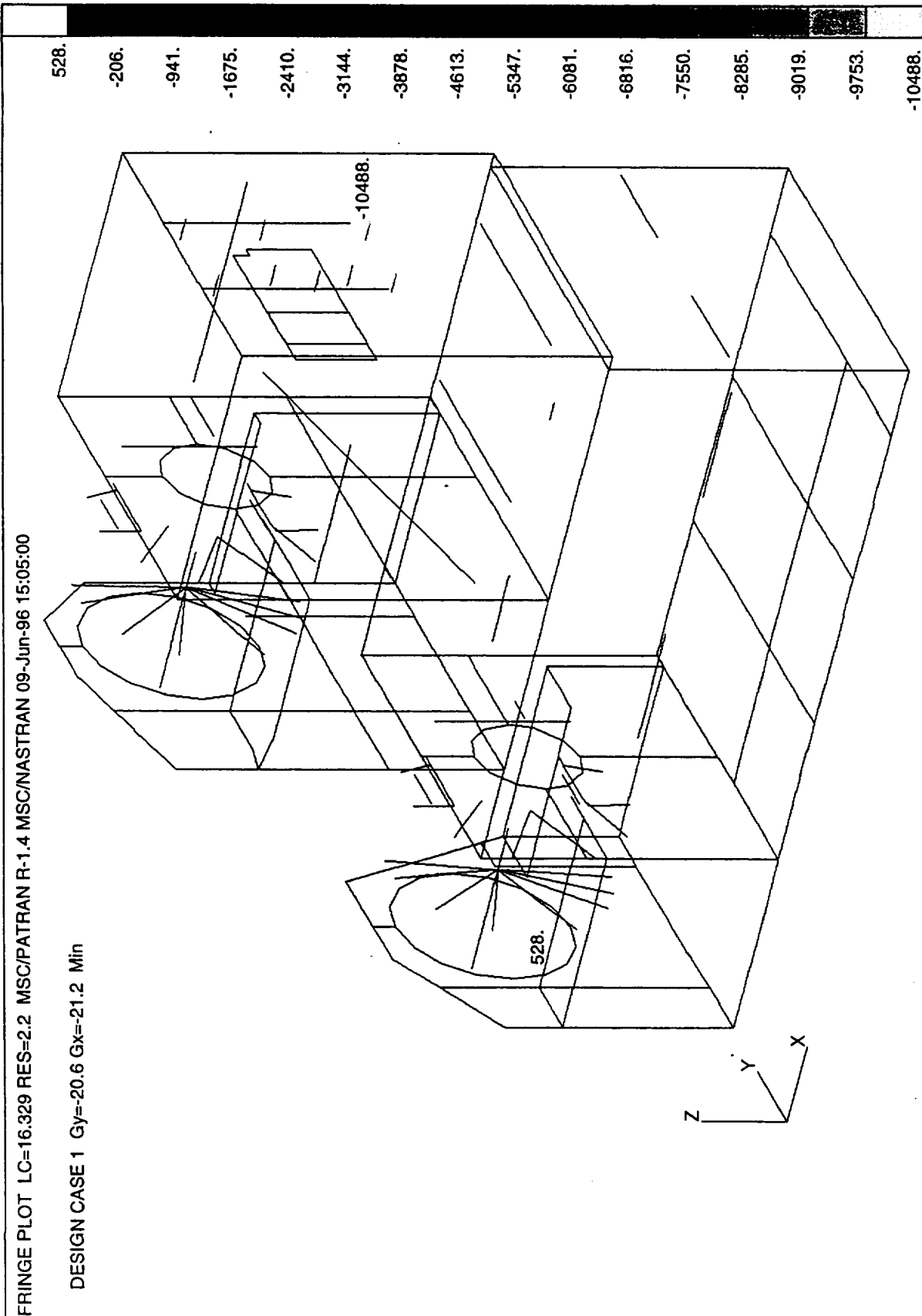


Figure 16 METSAT AMSU-A1 Beam Stresses, Case 1

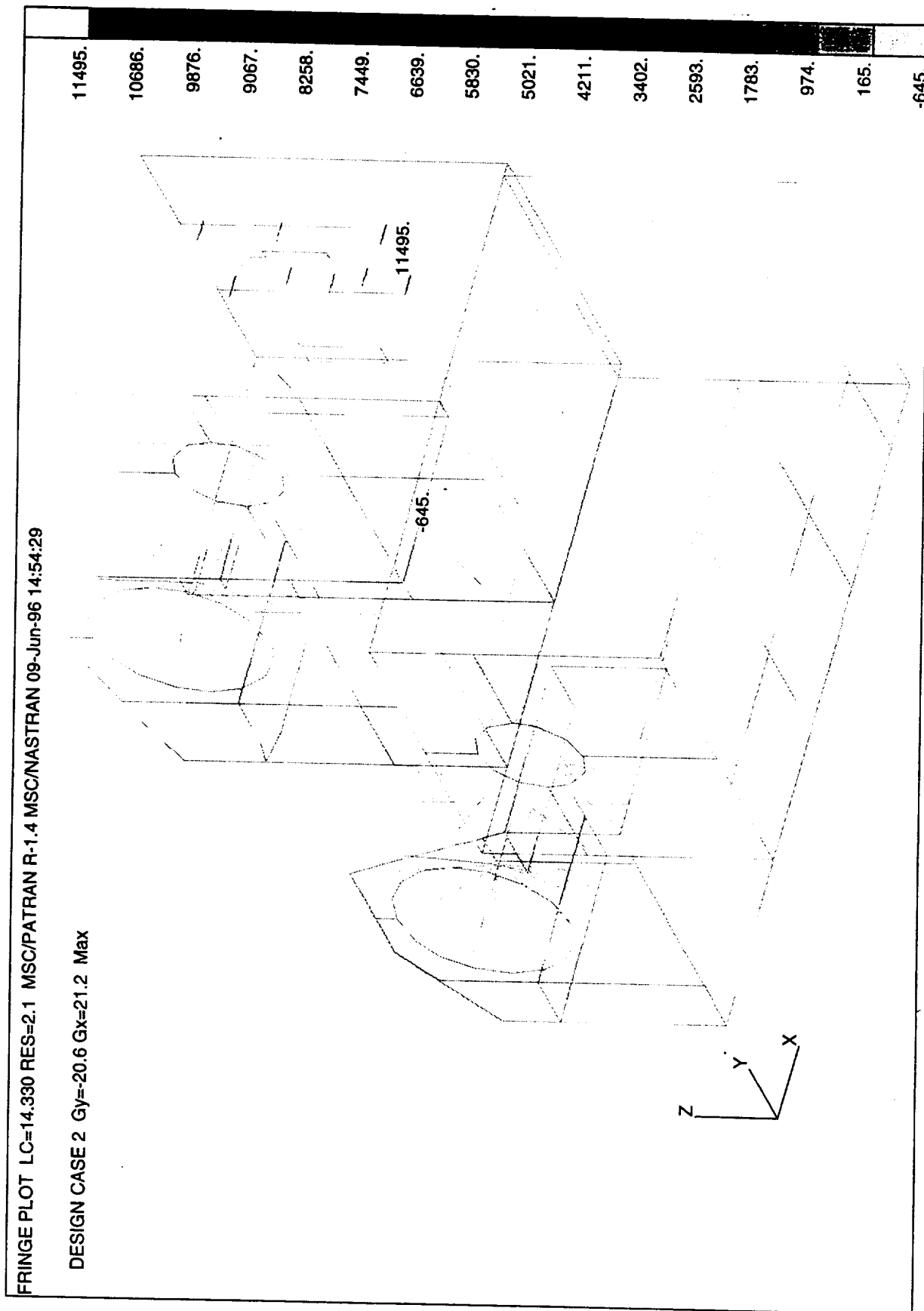


Figure 17 METSAT AMSU-A1 Beam Stresses, Case 2

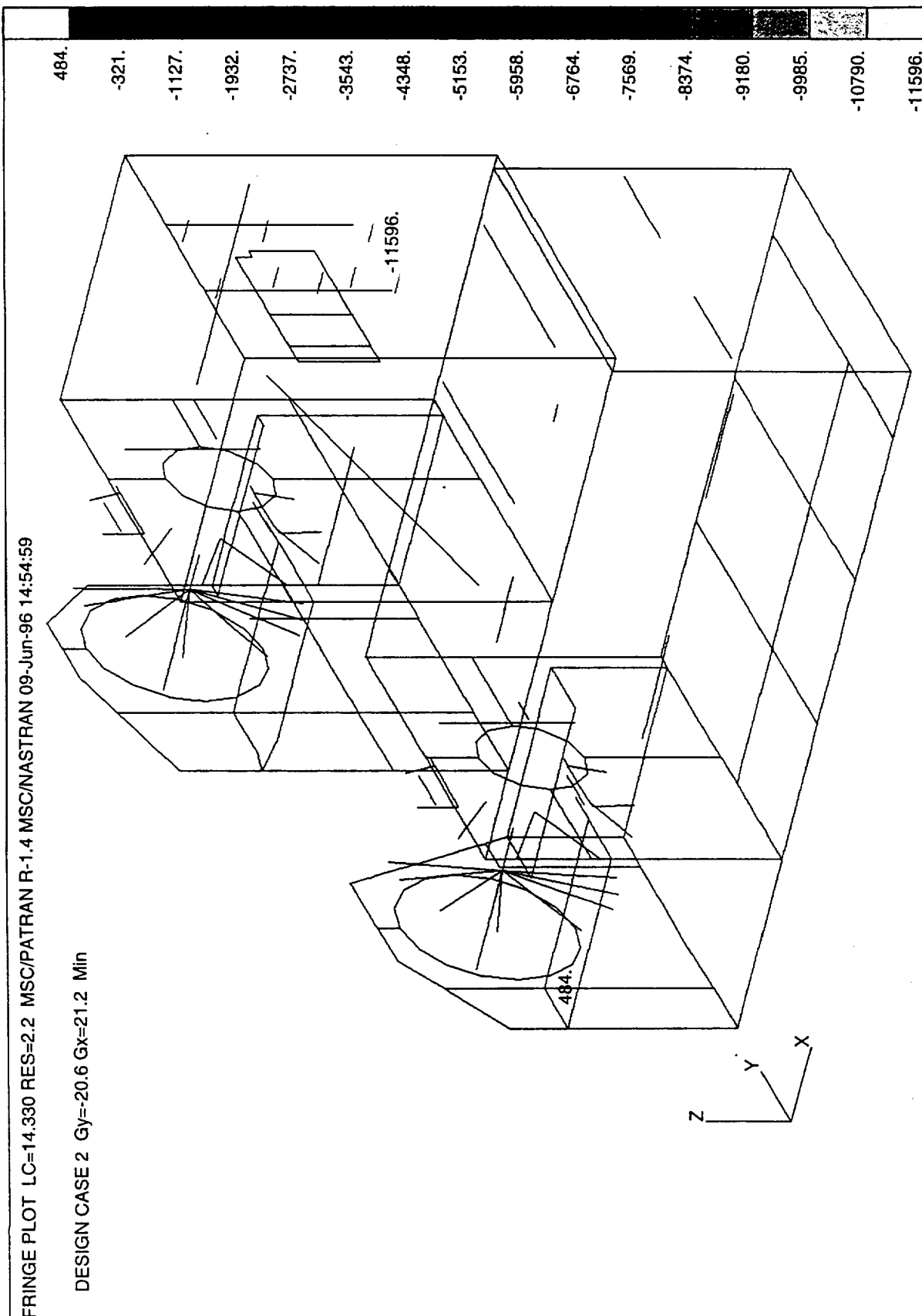


Figure 18 METSAT AMSU-A1 Beam Stresses, Case 2

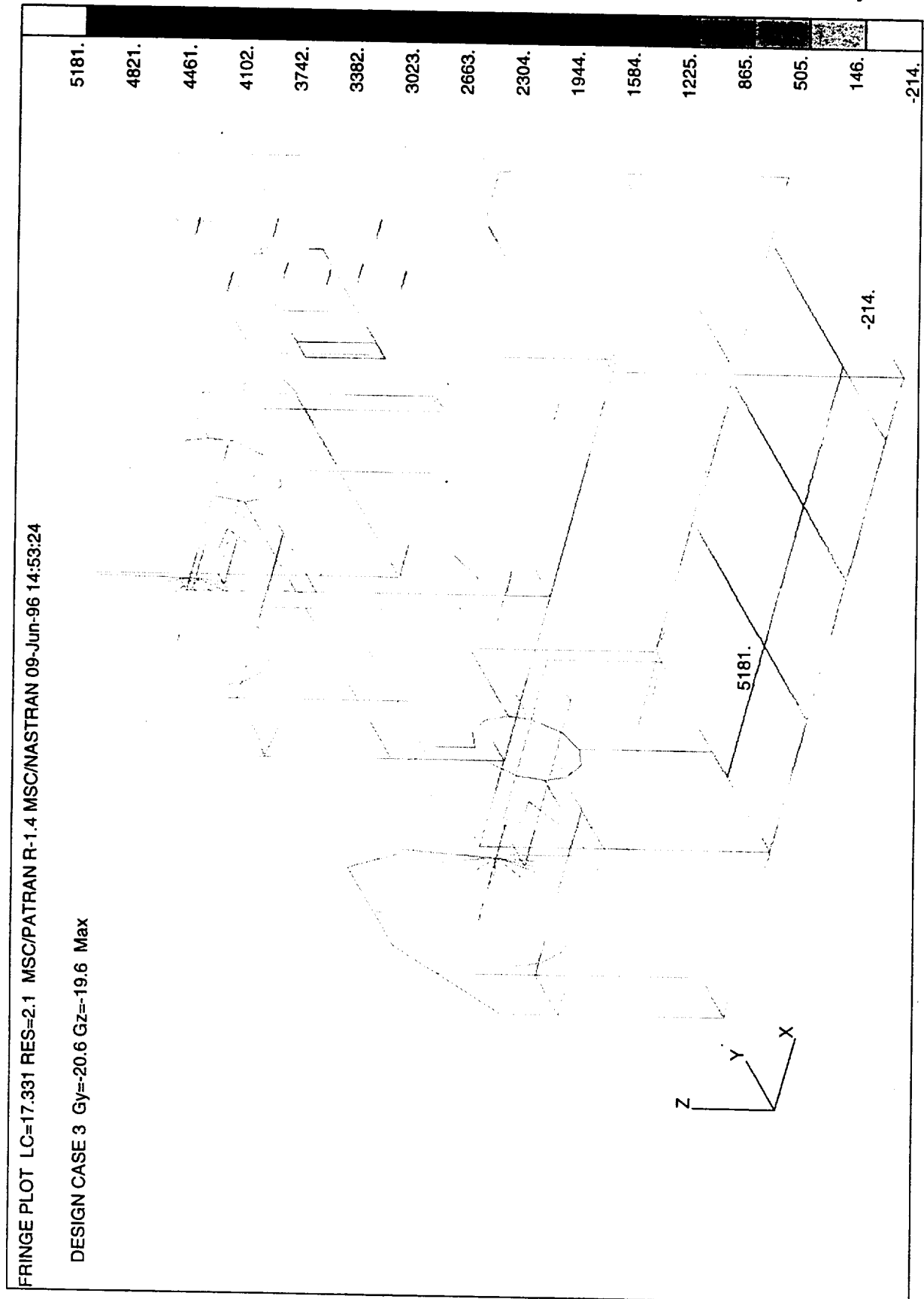


Figure 19 METSAT AMSU-A1 Beam Stresses, Case 3

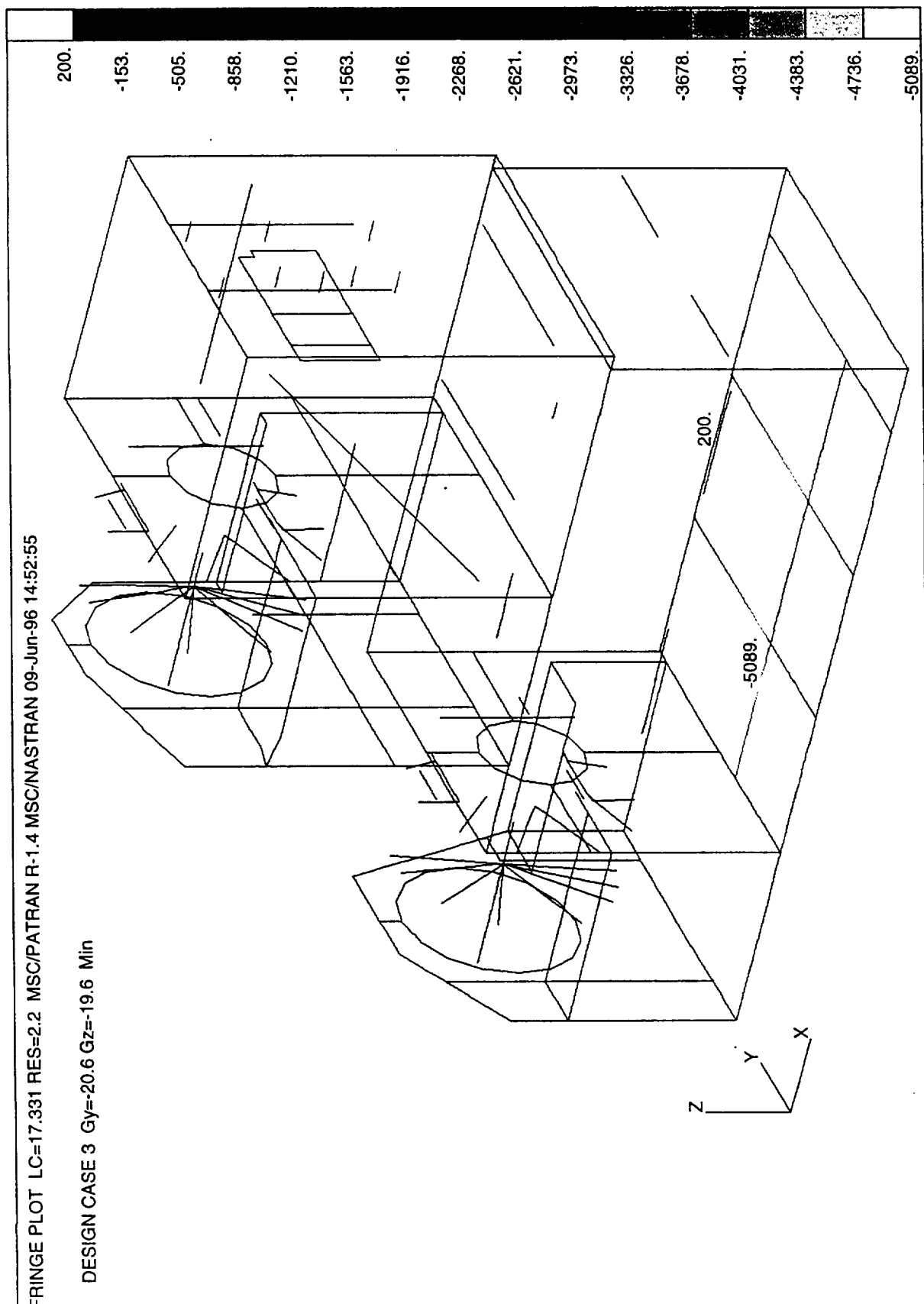


Figure 20 METSAT AMSU-A1 Beam Stresses, Case 3

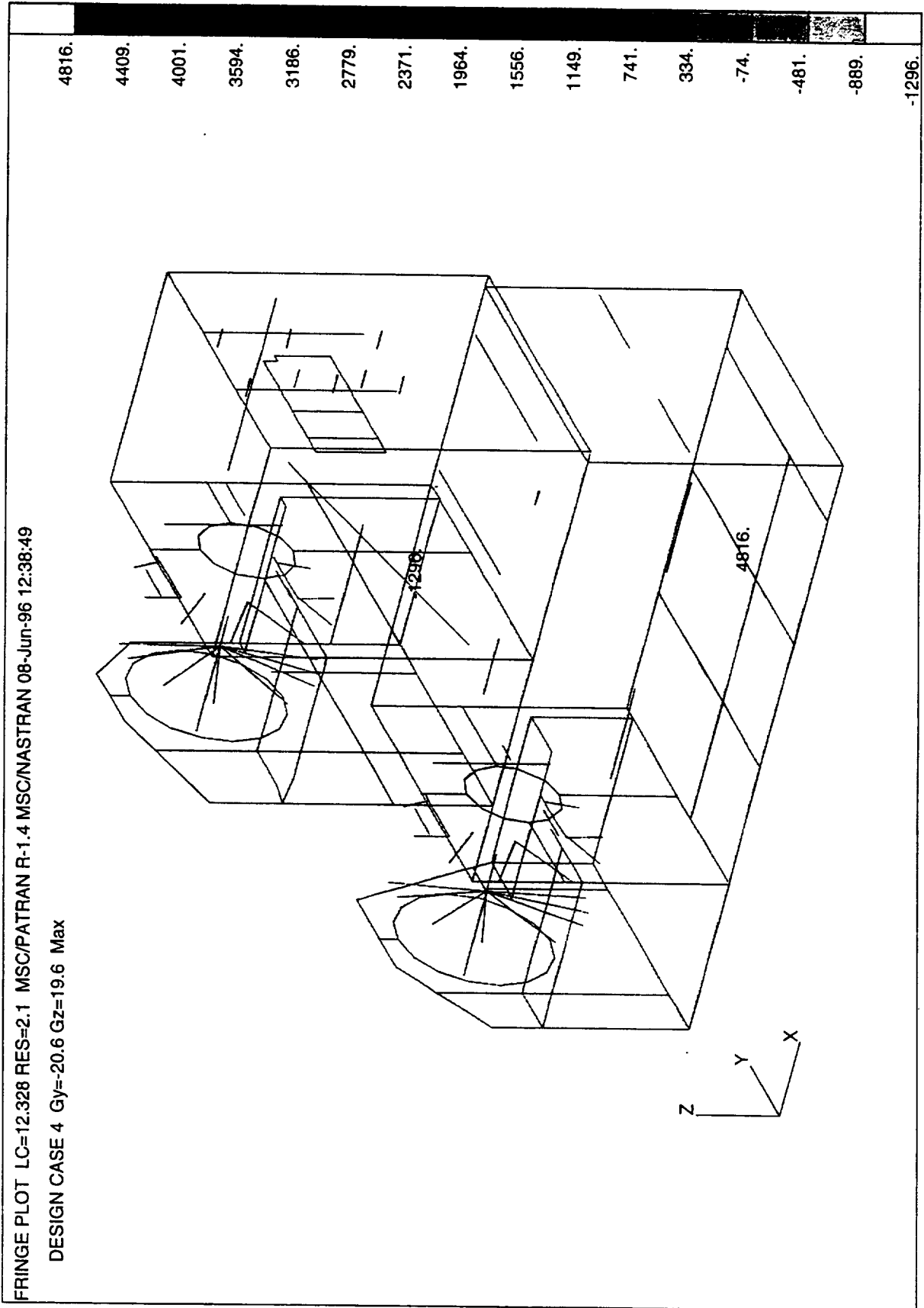


Figure 21 METSAT AMSU-A1 Beam Stresses, Case 4

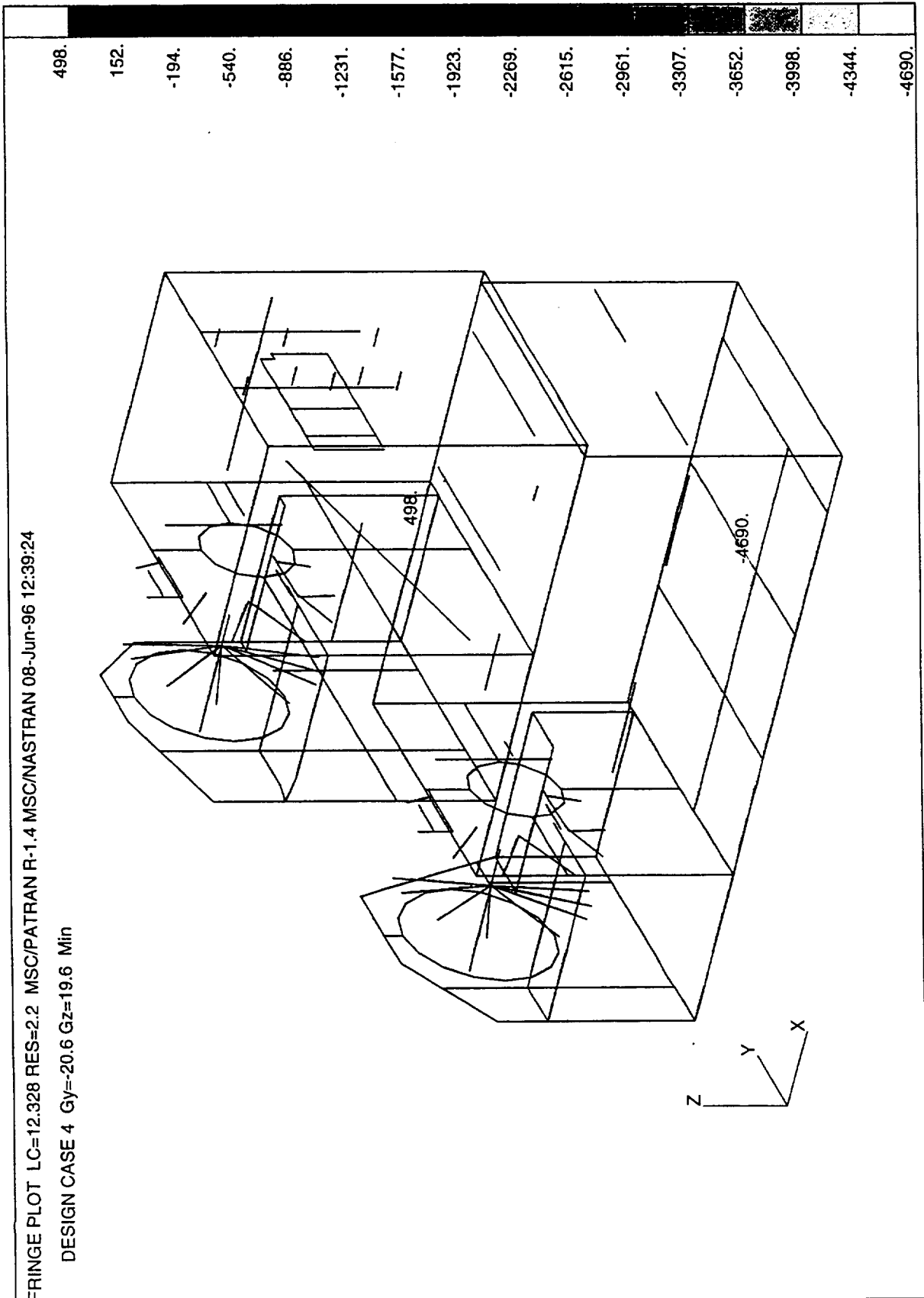


Figure 22 METSAT AMSU-A1 Beam Stresses, Case 4

Time: 15:08:03
Date: 07/03/96
Eigenvalues
Translational
Design_Gy=-20.6_Gz=19.6
Mode 1 : Frequency = 101.7
Max. Deformation =
9.230436E+01
@Node 11060

Report 10805A
July 1996

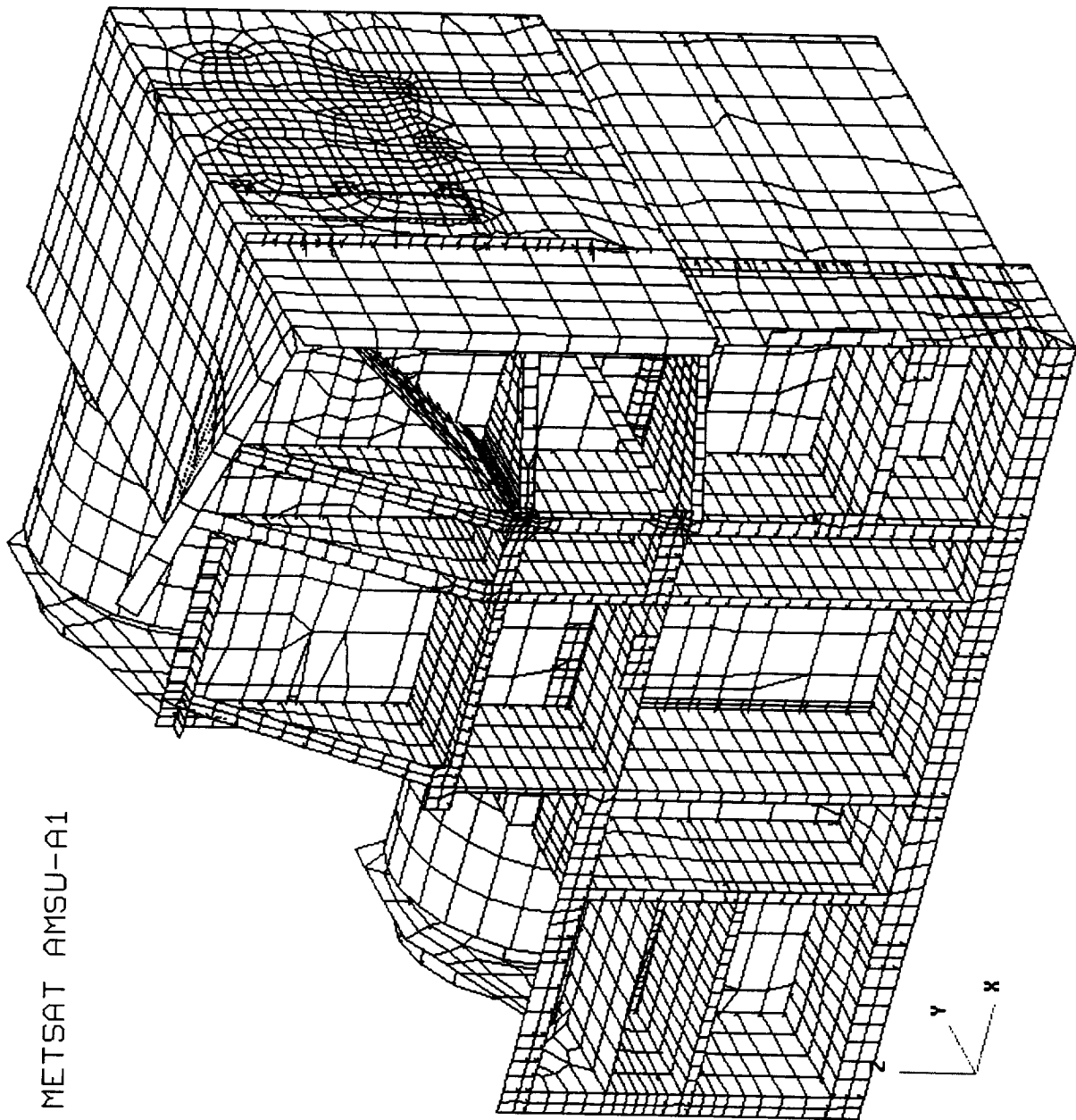


Figure 23 METSAT AMSU-A1 Mode 1, 101.8 Hz

Time: 15:12:33
Date: 07/03/96
Eigenvalues
Translational
Design_Gy=-20.6_Gz=19.6
Mode 2 : Frequency = 108.99
Max. Deformation =
4.339891E+01
@Node 4730

METSAT AMSU-A1

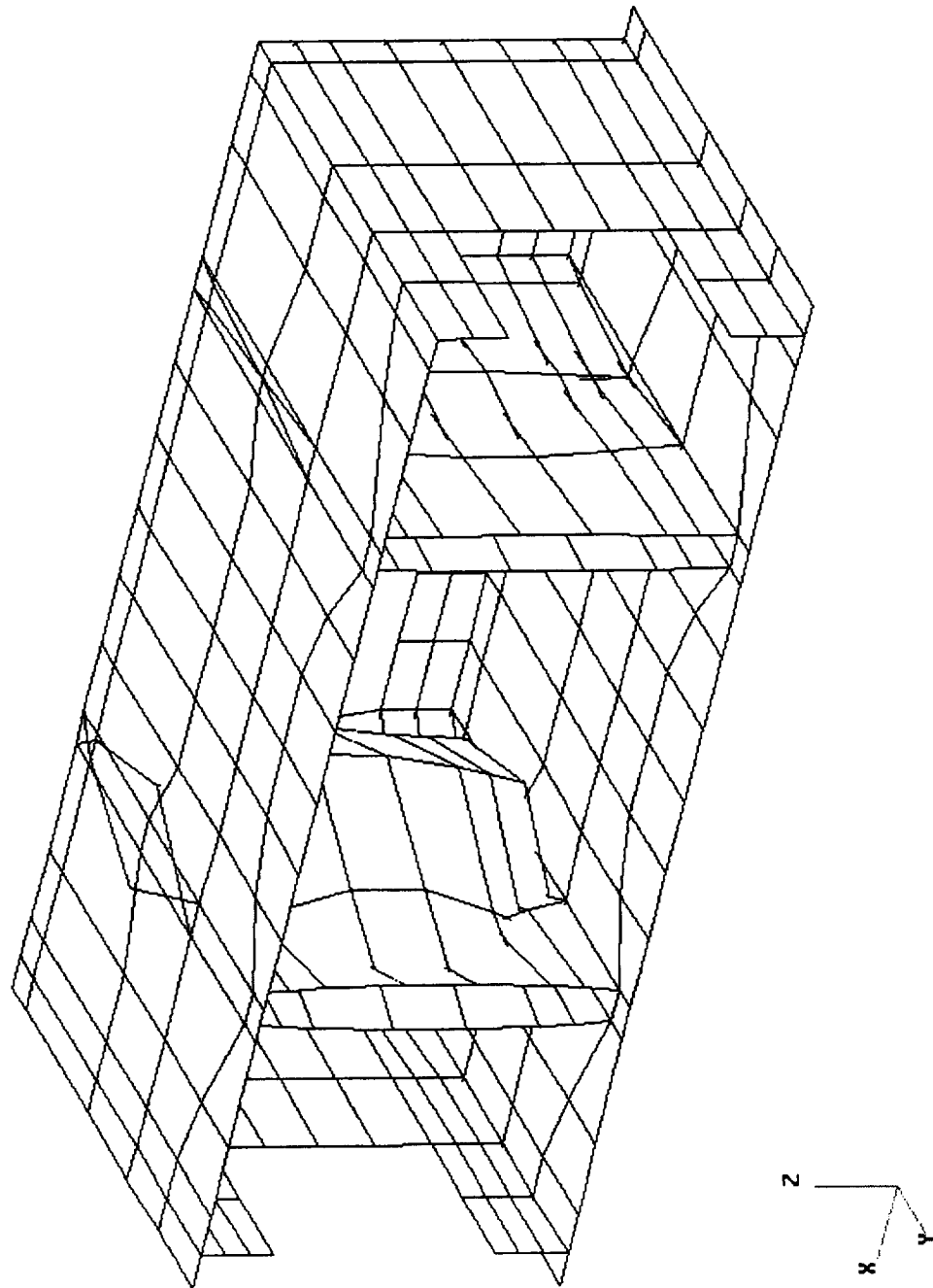


Figure 24 METSAT AMSU-A1 Mode 2, 109.0 Hz

Time: 15:13:37
Date: 07/03/96
Eigenvalues
Translational
Design_Gy=-20.6_Gz=19.6
Mode 3 : Frequency = 108.97
Max. Deformation =
4.282585E+01
@Node 4850

Report 10805A
July 1996

METSAT AMSU-A1

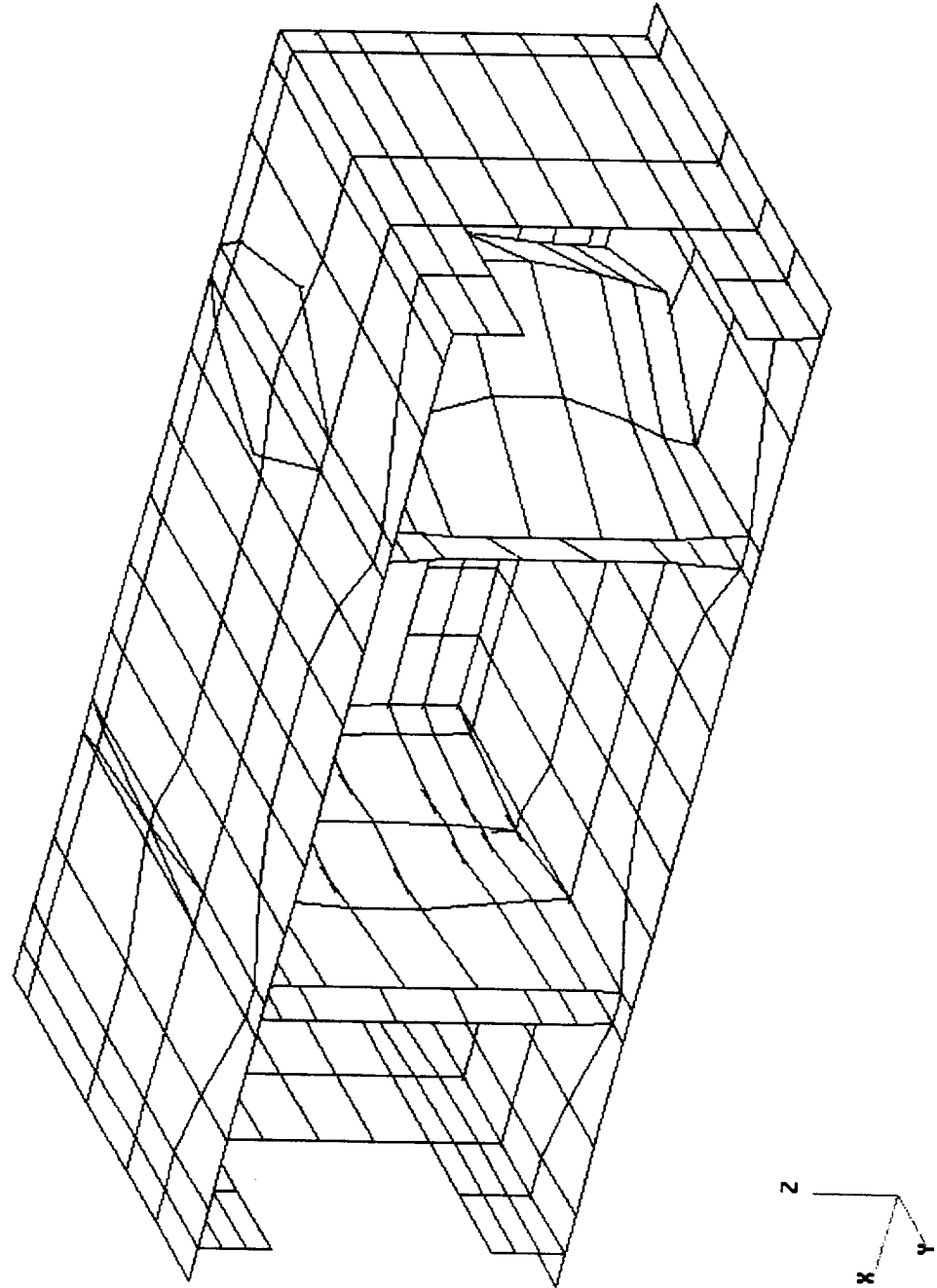


Figure 25 METSAT AMSU-A1 Mode 3, 109.0 Hz

Time: 15:14:37
Date: 07/03/96
Eigenvalues
Translational
Design_Gy=-20.6_Gz=19.6
Mode 4 : Frequency = 109.49
Max. Deformation =
3.202608E+01
eNode 4770

METSAT AMSU-A1

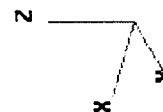
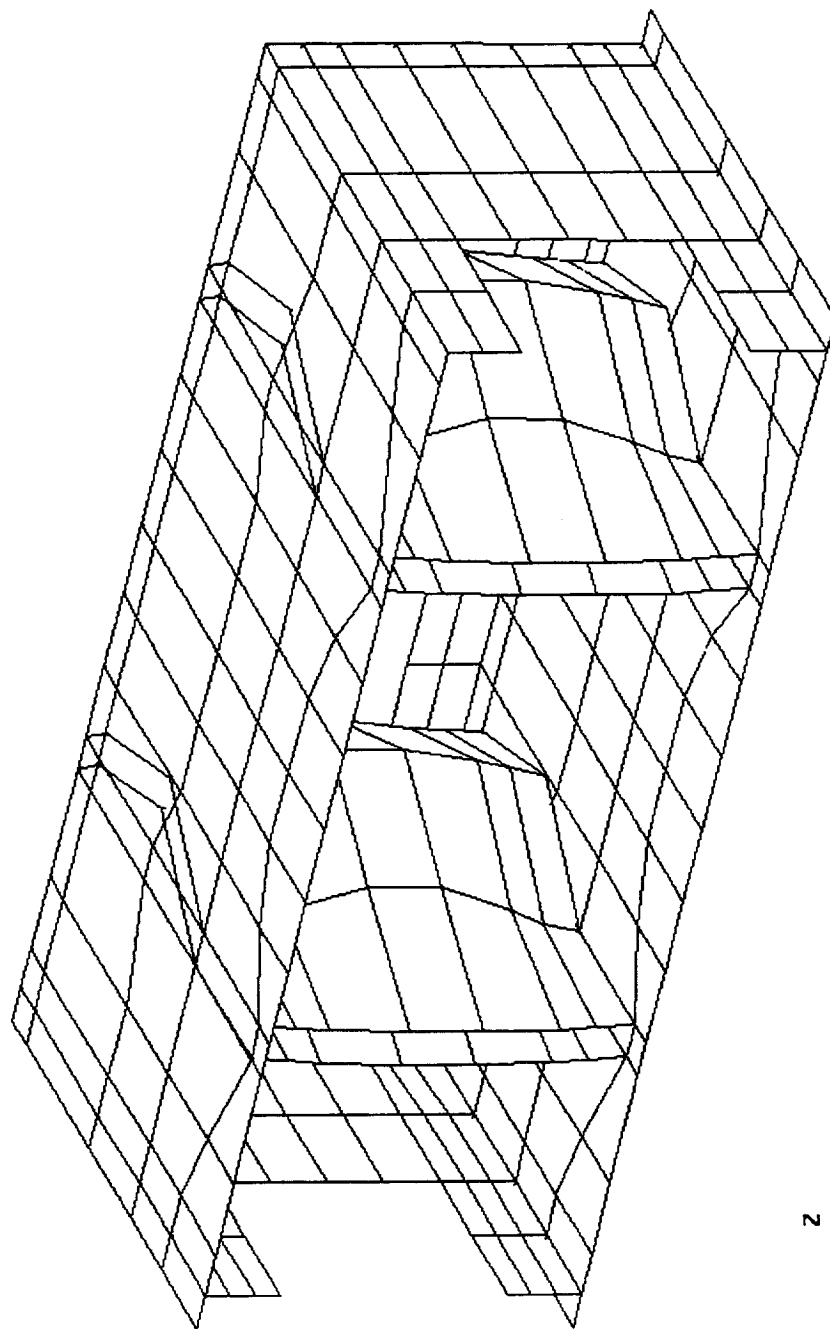


Figure 26 METSAT AMSU-A1 Mode 4, 109.5 Hz

Time: 15:15:44
Date: 07/03/96
EigenVectors
Translational
Design_Gy=-20.6_Gz=19.6
Mode 5 : Frequency = 109.74
Max. Deformation =
3.177974E+01
eNode 4810

Report 10805A
July 1996

METSAT AMSU-A1

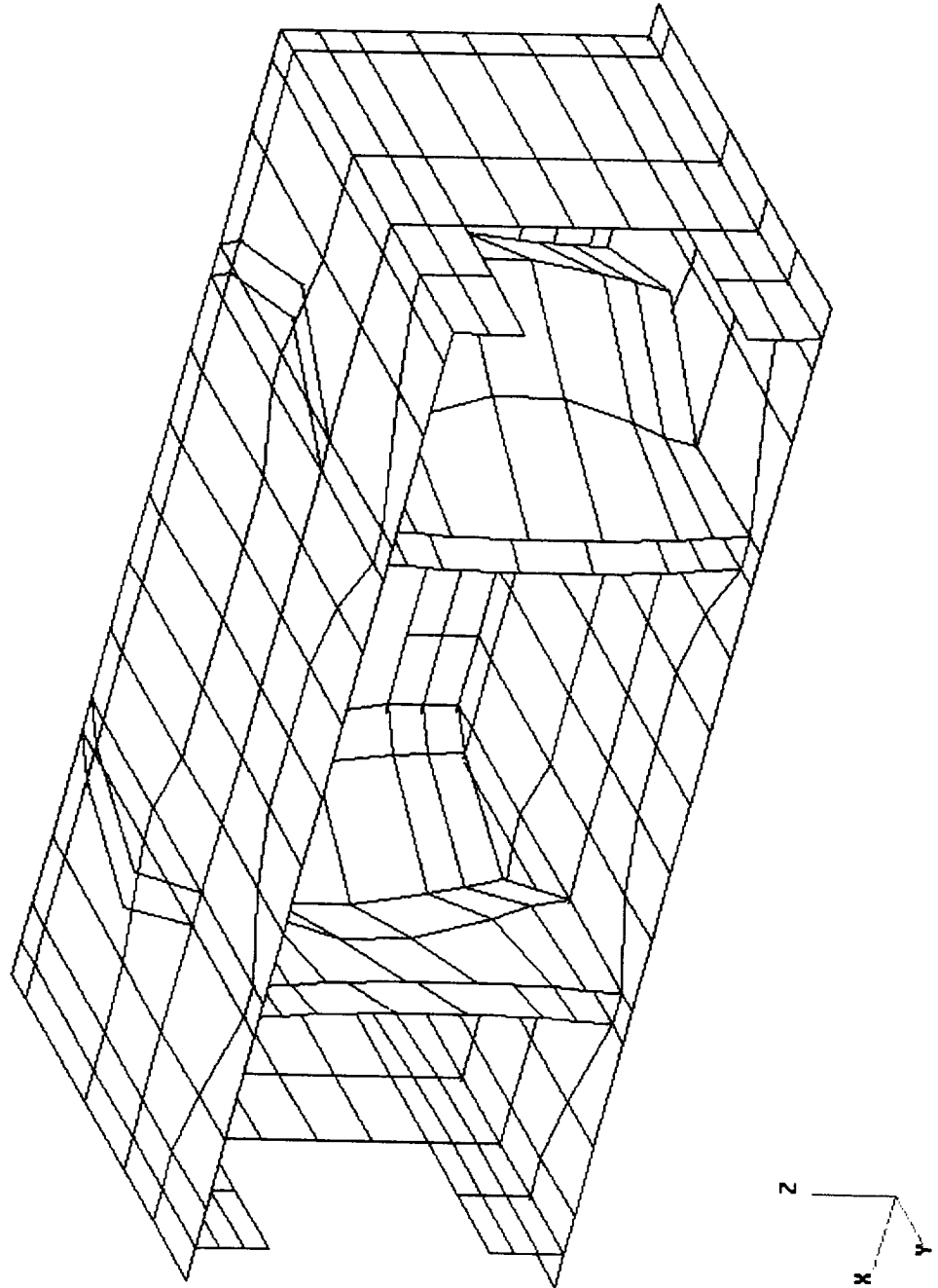


Figure 27 METSAT AMSU-A1 Mode 5, 109.7 Hz

Time: 15:09:25
Date: 07/03/96
Eigenvalues
Translational
Design_Gy=-20.6_Gz=19.6
Mode 6 : Frequency = 120.35
Max. Deformation =
1.391827E+01
@Node 6292

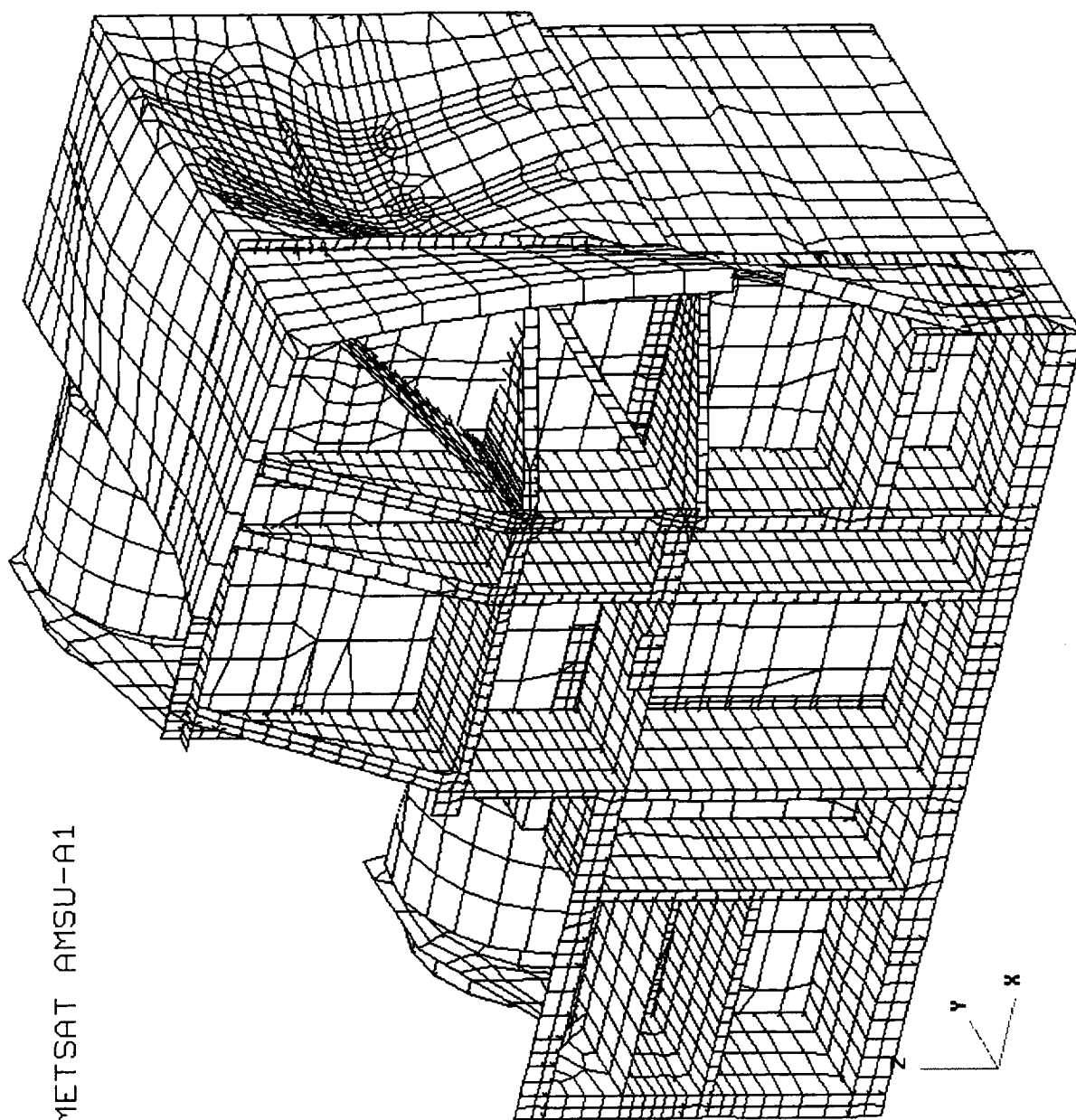


Figure 28 METSAT AMSU-A1 Mode 6, 120.4 Hz

Time: 15:30:54
Date: 07/03/96
Eigenvalues
Translational
Design_Gy=-20.6_Gz=19.6
Mode 7 : Frequency = 121.41
Max. Deformation =
4.582343E+01
@Node 4523

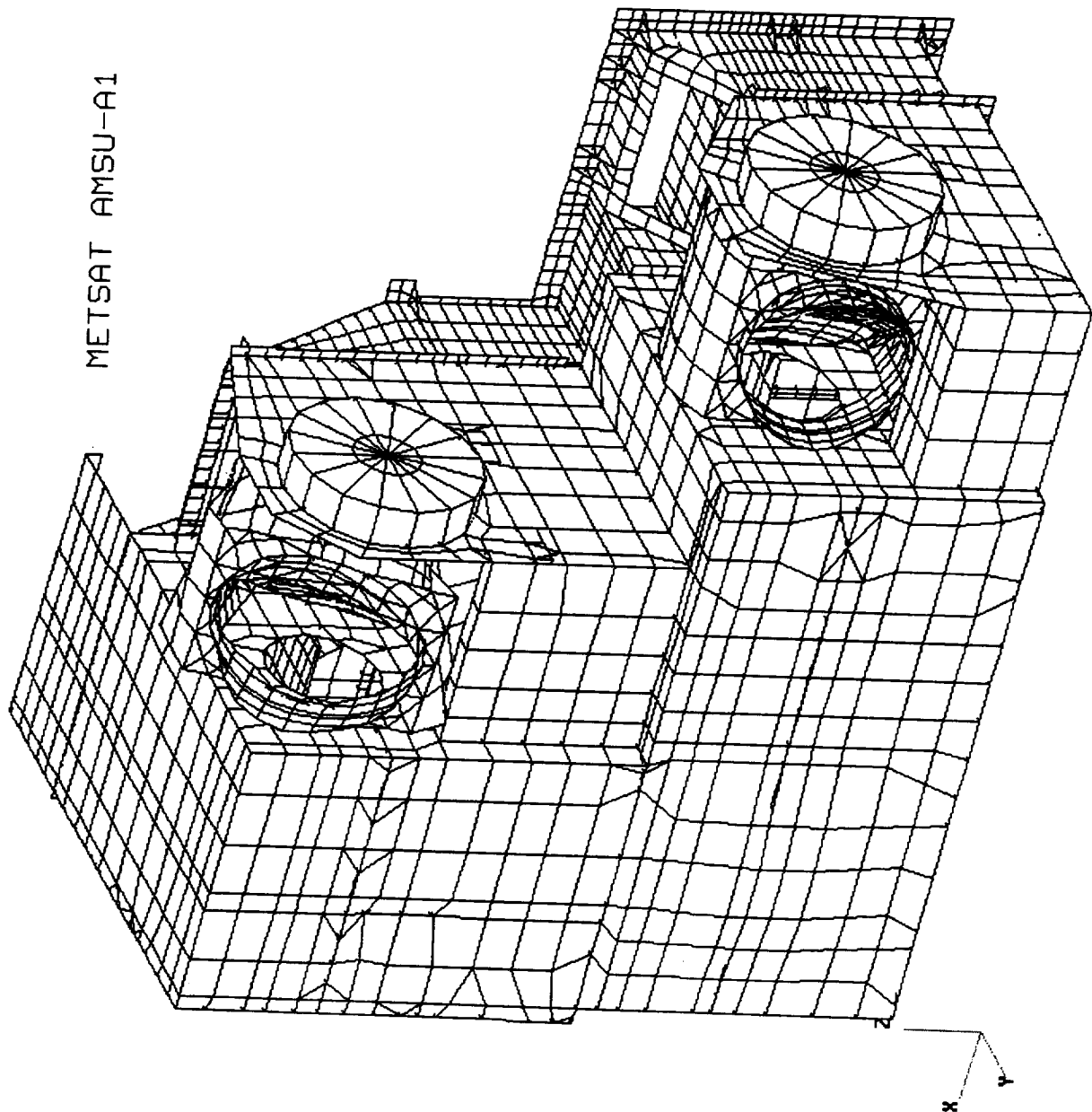


Figure 29 METSAT AMSU-A1 Mode 7, 121.4 Hz

Time: 15:03:46
Date: 07/03/96
Eigenvectors
Translational
Design_Gy=-20.6_Gz=19.6
Mode 8 : Frequency = 121.93
Max. Deformation =
3.847197E+01
@Node 1349

METSAT AMSU-A1

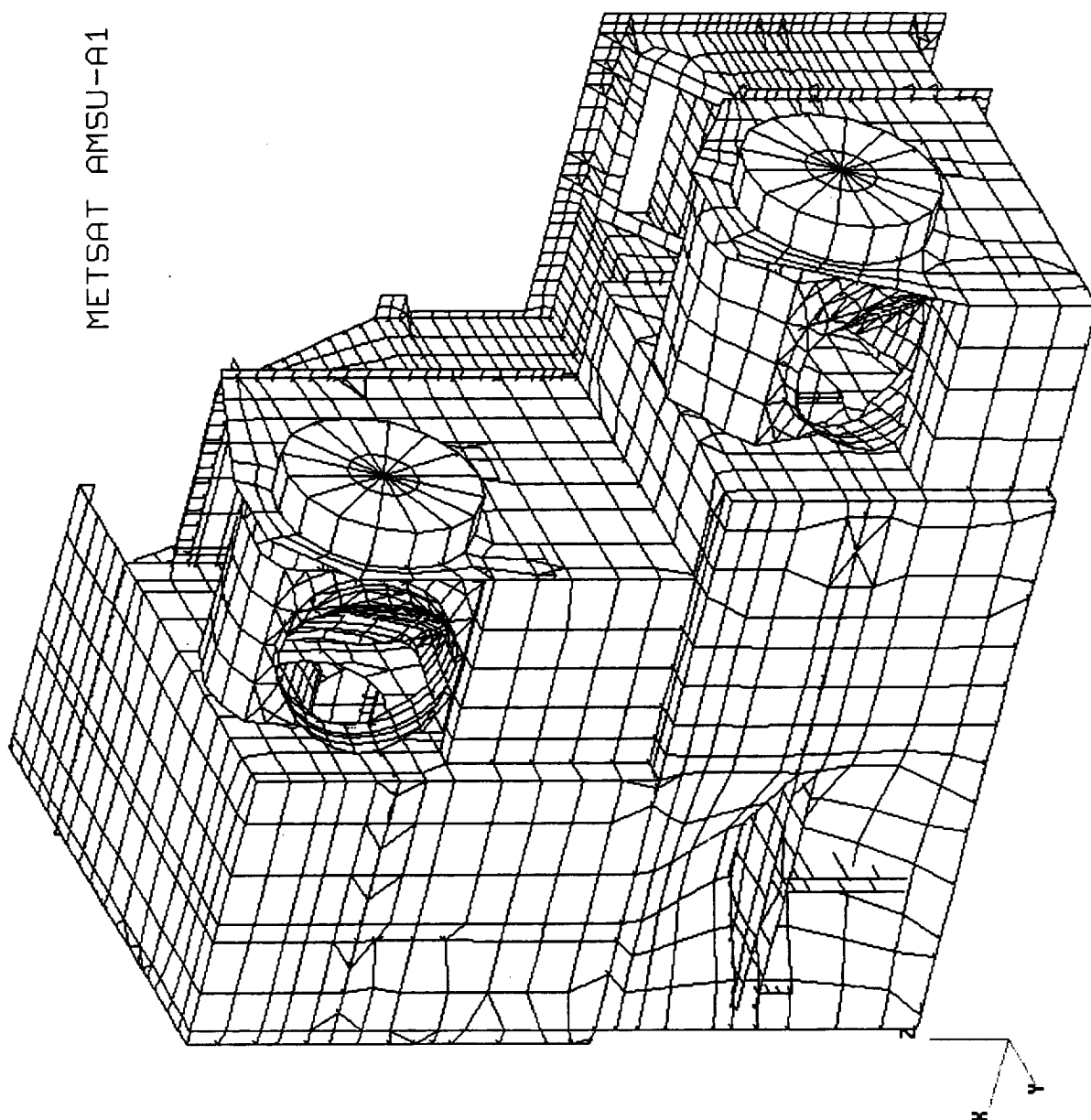


Figure 30 METSAT AMSU-A1 Mode 8, 121.9 Hz

Time: 15:23:14
Date: 07/03/96
Eigenvalues
Translational
Design_Gy=-20.6_Gz=19.6
Mode 9 : Frequency = 122.01
Max. Deformation =
3.923617E+01
@Node 4209

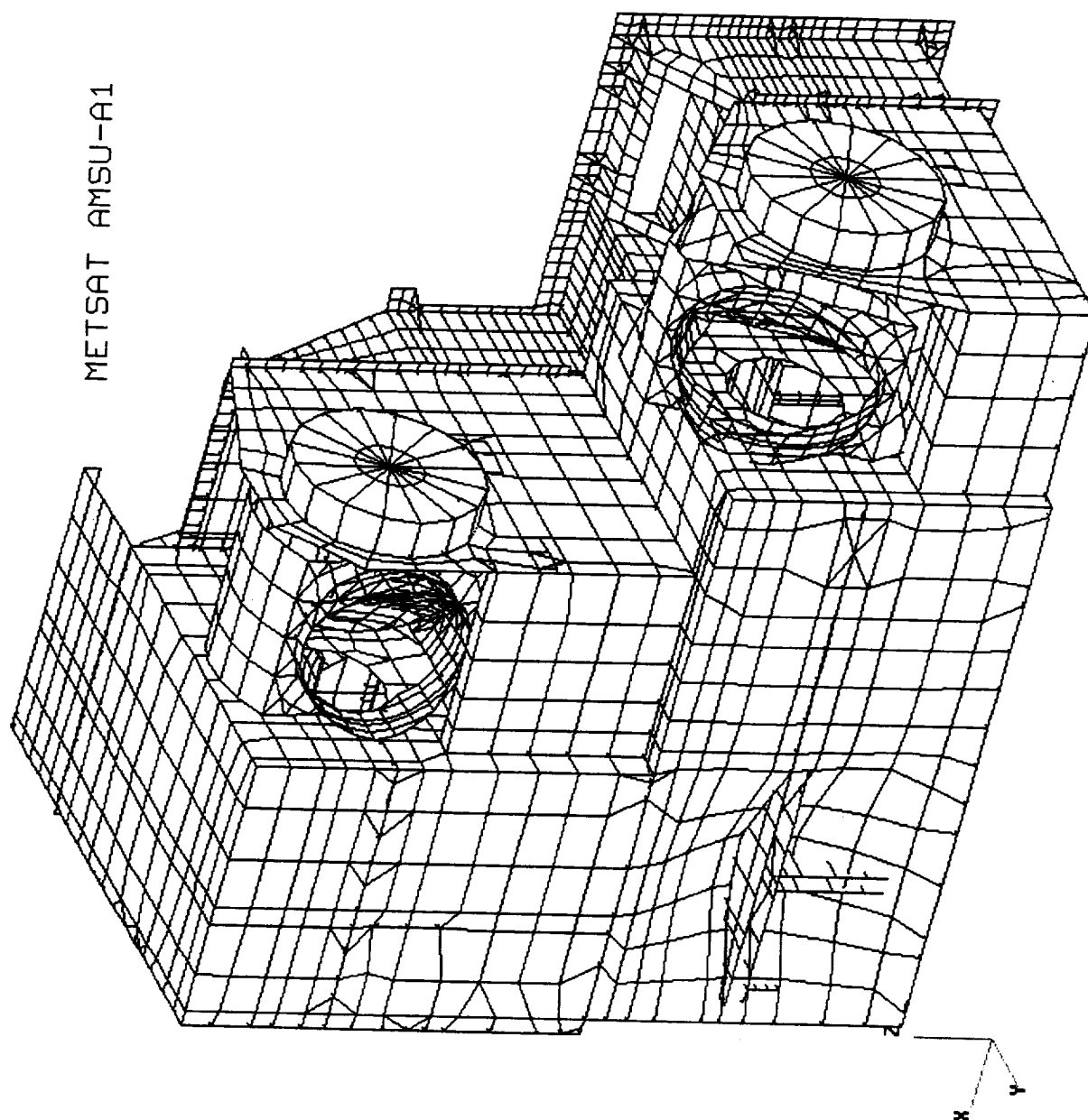


Figure 31 METSAT AMSU-A1 Mode 9, 122.0 Hz

Time: 15:32:31
Date: 07/03/96
Eigenvalues
Translational
Design_Gy=-20.6_Gz=19.6
Mode 10 : Frequency = 130.8
Max. Deformation =
3.556469E+01
@Node 11179

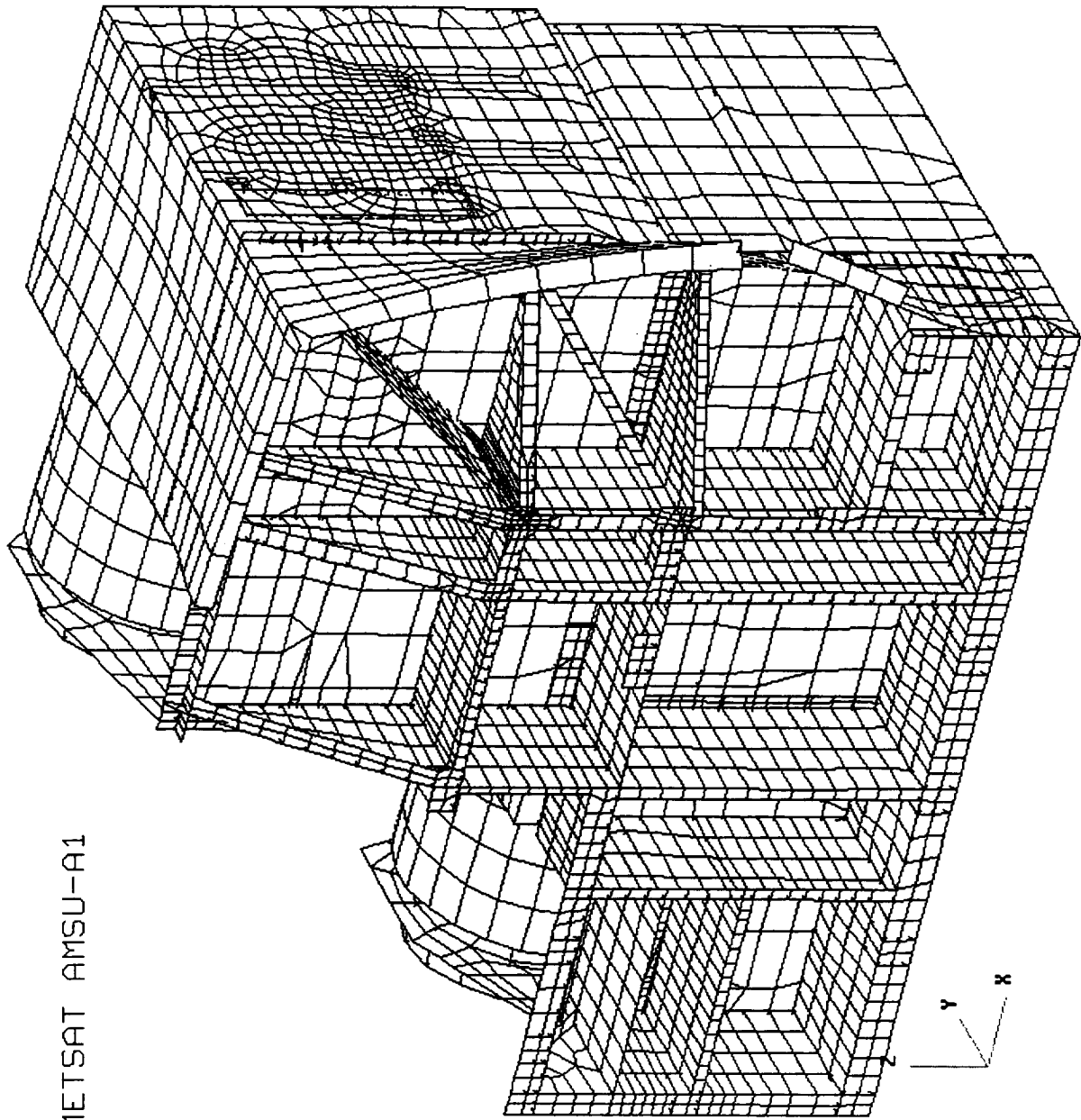


Figure 32 METSAT AMSU-A1 Mode 10, 130.8 Hz

Time: 15:33:57
Date: 07/03/96
Eigenvectors
Translational
Design_Gy=-20.6_Gz=19.6
Mode 11 : Frequency = 131.8
Max. Deformation =
3.565528E+01
@Node 11179

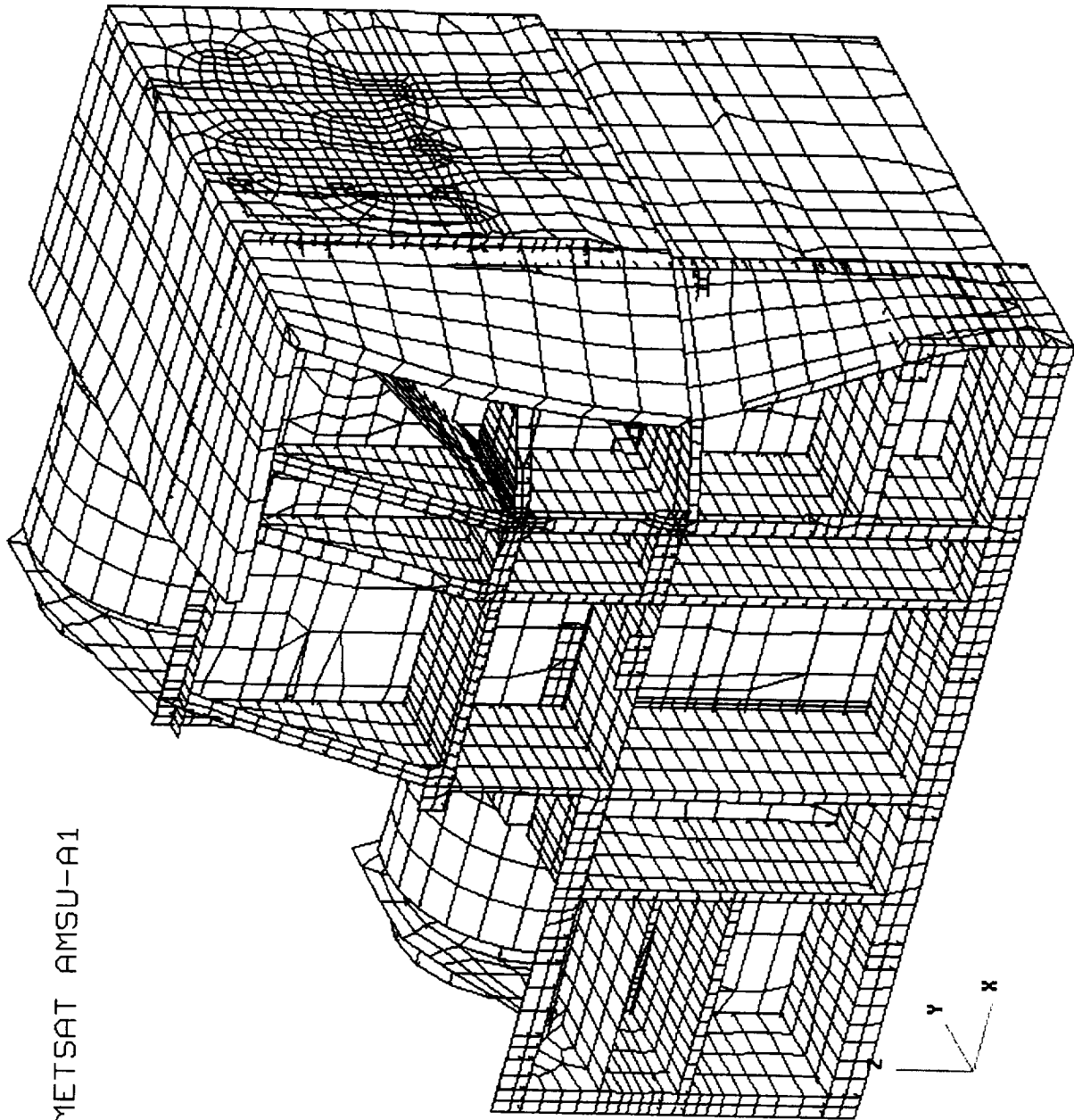


Figure 33 METSAT AMSU-A1 Mode 11, 131.8 Hz

Time: 15:35:26
Date: 07/03/96
Eigenvalues
Translational
Design_Gy=-20.6_Gz=19.6
Mode 12 : Frequency = 138.4
Max. Deformation =
4.224696E+01
@Node 4001

METSAT AMSU-A1

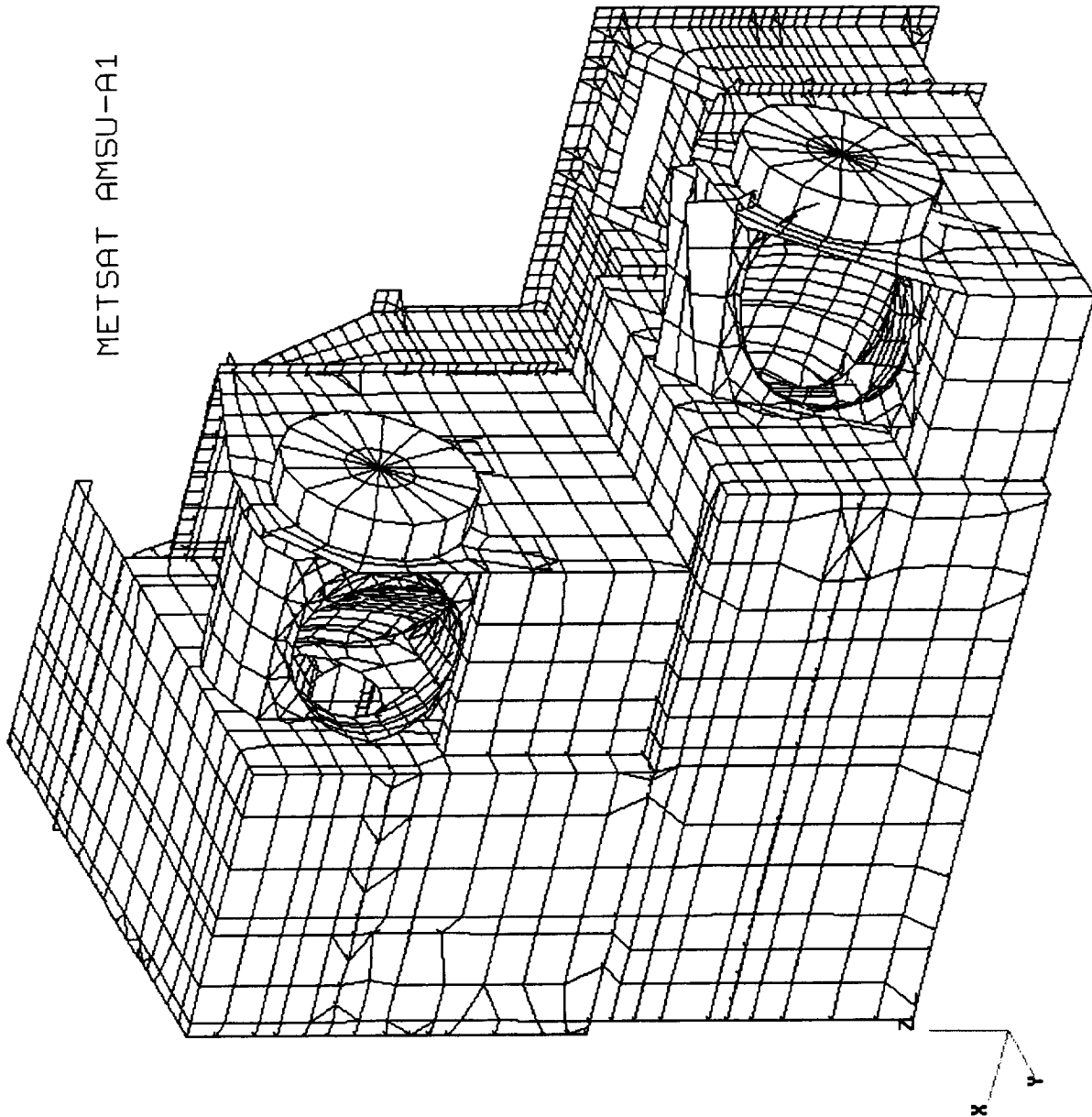
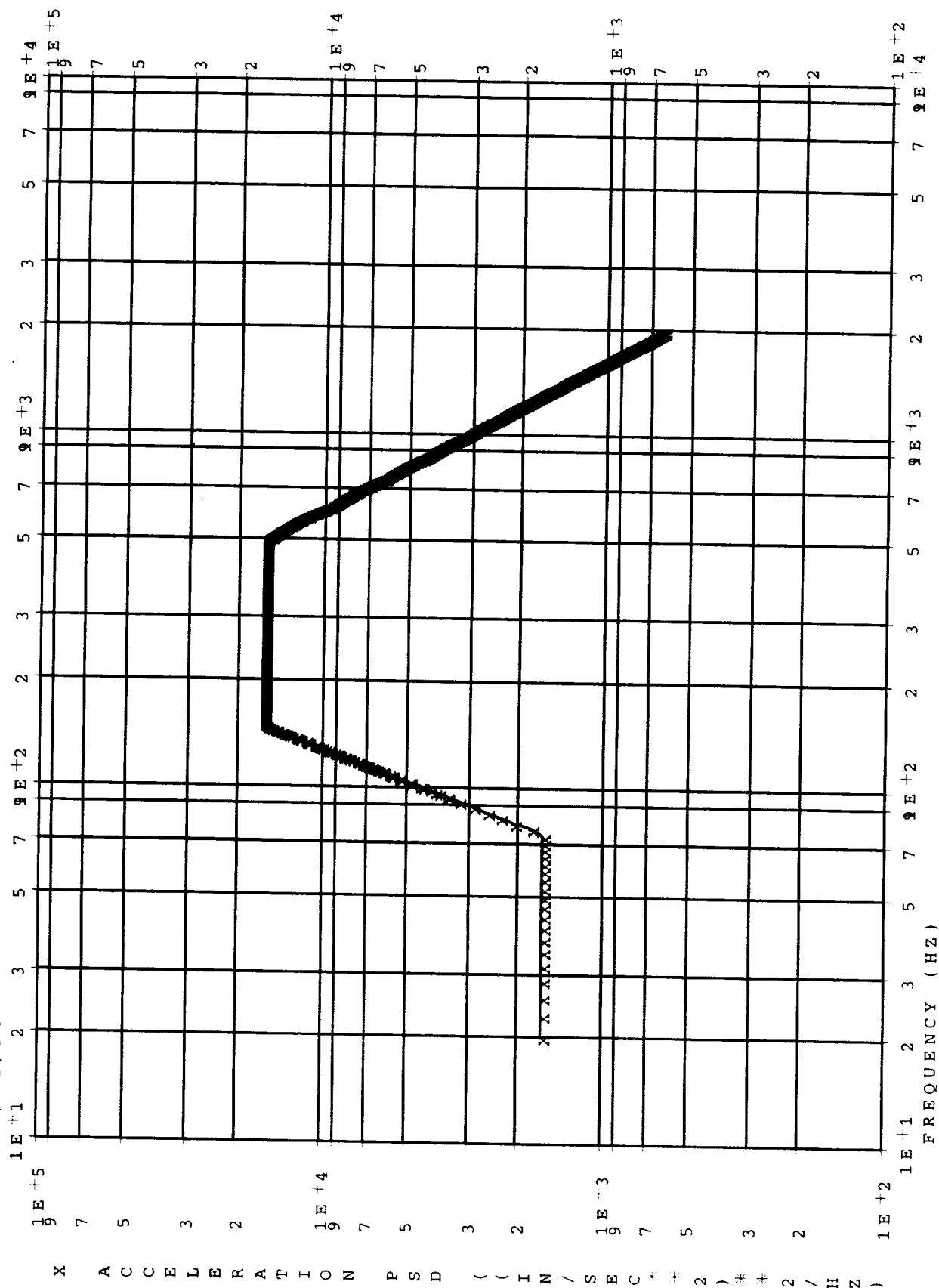


Figure 34 METSAT AMSU-A1 Mode 12, 138.4 Hz

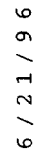
6/21/96

1



METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) X-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ

Figure 35 METSAT AMSU-A1 Input Random Vibration Input
Random Vibration non-Gaussian 0.04 sec V.D.

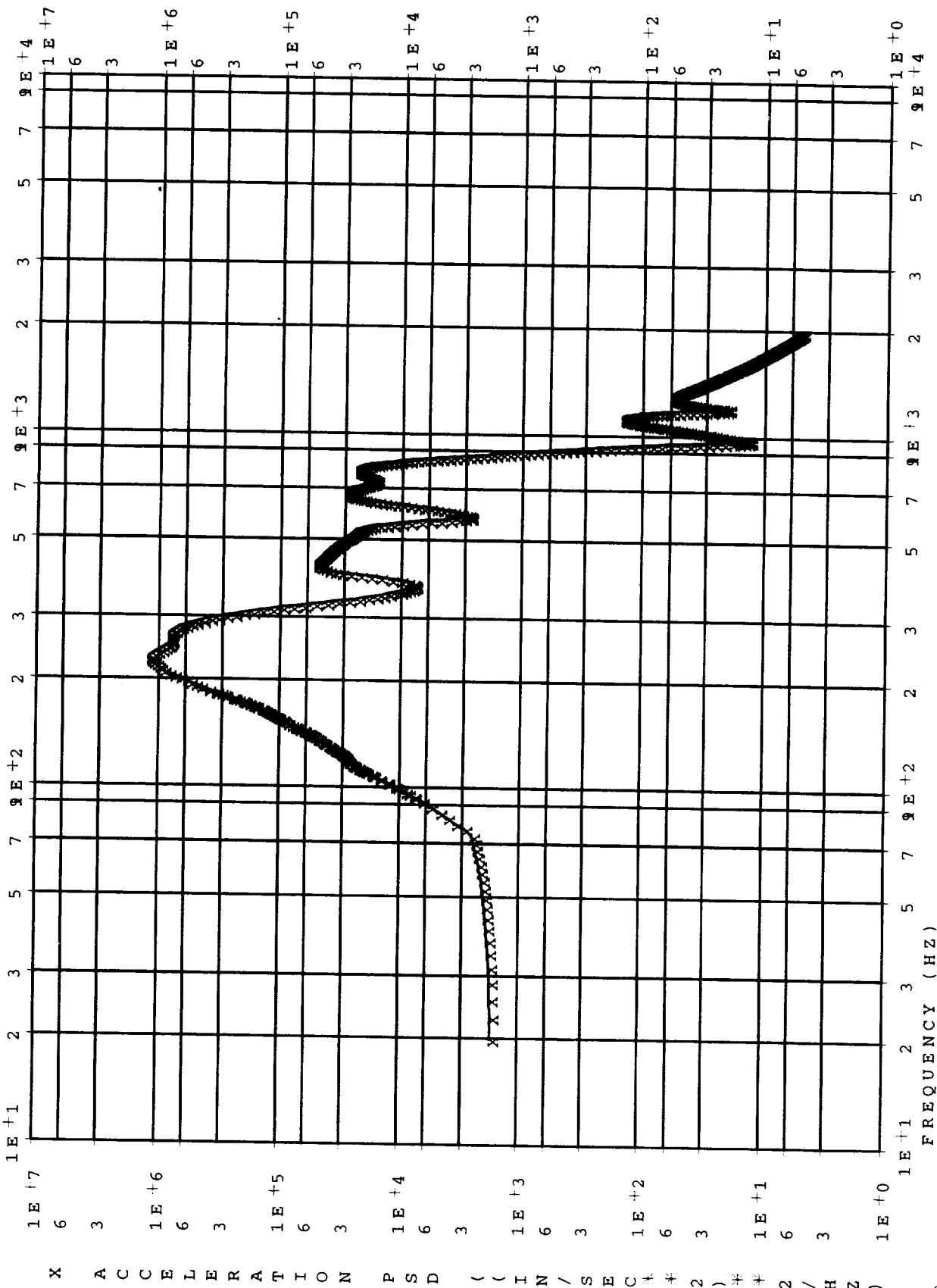


TEST NAME	TEST TYPE	TEST DATE	TEST TIME	TEST RESULT	TEST STATUS	TEST COMMENTS
AMTSU-A1	PSD INPUT	0-2000	HZ			

Figure 36 METSAT AMSU-A1 Random Vibration PSD Curve Response, Upper Motor Mount Panel X-Load X-Response, 28.11 grms

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METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) X-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ

Figure 37 METSAT AMSU-A1 Random Vibration PSD Curve Response,

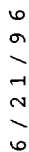


Figure 38 METSAT AMSU-A1 Random Vibration PSD Curve Response, Lower Right Front Support X-Load Y-Response, 26.48 grms

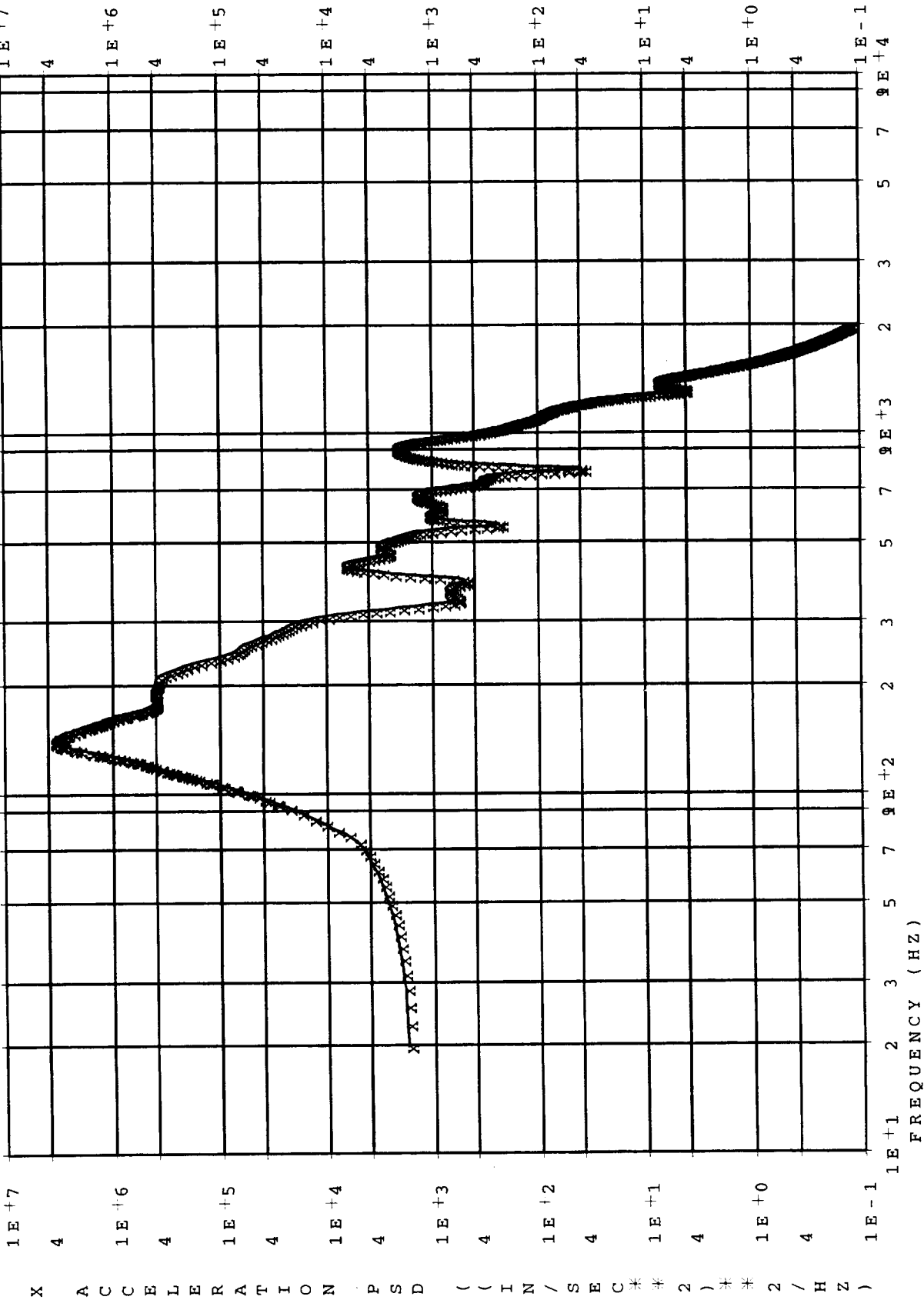


Figure 39 METSAT AMSU-A1 Random Vibration PSD Curve Response,
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METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) X-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ
Upper Card Cage Card X-Load X-Response, 29.14 grms

Figure 40 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Upper Card Cage Card X-Load X-Response, 29.14 grms

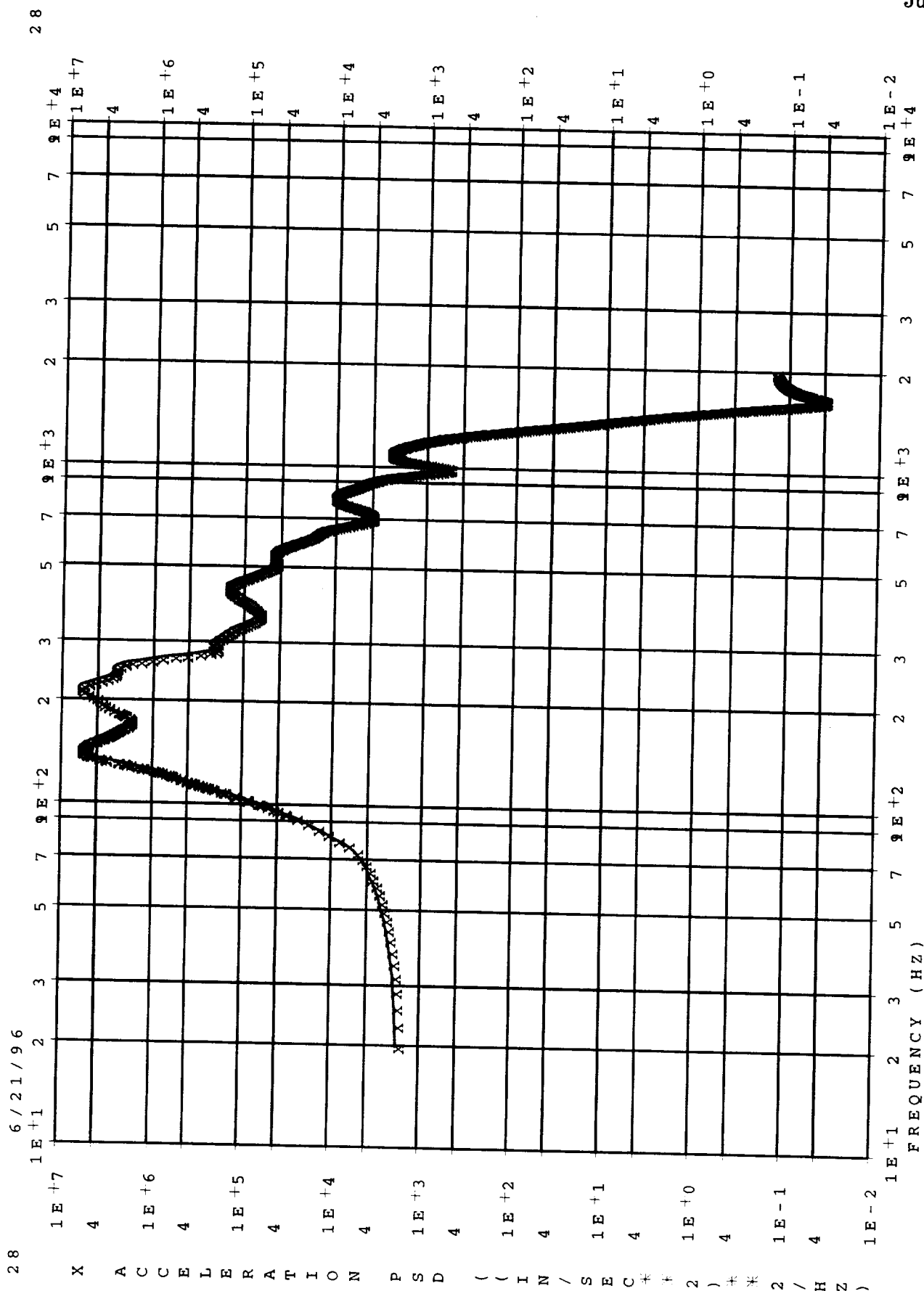
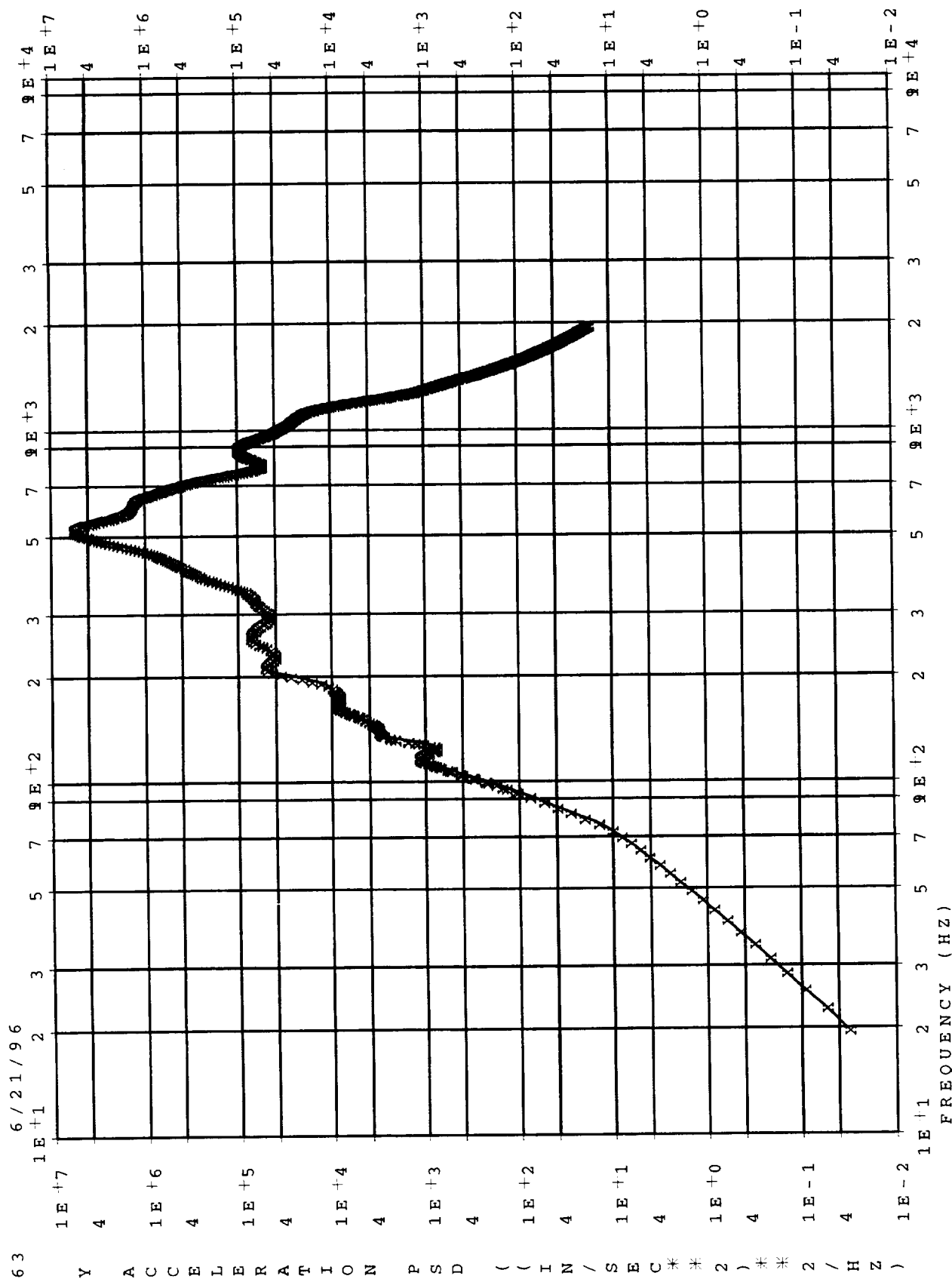
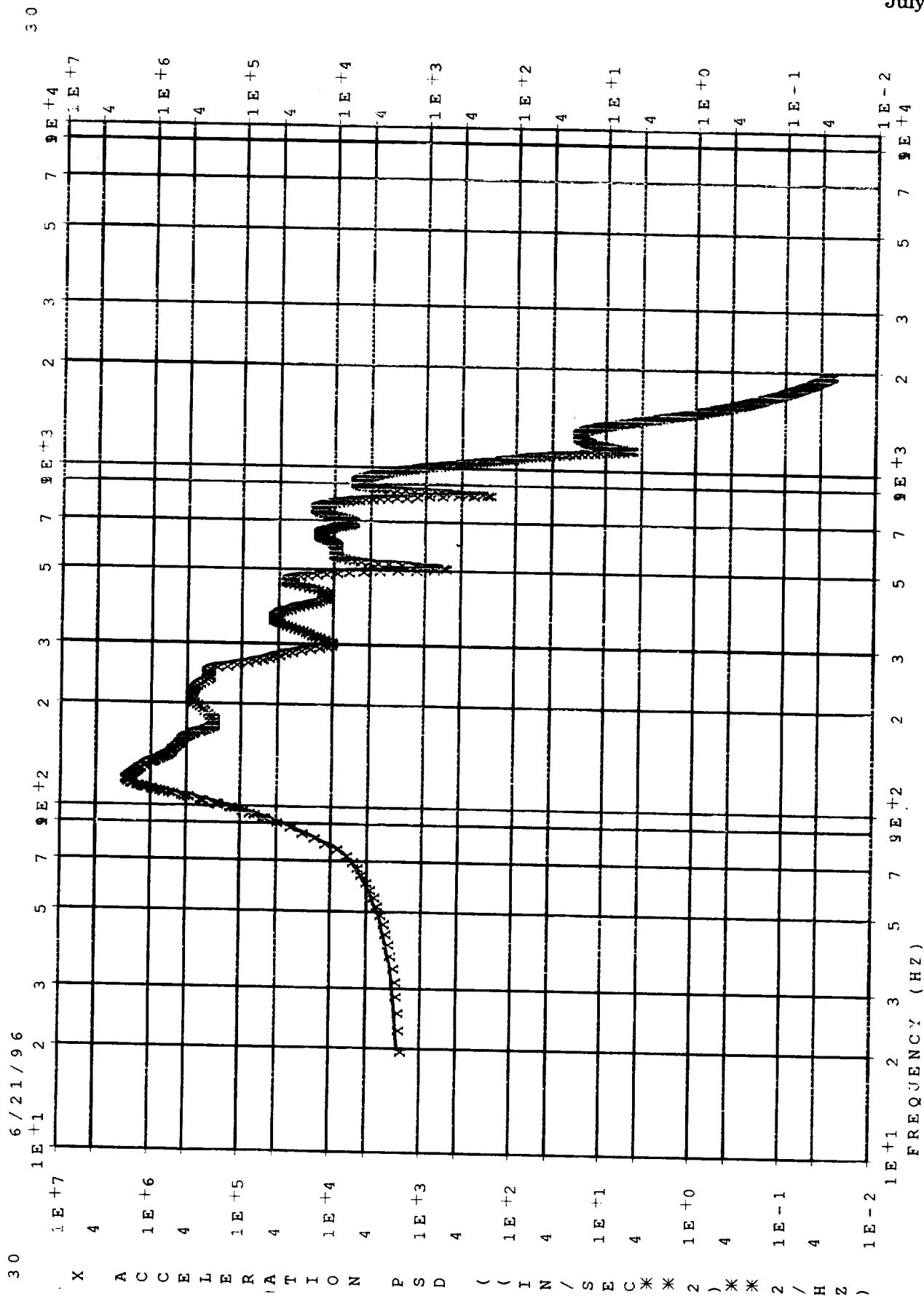


Figure 41 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Lower Reflector X-Load X-Response, 61.42 grms

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METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) X-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ
Figure 42 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Power Control/Monitor Bracket X-Load Y-Response, 68.91 grms



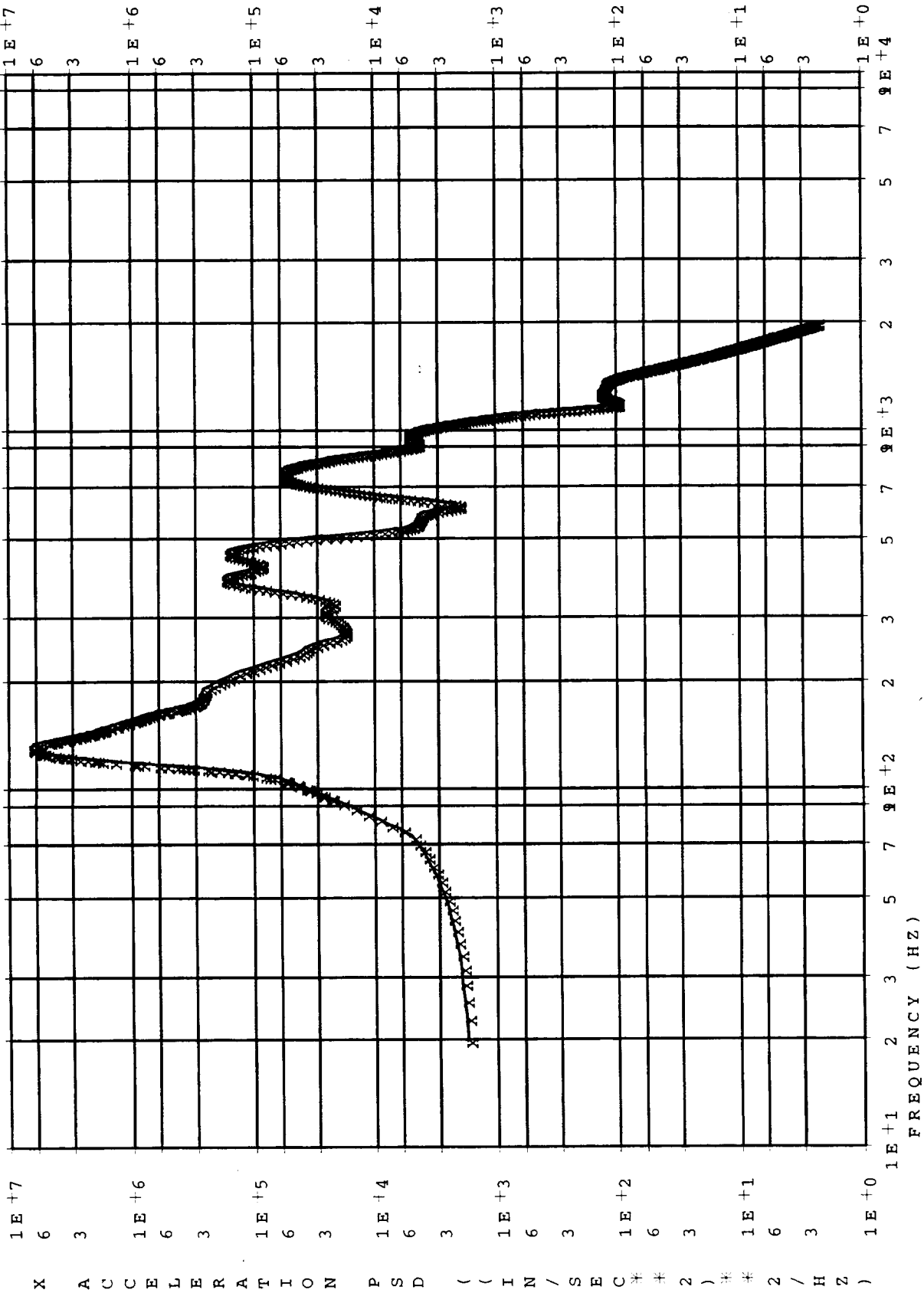
METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) X-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ

Figure 43 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Power Control/Monitor PWB X-Load X-Response, 25.95 grms

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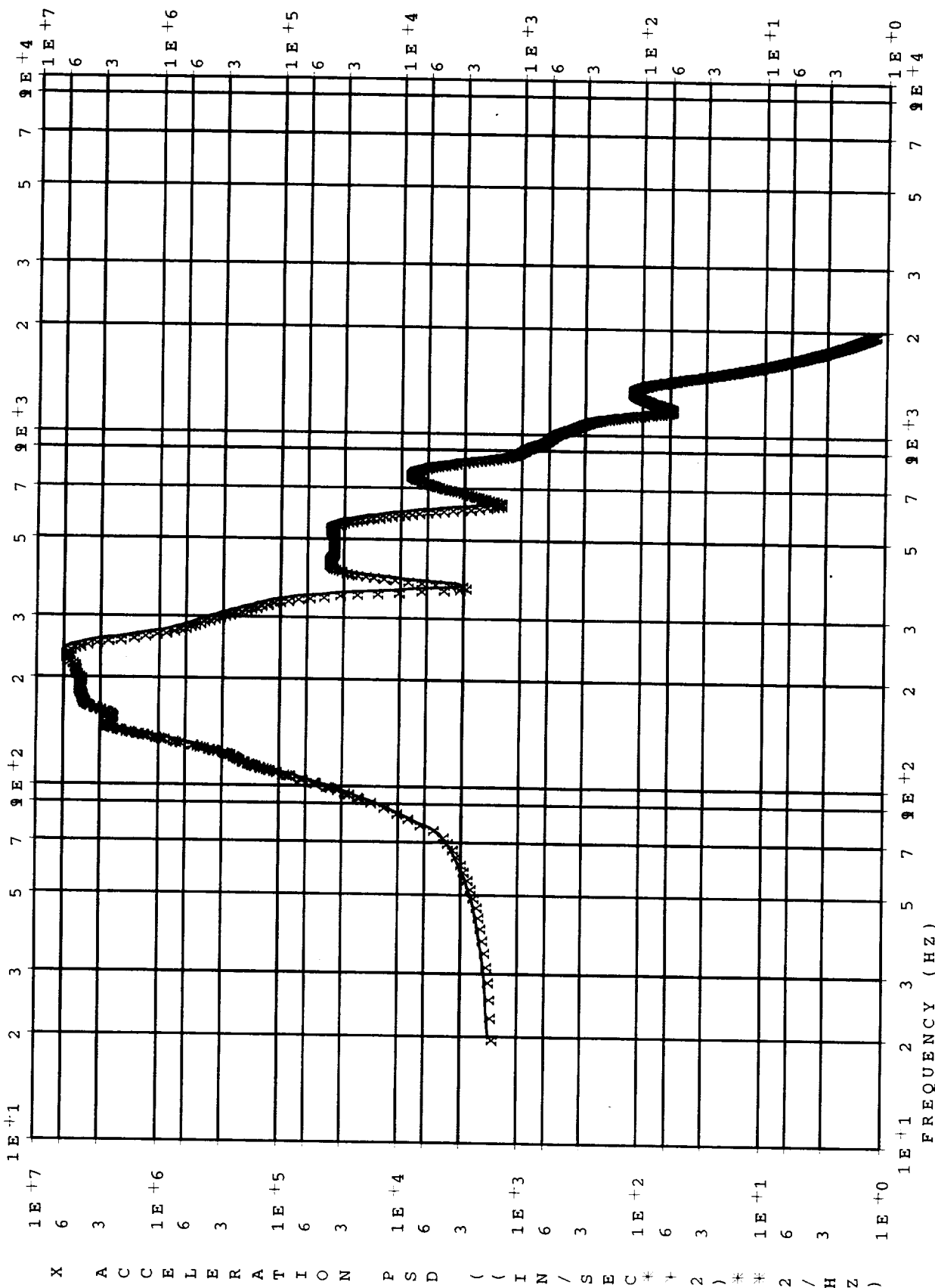


METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) X-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ
Figure 44 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Radiator Panel X-Load X-Response, 35.45 grms

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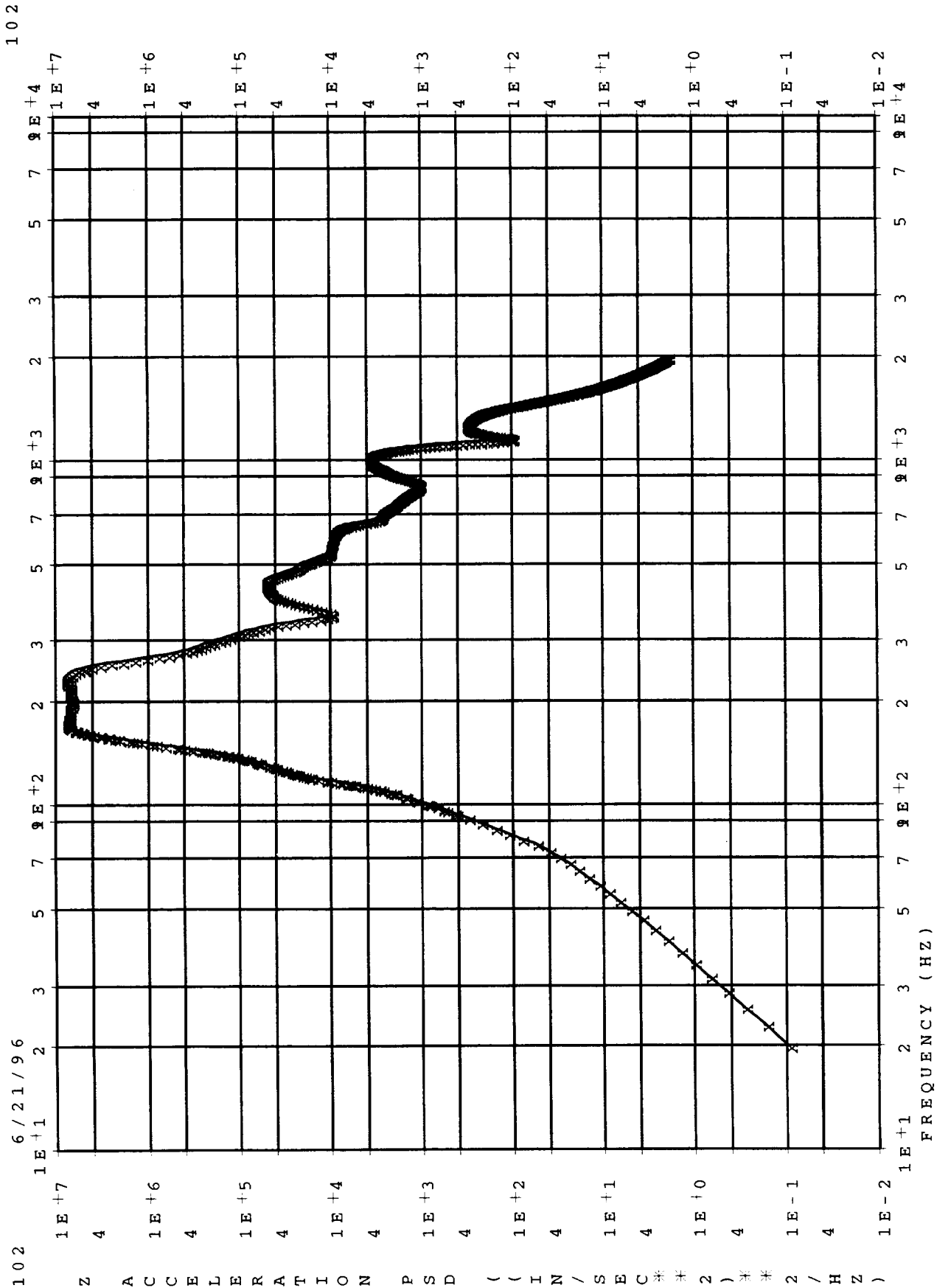
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METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) X-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ

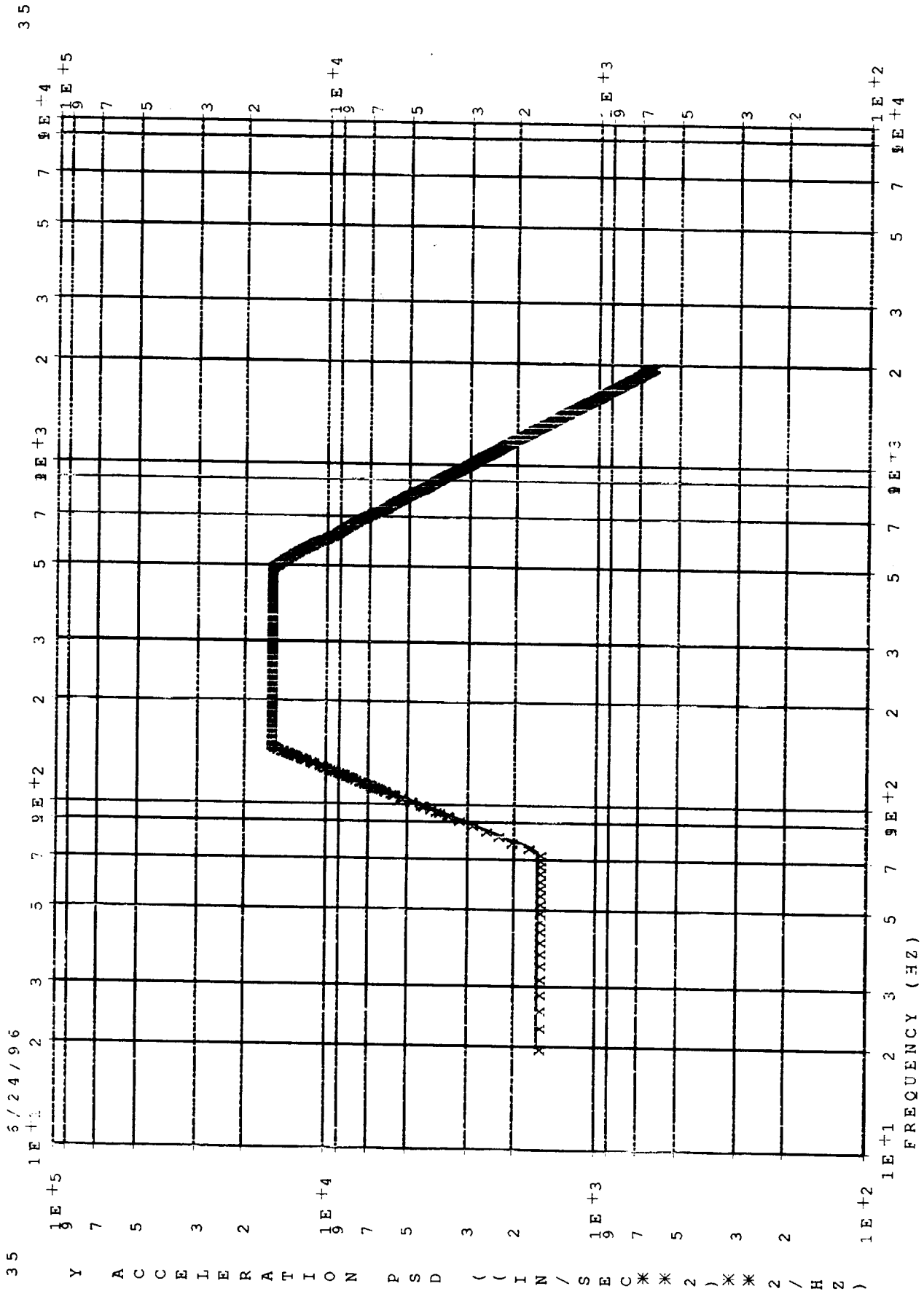
Figure 45 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Upper Reflector X-Load X-Response, 63.59 grms



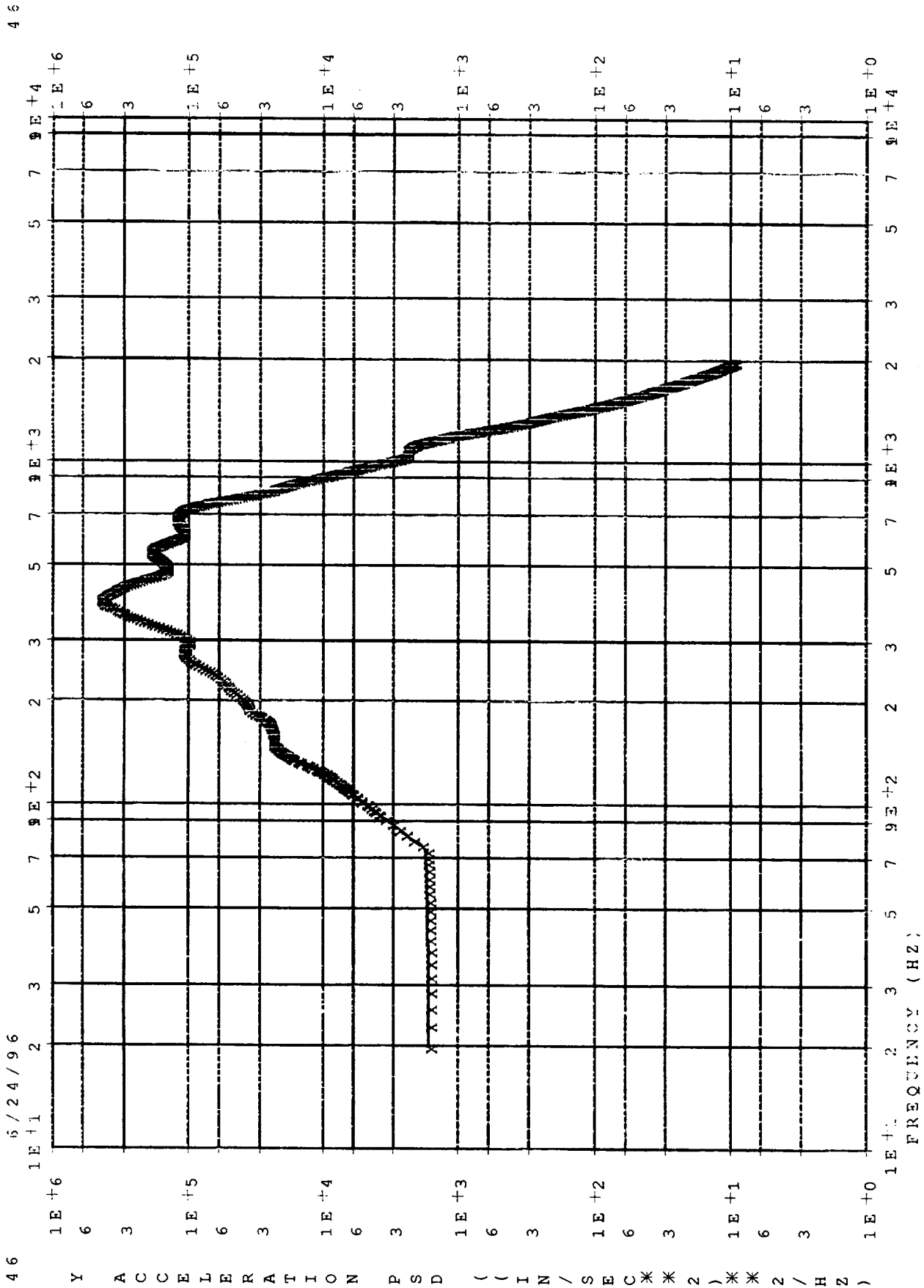
METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS :) X-DIRECTION

METSAT AMSU-A1 PSD INPUT 0-2000 HZ

Figure 46 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Upper Reflector X-Load Z-Response, 70.96 grms



METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) Y-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ
Figure 47 METSAT AMSU-A1 Input Random Vibration PSD Curve
8.84 grms, Y Direction

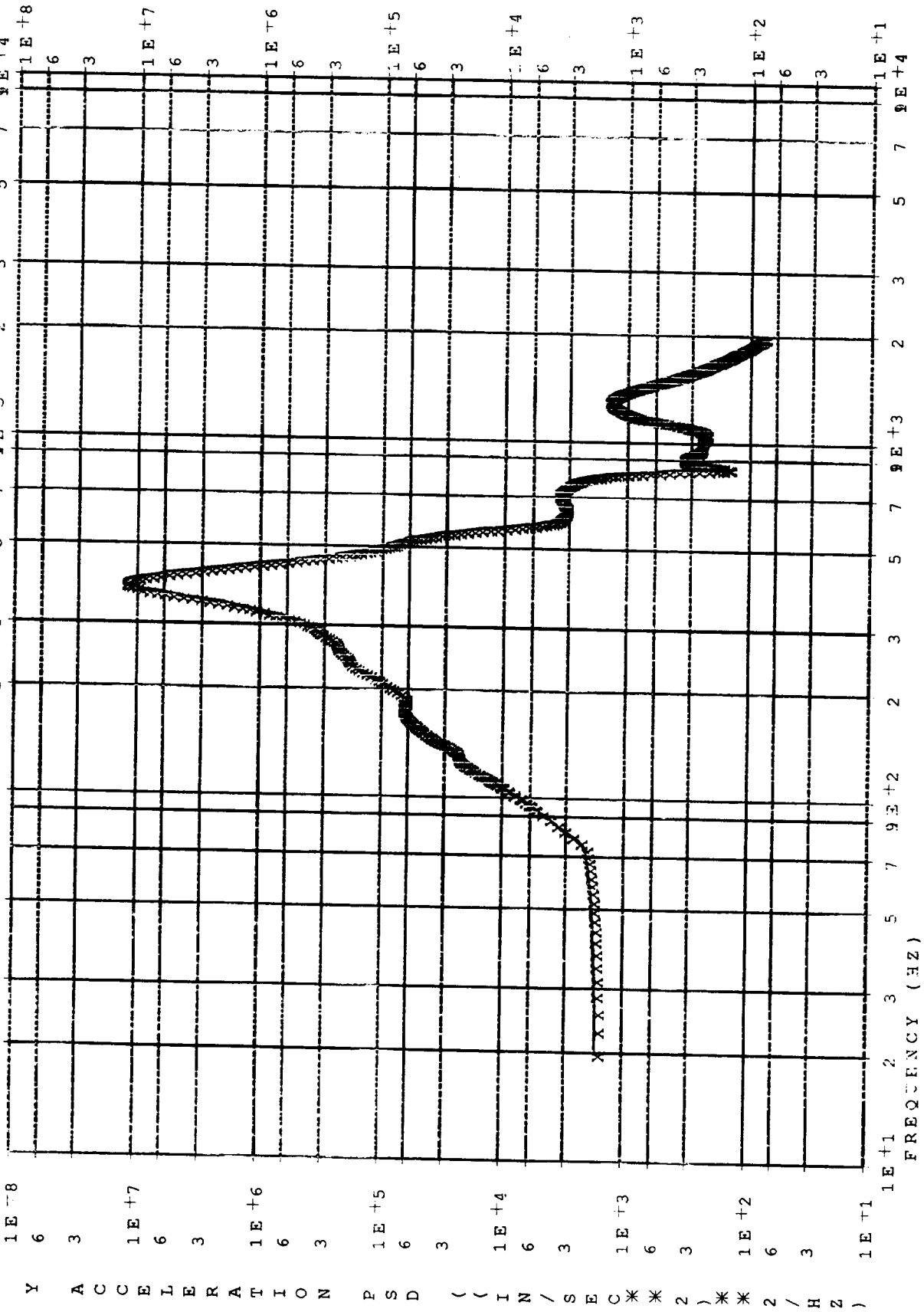


METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) Y-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ
Figure 48 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Lower Right Front Support Y-Load Y-Response, 26.60 grms

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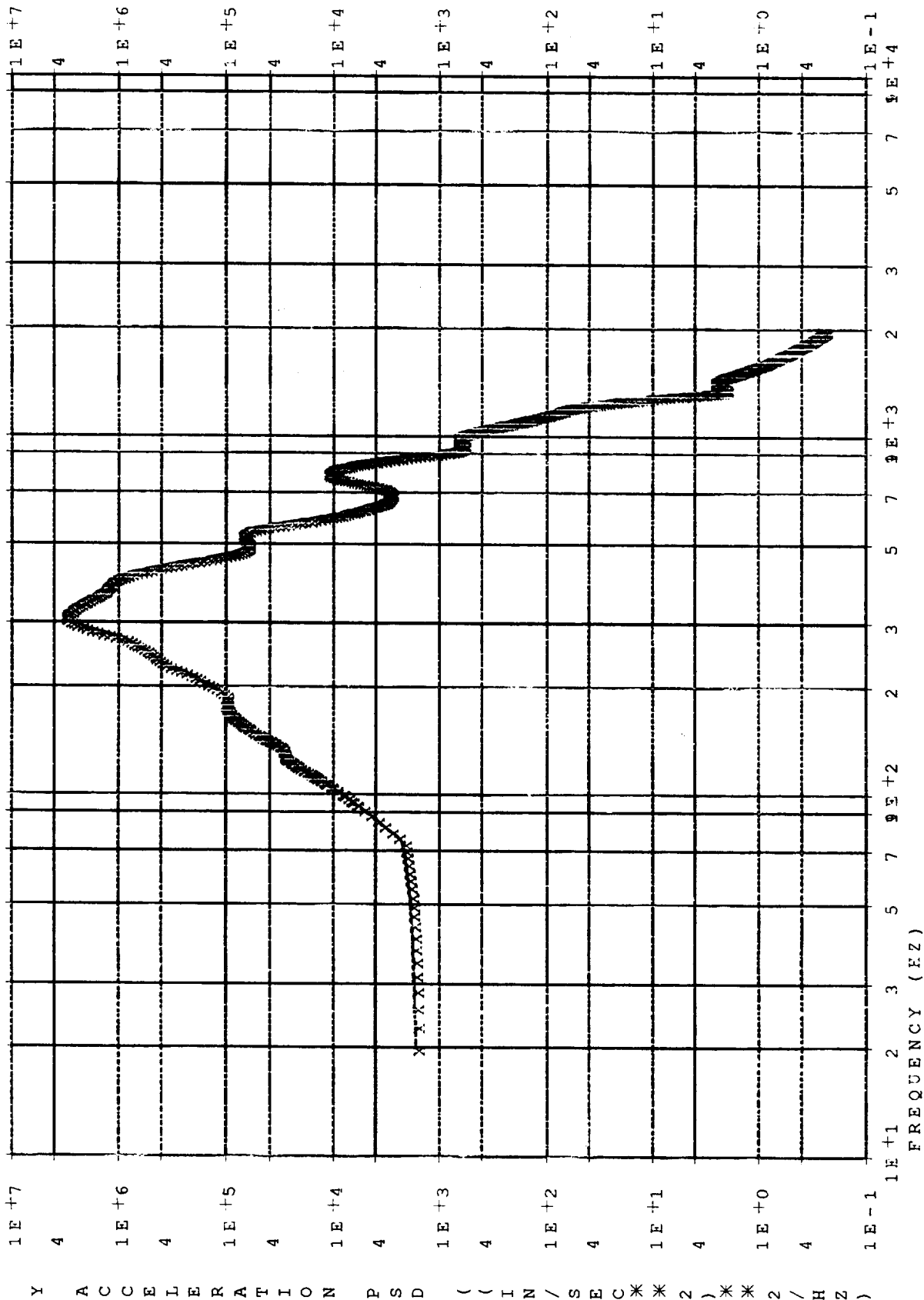
METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) Y-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ

Figure 49 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Upper Right Front Support Y-Load Y-Response, 70.66 grms

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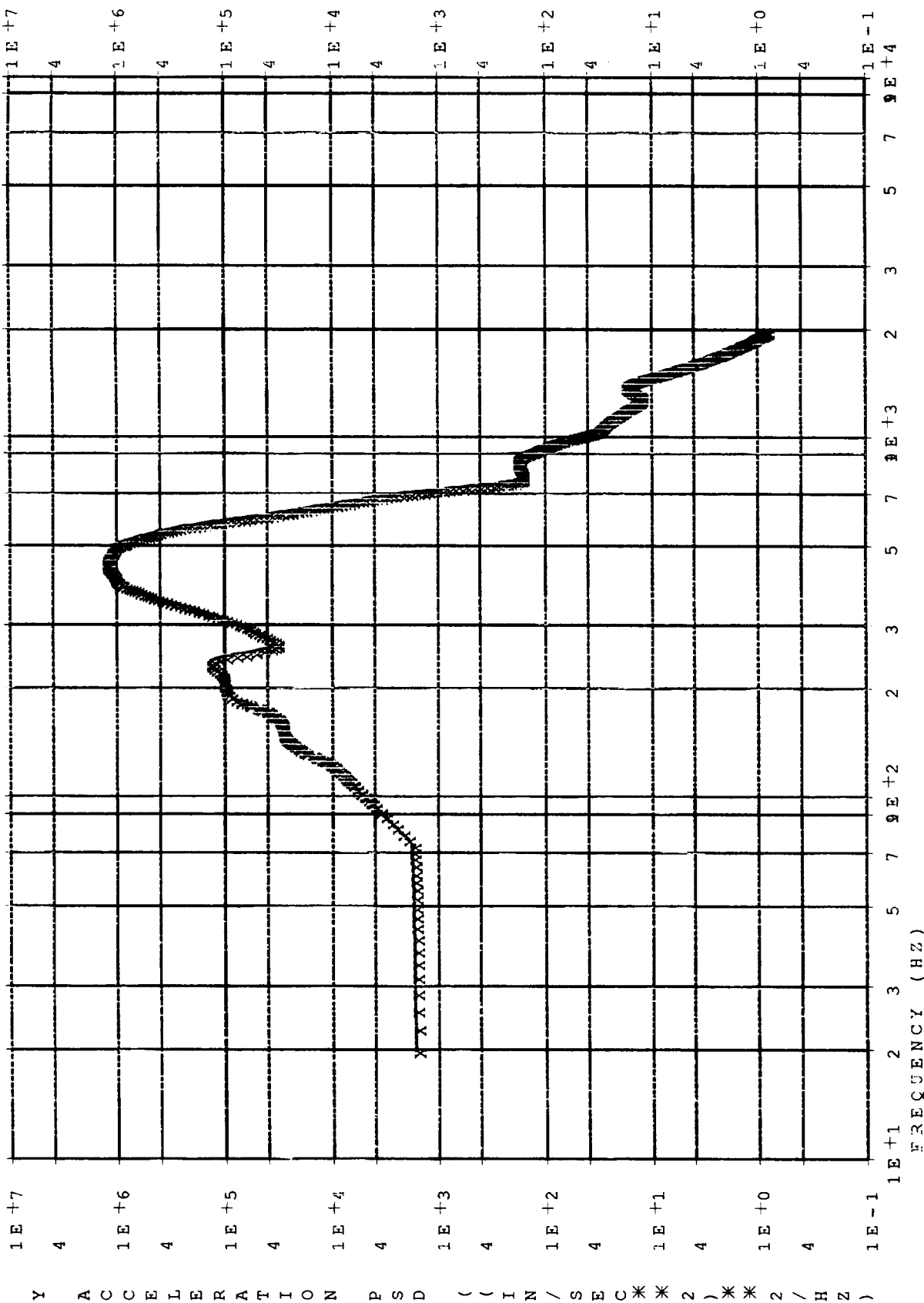


METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) Y-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ
Left Panel Y-Load Y-Response, 47.60 grms.



Figure 51 METSAT AMSU-A1 Random Vibration PSD Curve Response, Lower Card Cage Y-Load Y-Response, 35.25 grms

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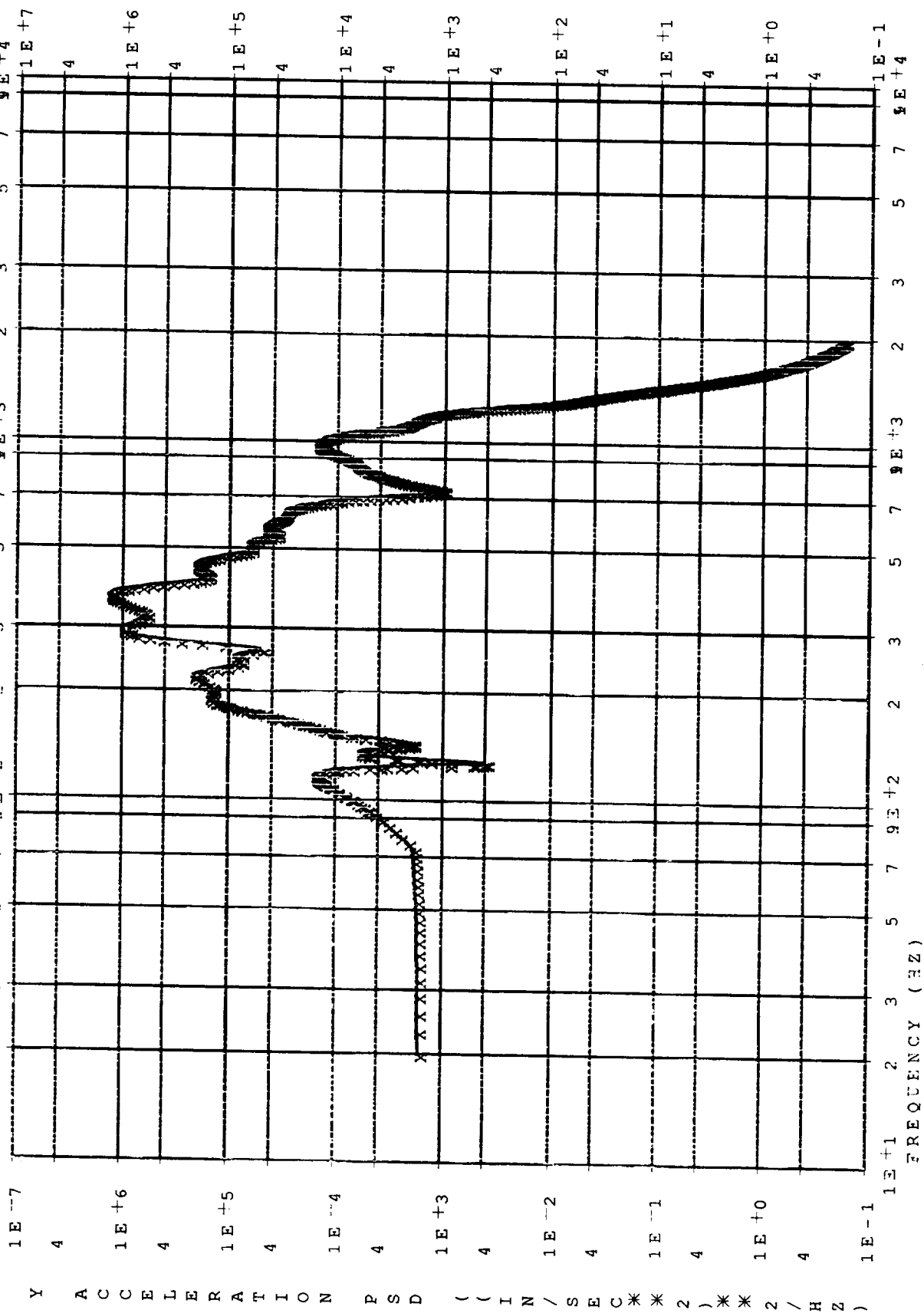


METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) Y-DIRECTION
METSAT AMSU-A1 PSD INPUT 3-2000 HZ
Figure 52 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Lower Warmload Y-Load Y-Response, 38.52 grms

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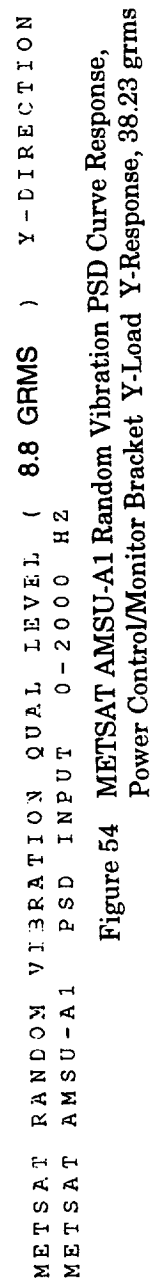
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METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) Y-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ

Figure 53 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Lower Reflector Y-Load Y-Response, 31.81 grms



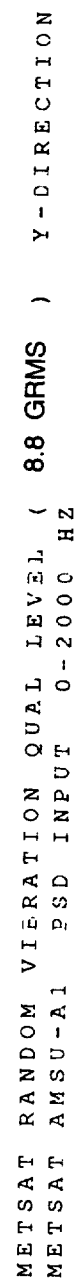
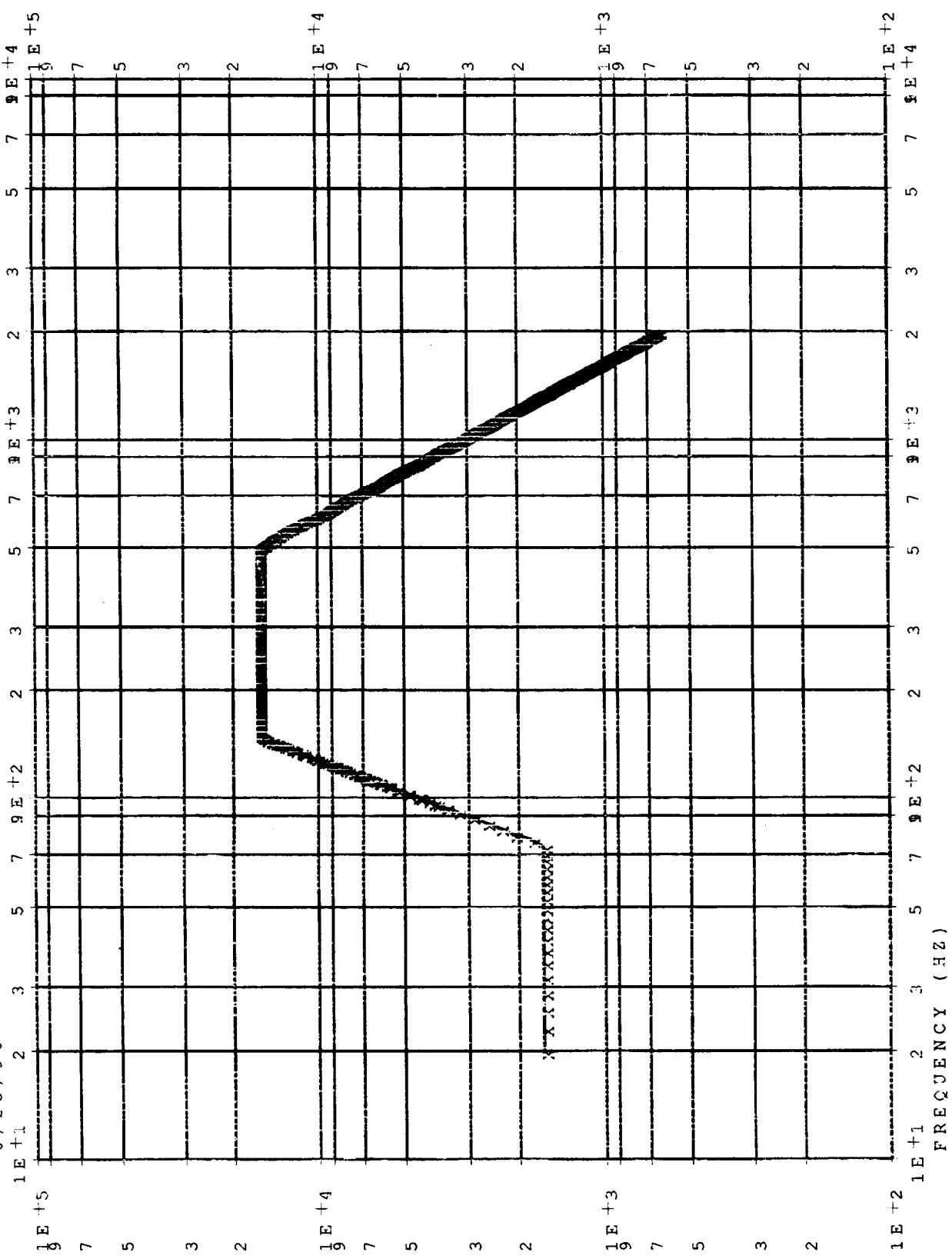


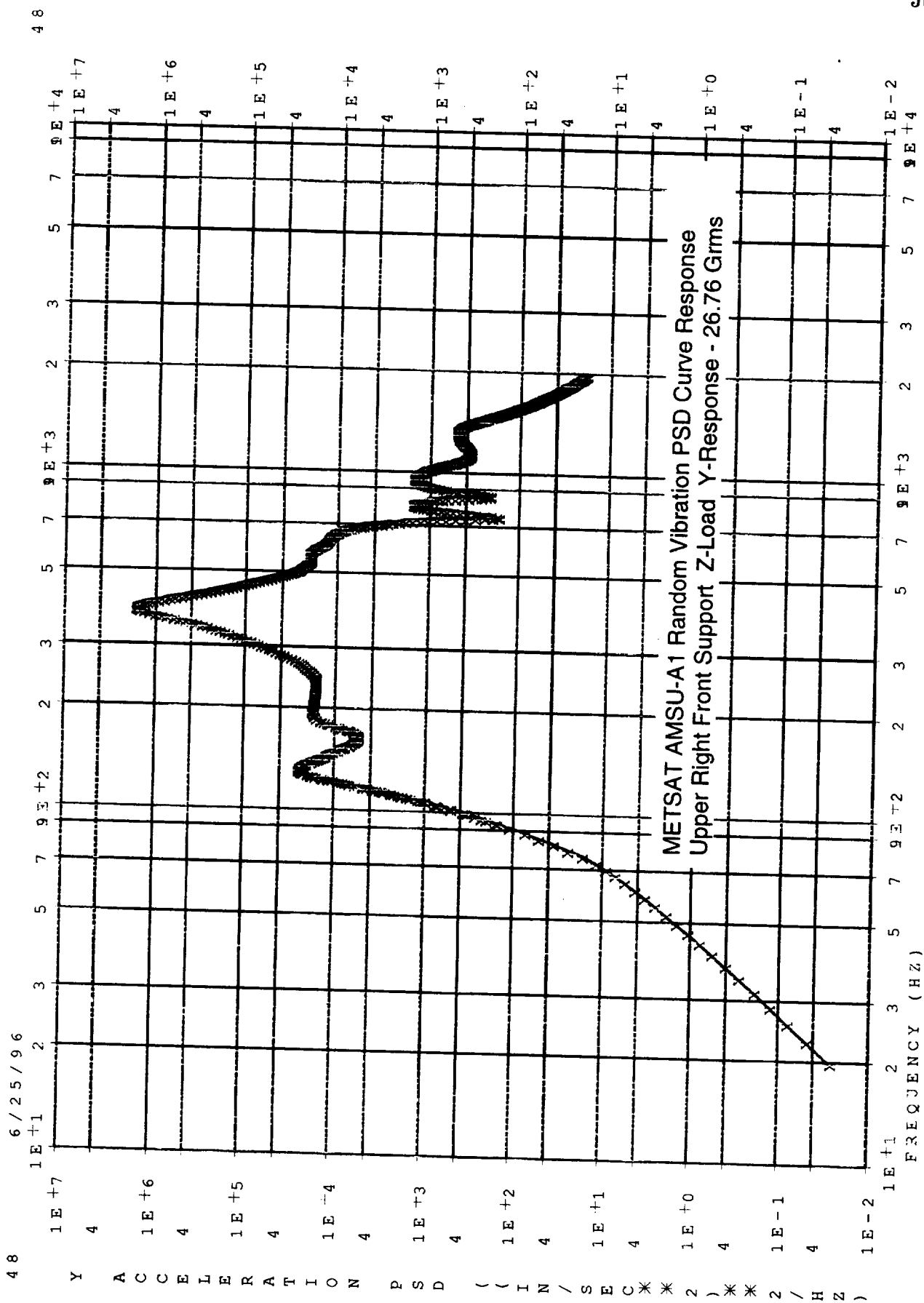
Figure 55 METSAT AMSU-A1 Random Vibration PSD Curve Response, Upper Reflector Y-Load Z-Response, 26.60 grms

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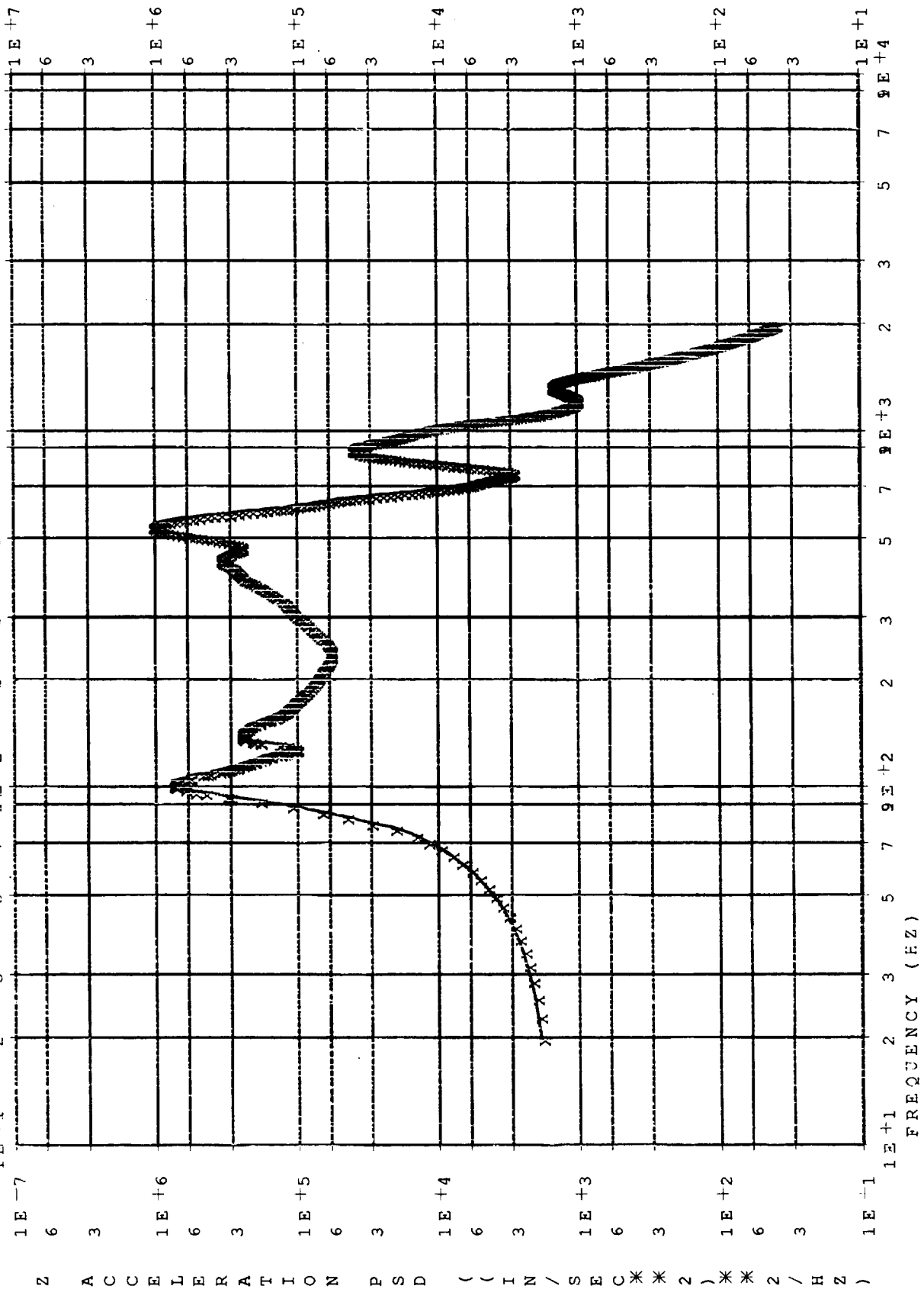


METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) Z-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ
Figure 56 METSAT AMSU-A1 Input Random Vibration PSD Curve
8.84 grms, Z Direction



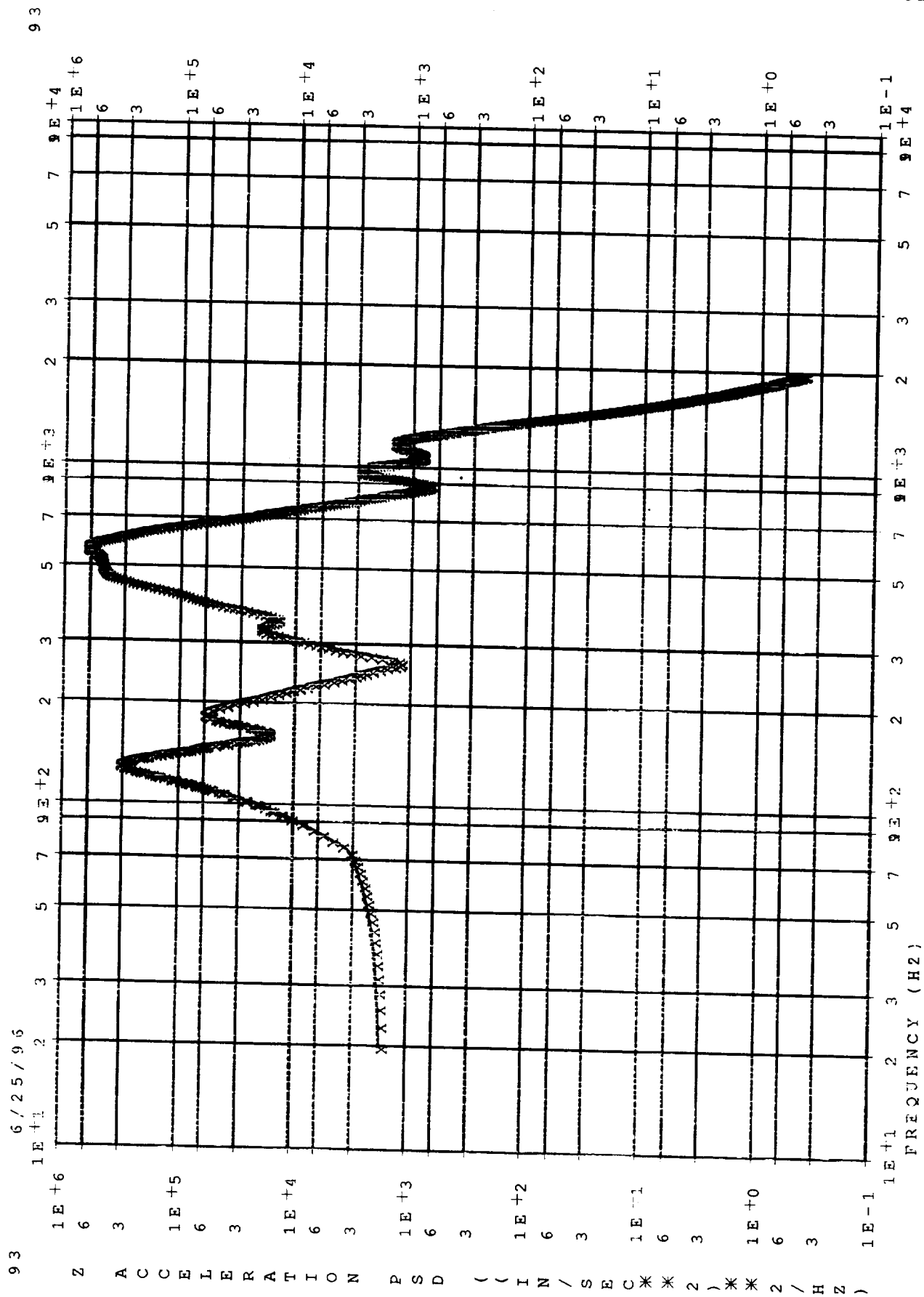
METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) Z-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ

Figure 57 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Upper Right Front Support Z-Load Y-Response, 26.76 grms



METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) Z-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ

Figure 58 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Top Panel Z-Load Z-Response, 32.98 grms



METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) Z-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ

Figure 59 METSAT AMSU-A1 Random Vibration PSD Curve Response,
Upper Card Cage Card Z-Load Z-Response, 28.25 grms

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Y A C C E L E R A T I O N P S D ((I N / S E C * *) * * 2 / H Z)



METSAT RANDOM VIBRATION QUAL LEVEL (8.8 GRMS) Z-DIRECTION
METSAT AMSU-A1 PSD INPUT 0-2000 HZ

Figure 60 METSAT AMSU-A1 Random Vibration PSD Curve Response
Power Control/Monitor Bracket Z-Load Y-Response - 75.97 grms

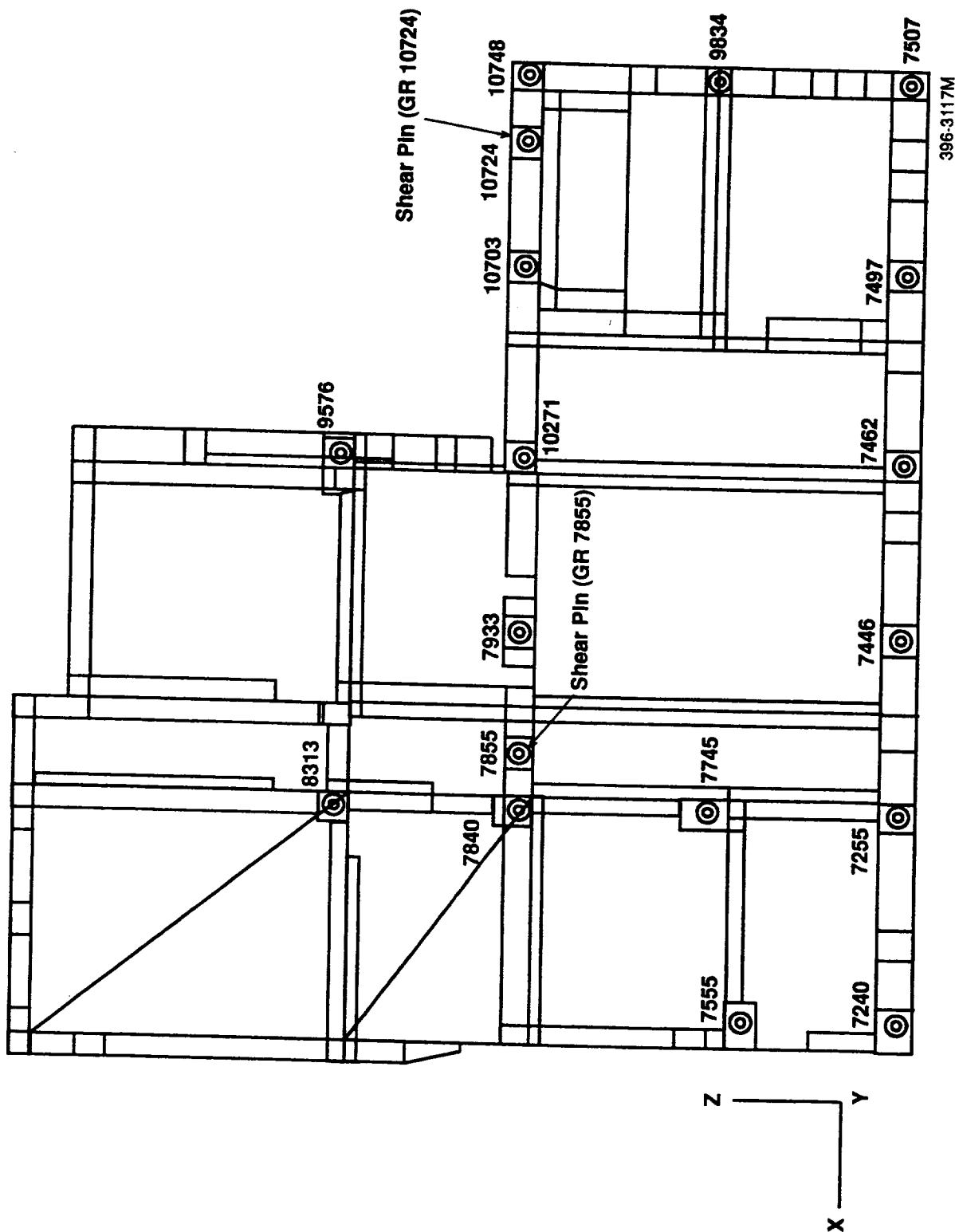


Figure 61 METSAT AMSU-A1 1331552-1 Sidemount Mounting Bolt Grids

Appendix A
NASTRAN FINITE ELEMENT MODEL

Appendix A

NASTRAN FINITE ELEMENT MODEL

The NASTRAN finite element model of the METSAT AMSU-A1 Module is shown as an assembly in Figures A-1 through A-4 in this appendix. Figures A-1, A-2, and A-3 show different views of the model. Figure A-4 is a section view of the model, showing the components modeled in the interior (i.e. shelves, card cages, warmload structures). Elements and grids are highlighted in the piece part models of Report 10738A.

METSAT AMSU-A1 1331720-1 MESH ONLY

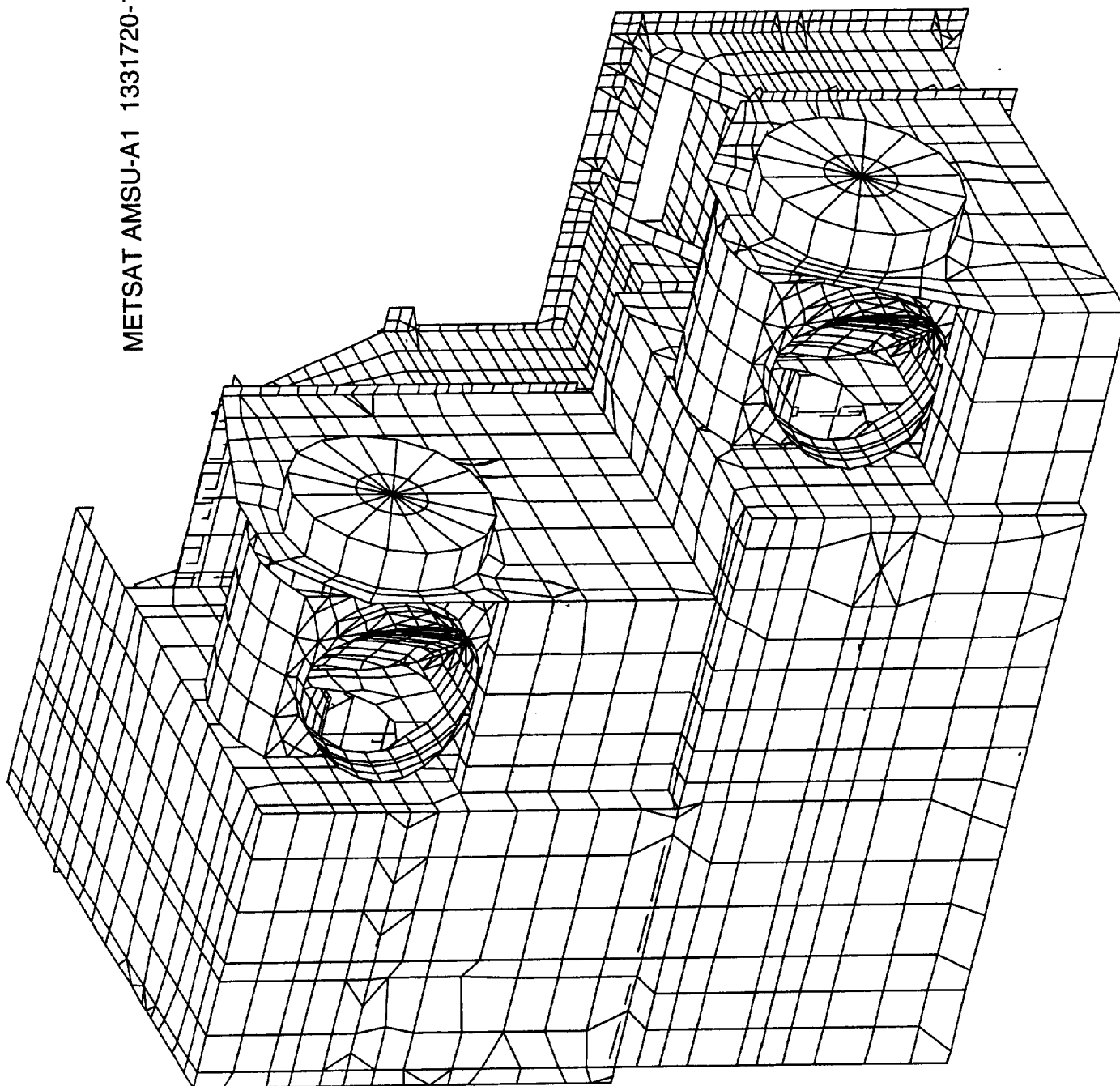
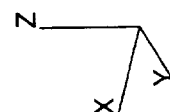
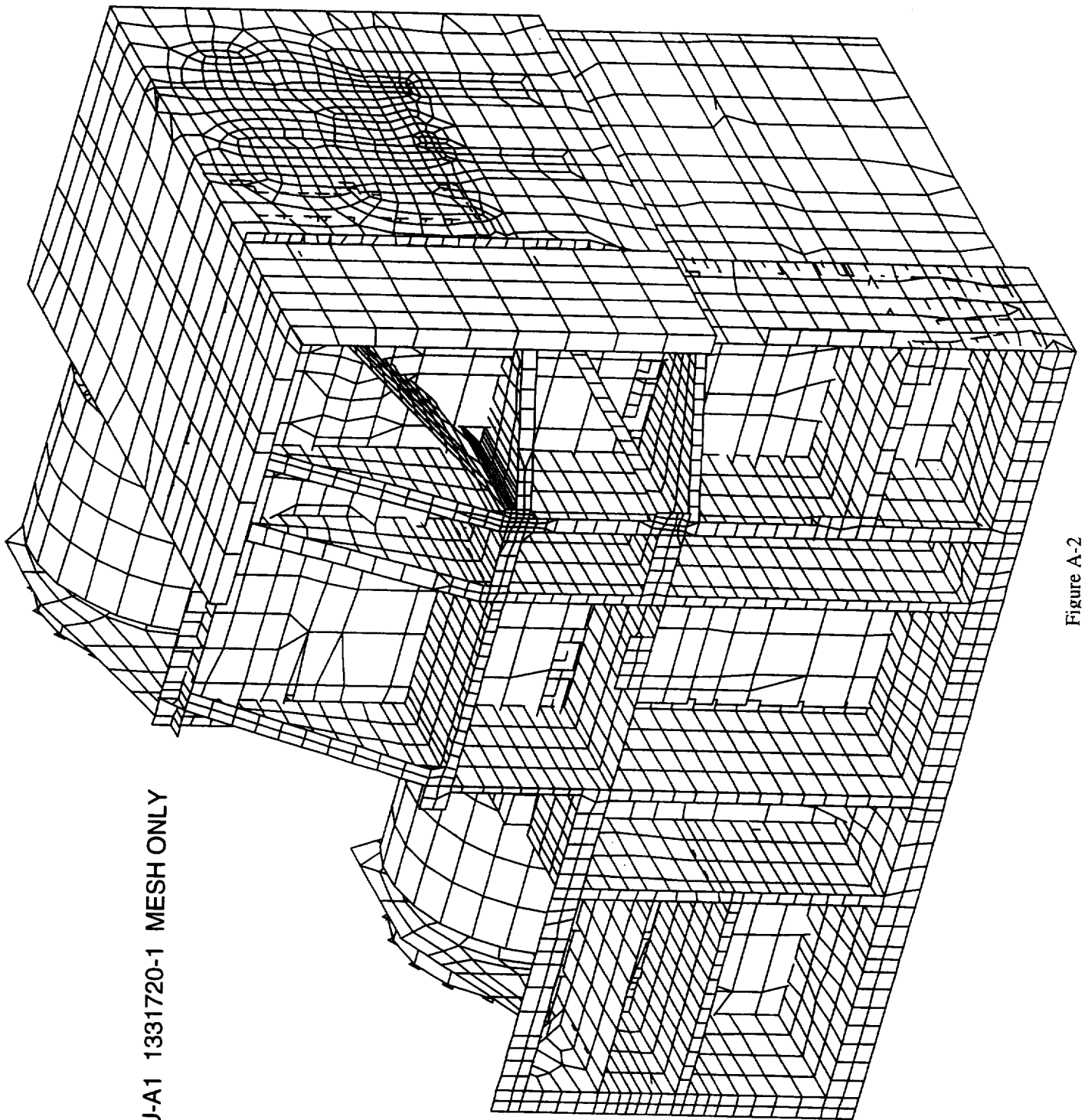


Figure A-1





METSAT AMSU-A1 1331720-1 MESH ONLY

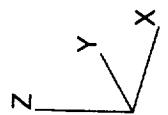


Figure A-2

METSAT AMSU-A1 1331720-1 MESH ONLY

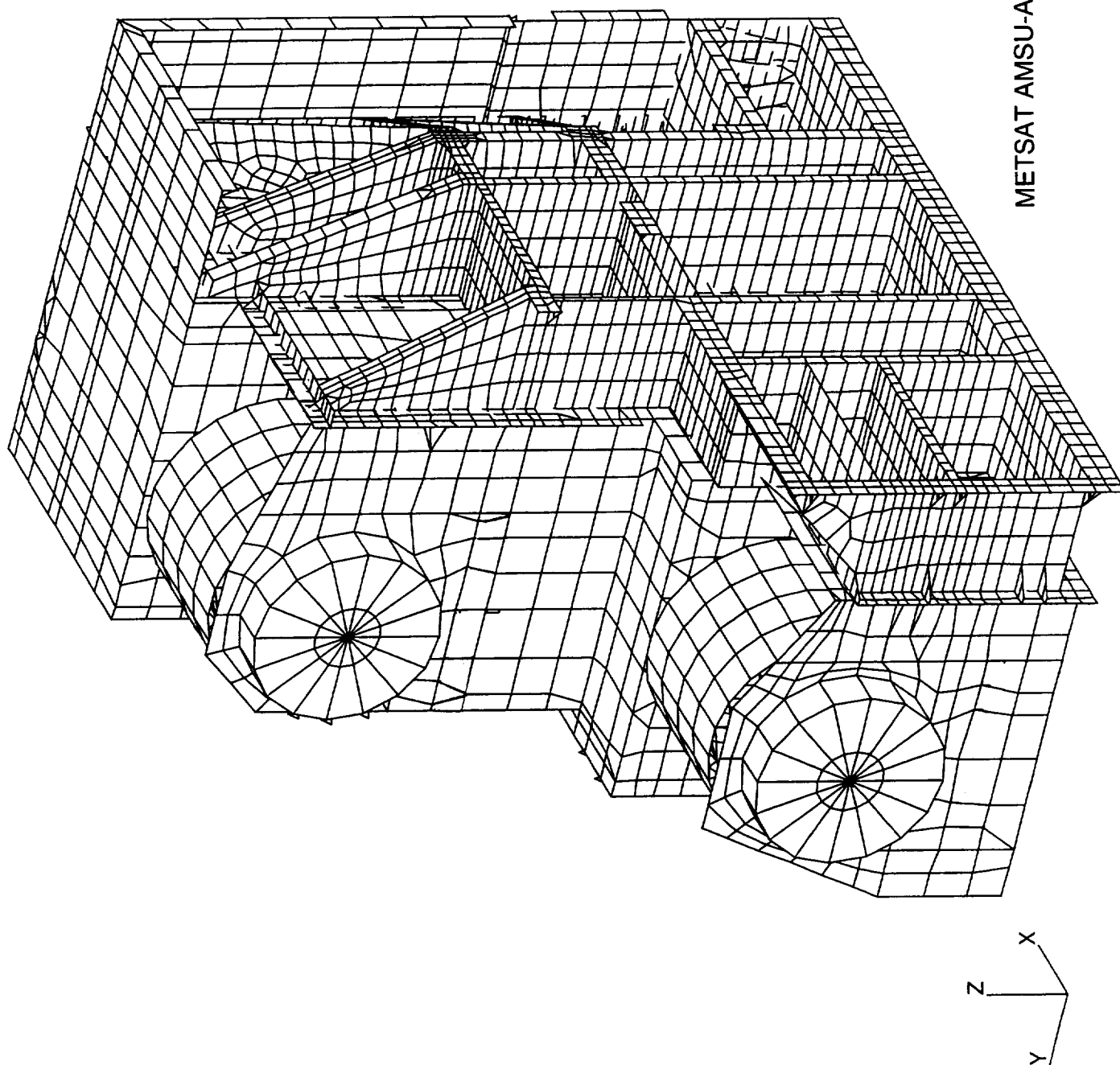


Figure A-3

METSAT AMSU-A1 INTERNAL VIEW - MESH ONLY

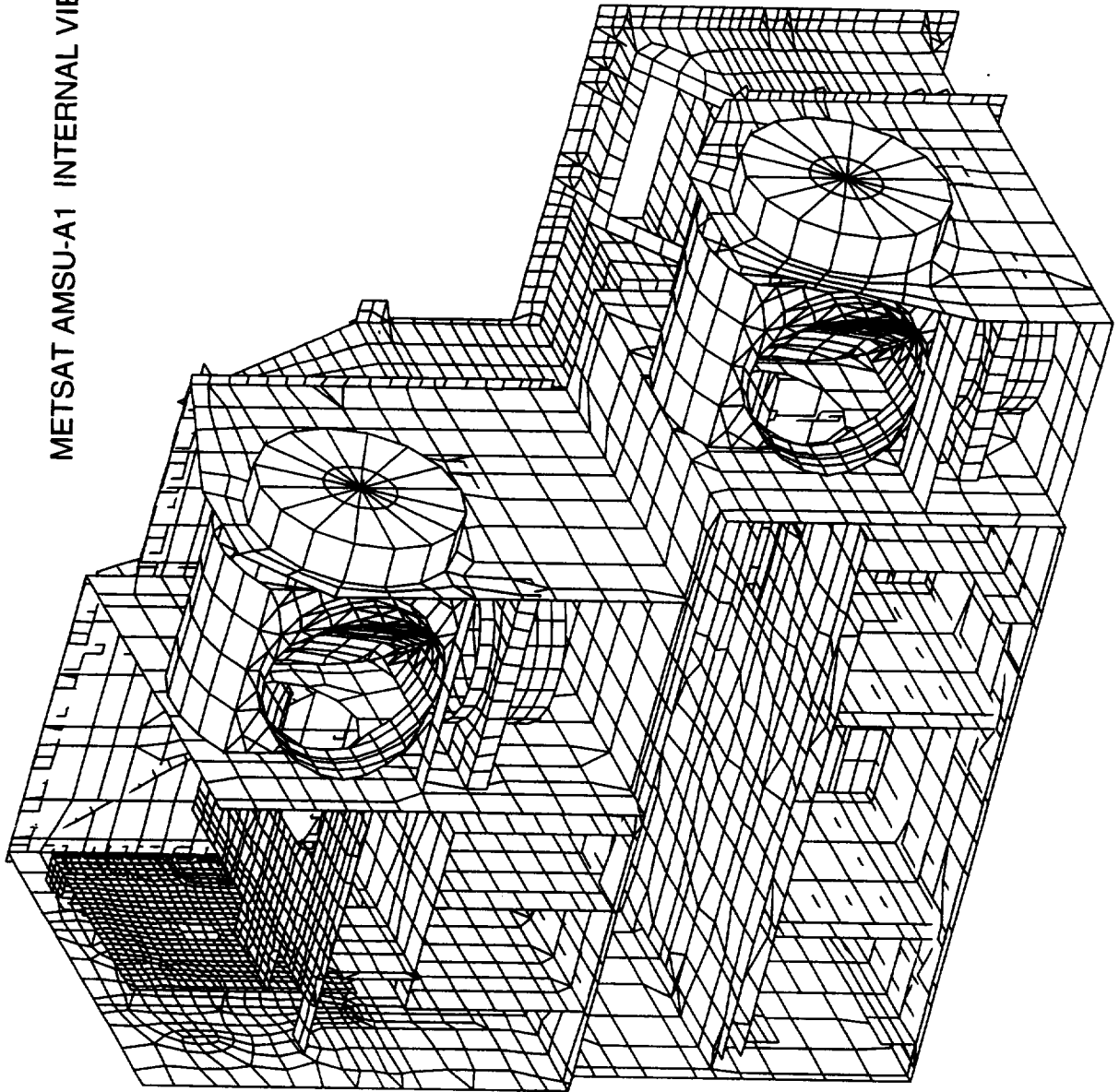
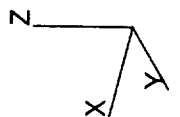


Figure A-4



Appendix B
NASTRAN FINITE ELEMENT MODEL
MATHEMATICAL VALIDITY CHECK

To demonstrate the mathematical soundness of the NASTRAN model, the model is subjected to the GSFC 422-11-12-01 Paragraph 11.1.4.i Deliverable Model Validity Check, where a rigid-body or stiffness-equilibrium check is performed. Using NASTRAN Solution 3, a DIAG 64 ALTER 126 DMAP is run. The NASTRAN Executive Control Data Deck consists of:

```
ID AMSU1,RANDOM
TIME 60
SOL 3
$
DIAG 64
$
COMPILE SOL3,SOUIN=MSCSOU
ALTER 126
VECPLOT, ,BGPDT,EQEXIN,CSTM,,,/RBGLOBAL/GRDPNT=0//4 $
VEC USET/V1/'G'/'F'/'COMP' $
PARTN RBGLOBAL,V1/RBFF,,,/0 $
TRNSP RBFF/RBFFT $
MPYAD KFF,RBFFT,/KFFR/ $
MATGPR GPL,USET,SIL,KFFR/'F'///1.E-2 $
DIAGONAL KFF/KFFD/OPT='SQUARE'/POWER=-1. $
MPYAD KFFD,KFFR,/KFFRN/ $
MATGPR GPL,USET,SIL,KFFRN/'F'///SMALL=1.E-5 $
ENDALTER
CEND
```

A NASTRAN correspondence on the rigid-body check describes the requirements of the test.

"The basic function of this check is to multiply through a cross product the free stiffness matrix by the model rigid body matrix. The matrix which results from this multiplication can be thought of as the internal forces which must be applied to the structure to overcome any model internal constraint to achieve the desired rigid body motion. This matrix is titled the KFFR matrix. The smaller the magnitude of the numbers in the matrix, the less internal constraint present in the model. The DMAP will print any values larger than 1.0E-2. In an attempt to evaluate the effect of any internal constraint, the KFFR matrix is divided by the diagonal stiffness term of each respective row. The resulting matrix is considered "normalized" and is titled the KFFRN matrix. A satisfactory KFFRN matrix will generally have terms less than 1.0E-5."

Thus terms of the KFFRN matrix need be less than 1.0E-5.

The NASTRAN METSAT AMSU-A1 finite element model has been checked to conform to GSFC 422-11-12-01 Paragraph 11.1.4.i requirements. All terms of the KFFRN matrix are less than 1.0E-5. The NASTRAN rigid-body check solution follows.

```
ID AVSUAL,RANDOM.
TIME 16
SOL 3 $
DIAG 64 $
COMPILE SOL3,SOUIN=MCSOU $
ALTER 126 $
VECPLOT,,BGPD,EQEXIN,CSTM,,,,/REGLOBAL/GRDPNT=0//4 $
VEC USET/V1/'G','F','COMP' $
PARTN REGLOBAL,V1,/RBFF,,,0 $
TRNSP RBFF/RBFF $
MEYAD KFF,RBFFT,KFFR/ $
MATGRP GPL,USET,SIL,KFFR///F///1.E-2 $
DIAGONAL KFF/KFFD/OPT='SQUARE'/POWER=-1. $
MEYAD KFFD,KFFR,/KFFRN/ $
MATGRP GPL,USET,SIL,KFFRN///F///SMALL=1.E-5 $
ENCALTER
$ $
$ $
CEND
```

LARGE FIELD DATA TRANS ALTER 126 EQUILIB CHECK - METSAT JUNE 7, 1996 MSC/NASTRAN 8/24/94 PAGE 31

C A S E C O N T R O L D E C K E C H O

CARD COUNT	
1	\$
2	TITLE = LARGE FIELD DATA TRANS ALTER 126 EQUILIB CHECK - METSAT
3	\$
4	\$
5	\$
6	\$
7	\$
8	\$
9	\$
10	\$
11	\$
12	\$
13	\$
14	\$
15	\$
16	\$
17	\$
18	\$
19	\$
20	\$
21	\$
22	\$
23	\$
24	\$
25	\$
26	\$
27	\$
28	\$
29	\$
30	\$
31	\$
32	\$
33	\$
34	\$
35	\$
36	\$
37	\$
38	\$
39	\$
40	\$
41	\$
42	\$
43	\$
44	\$
45	\$
46	\$
47	\$
48	\$
49	\$
50	\$

JUNE 7 1996 SIDEMOUNT UPDATE
 MESH IMPROVEMENT AT CONNECTIONS OF ANGLED STIFFENERS TO MOUNTING
 INCREASES STIFFNESS AT UPPER MOUNTING BOLTS BOLT FL

ADDDED CBARS TO UPPER AND LOWER REFLECTORS
 MODIFIED GRIDS IN UPPER WARMLOAD (PUT IN PLANE)

SIDEMOUNT - PUT GRIDS IN LOCAL ANALYSIS COORD SYSTEMS
 FOR SKEWED PLANES
 SURFACE LOCAL ANALYSIS COORD SYS

LOWER BASEPLATE T=.200 RIDGE BELOW LOWER AFT PANEL REMOVED
 IN METSAT (SET TO T=.060)

PANEL FLANGES INCREASED TO REQ'D EOS/METSAT LEVELS

UPPER FRONT PANEL (1331352) RIB PUT BACK TO T=.07 (WAS .08)

UP CARD CAGE TOP PANEL (1331162-3) PUT BACK TO T=.05 (WAS .06)

LOWER BASEPLATE CBAR OFFSETS ADDED
 PBAR 5 COORD 0 <0,.3125,0> PBAR 6 COORD 0 <0,.156,0>

1331404 LOWER BASEPLATE CONCEPT (METSAT)

1331552 SIDEMOUNT ADDED W/RATIGED DENSITY TO GIVE 12.50 LB

SIGNAL PROCESSOR/CARD CAGES
 NO CHANGE TO LOWER CARD CAGE FROM EOS
 2 CARDS PUT IN UPPER CARD CAGE (W = 2 X .30 LB)

1331648 TOP PANEL MODELED (ADDED FLAPS AROUND SIDEMOUNT)

1331640 LEFT PANEL NEW T=.050 (WAS .062 ON EOS) ALSO REMOVED NON
 STRUCTURAL MASS SINCE NO MIRRORS ARE APPLIED TO METSAT LEFT PNL

NEW POWER CONTROL/MONITOR FILTER BOX 1356907 REPLACES 1356761
 NO FILTERS IN METSAT, ALSO ENCLOSURE LOWER COVER REMOVED

ADD 1331336 ALIGNMENT CUBE MASS TO UPPER MOTOR MOUNT

ADD 1331097 CARD RETAINER MASS TO UPPER RIGHT PANEL

LARGE FIELD DATA TRANS ALTER 126 EQUILIB CHECK - METSAT JUNE 7, 1996 MSC/NASTRAN 3/24/94 PAGE 32

```

CARD
COUNT
51 $
52 $ ADD INSULATION MASS TO SIDEMOUNT (8 CONM2'S 10677-10684)
53 $
54 $ REMOVED INSULATION NSM FROM LEFT PANEL AND COVERS
55 $
56 $ REVISED UPPER CARDS RETENTION AT TOP AND BOTTOM OF CARDS
57 $ (EXTENDED THE RETENTION LENGTH)
58 $
59 $ CORRECTED SOME FREE SURFACES IN UPPER CARDS
60 $
61 $ PROVIDED AN EFFECTIVE THICKNESS FOR TOP PANEL W/MIRRORS
62 $ THAT TAKES INTO EFFECT THE .010 EPOXY-GLASS SHEET
63 $ (INCREASES EFFECTIVE T TO .064 (WAS .062) AND RAISES
64 $ 1ST MODE TO 101.7 HZ)
65 $
66 $ REMOVE ALL SPC'S
67 $
68 $ ECHO = NONE
69 $
70 $ SUBCASE 1
71 $ SUBCASE NAME : EQUILIBRIUM
72 $ SUBTITLE=EQUILIBRIUM
73 $ METHOD = 1
74 $ VECTOR(SORT1,REAL)=ALL
75 $ SPCFORCES(SORT1,REAL)=ALL
76 $ BEGIN BULK

INPUT BULK DATA CARD COUNT = 42728

TOTAL COUNT= 42122

```

SEQUENCE PROCESSOR OUTPUT

THERE ARE 9307 POINTS DIVIDED INTO 1 GROUP(S).

CONNECTION DATA

ELEMENT TYPE	NUMBER	ASSEMBLY TIME (SEC)
BEAM	124	0.13
ELAS	28	0.00
QUAD4	8288	17.40
BAR	894	0.80
TRIA3	688	0.69

TOTAL MATRIX ASSEMBLY TIME FOR 10022 ELEMENTS IS 19.03 SECONDS.

RIGID ELEMENT PROCESSING COMPLETED.

ORIGINAL PERFORMANCE DATA

SUPER (GROUP) ID	NO. GRIDS	AV. CONNECTIVITY	C-AVERAGE	C-RMS	C-MAXIMUM	P-GROUPS	P-AVERAGE	DECOMP TIME (SEC) (6.0 DOF/GRID)
0	9307	8.67	514.46	609.27	1071	97	579.43	62629.938

RESEQUENCED PERFORMANCE DATA

SUPER (GROUP) ID	NO. GRIDS	AV. CONNECTIVITY	C-AVERAGE	C-RMS	C-MAXIMUM	P-GROUPS	P-AVERAGE	DECOMP TIME (SEC) (6.0 DOF/GRID)	METHOD
0	9307	8.67	141.32	162.27	264	0	0.00	4679.772	ACTIVE

C U T P U T F R O M G R I D P O I N T W E I G H T G E N E R A T O R

REFERENCE POINT = 0					
M O					
* 3.860000E+10	3.597437E-18	3.924477E-18	-4.742077E-17	3.637418E+11	-1.975457E+11
* 3.597437E-18	3.860000E+10	-2.616318E-18	-3.637418E-11	2.076702E-17	6.055359E+11
* 3.924477E-18	-2.616318E-18	3.860000E+10	1.975457E-11	-6.055359E+11	3.556557E-16
* -4.742077E-17	-3.637418E+11	1.975457E+11	4.438653E+12	-3.098939E+12	-5.706185E+12
* 3.637418E+11	2.076702E-17	-6.055359E+11	-3.098939E+12	1.292459E+13	-1.861544E+12
* -1.975457E+11	6.055359E+11	3.556557E-16	-5.706185E+12	-1.861544E+12	1.051031E+13
S					
* 1.000000E+00	0.000000E+00	0.000000E+00	0.000000E+00	* * * * *	
* 0.000000E+00	1.000000E+00	0.000000E+00	0.000000E+00	* * * * *	
* 0.000000E+00	0.000000E+00	0.000000E+00	1.000000E+00	* * * * *	
DIRECTION					
MASS AXIS SYSTEM (S)					
MASS					
X	3.860000E+10	X-C.G.		Y-C.G.	Z-C.G.
Y	3.860000E+10	-1.223517E-27		5.117764E+00	9.423364E+00
Z	3.860000E+10	1.568746E+01		5.380058E-28	9.423364E+00
		1.568746E+01		5.117764E+00	9.213880E-27
I (S)					
* -6.553600E+05	-2.621440E+05	0.000000E+00	* * * * *		
* -2.621440E+05	-5.242880E+05	-2.621440E+05	* * * * *		
* 0.000000E+00	-2.621440E+05	0.000000E+00	* * * * *		
I (Q)					
* -8.918806E+05	-4.107130E+05	* * * * *			
* 1.229460E+05	1.229460E+05				
Q					
* 7.284717E-01	-6.702843E-01	-1.415905E-01	* * * * *		
* -6.572776E-01	-6.255454E-01	-4.203317E-01	* * * * *		
* -1.931704E-01	3.392640E-01	-8.962552E-01	* * * * *		

POINT	KFFR	VALUE	POINT	VALUE	POINT	VALUE	POINT	VALUE
COLUMN	1 (4003-T1) .							
COLUMN	2 (4003-T2) .							
COLUMN	3 (4003-T3) .							
COLUMN	4 (4003-R1) .							
2166 T1	3.39045E-02		2166 R2	1.15137E-02	2167 R2	1.44013E-02	3343 T1	-2.42962E-02
3357 T1	-4.76191E-02		3357 R2	-1.02337E-02	3342 T1	8.17207E-02	2168 R2	-3.95110E-02
7248 T1	2.09117E-02		7238 T1	2.31287E-02	7247 R2	-1.95223E-02	7247 R3	1.62478E-02
7236 T1	1.02005E-02		7104 T1	-1.08867E-02	7283 T1	-4.31142E-02	7288 R2	1.18344E-02
7289 R3	2.37814E-02		7290 T1	6.39906E-02	7103 T1	-3.44161E-02	7103 T1	1.62626E-02
11998 T2	1.28618E-02		11997 T2	-3.83302E-02	11981 T2	-1.38390E-02	7136 T1	-1.08866E-02
7321 R2	1.18344E-02		7322 R2	2.91412E-02	7322 R3	2.37813E-02	7321 T1	-4.31142E-02
7135 T1	1.62626E-02		11996 T2	-2.19104E-02	11980 T2	2.19104E-02	7134 T1	3.44161E-02
11994 T2	1.38389E-02		11978 T2	-1.38389E-02	11993 T2	1.18847E-02	11979 T2	2.9104E-02
11975 T2	-1.38390E-02		11990 T2	1.38390E-02	11974 T2	-1.38390E-02	11991 T2	1.38390E-02
11987 T2	1.38390E-02		11971 T2	-1.38390E-02	12893 T3	1.16542E-02	11973 T2	1.99562E-02
12894 R2	-1.23889E-02		12923 T3	-1.38943E-02	12643 T2	1.81689E-02	12893 T2	-1.01903E-02
13017 T3	-1.78033E-02		12982 T1	-1.58070E-02	12895 T2	-2.09410E-02	12627 T2	-1.03372E-02
8213 R2	-1.23519E-02		8219 T1	3.50399E-02	12893 T2	2.52854E-02	12895 R1	-1.49388E-02
12645 R3	-1.05974E-02		12977 T3	-4.55781E-02	12930 T3	3.06609E-02	12623 T2	1.60772E-02
13016 T3	-1.08438E-02		7661 T1	3.95770E-02	12978 T3	3.06609E-02	12939 T3	-1.27185E-02
8216 T1	-1.02793E-02		7848 T1	-6.49823E-02	7849 T1	2.54053E-02	7847 T1	5.82627E-02
12975 R2	2.08334E-02		12976 T1	-1.26841E-02	12976 T2	4.51377E-02	12974 T3	1.45524E-02
12909 R2	1.04218E-02		12907 T3	1.32432E-02	12907 R1	-1.99648E-02	12974 T3	1.16628E-02
12910 R1	1.46586E-02		12910 R2	1.52280E-02	12912 R2	-1.15645E-02	13004 T3	1.16481E-02
12921 T3	-1.11552E-02		12926 T3	1.55581E-02	13006 T2	-1.73237E-02	12910 T3	-2.40597E-02
13007 T3	-2.58405E-02		13009 T3	-2.08578E-02	12613 T2	-1.00058E-02	13003 T3	2.97717E-02
12965 T3	1.93848E-02		13014 T3	3.35115E-02	12951 T3	-2.13440E-02	12927 T1	1.42438E-02
12607 T2	1.03881E-02		12648 T2	-2.77316E-02	12956 T3	-1.23686E-02	12647 T3	-1.82002E-02
12962 T3	1.92964E-02		12961 T3	-2.23872E-02	12956 T3	-1.23686E-02	12953 T3	-1.42717E-02
13011 T3	1.37934E-02		12945 T1	-2.28372E-02	13012 T3	-2.26145E-02	12953 T3	-1.13449E-02
12569 R3	-1.19484E-02		12955 T1	1.60255E-02	12493 T2	-2.61771E-02	12968 T3	-2.14831E-02
10504 R3	1.35789E-02		10575 R3	-1.33035E-02	10503 T1	2.10348E-02	12569 T2	-1.30904E-02
11836 R2	2.26836E-02		12456 T3	-1.38494E-02	12456 R2	6.50668E-02	10593 T1	1.19863E-02
12458 R2	8.79664E-02		12459 T3	6.85071E-02	12459 R2	4.48703E-02	11896 T3	-2.61292E-02
10667 R3	1.566718E-02		10454 T1	-1.38416E-02	10560 T2	-1.41045E-02	12458 T3	1.02024E-02
8322 T3	-6.05111E-02		8322 R2	4.29923E-02	8312 T3	4.68301E-02	10869 T1	2.35442E-02
12451 R2	-9.47084E-02		12452 R2	-1.22856E-01	12453 T3	2.55604E-02	8324 R2	-2.35083E-02
12454 R2	-5.81074E-02		12446 R2	-6.02087E-02	12441 R2	7.54175E-02	12451 T3	4.69296E-02
12448 R1	-1.07286E-02		12448 R2	3.07002E-02	12449 T3	1.46613E-02	12454 T3	-8.60774E-02
12443 T3	-1.10265E-02		12443 R2	-3.98017E-02	12437 T3	-7.36888E-02	12448 T3	-1.11894E-02
8320 R2	-2.70955E-02		12436 R2	-5.78040E-02	12439 R2	-7.46907E-02	12442 T3	8.27383E-02
12436 R2	-2.09936E-02		12439 T3	-3.20923E-02	10465 T1	1.0465 T2	12438 T3	4.42928E-02
10874 R3	1.93559E-02		10465 T1	-1.26481E-01	10466 R2	-1.33429E-02	10874 R2	6.40415E-02
10553 R2	-1.05479E-02		10466 T1	6.07909E-02	9179 T1	2.19995E-02	10553 T1	-7.65151E-02
12776 R2	-1.10890E-02		9179 T1	1.73571E-02	12785 T3	3.49428E-02	12776 T3	1.95978E-02
12784 R1	-1.09753E-02		12783 T3	4.02576E-02			12784 T3	-2.65979E-02
							9187 R2	1.22774E-02

POINT	KFFR	VALUE	POINT	VALUE	POINT	VALUE	POINT	VALUE	POINT	VALUE
9188 R2	1.18990E-02		12789 T3	1.91351E-02	12790 R2	1.01104E-02	12792 R1	-1.06349E-02	12793 T3	-1.09965E-02
12793 R2	-1.24027E-02		12794 T3	-1.08134E-02	9197 T1	-1.43412E-02	9198 T1	-1.43412E-02	12794 T1	-1.72939E-02
10664 R3	1.02857E-02		10610 T1	2.23465E-02	10665 T1	4.17255E-02	12797 R1	-1.89017E-02	12797 R2	-1.21638E-02
12799 R2	1.31105E-02		12800 R2	1.23034E-02	12801 R2	1.18256E-02	12804 T3	-1.68224E-02	12805 T3	-1.03398E-02
12806 R2	-1.09334E-02		12807 R1	-1.19277E-02	12807 R2	-1.22367E-02	12803 R1	1.06447E-02	12802 T3	-3.40803E-02
9223 R3	1.15167E-02		9222 T1	-1.87614E-02	9222 R1	1.43540E-02	12812 T3	-2.09189E-02	12813 T3	1.01297E-02
12814 R1	1.00498E-02		12815 T1	-1.33706E-02	12811 R1	-1.09403E-02	12811 R2	-1.34855E-02	12810 R1	1.56802E-02
12809 R1	1.01699E-02		9227 R2	-1.03309E-02	12820 T3	2.47636E-02	12821 R2	-1.38104E-02	12824 R1	1.41028E-02
12824 R2	-1.93738E-02		12823 T3	-3.84991E-02	12823 R2	-1.02901E-02	9237 T1	1.09608E-02	10656 T1	-5.58442E-02
10656 R2	1.63810E-02		10656 R3	-1.26930E-02	10786 T1	9.38752E-02	10786 R3	-3.62797E-02	12826 T3	2.48499E-02
12827 R1	1.35205E-02		12828 R2	-1.21802E-02	12829 T3	1.14594E-02	12829 R2	-1.18285E-02	12831 R1	1.45009E-02
12830 T3	4.22777E-02		12833 R1	1.16144E-02	10655 T1	-3.32389E-02	10655 R2	-2.17256E-02	10647 T1	-1.54903E-02
10647 R2	-1.59271E-02		10647 R3	-2.27672E-02	10790 T1	8.82246E-02	10790 R2	1.52235E-02	10790 R3	3.51950E-02
10793 T1	1.42671E-02		10793 R2	6.35365E-02	10793 R3	3.49977E-02	10796 T1	-1.51186E-01	10796 R2	1.07697E-02
10796 R3	1.62465E-02		12839 R2	1.18375E-02	12840 T3	-1.74060E-02	12841 T3	-4.05141E-02	10628 T1	6.33417E-02
10628 R2	-1.87824E-02		12842 T3	-1.41797E-02	12843 T3	2.54292E-02	12844 T3	1.74271E-02	12844 R2	1.10639E-02
12845 T3	4.42351E-02		12845 R2	1.01374E-02	12852 T3	-4.34285E-02	10458 T1	-4.83646E-02	10452 T1	1.34348E-02
12851 T3	-2.63814E-02		12855 T3	-1.59535E-02	12855 R1	1.17569E-02	12857 T3	1.50149E-02	12857 R1	1.23791E-02
12859 R1	1.66855E-02		12858 T3	1.13060E-02	12860 T3	3.94993E-02	12860 R1	1.01924E-02	9802 R3	1.06947E-02
8774 T1	2.93907E-02		8804 T1	-7.38691E-02	8804 R3	1.66575E-02	8744 T1	4.44785E-02	8744 T1	-2.93907E-02
8800 T1	7.38691E-02		8800 R3	-1.66575E-02	8741 T1	-4.44785E-02	8700 T1	-4.44785E-02	8788 T1	-8.50372E-02
8788 R2	3.81033E-02		8788 R3	-1.66575E-02	8695 T1	5.00622E-02	8691 T1	7.94532E-02	3795 T1	-1.14123E-02
3775 T1	1.20185E-02		3796 T1	-1.14182E-02	3776 T1	1.20185E-02	3797 T1	1.14269E-02	3777 T1	-1.20185E-02
3794 T1	1.14064E-02		3771 T1	-1.20185E-02	6336 R3	-1.50683E-02	6338 R3	1.50683E-02	6340 R2	-1.42288E-02
6333 T1	4.22603E-02		6335 T1	-4.22603E-02	6323 T1	4.22603E-02	6321 T1	-4.22605E-02	6320 R3	-1.50683E-02
6318 R3	1.50683E-02		6316 R2	-1.42288E-02						
5 (4003-R2) .										
7256 T2	-1.26970E-02		7365 T2	-1.25634E-02	7365 R3	1.03491E-02	7363 R3	1.32247E-02	7361 T2	1.14716E-02
7361 R3	1.02604E-02		7359 T2	1.53098E-02	7622 T2	4.50802E-02	7626 T2	-4.50806E-02	8189 T2	4.16864E-02
11984 T2	-1.54552E-02		12000 T2	1.54552E-02	8191 T2	-1.16699E-02	8191 R3	-1.60897E-02	11939 T2	-3.08964E-02
11982 T2	2.21442E-02		11998 T2	-2.21442E-02	11997 T2	-2.38265E-02	11981 T2	2.38265E-02	11996 T2	3.77231E-02
11980 T2	-3.77231E-02		11995 T2	3.77231E-02	11979 T2	-3.77231E-02	11994 T2	-2.38264E-02	11978 T2	2.38265E-02
11993 T2	-2.04619E-02		11977 T2	2.04619E-02	11991 T2	-2.38265E-02	11975 T2	2.38265E-02	11990 T2	-2.38265E-02
11974 T2	2.38265E-02		11989 T2	3.43365E-02	11973 T2	-3.43585E-02	11987 T2	-2.38265E-02	11971 T2	2.38265E-02
8681 T2	2.37639E-02		8682 R1	-1.83044E-02	8121 T2	-3.87163E-02	2231 T2	1.52732E-02	8139 R1	1.80068E-02
8140 T2	4.18293E-02		5522 T2	-2.15883E-02	12643 T2	-2.36785E-02	12631 T2	1.49632E-02	12627 T1	-1.03372E-02
12627 T2	1.68856E-02		12644 T2	-3.26462E-02	12644 R3	1.59659E-02	13017 T3	2.90813E-02	12982 T1	-2.58204E-02
12982 T2	-1.77398E-02		12623 T1	1.60772E-02	12623 T2	-2.62618E-02	12645 T3	4.29261E-02	12645 R3	1.73106E-02
13016 T3	1.27131E-02		12646 T2	-1.33221E-02	12646 R3	1.42774E-02	12976 T1	2.07192E-02	12976 T2	1.02227E-02
13015 T3	-2.31117E-02		13004 T3	-1.90269E-02	12911 T1	2.85386E-02	12911 T2	1.40718E-02	12911 T3	-1.03628E-02
13005 T3	4.86314E-02		13006 T3	2.82980E-02	13006 R2	2.07910E-02	12927 T1	-2.32669E-02	12927 T2	-1.14799E-02
13007 T3	4.22099E-02		12943 T1	-1.10494E-02	13009 T3	3.40707E-02	12613 T2	-1.00056E-02	12613 T2	1.63439E-02
12647 T2	-3.05834E-02		12647 R3	2.97296E-02	12972 T1	1.41777E-02	13014 T3	-5.47403E-02	7871 T2	-2.37639E-02
12950 T1	-1.45831E-02		13010 T3	1.57287E-02	13010 R2	-1.16753E-02	12607 T2	1.03881E-02	12607 T2	-1.69688E-02
12648 T2	4.52989E-02		12648 R3	2.02038E-02	12968 T1	-1.19722E-02	13013 T3	1.28959E-02	13013 R2	-1.17567E-02
13011 T3	-2.25312E-02		12964 T1	-1.36958E-02	13012 T3	3.69402E-02	12649 T2	-1.16819E-02	12649 R3	1.28469E-02
12650 T2	-1.37819E-02		12650 R3	2.67161E-02	12569 T2	2.13829E-02	12569 R3	1.95174E-02	12495 T1	-2.61773E-02
12495 T2	4.27597E-02		12504 T2	-2.12228E-02	12504 R1	-1.09750E-02	8318 T2	1.68433E-02	11909 T2	-1.07171E-02
12692 T2	-2.60185E-02		12696 T2	1.67042E-02	12693 T2	-3.56670E-02	12694 T2	5.80667E-02	12698 T2	-3.41819E-02

POINT	KFFR	VALUE	POINT	VALUE	POINT	VALUE	POINT	VALUE	POINT	VALUE
10691 T2	-2.80034E-02		10692 T2	2.30202E-02	10660 T2	1.03582E-01	10660 R1	2.70417E-02	10660 R3	3.67681E-02
10661 T2	-1.04599E-01		10651 R1	-3.54153E-02	10706 R3	-7.46095E-02	10665 T2	-1.03434E-01	10665 R1	-2.80014E-02
10665 R1	3.75712E-02		10656 T2	1.03435E-01	10666 R1	3.44705E-02	12753 T2	3.90570E-02	12771 T2	-1.95286E-02
12771 R1	1.55282E-02		10713 T2	2.16411E-02	10717 T2	-2.16411E-02	10718 T2	4.05787E-02	12772 T2	-1.93901E-02
12772 R1	1.19933E-02		12772 R3	-1.06700E-02	10714 T2	-2.16411E-02	10718 T2	2.16411E-02	12755 T2	3.22233E-02
9275 T2	-1.56640E-02		10670 T2	1.76203E-01	10670 R1	8.39874E-02	10670 R3	8.86926E-02	10671 T2	-1.76201E-01
10671 R1	-6.52971E-02		12756 T2	-2.62096E-02	12768 T2	3.09801E-02	12218 T2	1.44861E-02	10723 R3	-1.76852E-01
12757 T2	-4.69913E-02		12767 R3	1.13374E-02	10675 T2	-1.79890E-01	10675 R1	-8.29575E-02	10675 R3	8.86483E-02
10676 T2	1.79895E-01		10676 R1	6.63246E-02	12758 T2	3.63610E-02	12766 T2	-2.55740E-02	12237 T2	3.11392E-02
12242 T2	1.33158E-02		12247 T2	2.47519E-02	12280 T2	1.00347E-02	12280 R3	1.40339E-02	12274 T2	-5.61814E-02
12279 T2	1.33167E-02		12279 R1	1.68491E-02	12273 T2	-2.66325E-02	12278 T2	5.18567E-02	12278 R3	1.10490E-02
12272 T2	-6.76010E-02		12252 T2	3.38005E-02	12277 T2	3.38005E-02	12277 R3	1.70566E-02	12271 T2	-6.76010E-02
12266 T2	-2.30761E-02		12283 T2	2.30761E-02	12283 R3	1.16448E-02	12281 T2	-4.61523E-02	10681 T2	-9.35206E-02
10681 R1	-2.51215E-02		10680 T2	1.02732E-01	10680 R1	3.53209E-02	10680 R3	4.70608E-02	12759 T2	4.55219E-02
12765 T2	-2.63144E-02		12765 R3	-1.10091E-02	12296 T2	2.21579E-02	12292 T2	4.37839E-02	12292 R3	1.30070E-02
12287 T2	-8.26725E-02		12295 T2	1.21371E-02	10748 R1	-3.68590E-02	12291 R3	2.21582E-02	12291 R1	-1.1815E-02
12286 T2	-4.43161E-02		10749 T2	2.20960E-02	10749 R1	4.79245E-01	10749 R3	1.67683E-01	10685 T2	-1.48444E-01
10751 R3	-5.33418E-02		10685 R3	-4.39428E-02	10686 T2	1.32609E-01	10686 R3	1.18619E-02	10781 T2	3.03453E-02
10685 R1	-4.52337E-02		12760 T2	1.00730E-02	12764 T2	-3.86555E-02	12764 R1	1.27546E-02	12761 T2	4.69362E-02
10782 T2	-3.03453E-02		12763 R1	3.31880E-02	12763 R3	-1.45879E-02	9924 T2	3.69104E-02	9886 R1	1.53992E-02
12763 T2	-3.27402E-02		11911 T2	-2.37301E-02	11911 R3	3.52173E-01	12348 T2	-1.63768E-02	12343 T2	-6.15406E-02
2826 T2	-3.76525E-02		12404 T2	2.45495E-02	12400 T2	5.72643E-02	12400 R1	-1.34901E-02	12403 T2	-1.4508E-02
12401 T2	1.52689E-02		12425 T2	2.95107E-02	12424 R1	-1.31337E-02	12384 T2	-2.53896E-02	12423 T2	-3.32152E-02
12394 T2	-2.29739E-02		12425 T2	2.95107E-02	12370 T2	-1.12234E-02	12420 T2	1.12234E-02	12681 T2	1.78393E-02
12375 T2	-1.14879E-02		12421 T2	1.14879E-02	12725 T2	-2.45861E-02	12730 T2	-1.63693E-02	12735 T2	3.11084E-02
12715 T2	-1.95284E-02		12720 T2	2.13233E-02	12748 T2	2.60015E-02	5505 T2	-2.23692E-02	12750 R1	-1.22044E-02
12740 T2	-1.16373E-02		12745 T2	-2.11022E-02	2009 T2	1.28797E-02	2952 R3	-1.47497E-02	2953 T2	1.11654E-02
8552 T2	3.16134E-02		8552 R1	1.34497E-02	5512 R3	2.70713E-02	5515 T2	1.14591E-02	5513 R3	2.75002E-02
5511 T2	-2.09972E-02		5511 R3	2.07200E-02	2959 R3	-1.57665E-02	2957 R3	-1.00295E-02	5517 R3	1.37293E-02
2960 R3	-1.57683E-02		5514 T2	2.74637E-02	5516 R3	1.67951E-02	2738 T2	-1.33383E-02	3721 T2	1.74885E-02
2958 R3	-1.36213E-02		5516 T2	1.13013E-02	5520 R3	1.10260E-02	4718 T2	3.35955E-02	3721 T2	-1.91974E-02
5518 R3	1.37458E-02		5519 R3	1.35386E-02	4718 T2	3.35955E-02	4722 T3	3.03393E-02	4703 T2	-3.35952E-02
3733 T2	1.91973E-02		3733 R3	-5.86642E-02	3727 T1	1.10843E-02	4816 T3	-3.31954E-02	3744 T3	7.83747E-02
3721 T3	-5.35924E-02		3739 T3	-2.67964E-02	4816 T2	1.67977E-02	4834 T2	-1.67977E-02	4834 T3	-4.24116E-02
4703 T3	-6.63908E-02		4830 T3	-1.83829E-02	3751 T3	7.17331E-02	4834 T2	-1.67977E-02	4851 T3	-3.78195E-02
4830 T2	1.67977E-02		4846 T2	-3.67772E-02	4852 T3	1.01225E-02	4851 T2	1.67977E-02	3757 T2	-2.87960E-02
3762 T2	2.87960E-02		4846 T3	-3.67772E-02	4852 T3	1.01225E-02	4838 T3	-1.11642E-01	4842 T3	1.19564E-02
3757 T3	-8.03888E-02		3762 T3	1.35696E-02	3767 T2	2.93788E-01	4842 T2	-1.67975E-02	4758 T2	3.35955E-02
4823 T2	-5.03929E-02		3767 T1	1.10840E-02	3767 T3	2.93788E-01	3734 T3	-5.86642E-02	3722 T3	-5.35925E-02
4758 T3	-6.71053E-02		4823 T3	-9.95863E-02	3734 T2	1.91973E-02	3734 T3	-5.86642E-02		
4762 T3	3.03393E-02		3728 T1	1.10843E-02	3728 T2	2.15413E-01	3722 T2	-1.91974E-02		
			4743 T2	-3.35952E-02	4743 T3	-6.63908E-02				
3029 T3	1.68668E-02		3028 R2	-1.36274E-02	3027 R2	-1.90478E-02	3026 R2	-2.07030E-02	3025 R2	-1.93395E-02
3024 R2	-1.67805E-02		3023 T3	-1.30187E-02	3023 R1	1.08545E-02	3023 R2	-1.71642E-02	3022 T3	-1.62843E-02
3010 T3	1.22726E-02		1579 T1	4.37142E-02	1579 T2	1.62182E-01	1579 R1	6.16600E-02	1579 R1	4.40275E-02
1435 T2	-1.36318E-01		1435 T3	-6.64114E-02	1425 T1	-1.87819E-01	1425 T2	-1.31186E-02	1425 T3	-3.41147E-02
1425 R2	4.97539E-02		5582 T1	-1.50402E-01	5582 T2	6.30915E-02	5582 R1	1.07104E-02	5582 R1	1.22251E-02
5582 R2	2.30646E-02		1415 T1	-2.81065E-01	1415 T2	7.44386E-02	1415 T3	-1.60118E-02	1415 R1	1.11528E-02

COLUMN 6 (4003-R3).

POINT	KFR	VALUE	POINT	VALUE	POINT	VALUE	POINT	VALUE	
1415 R2	9.10848E-02	5585 T1	-3.79029E-01	5585 R2	5.81198E-02	1405 T1	-3.66451E-01	1405 R2	1.09624E-01
5586 T1	-4.56597E-01	5586 R2	7.00124E-02	5586 T1	-3.26150E-01	1395 T3	1.05597E-02	1395 R2	9.19131E-02
5587 T1	-4.06381E-01	5587 R2	6.23101E-02	1385 T1	-1.12768E-01	1385 T2	-7.44387E-02	1385 T3	2.06480E-02
1385 T1	-1.11527E-02	1385 R2	4.23373E-02	5588 T1	-2.50413E-01	5588 R2	3.83925E-02	1375 T1	-1.06256E-01
1375 T2	-7.44384E-02	1375 R1	-1.11527E-02	1375 R2	1.93812E-02	5589 T1	-7.23305E-02	5589 R2	1.10886E-02
1569 T1	2.94835E-01	1569 T2	1.11559E-01	1569 T3	2.34041E-02	5589 R1	1.64309E-02	1569 R2	-6.63512E-02
1559 T1	6.60094E-01	1559 T2	1.11558E-01	1559 T3	1.65946E-02	5593 R1	1.67144E-02	1559 R2	-1.49205E-01
1549 T1	8.23048E-01	1549 R2	-1.79636E-01	1553 T1	7.32531E-01	1539 T3	-1.05641E-02	1539 R2	-1.54233E-01
1529 T1	4.03181E-01	1529 T2	-1.11559E-01	1529 T3	-2.12319E-02	1529 R1	-1.67141E-02	1529 R2	-8.07298E-02
1519 T1	1.78587E-01	1519 T2	-1.11558E-01	1519 R1	-1.67141E-02	1519 R2	-3.07697E-02	7259 T3	-1.31467E-02
7261 T3	1.26082E-02	1214 T3	5.22577E-02	1216 R2	1.00082E-02	12048 T3	4.13333E-02	8066 T3	-1.67203E-02
12026 T3	-2.06176E-02	11961 T3	1.27140E-02	12057 T3	2.20233E-02	12057 R1	-1.24770E-02	12041 R1	-1.46758E-02
11993 T3	-2.69959E-02	11962 T3	1.07992E-02	12056 R1	-1.26358E-02	12040 R1	-1.50886E-02	12024 R1	-1.47275E-02
12893 T3	-1.58667E-02	12899 T2	1.38736E-02	12894 T3	-4.83586E-02	12394 R2	1.68670E-02	2923 T3	1.89166E-02
12895 T2	2.85102E-02	12895 R1	1.56483E-02	2895 R3	1.17017E-02	12350 T3	-3.44250E-02	2297 T3	6.23527E-02
12978 T3	-4.17435E-02	12978 R1	-1.35293E-02	12979 T3	1.73157E-02	12350 T3	2.34321E-02	22904 R1	-1.98935E-02
12619 T3	2.21495E-02	12620 T3	1.50663E-02	12620 R1	1.57450E-02	12350 T3	-1.07686E-02	22621 R1	1.93962E-02
12974 T3	1.58784E-02	12974 R1	-1.13500E-02	12975 R2	-2.93638E-02	12376 T3	-5.51948E-02	12908 T3	-1.43187E-02
12908 R2	-1.62663E-02	12907 T3	-1.80301E-02	12907 R1	2.71512E-02	12307 R2	-2.52239E-02	12910 T3	3.21562E-02
12910 R1	-1.99570E-02	12910 R2	-2.20937E-02	12932 T3	-1.34454E-02	12312 R2	1.57446E-02	12911 T3	1.49809E-02
12921 T3	1.51873E-02	12926 T3	-2.11817E-02	12541 T3	-1.53875E-02	12317 R2	-1.76908E-02	12615 T3	-1.74684E-02
12615 R1	1.19341E-02	12614 T3	2.58668E-02	12614 R1	1.39568E-02	12313 T3	1.28022E-02	12969 T3	-2.63916E-02
12970 R1	1.05464E-02	12971 T3	-1.00900E-02	22951 T3	2.59059E-02	12313 T3	-1.99330E-02	12950 T3	1.59255E-02
12958 T3	2.40507E-02	12958 R1	1.62624E-02	22962 T3	-2.62713E-02	12362 R1	1.13483E-02	12961 T3	1.68647E-02
12967 R2	1.14330E-02	12956 T3	-1.56299E-02	22963 T3	2.53077E-02	12363 R2	-1.11301E-02	12968 T3	2.56321E-02
12964 T3	-2.16147E-02	11896 T3	2.02496E-02	22966 R2	-1.75794E-02	12456 T3	1.07330E-02	12456 R2	-5.04257E-02
12457 R2	-6.81795E-02	12458 R2	-6.81724E-02	22459 R2	-3.47737E-02	5324 T3	-2.96022E-02	5324 T3	-2.96022E-02
8324 R2	1.82960E-02	8322 T2	4.68951E-02	12459 R2	-5.30919E-02	8312 R2	-3.62925E-02	8312 R2	2.13918E-02
12451 R2	-3.63696E-02	12451 R2	7.33974E-02	12452 R2	9.52109E-02	12453 T3	-1.98089E-02	12453 R2	9.43007E-02
12454 T3	6.67085E-02	12444 R2	4.50323E-02	12446 R2	4.67204E-02	12441 R2	-5.84473E-02	12447 R2	3.78551E-02
12448 R2	2.37921E-02	12449 T3	-1.13623E-02	12449 R2	1.29156E-02	12442 T3	-6.41208E-02	12443 R2	3.08456E-02
12444 T3	5.78840E-02	12444 R2	1.66808E-02	8320 T3	-3.43262E-02	8320 R2	2.09985E-02	12436 R2	4.47972E-02
12437 T3	5.71076E-02	12437 R2	3.45377E-02	22438 T3	1.66548E-02	12438 R2	1.62697E-02	12439 T3	2.48710E-02
12776 T3	-1.51879E-02	12692 T2	1.23442E-02	12781 T3	-1.49697E-02	12784 T3	2.06129E-02	12783 T3	-3.11989E-02
12785 T3	-2.70801E-02	12785 R1	1.05043E-02	12693 T2	1.69217E-02	12784 T3	-1.48294E-02	12694 T2	-2.75490E-02
12768 T2	1.62172E-02	12797 R1	1.46485E-02	12799 R2	-1.01605E-02	12753 T2	-1.85301E-02	12804 T3	1.30371E-02
12802 T3	2.64117E-02	12754 T2	-1.92521E-02	12812 T3	1.62118E-02	12753 T2	1.03620E-02	12811 R2	1.04356E-02
12810 R1	-1.21519E-02	12755 T2	1.52879E-02	12820 T3	-1.91914E-02	12815 R1	1.07028E-02	12824 R1	-1.09294E-02
12824 R2	1.50144E-02	12823 T3	2.98362E-02	12756 T2	1.24348E-02	12768 T2	-1.46981E-02	9533 T3	-1.54747E-02
12826 T3	-1.92582E-02	12827 R1	-1.04782E-02	12831 R1	-1.12379E-02	12830 T3	-3.27645E-02	12757 T2	2.22344E-02
9247 T3	1.56409E-02	12840 T3	1.34893E-02	12841 T3	3.13978E-02	12758 T2	1.72513E-02	12766 T2	2.21333E-02
12237 T2	-2.18030E-02	12247 T2	-1.45624E-02	12274 T2	3.30535E-02	12273 T2	1.56688E-02	12273 T2	-3.05091E-02
12278 R3	-1.00305E-02	12272 T2	3.97721E-02	12252 T2	-1.38860E-02	12277 T2	-1.98860E-02	12277 R3	-1.00350E-02
12271 T2	3.97721E-02	12266 T2	-1.35765E-02	12283 T2	-1.35765E-02	12281 T2	2.71530E-02	12842 T3	1.09890E-02
12843 T3	-1.97072E-02	12844 T3	-1.35057E-02	12845 T3	-3.42814E-02	12759 T2	-2.15973E-02	12765 T2	2.14845E-02
12297 T2	-1.18979E-02	12292 T2	-2.57595E-02	12287 T2	4.86391E-02	12296 T2	-1.30363E-02	12291 T2	-1.30364E-02
12296 T2	2.60727E-02	12295 T2	-1.29998E-02	12852 T3	3.35664E-02	12764 T2	1.83396E-02	12761 T2	-2.2683E-02
12763 T2	1.55332E-02	12763 R1	-1.57456E-02	12851 T3	2.04451E-02	12855 T3	1.23637E-02	12857 T3	-1.16363E-02
12859 R1	-1.29310E-02	12860 T3	-3.06113E-02	12343 T2	2.91971E-02	12404 T2	-1.16472E-02	12400 T2	-2.71683E-02
12394 T2	1.08997E-02	12425 T2	-1.40010E-02	12384 T2	-1.20458E-02	12423 T2	1.57585E-02	12720 T2	1.01166E-02

LARGE FIELD DATA TRANS ALTER 126 EQUILIB CHECK - NETSAT

JUNE 7, 1996 MSC/NASIRAN

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POINT	KFFR	VALUE	POINT	VALUE	POINT	VALUE	POINT	VALUE
12725 T2	1.16646E-02		12735 T2	-1.47590E-02	12745 T2	1.60117E-02	12748 T2	-1.23361E-02
8185 T3	-7.66029E-02		8185 R2	-1.59037E-02	8164 T3	4.53143E-02	8169 T3	3.12885E-02
8722 T3	2.37850E-02		8710 T3	1.32164E-02	8707 T3	-1.32164E-02	8830 T3	-2.47215E-02
8828 T3	2.47216E-02		3672 T3	-2.46220E-02	5539 T3	1.85347E-02	3847 T3	-1.21990E-02
3869 R1	1.49058E-02		3834 T3	1.23669E-02	3834 R1	-1.04679E-02	3882 R1	1.11316E-02
3833 T3	1.22011E-02		3790 T3	-1.18242E-02	3759 T3	1.20185E-02	3798 T3	1.06552E-02
1511 R1	-1.31340E-02		3632 R1	1.34532E-02	1512 R1	-1.32146E-02	3626 R1	1.35781E-02
3620 R1	1.17712E-02		3452 T3	1.78575E-02	1472 R1	1.32146E-02	1471 R1	1.25292E-02
3625 T3	1.05536E-02		3625 R1	-1.17712E-02	3631 R1	-1.35781E-02	3637 R1	-1.29160E-02
2949 T3	-1.28629E-02		2970 T3	1.02102E-02	2952 T3	1.00558E-02	2969 T3	-4.43300E-02
2968 T3	2.60670E-02		2968 R1	-1.71141E-02	2968 R2	-2.51714E-02	2959 T3	-1.80178E-02
2967 R2	-1.33452E-02		2958 T3	-3.56672E-02	2965 R1	1.32235E-02	2957 T3	-1.47566E-02
2964 T3	-2.69233E-02		2964 R2	-1.09935E-02	2963 T3	-5.49578E-02	2963 R2	-1.72115E-02
2951 T3	-1.47455E-02		2962 T3	6.00111E-02	2962 R1	1.30676E-02	2950 T3	-1.01896E-02
2954 T3	1.01896E-02		3757 T3	-1.20186E-02	3004 R2	1.25410E-02	3005 R2	1.43593E-02
3007 T3	1.30056E-02		3008 T3	-2.22144E-02	3001 T3	-1.18551E-02	4967 R2	-1.38073E-02
4968 R2	-1.43314E-02		4959 T3	-1.03222E-02	4970 T3	-1.01044E-02	4965 T3	1.28512E-02
5104 T3	-3.96243E-02		5117 T3	-3.10734E-02	5117 R1	1.72457E-02	5117 R2	2.48284E-02
5130 R2	3.57801E-02		5145 R2	3.79706E-02	5160 T3	2.98237E-02	5160 R2	2.94745E-02
5176 T3	3.20704E-02		5180 R1	1.85443E-02	5160 R2	-1.30018E-02	5171 T3	3.24315E-02
5052 R1	1.16948E-02		5052 R2	-1.61649E-02	5180 R2	-1.30018E-02	5067 T3	-3.50065E-02
2944 T3	-2.28014E-02		2943 T3	4.91049E-02	2948 T3	-3.66167E-02	2945 T3	2.04348E-02
					2942 T3	2.32512E-02	11648 T3	-1.20889E-02

LARGE FIELD DATA TRANS ALTER 126 EQUILIB CHECK - METSAT

POINT	KFFRN	VALUE		POINT		VALUE		POINT		VALUE		POINT		VALUE	
		VALUE	POINT	VALUE	POINT	VALUE	POINT	VALUE	POINT	VALUE	POINT	VALUE	POINT	VALUE	POINT
COLUMN	1 (	4003-T1).													
COLUMN	2 (	4003-T2).													
COLUMN	3 (	4003-T3).													
COLUMN	4 (	4003-R1).													
COLUMN	5 (	4003-R2).													
COLUMN	6 (	4003-R3).													

Appendix C
DESIGN LOAD CASES 1 THROUGH 8
MAXIMUM CONTOUR & STRESS PLOTS

Time: 10:38:33
Date: 06/12/96

Contour
Node Scalar1

Color Index

—	0	0.186E+04
—	9	0.166E+04
—	8	0.146E+04
—	7	0.125E+04
—	6	0.105E+04
—	5	0.846E+03
—	4	0.642E+03
—	3	0.439E+03
—	2	0.236E+03
—	1	0.326E+02

Min = 3.258430E+01

Max = 2.064914E+03

Min ID= 1172

Max ID= 1094

Contour_6:

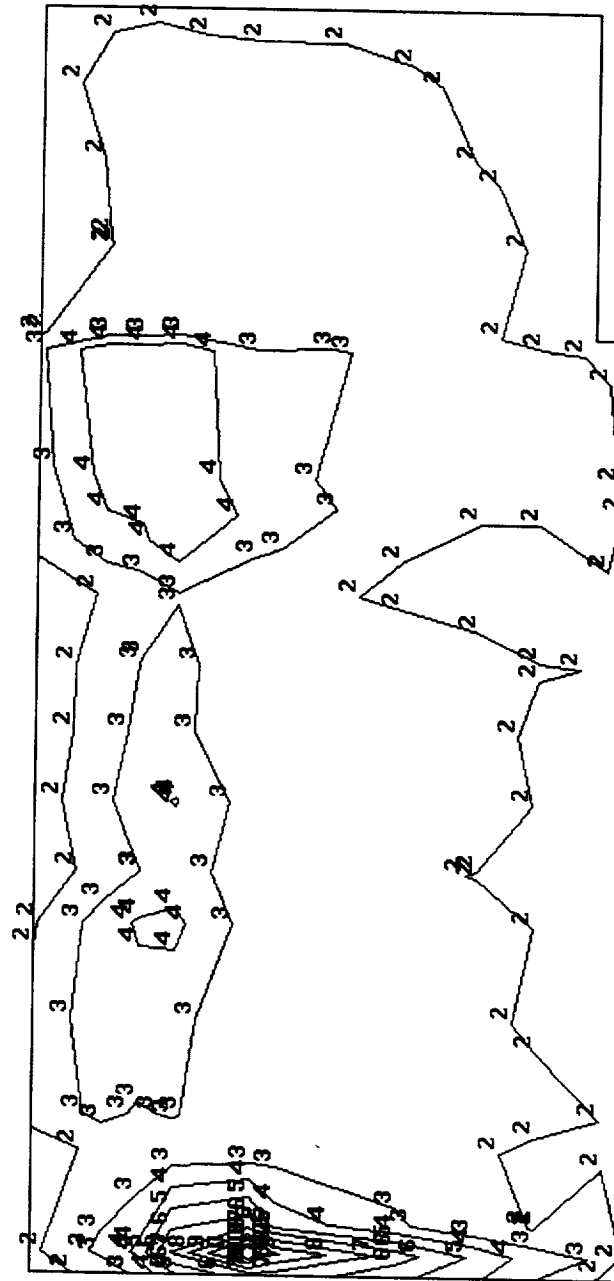
Stress Tensor

Von Mises
At 22

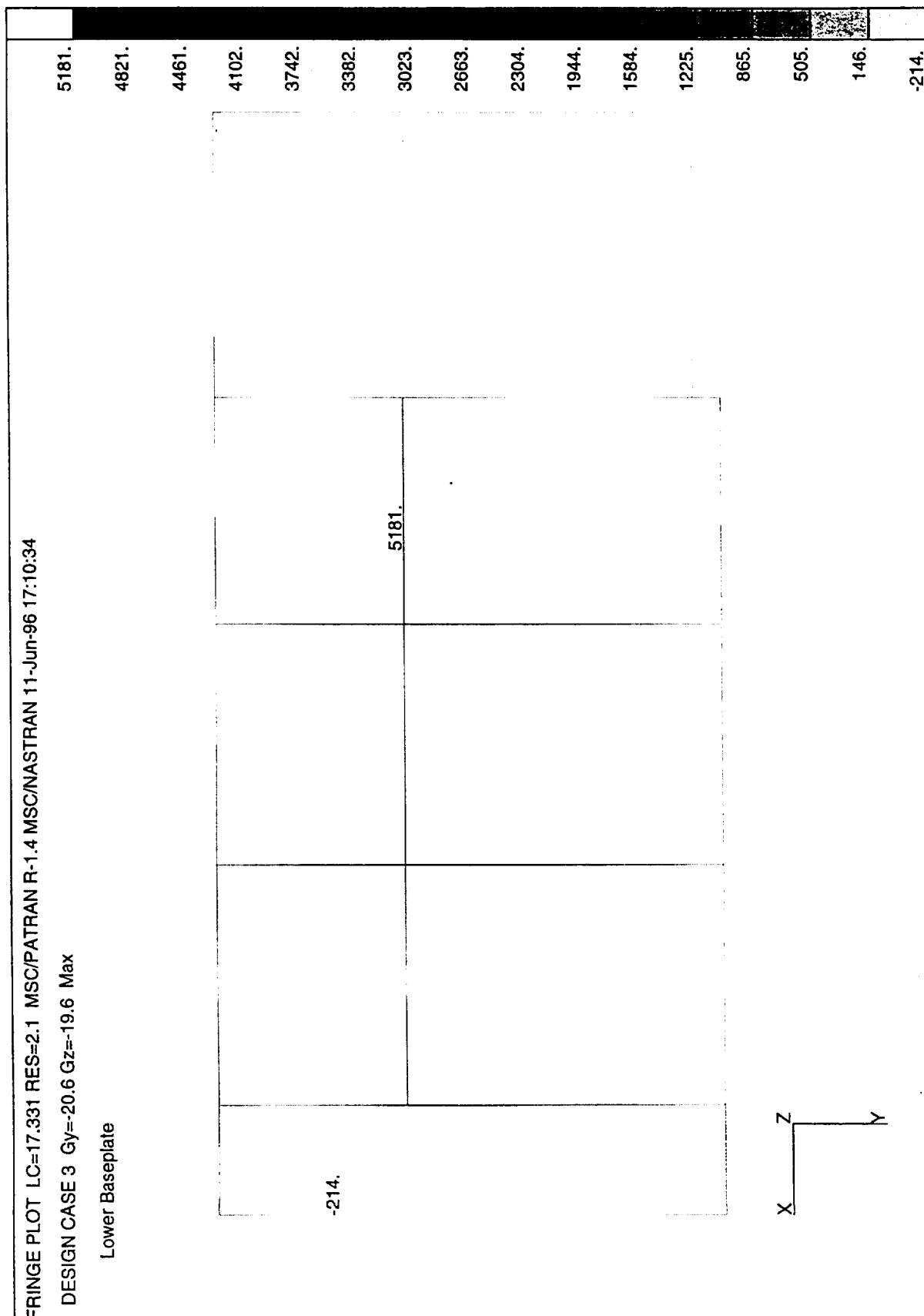
Design_Gy=-20.6_Gz=-19.6
Static Subcase

METSAT AMSU-A1

Lower Baseplate



X
Y
Z



Time: 11:19:26
Date: 06/12/96

Contour
Node Scalar1

Color Index

—	0	0.369E+04
—	9	0.329E+04
—	8	0.288E+04
—	7	0.248E+04
—	6	0.207E+04
—	5	0.167E+04
—	4	0.126E+04
—	3	0.855E+03
—	2	0.450E+03
—	1	0.442E+02

Min = 4.418999E+01

Max = 4.100530E+03

Min ID= 1172

Max ID= 2880

Contour_4:

Stress Tensor

Von Mises

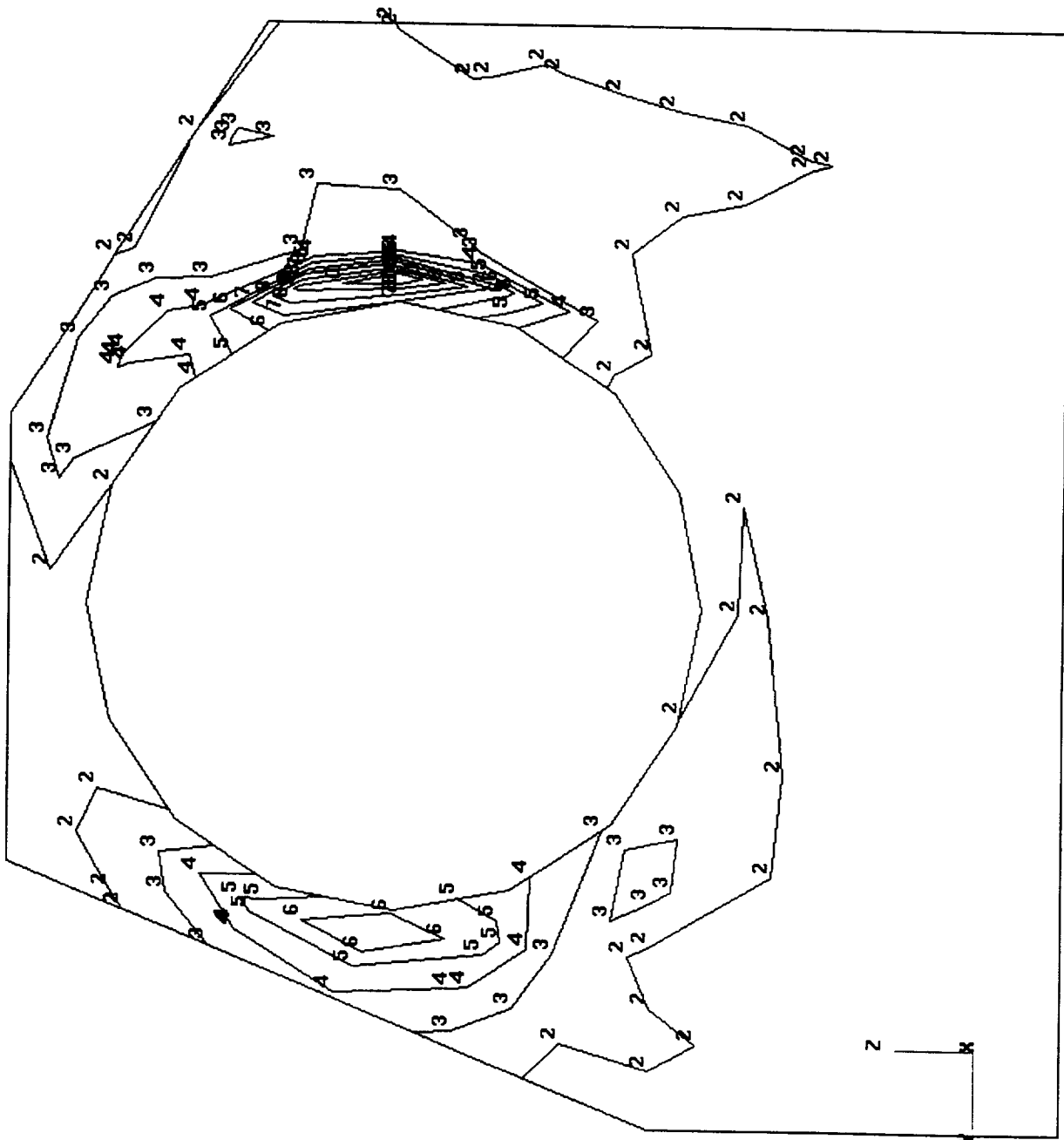
At Z2

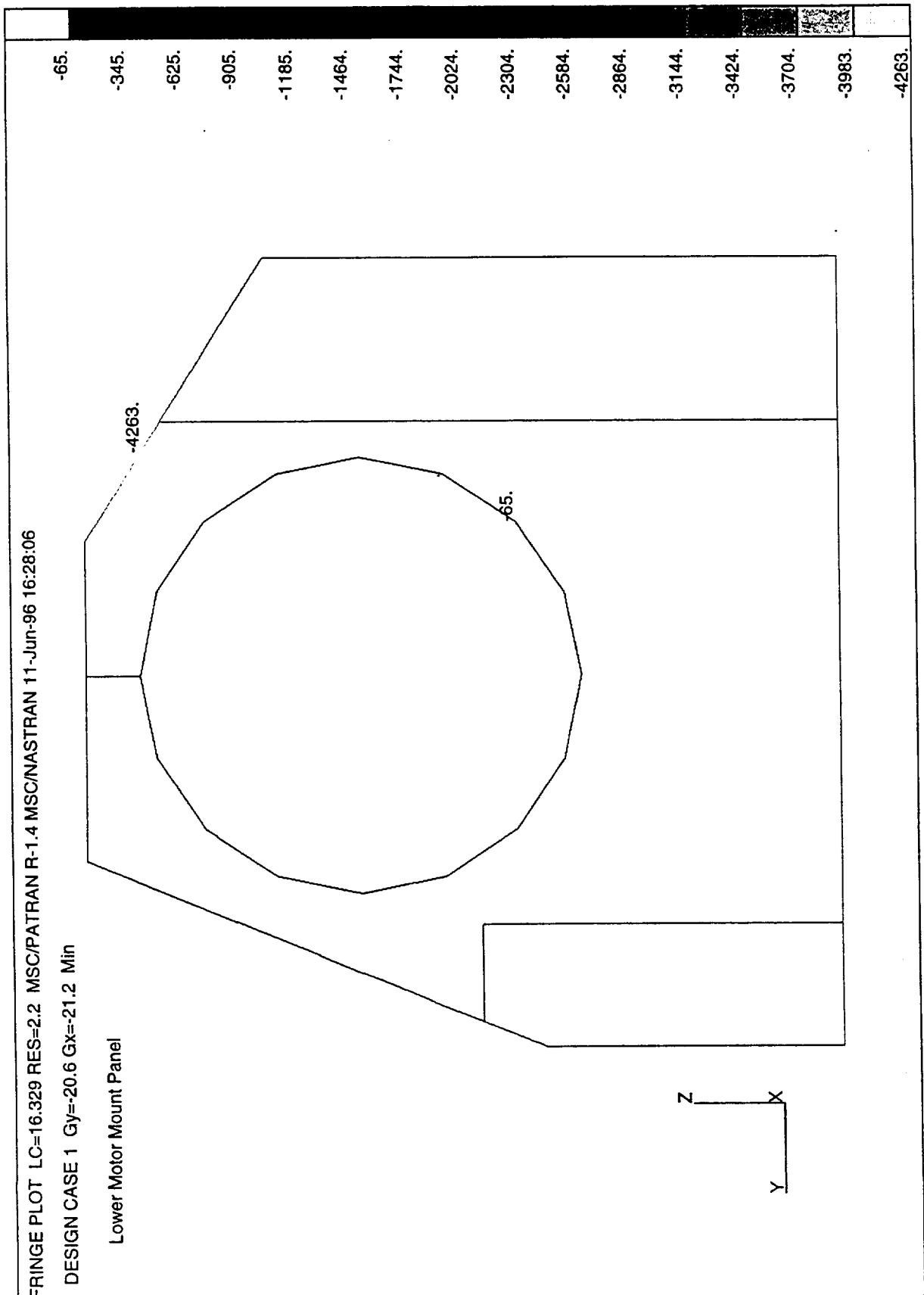
Design_Qy=-20.6_Qx=21.2

Static Subcase

METSAT AMSU-A1

Lower Motor Mount





Time: 11:24:22
Date: 06/12/96

Contour
Node Scalar1

Color Index	
0	0.178E+04
9	0.158E+04
8	0.139E+04
7	0.120E+04
6	0.100E+04
5	0.810E+03
4	0.616E+03
3	0.422E+03
2	0.229E+03
1	0.348E+02

Min = 3.480820E+01

Max = 1.972071E+03

Min ID= 1569

Max ID= 2773

Contour_2:

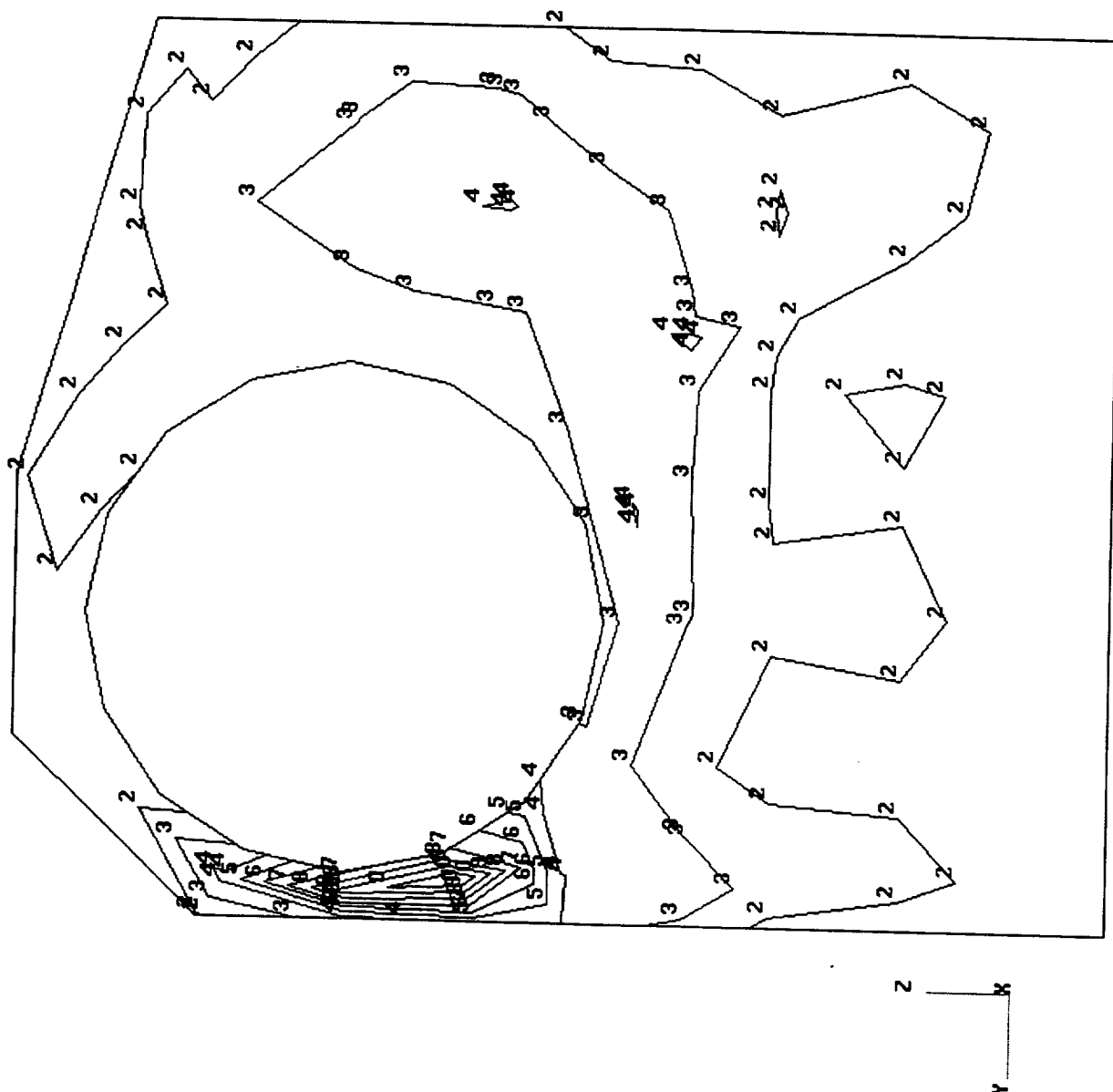
Stress Tensor

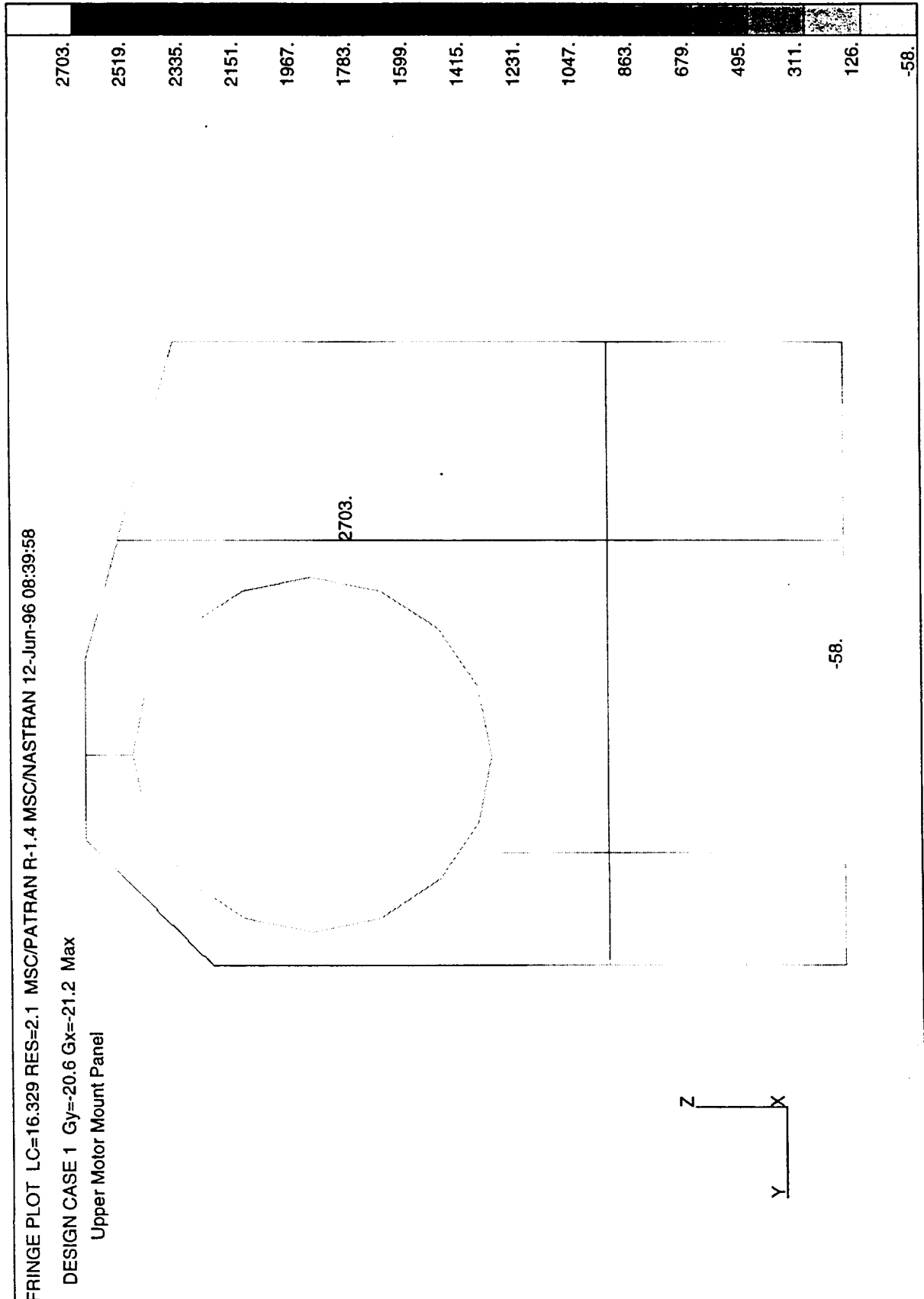
Von Mises

At 22

Design_gy=-20.6_gx=-21.2
Static Subcase

METSAT AMSU-A1
Upper Motor Mount





Time: 11:28:57
Date: 06/12/96

Contour
Node Scalar1

Color Index

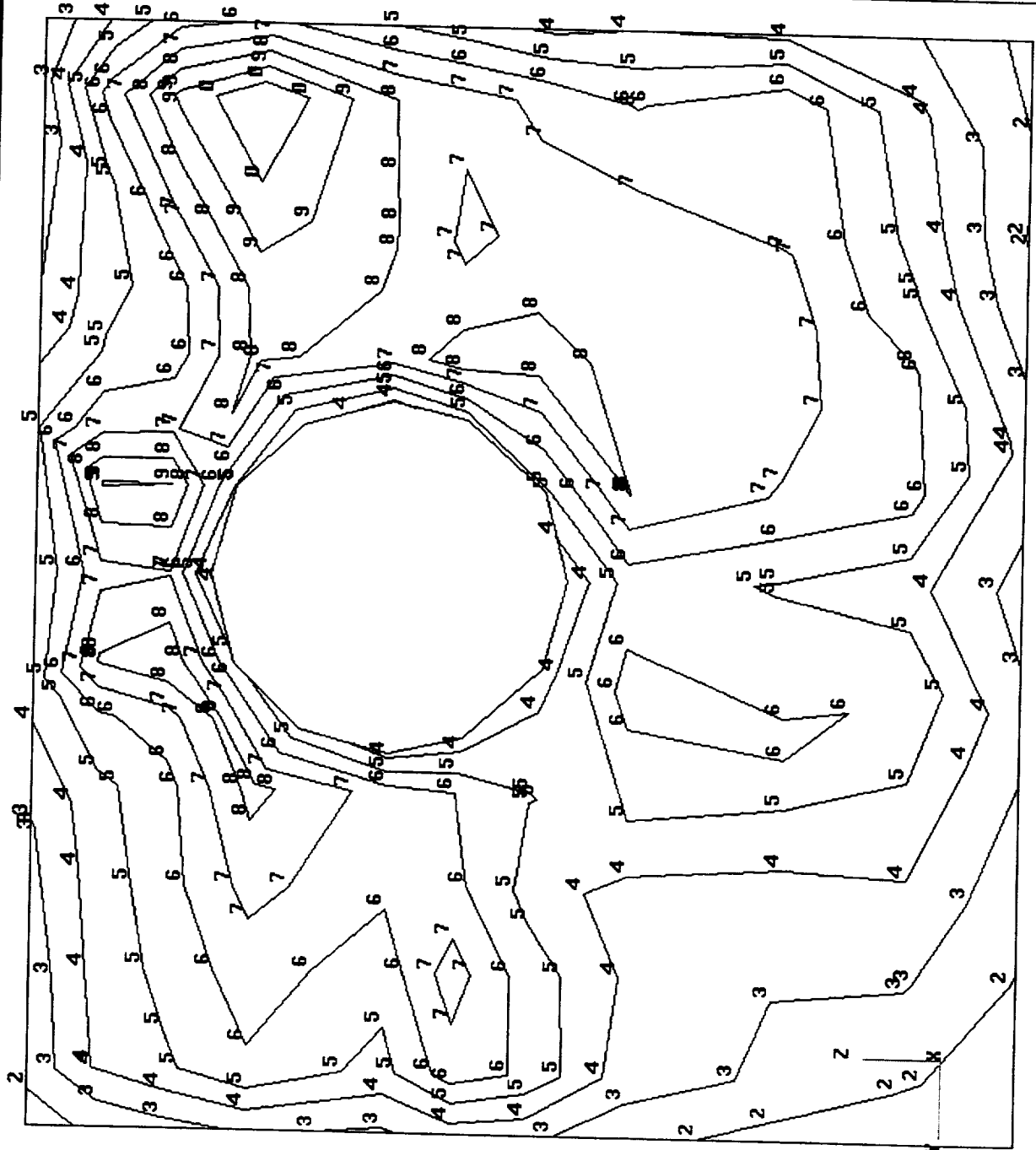
—	0	0.182E+04
—	9	0.163E+04
—	8	0.143E+04
—	7	0.123E+04
—	6	0.104E+04
—	5	0.843E+03
—	4	0.647E+03
—	3	0.452E+03
—	2	0.256E+03
—	1	0.605E+02

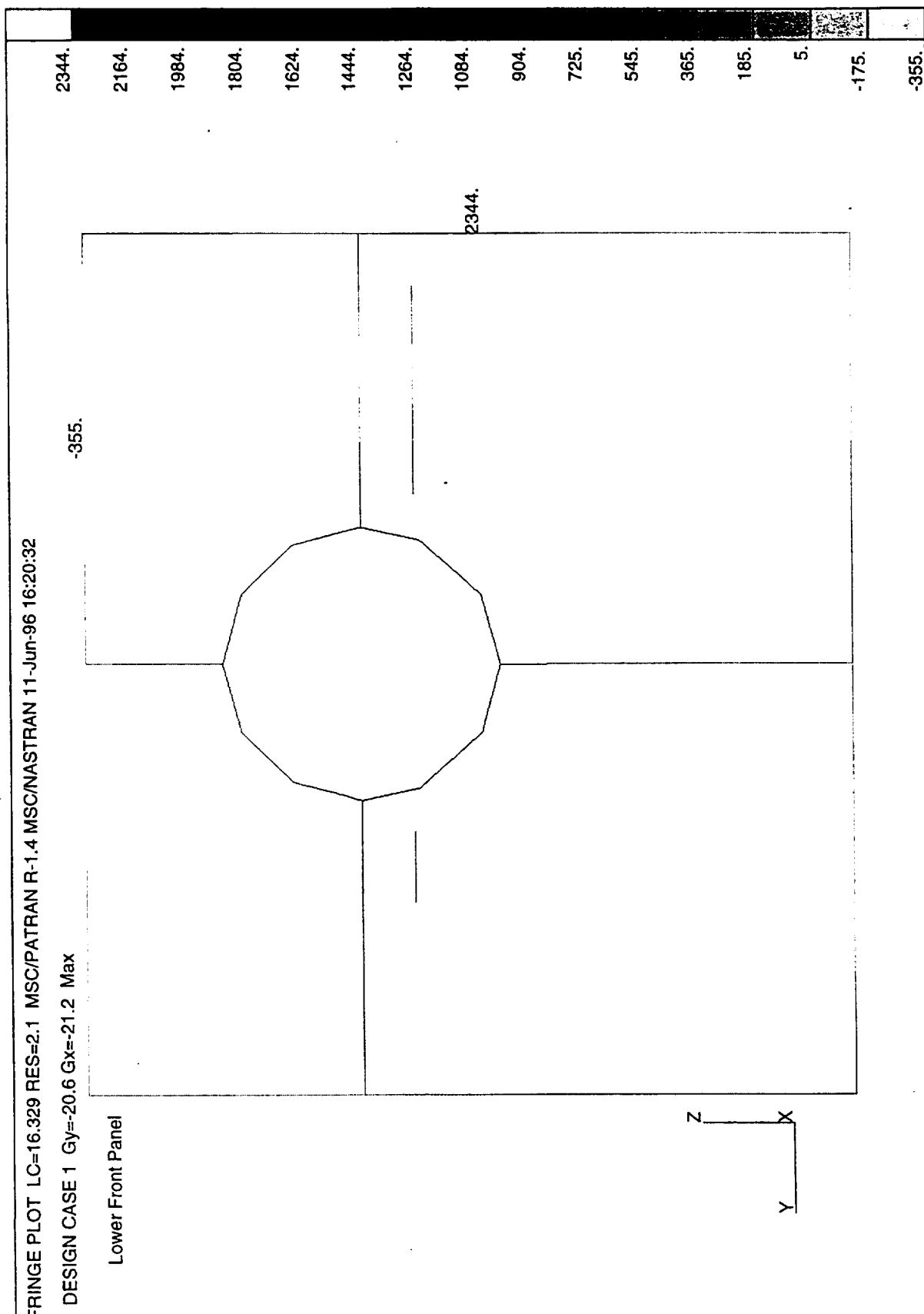
Min = 6.047171E+01
Max = 2.016662E+03
Min ID= 1191
Max ID= 1888
Contour_7:
Stress Tensor

Von Mises
At Z1

Design_Gy=-20.6_Gz=19.6
Static Subcase

METSAT AMSU-A1
Lower Front Panel





Time: 11:30:32
Date: 06/12/96

Contour
Node Scalar1

Color Index

—	0	0.158E+04
—	9	0.141E+04
—	8	0.124E+04
—	7	0.107E+04
—	6	0.903E+03
—	5	0.734E+03
—	4	0.566E+03
—	3	0.398E+03
—	2	0.229E+03
—	1	0.605E+02

Min = 6.055479E+01

Max = 1.745420E+03

Min ID= 1205

Max ID= 1227

Contour_5:

Stress Tensor

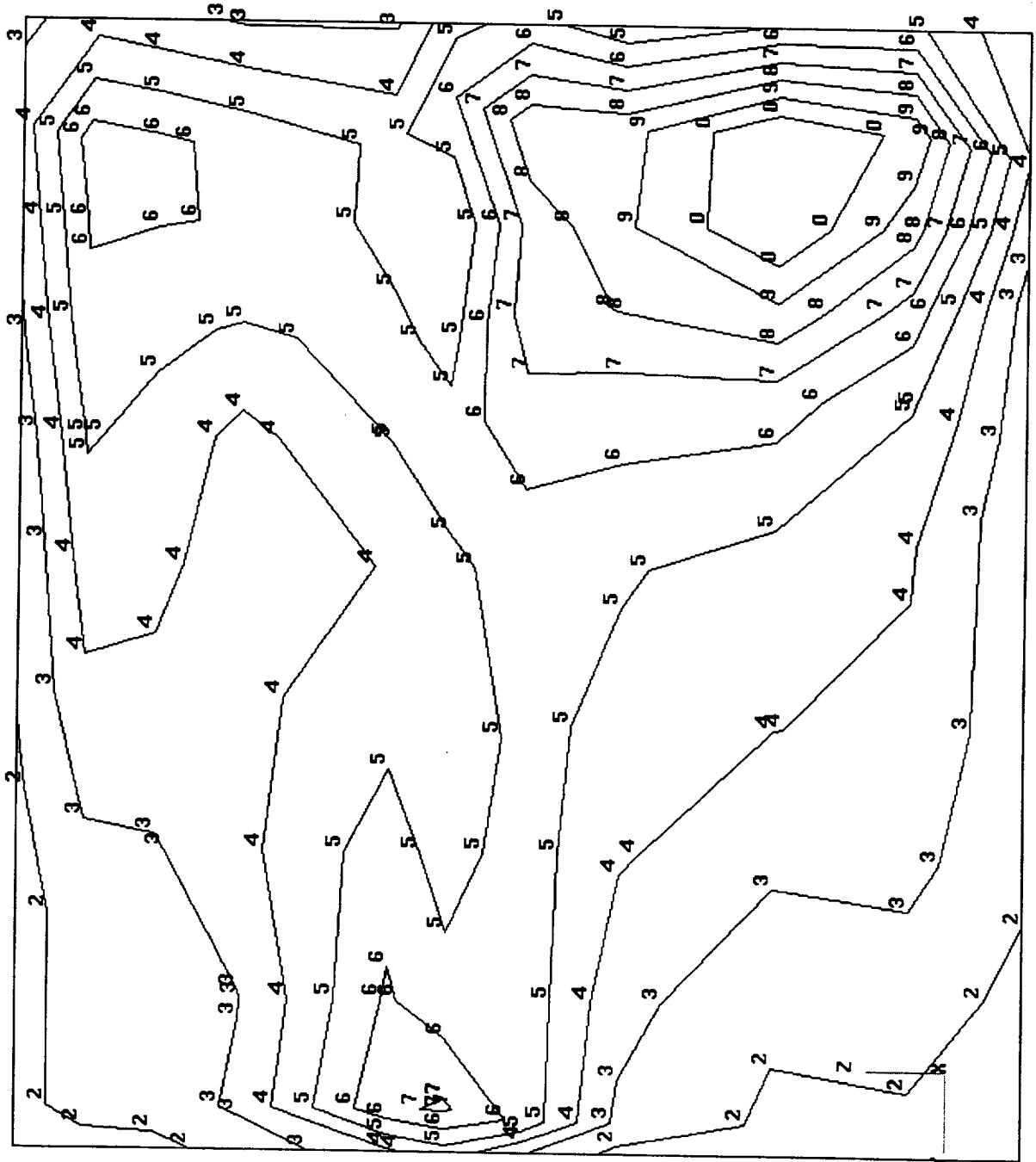
Von Mises

At Z1

Design_Gy=-20.6_Gz=-19.6

Static Subcase

METSAT AMSU-A1
Lower Aft Panel



Time: 10:44:47
Date: 06/12/96

Contour
Node Scalar1

Color Index

0	0.163E+04
9	0.145E+04
8	0.127E+04
7	0.109E+04
6	0.911E+03
5	0.731E+03
4	0.551E+03
3	0.371E+03
2	0.191E+03
1	0.112E+02

Min = 1.120488E+01
Max = 1.811265E+03

Min ID= 1457

Max ID= 1477

Contour_7:

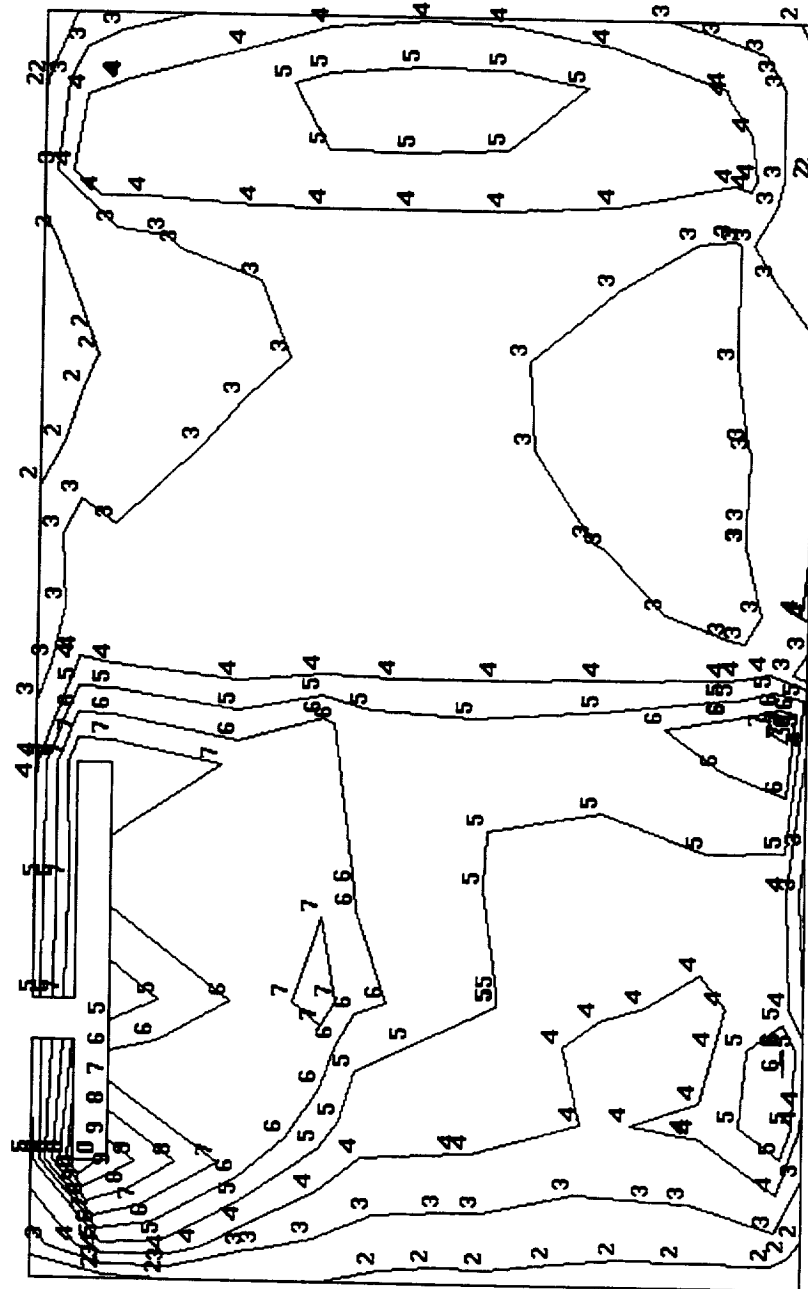
Stress Tensor

Von Mises

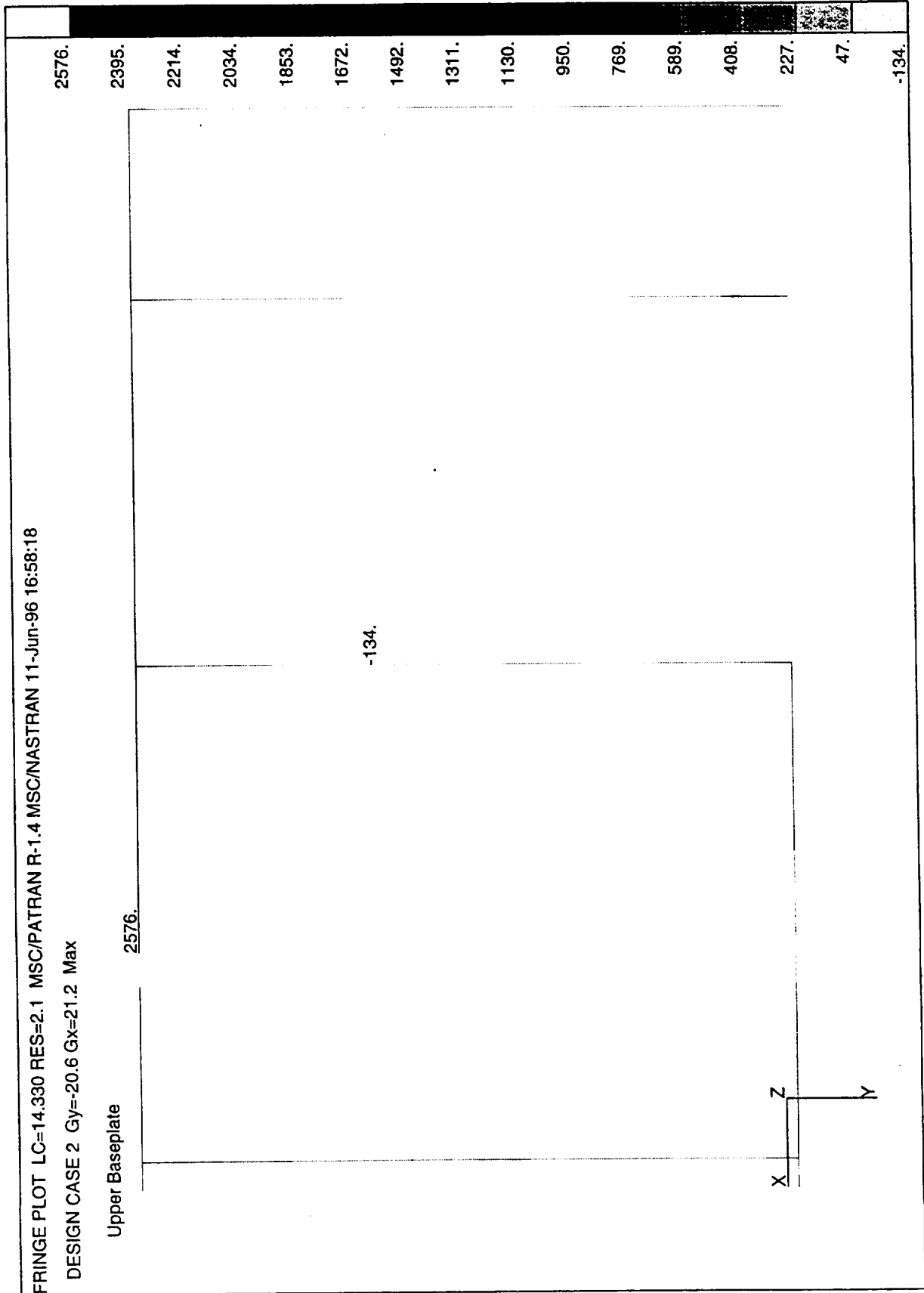
At 21

Design_Qy=-20.6_Gz=19.6
Static Subcase

METSAT AMSU-A1
Upper Baseplate



X
Y



Time: 12:57:28
Date: 06/13/96

Contour
Node Scalar1

Color Index

0	0.167E+04
9	0.149E+04
8	0.131E+04
7	0.112E+04
6	0.938E+03
5	0.754E+03
4	0.571E+03
3	0.387E+03
2	0.203E+03
1	0.190E+02

Min = 1.898106E+01

Max = 1.857719E+03

Min ID= 1993

Max ID= 2101

Contour_8:

Stress Tensor

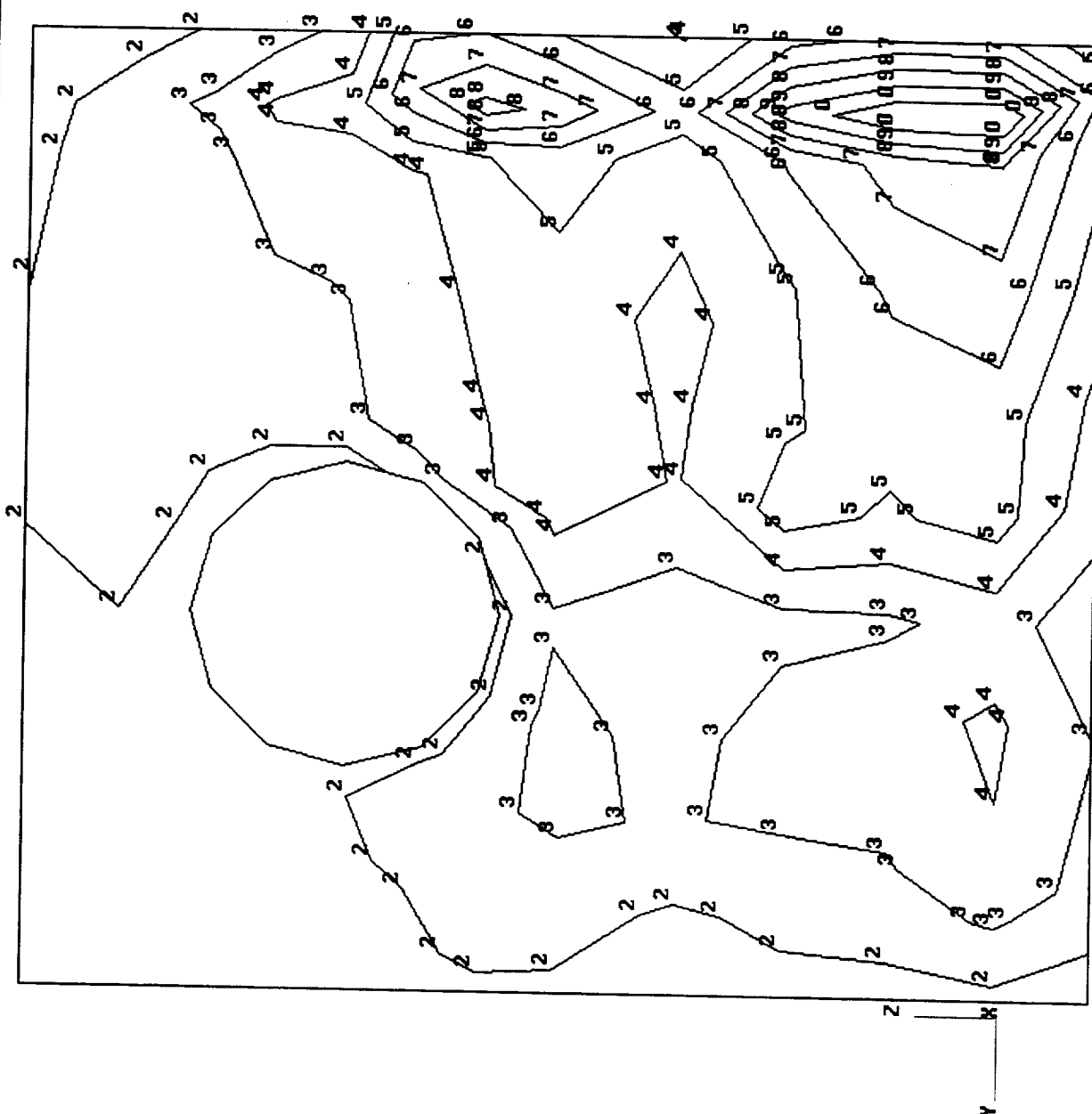
Von Mises

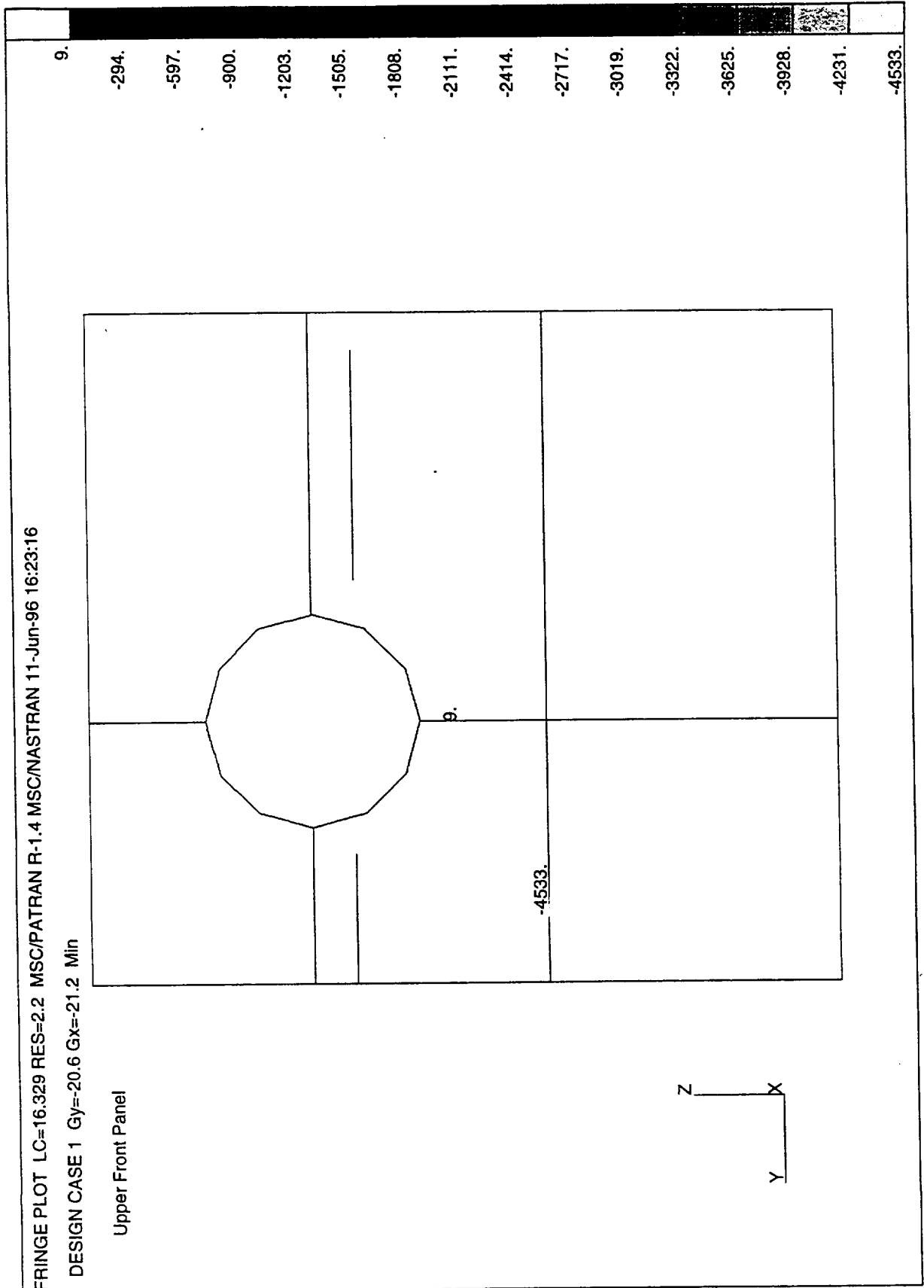
At 22

Design_Gy=-20.6_Gz=19.6

Static Subcase

METSAT AMSU-A1
Upper Front Panel





Time: 10:29:22
Date: 06/11/96

Contour
Node Scalar1

Color Index

0	0.460E+04
9	0.411E+04
8	0.363E+04
7	0.314E+04
6	0.266E+04
5	0.217E+04
4	0.168E+04
3	0.120E+04
2	0.712E+03
1	0.225E+03

Min = 2.254730E+02

Max = 5.086936E+03

Min ID= 6878

Max ID= 5594

Contour_1:

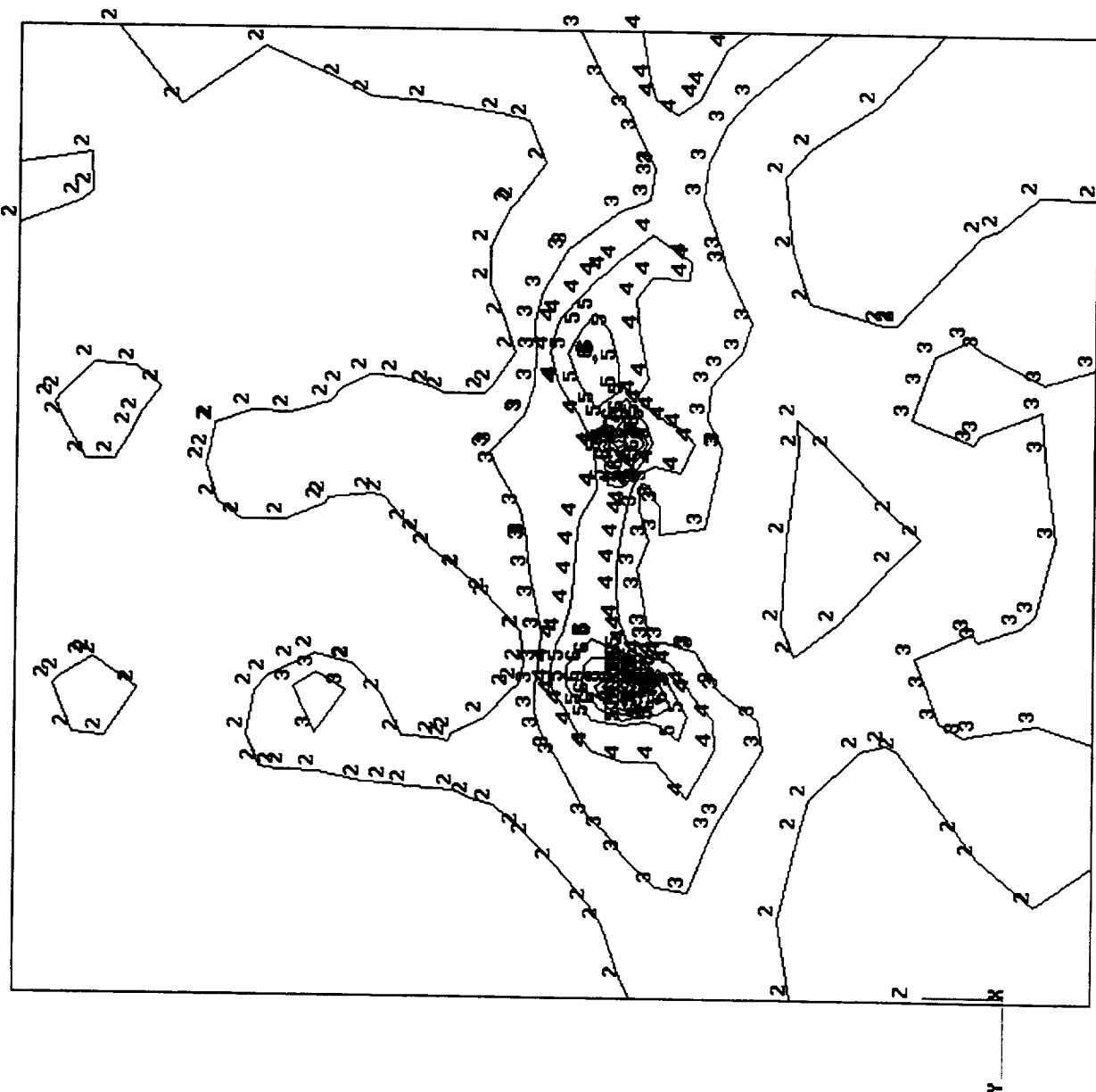
Stress Tensor

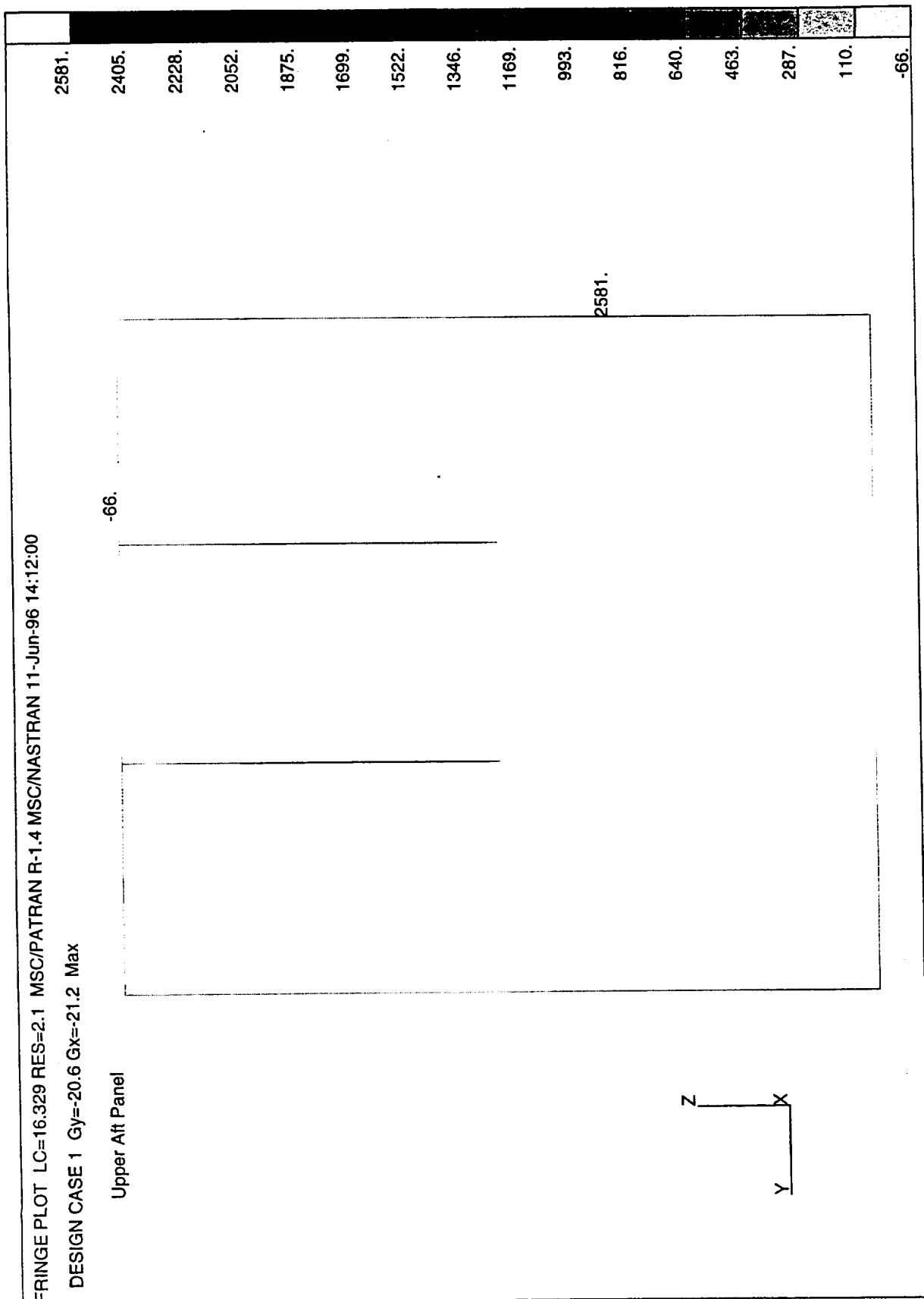
Von Mises

At 21

Design_Gy=-20.6_Gx=-21.2
Static Subcase

METSAT AMSU-A1
Upper Aft Panel





Time: 10:49:47
Date: 06/12/96

Contour
Node Scalar1

Color Index

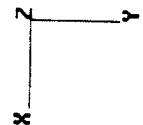
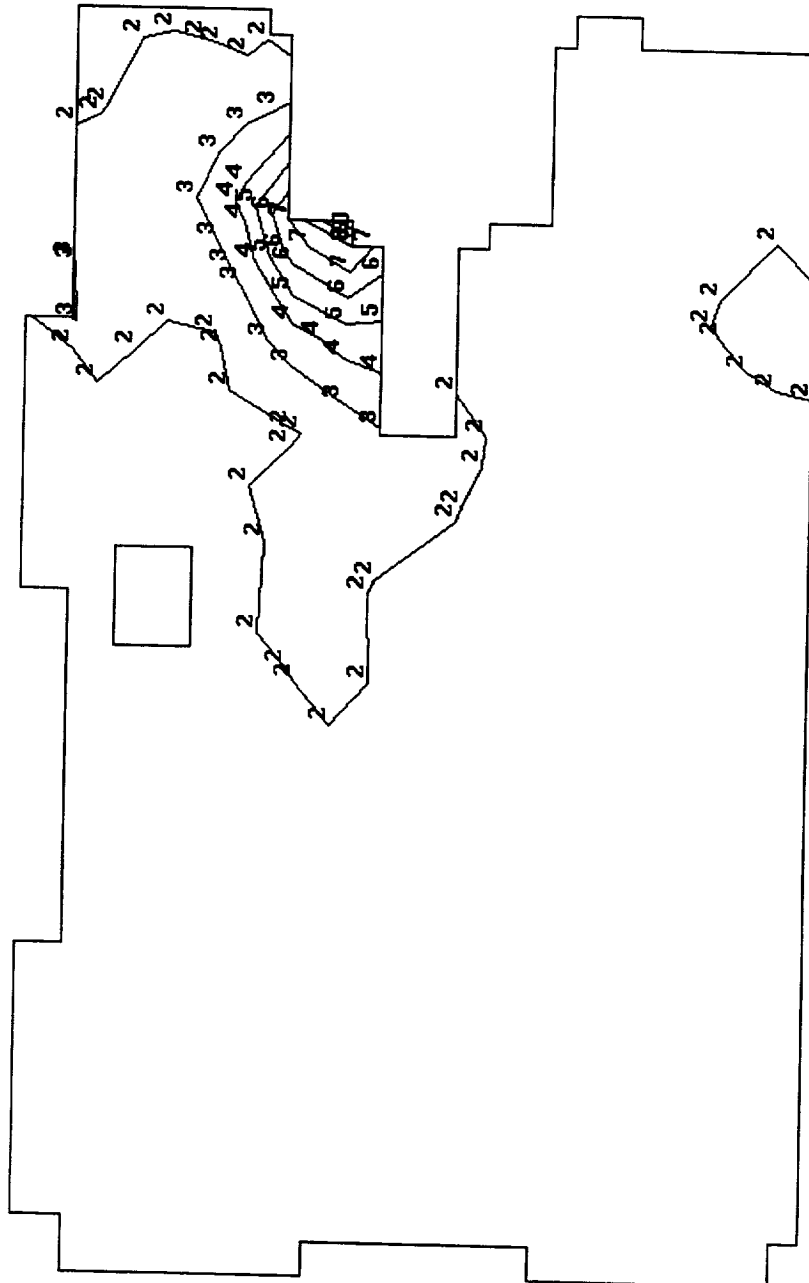
0	0.788E+04
9	0.701E+04
8	0.614E+04
7	0.526E+04
6	0.439E+04
5	0.352E+04
4	0.265E+04
3	0.178E+04
2	0.915E+03
1	0.446E+02

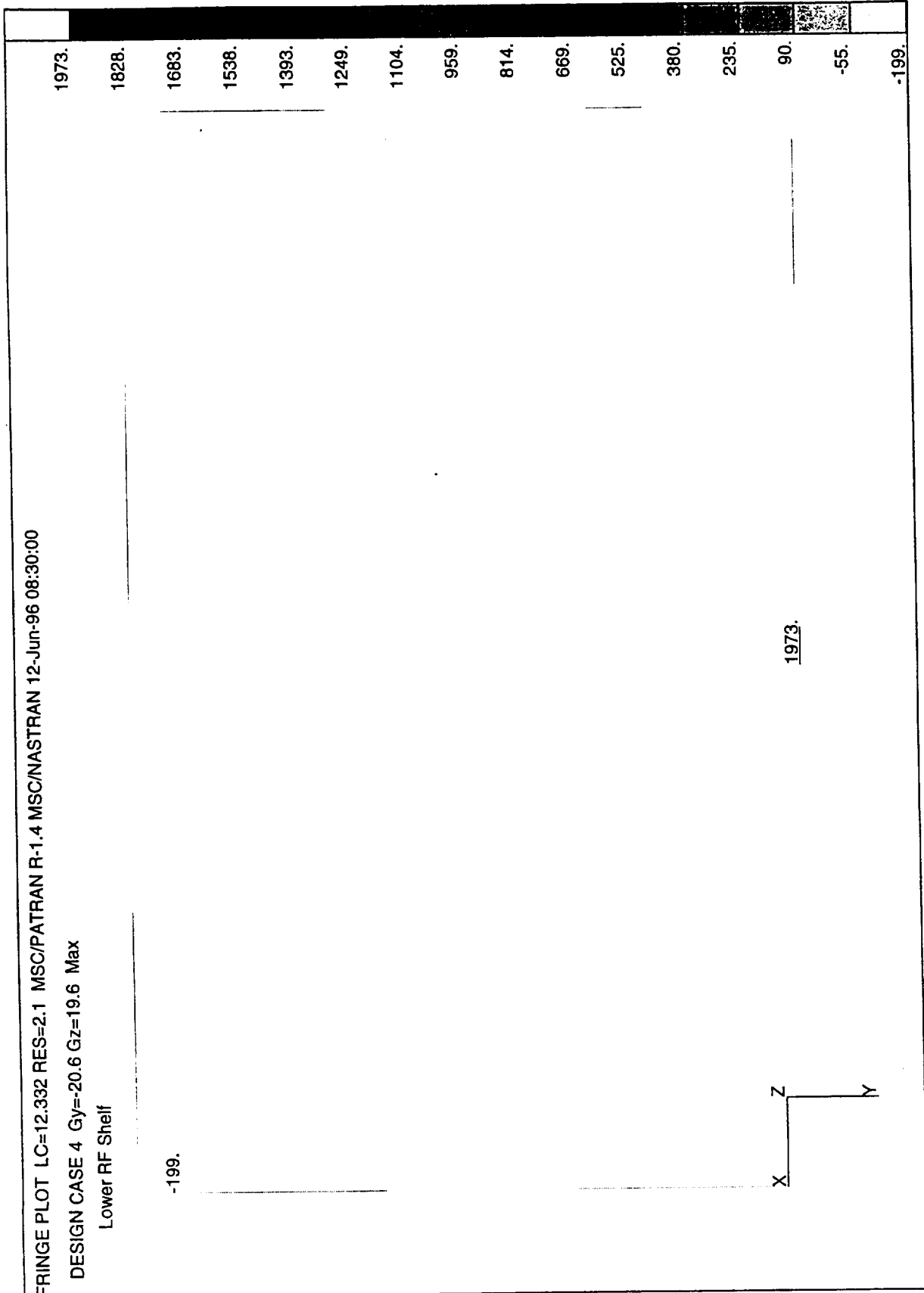
Min = 4.458046E+01
Max = 8.745224E+03
Min ID= 1861
Max ID= 1772
Contour_1:
Stress Tensor

Von Mises
At Z1

Design_Gy=-20.6_Gx=-21.2
Static Subcase

METSAT AMSU-A1
Lower RF Shelf





Time: 10:53:58
Date: 06/12/96

Contour
Node Scalar1

Color Index

0	0.343E+04
9	0.305E+04
8	0.268E+04
7	0.230E+04
6	0.192E+04
5	0.155E+04
4	0.117E+04
3	0.797E+03
2	0.421E+03
1	0.456E+02

Min = 4.559451E+01

Max = 3.804087E+03

Min ID= 2258

Max ID= 2569

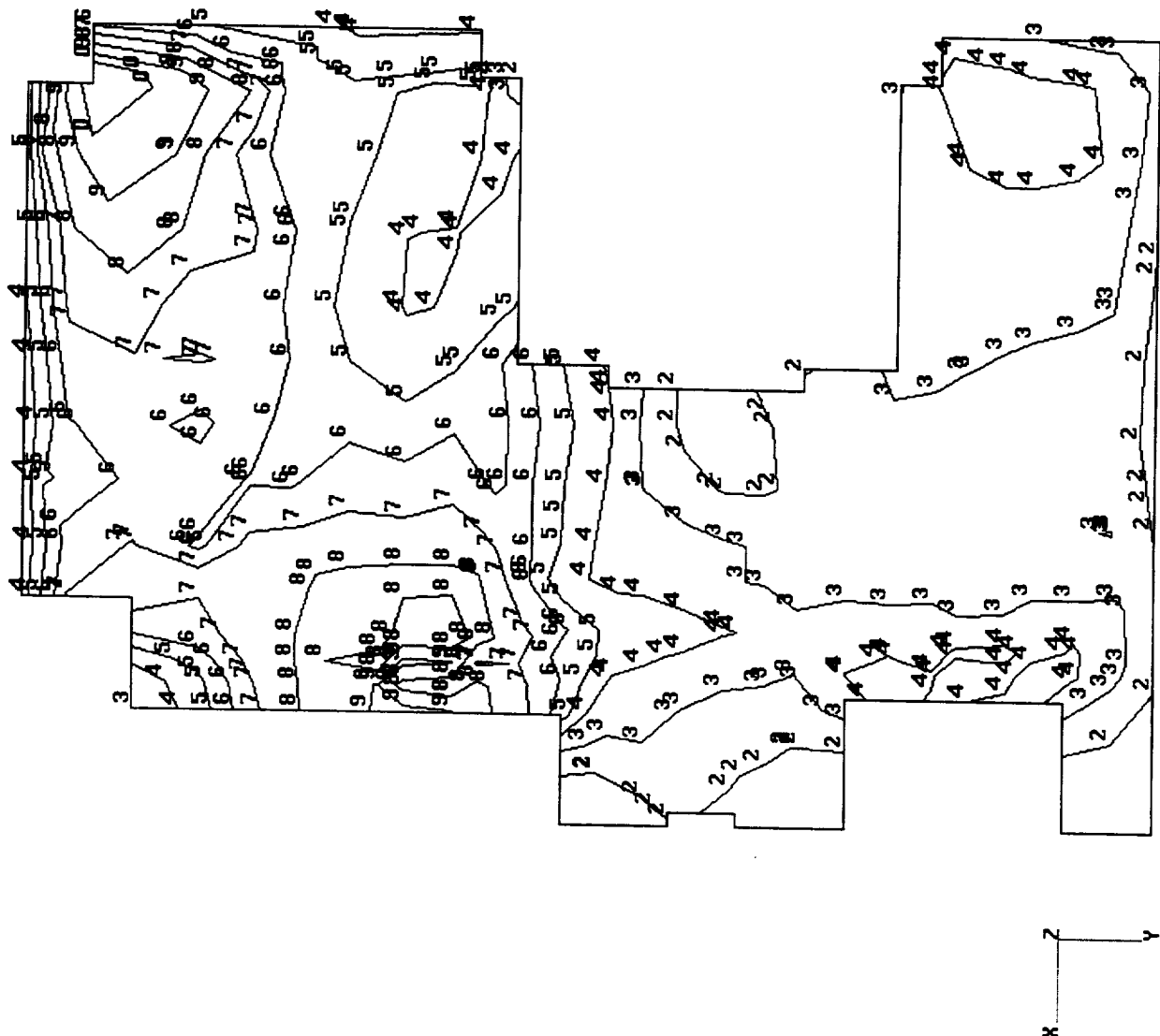
Contour_8:

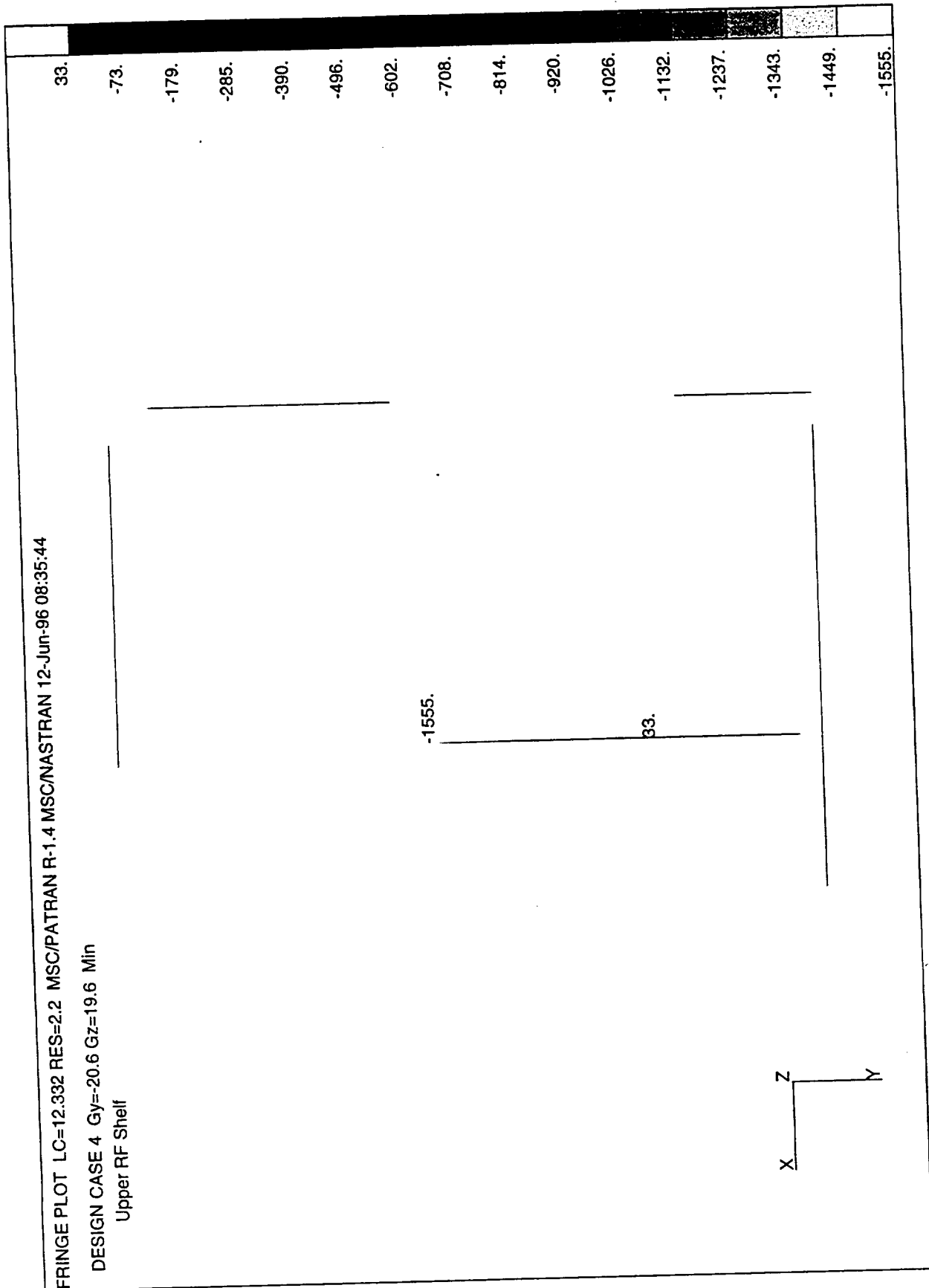
Stress Tensor

Von Mises
At 22

Design_Qy=-20.6_Gz=19.6
Static Subcase

METSAT AMSU-A1
Upper RF Shelf





Time: 07:38:09
Date: 06/14/96

Contour
Node Scalar1

Color	Index
—	0 0.523E+03
—	9 0.473E+03
—	8 0.423E+03
—	7 0.373E+03
—	6 0.323E+03
—	5 0.273E+03
—	4 0.223E+03
—	3 0.173E+03
—	2 0.123E+03
—	1 0.727E+02

Min = 7.268614E+01

Max = 5.733126E+02

Min ID= 1172

Max ID= 2658

Contour_1:

Stress Tensor

Von Mises

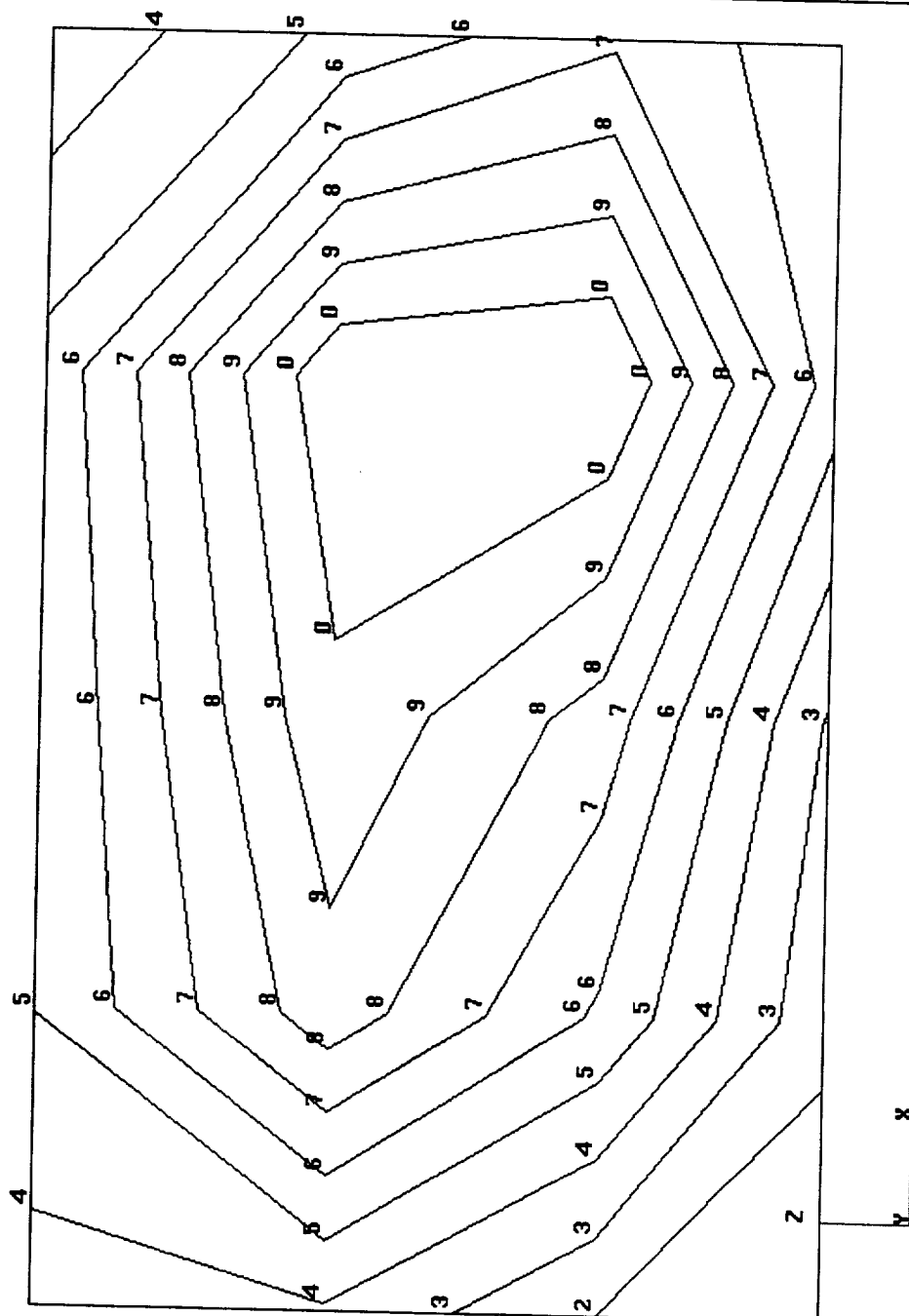
At 21

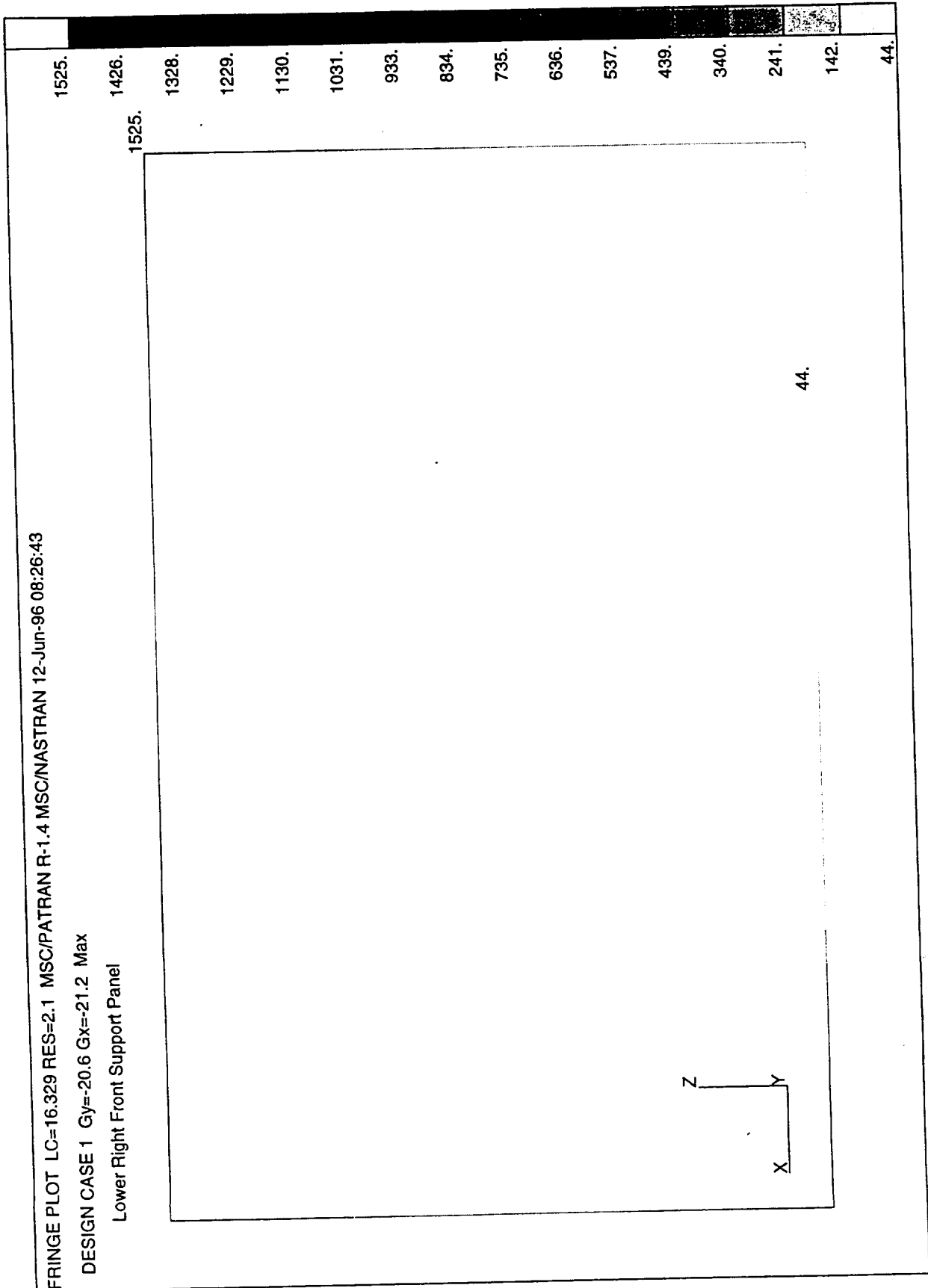
Design_Gy=-20.6_Gx=-21.2
Static Subcase

METSAT AMSU-A1

Lower Right Front

Support Panel





Time: 07:46:59
Date: 06/14/96

Contour
Node Scalar1

Color Index

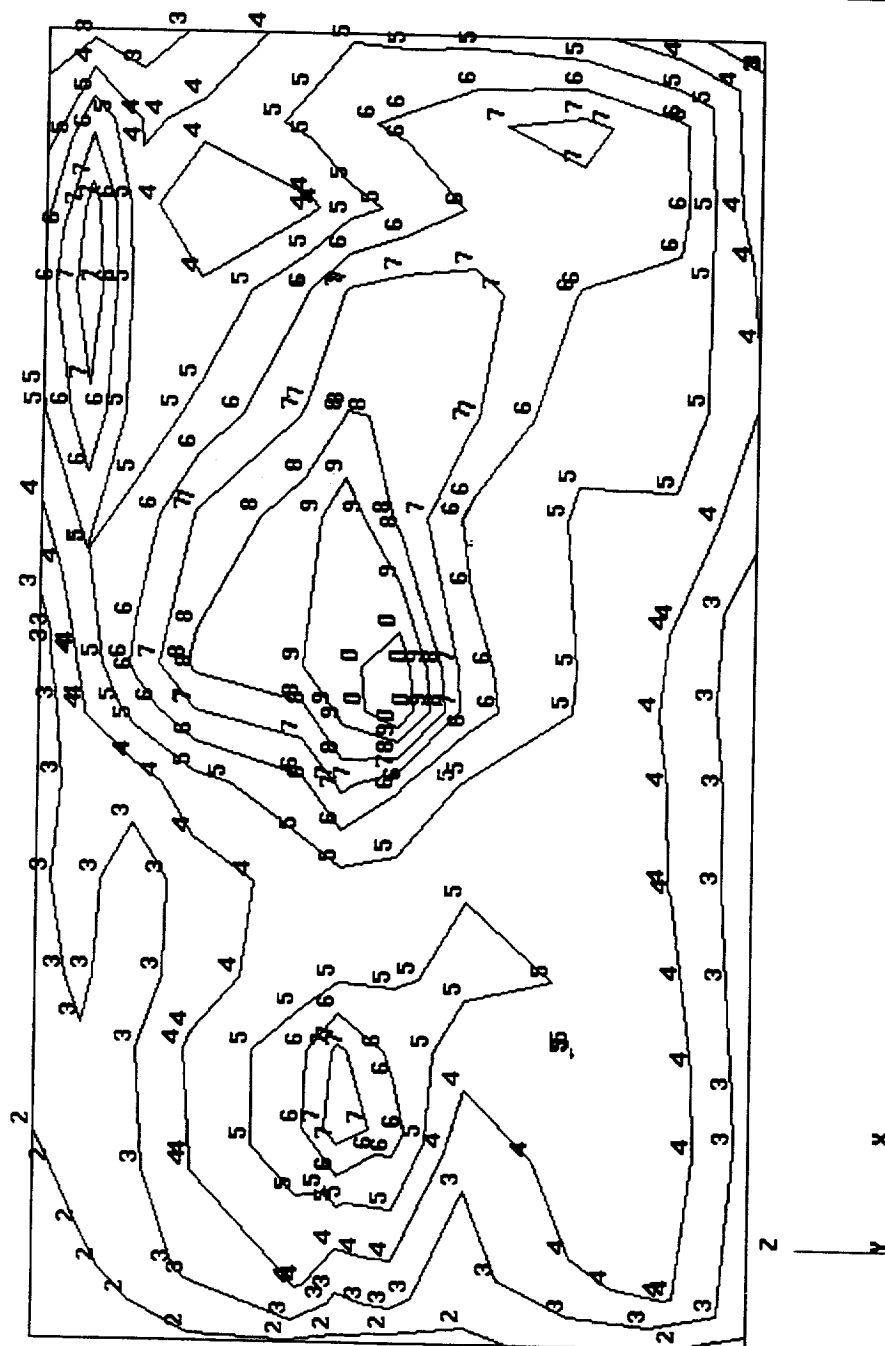
0	0.140E+04
9	0.125E+04
8	0.110E+04
7	0.942E+03
6	0.789E+03
5	0.636E+03
4	0.483E+03
3	0.331E+03
2	0.178E+03
1	0.247E+02

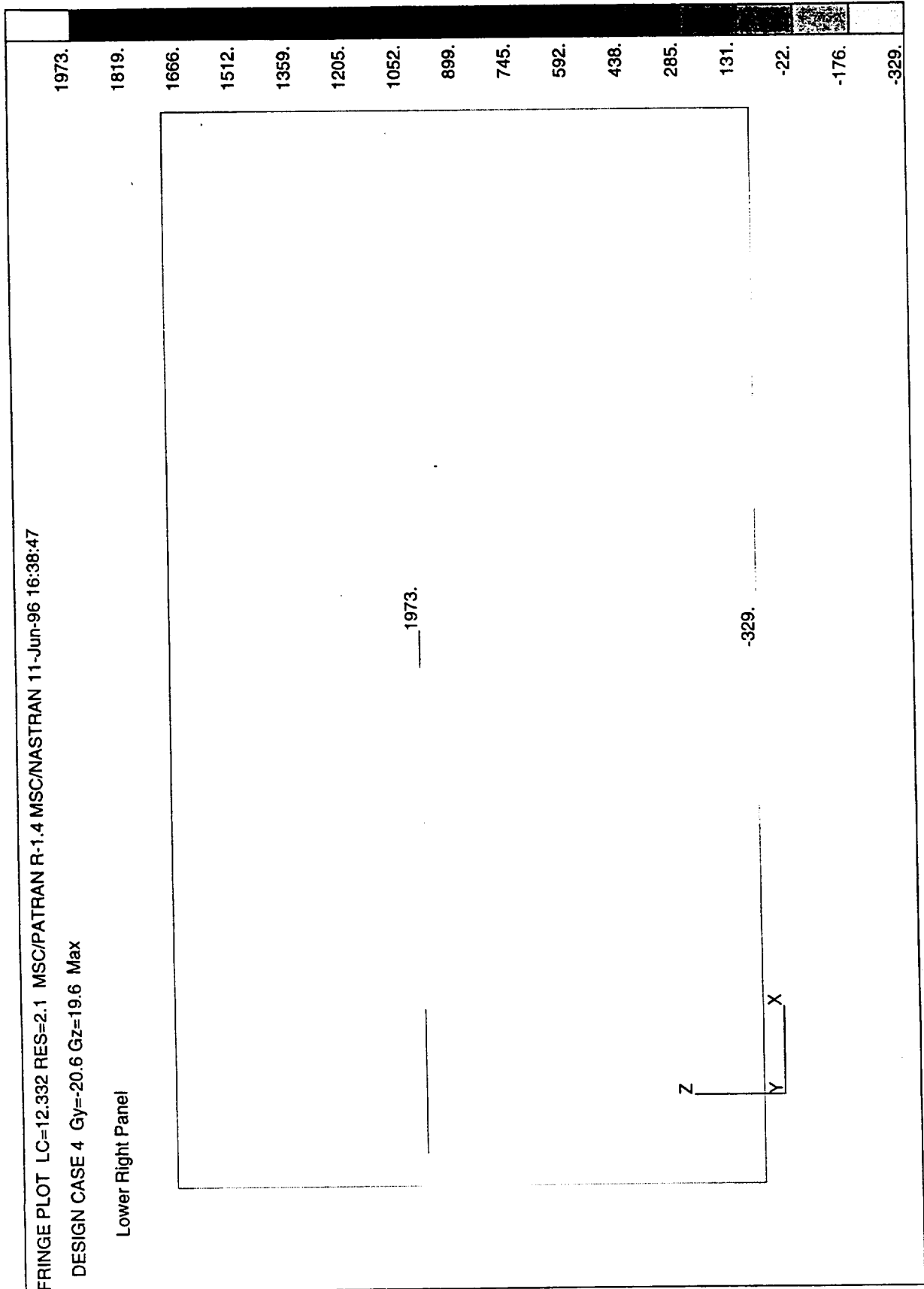
Min = 2.466981E+01
Max = 1.553853E+03
Min ID= 1455
Max ID= 1379
Contour_5:
Stress Tensor

Von Mises
At 21

Design_Qy=-20.6_Gz=-19.6
Static Subcase

METSAT AMSU-A1
Lower Right Panel





Time: 07:57:13
Date: 06/14/96

Contour
Node Scalar1

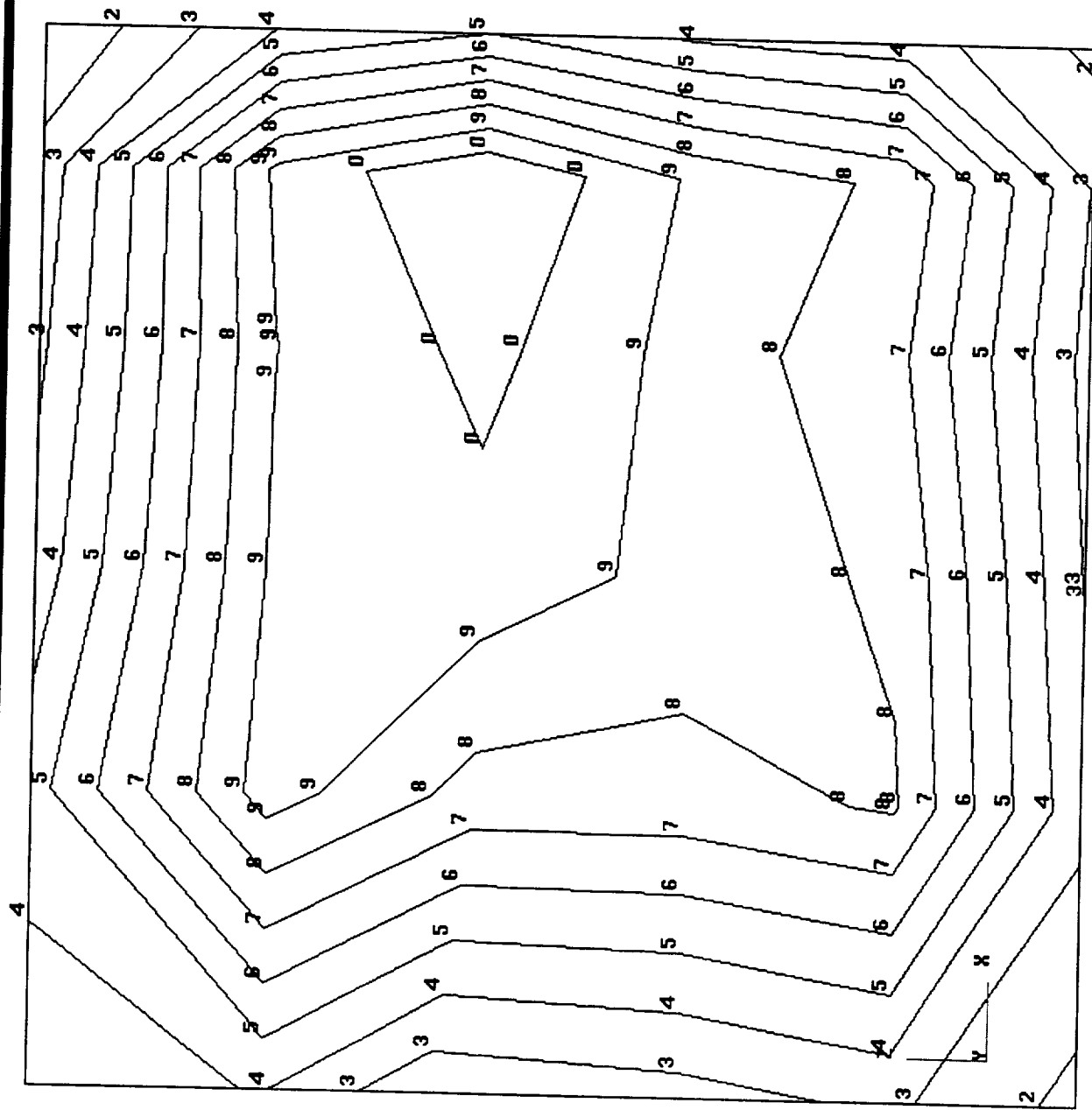
Color Index	
0	0.854E+03
9	0.780E+03
8	0.706E+03
7	0.632E+03
6	0.558E+03
5	0.483E+03
4	0.409E+03
3	0.335E+03
2	0.261E+03
1	0.187E+03

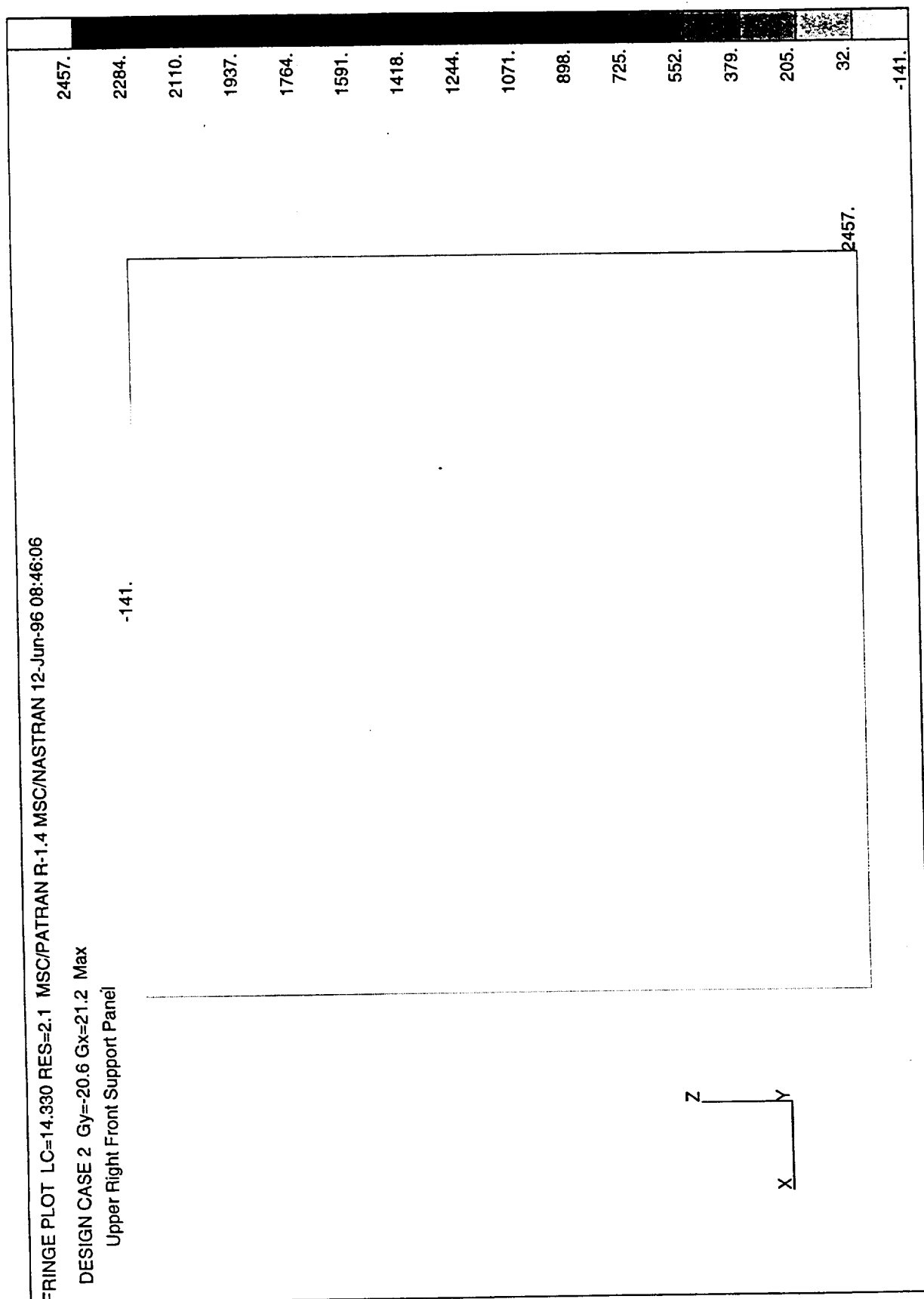
Min = 1.870038E+02
Max = 9.280908E+02
Min ID= 1998
Max ID= 2677
Contour_1:
Stress Tensor

Von Mises
At 21

Design_Gy=-20.6, Gx=-21.2
Static Subcase

METSAT AMSU-A1
Upper Right Front
Support Panel





Time: 08:02:45
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.156E+04
9	0.139E+04
8	0.122E+04
7	0.106E+04
6	0.890E+03
5	0.724E+03
4	0.558E+03
3	0.392E+03
2	0.226E+03
1	0.595E+02

Min = 5.951554E+01

Max = 1.721185E+03

Min ID= 1993

Max ID= 2205

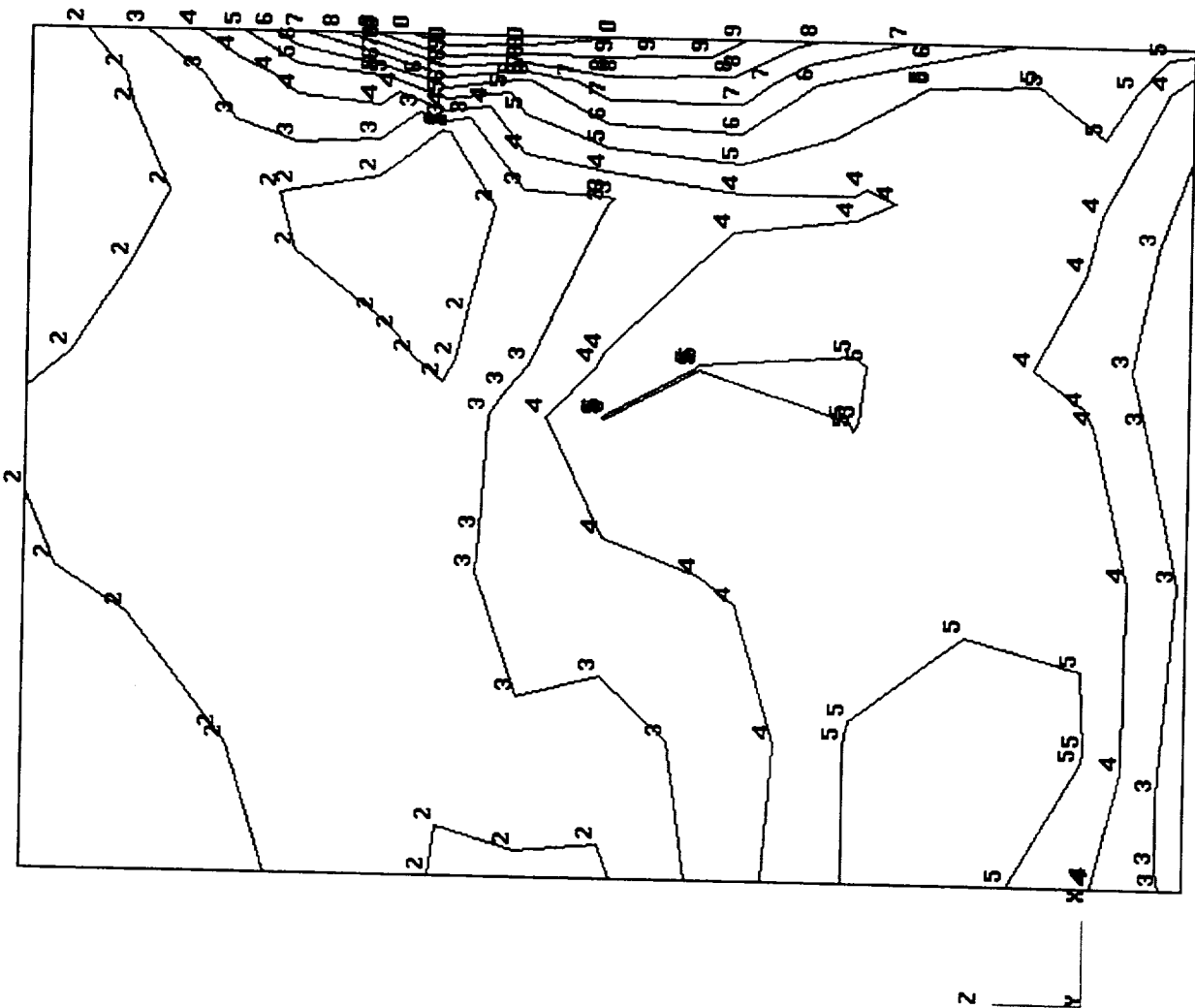
Contour_1:

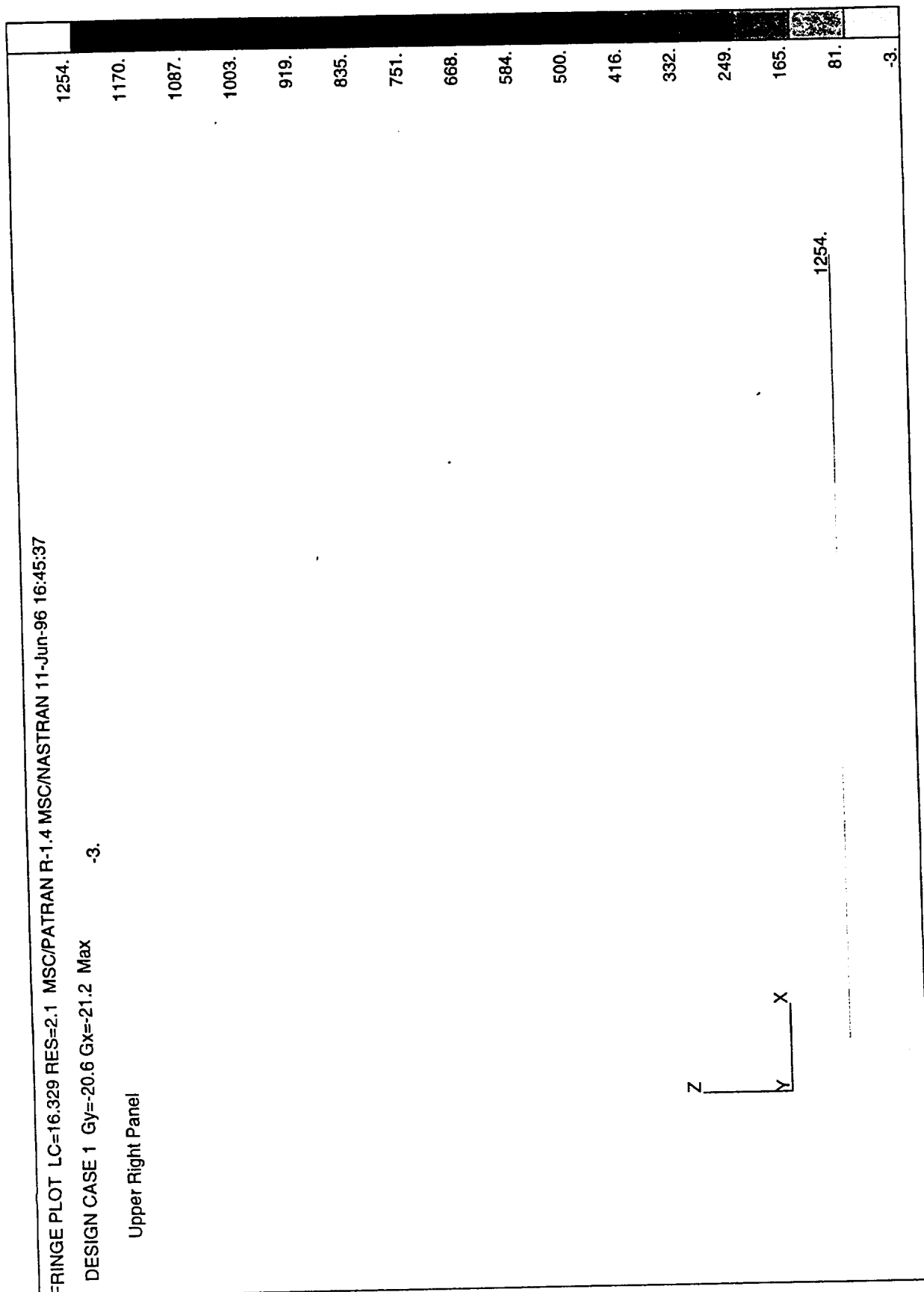
Stress Tensor

Von Mises
At 21

Design_Gy=-20.6_Gx=-21.2
Static Subcase

METSAT AMSU-A1
Upper Right Panel





Time: 08:27:49
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.207E+04
9	0.185E+04
8	0.162E+04
7	0.139E+04
6	0.116E+04
5	0.938E+03
4	0.710E+03
3	0.483E+03
2	0.256E+03
1	0.288E+02

Min = 2.882922E+01

Max = 2.300667E+03

Min ID= 2210

Max ID= 2231

Contour_3:

Stress Tensor

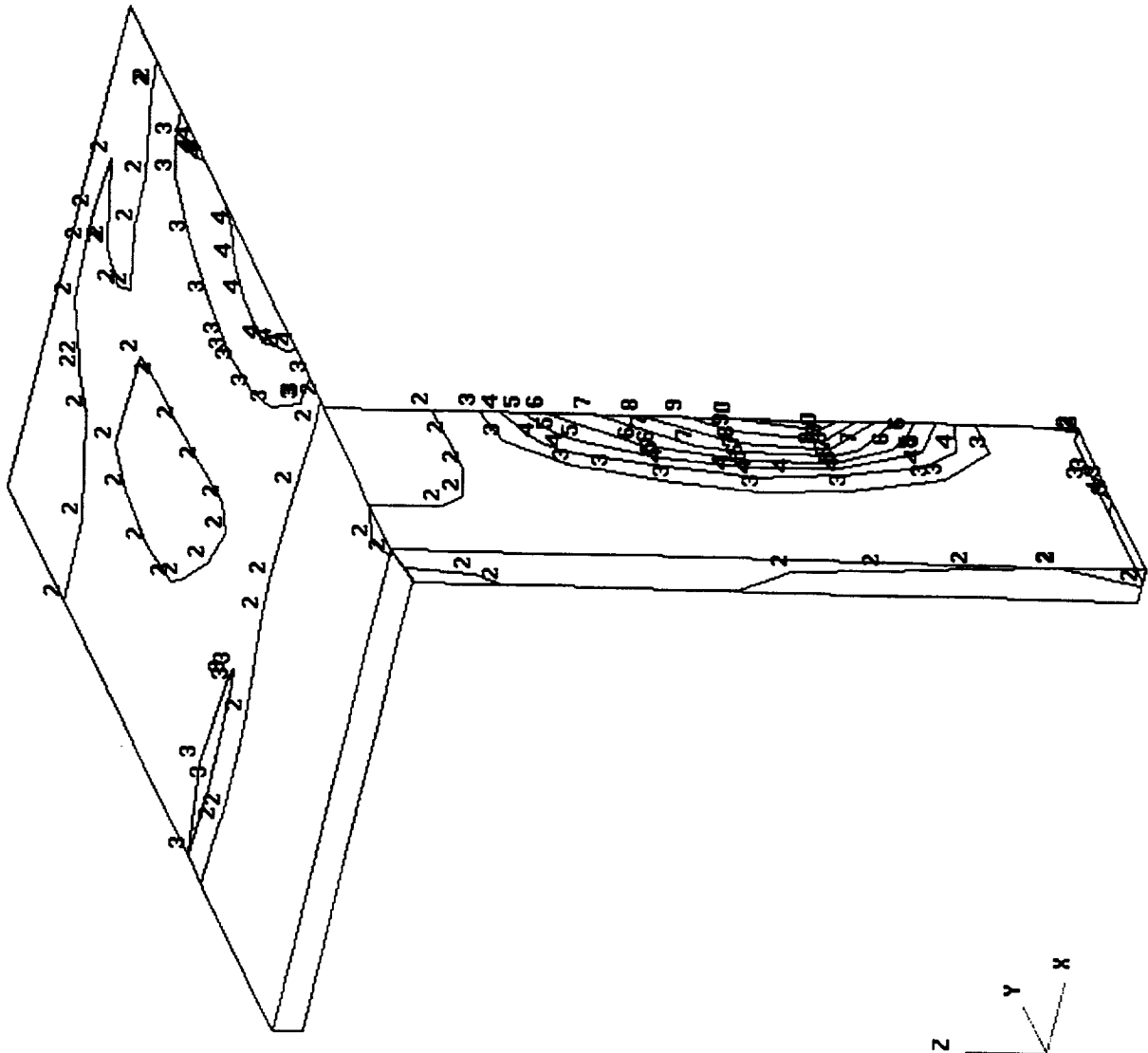
Von Mises

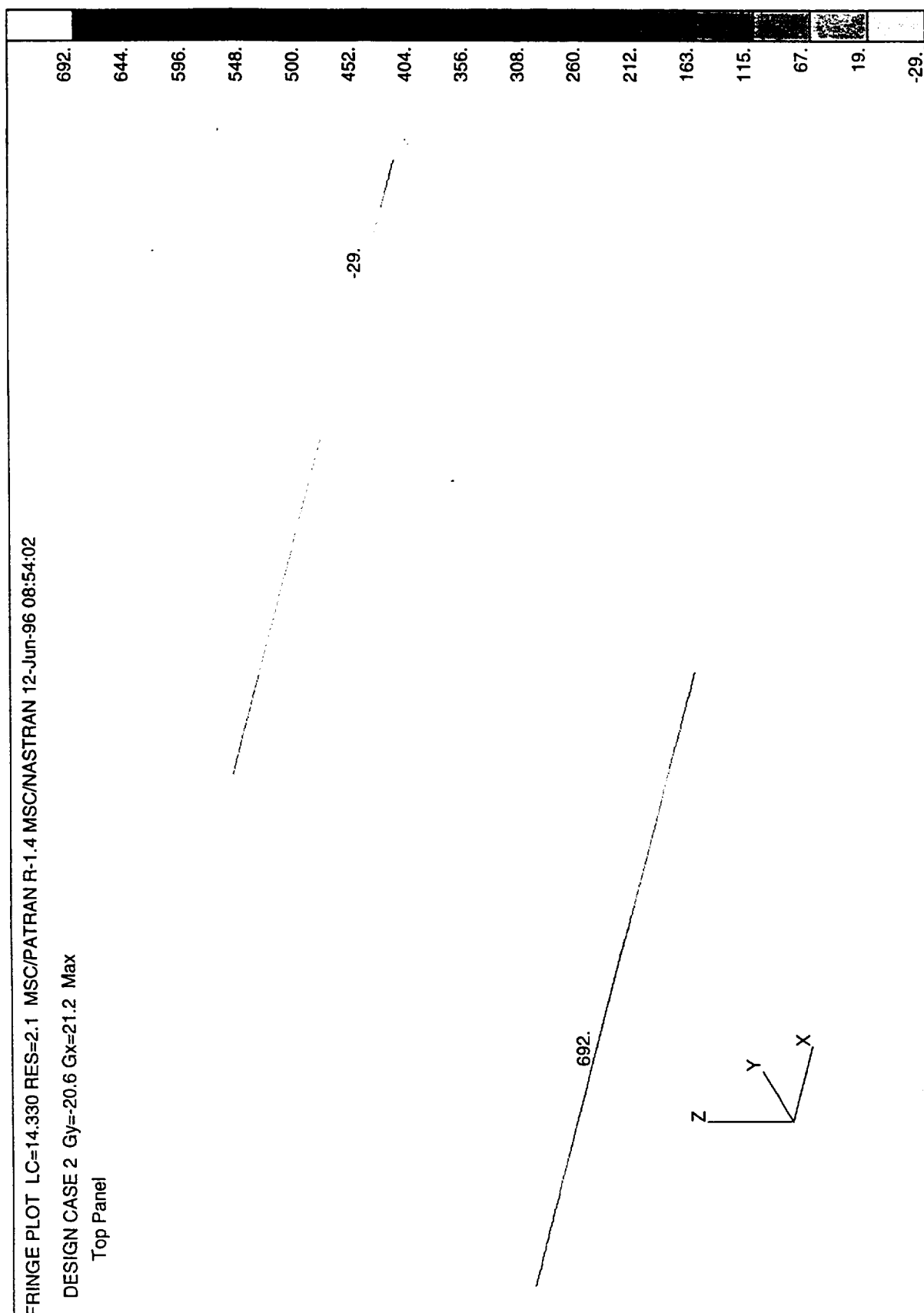
At Z1

Design_Gy=-20.6_0x=21.2
Static Subcase

METSAT AMSU-A1

Top Panel





Time: 08:07:16
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.176E+04
9	0.158E+04
8	0.140E+04
7	0.121E+04
6	0.103E+04
5	0.848E+03
4	0.665E+03
3	0.481E+03
2	0.298E+03
1	0.115E+03

Min = 1.149367E+02

Max = 1.947778E+03

Min ID= 1468

Max ID= 2231

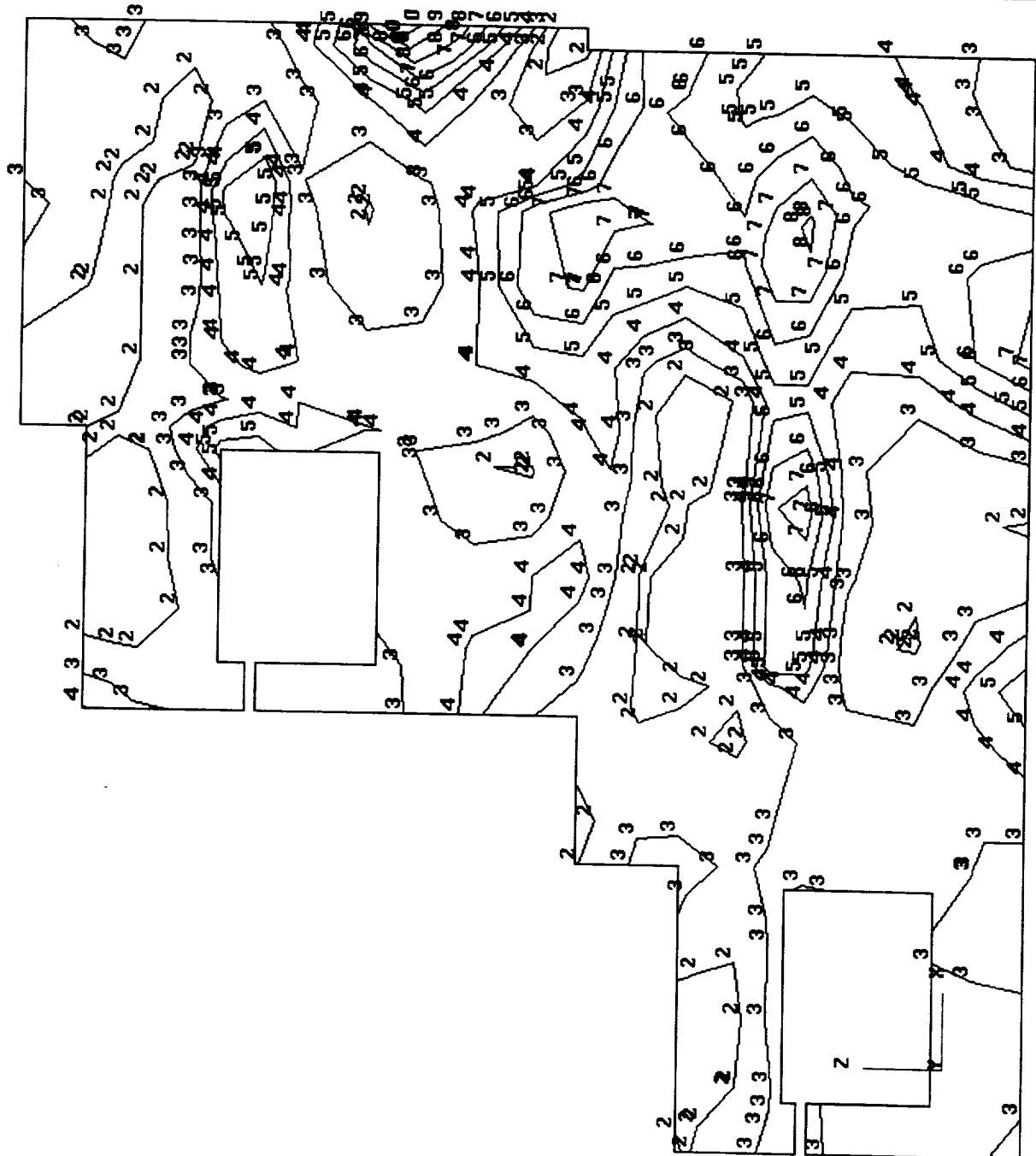
Contour_7:

Stress Tensor

Von Mises
At 21

Design_Gy=-20.6_Gz=19.6
Static Subcase

METSAT AMSU-A1
Left Panel



Time: 11:02:49
Date: 06/12/96

Contour
Node Scalar1

Color Index

0	0.186E+04
9	0.171E+04
8	0.155E+04
7	0.140E+04
6	0.125E+04
5	0.109E+04
4	0.940E+03
3	0.787E+03
2	0.634E+03
1	0.481E+03

Min = 4.810337E+02

Max = 2.011312E+03

Min ID= 2829

Max ID= 2914

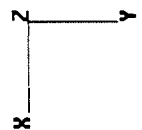
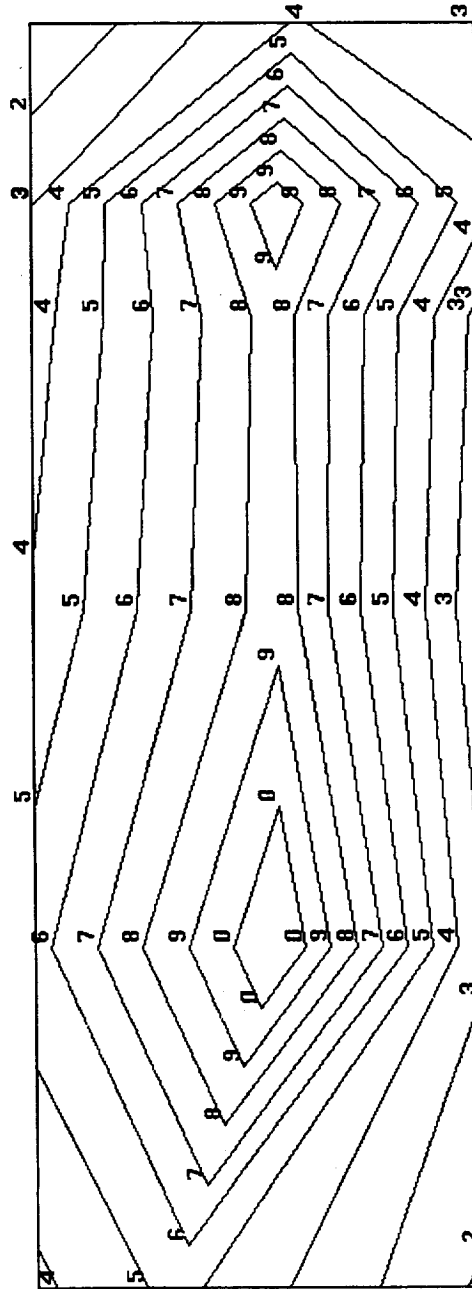
Contour_1:

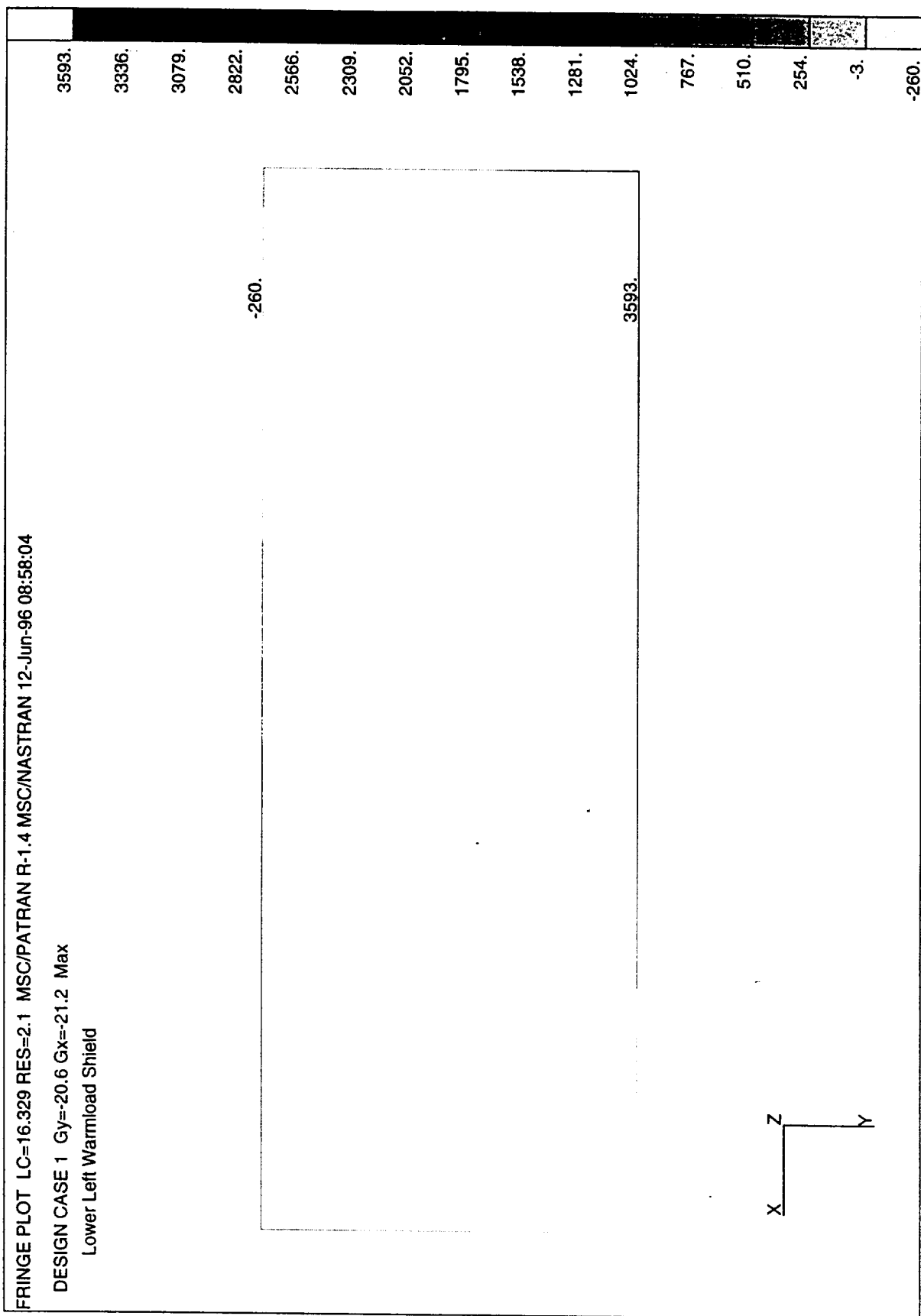
Stress Tensor

Von Mises
At 21

Design_Gy=-20.6_Gx=-21.2
Static Subcase

METSAT AMSU-A1
Lower Left
Warmload Shield





Time: 10:59:16
Date: 06/12/96

Contour
Node Scalar1

Color Index

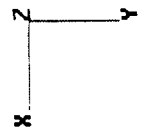
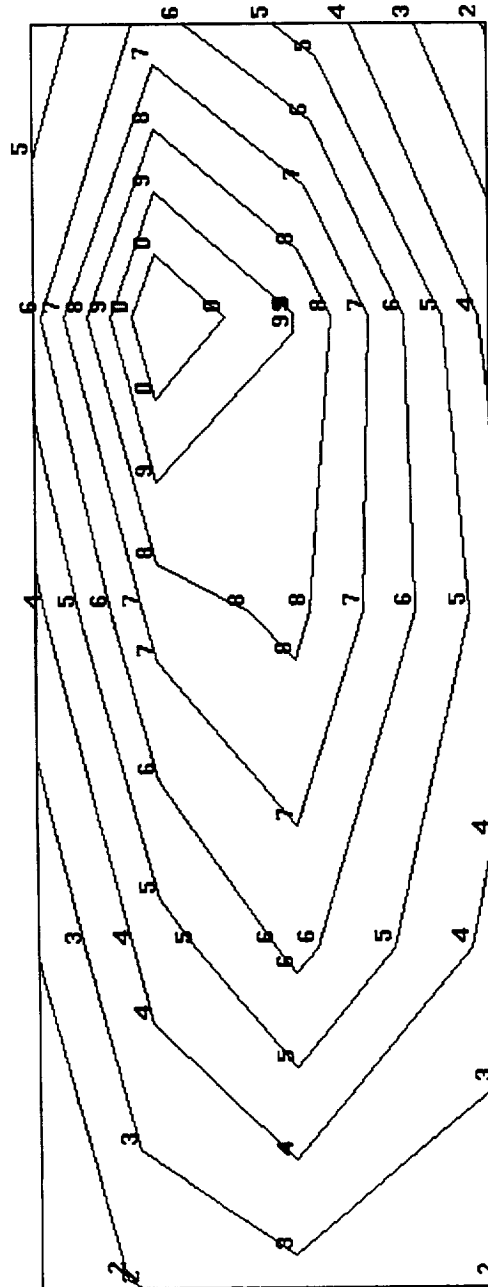
— 0 0.696E+03
— 9 0.625E+03
— 8 0.555E+03
— 7 0.484E+03
— 6 0.414E+03
— 5 0.344E+03
— 4 0.273E+03
— 3 0.203E+03
— 2 0.132E+03
— 1 0.616E+02

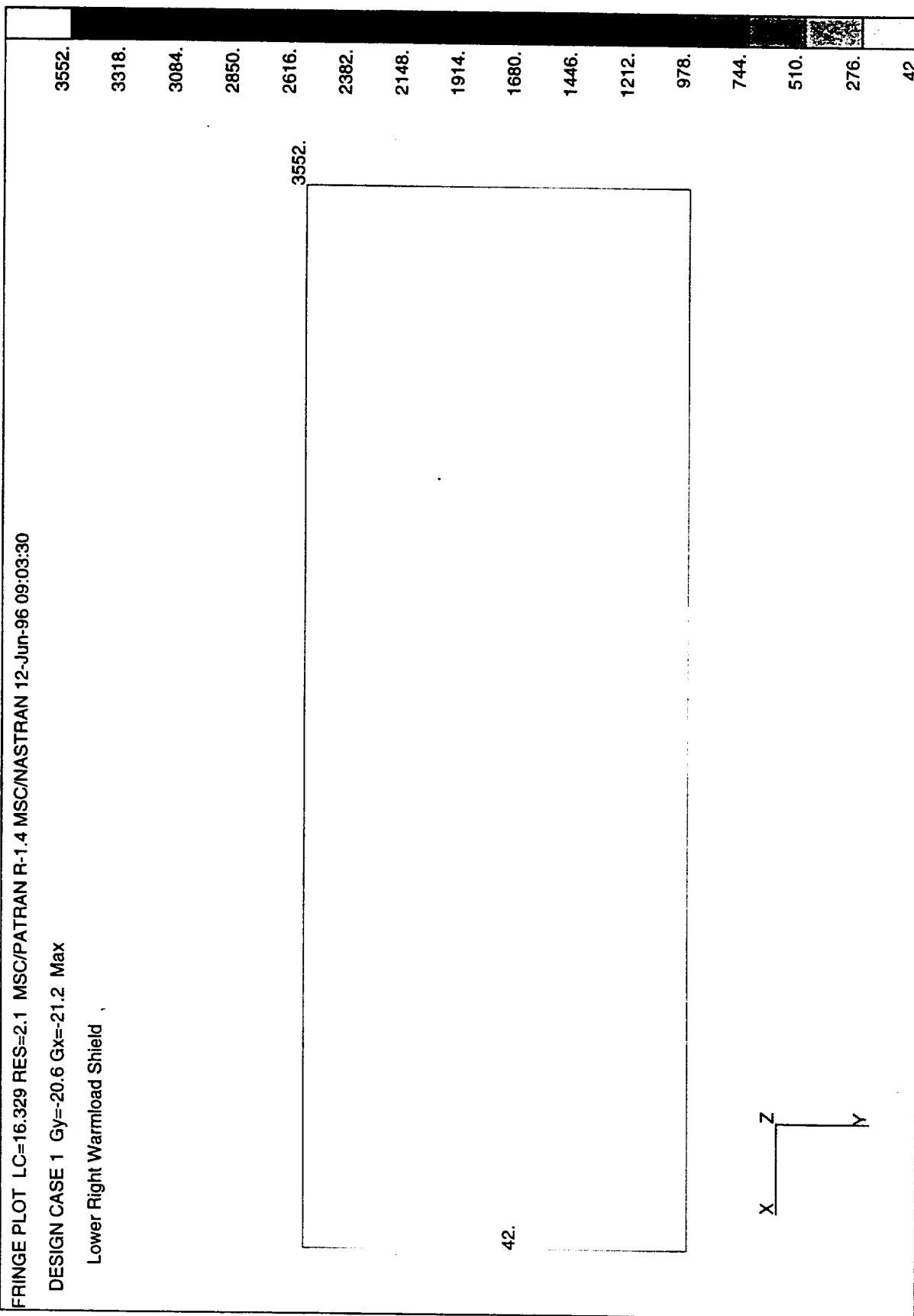
Min = 6.165003E+01
Max = 7.663430E+02
Min ID= 1938
Max ID= 2903
Contour_2:
Stress Tensor

Von Mises
At 22

Design_Gy=-20.6_Gx=-21.2
Static Subcase

METSAT AMSU-A1
Lower Right
Warmload Shield





Time: 11:13:00
Date: 06/12/96

Contour
Node Scalar1

Color Index

0	0.135E+04
9	0.123E+04
8	0.111E+04
7	0.996E+03
6	0.878E+03
5	0.761E+03
4	0.643E+03
3	0.526E+03
2	0.408E+03
1	0.291E+03

Min = 2.907404E+02

Max = 1.466241E+03

Min ID= 2763

Max ID= 2929

Contour_4:

Stress Tensor

Von Mises

At 22

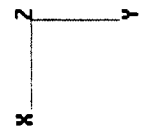
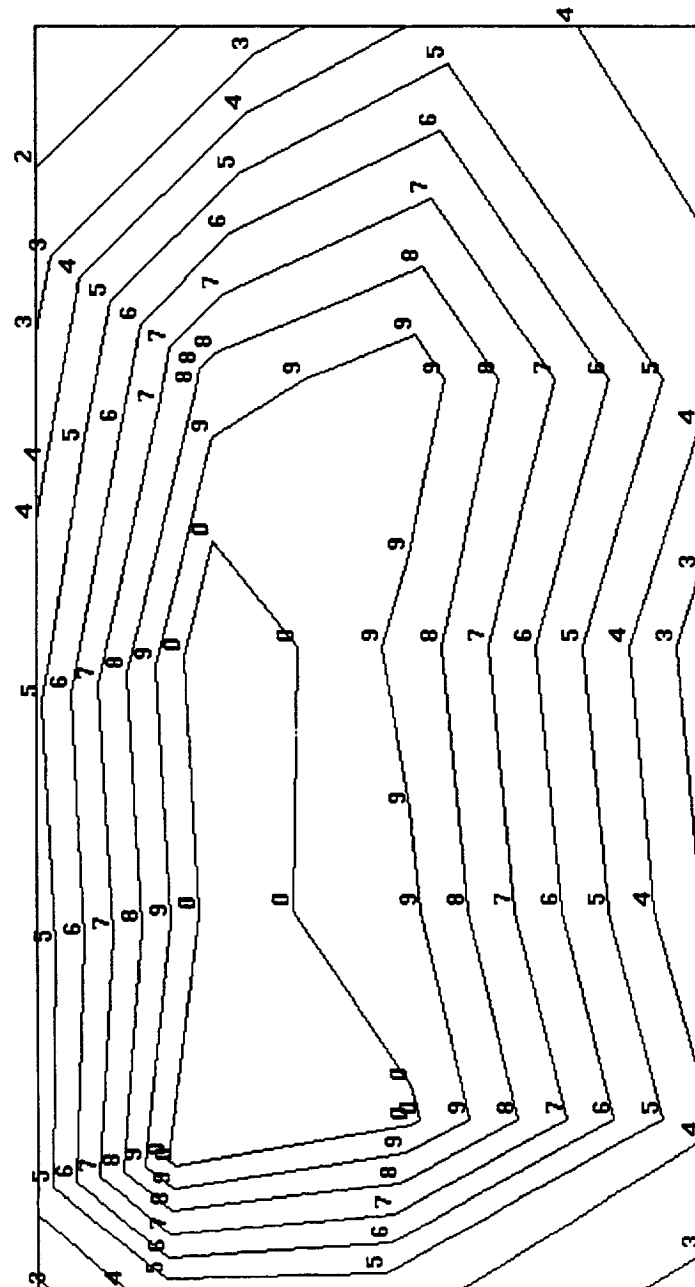
Design_Gy=-20.6_Gx=21.2

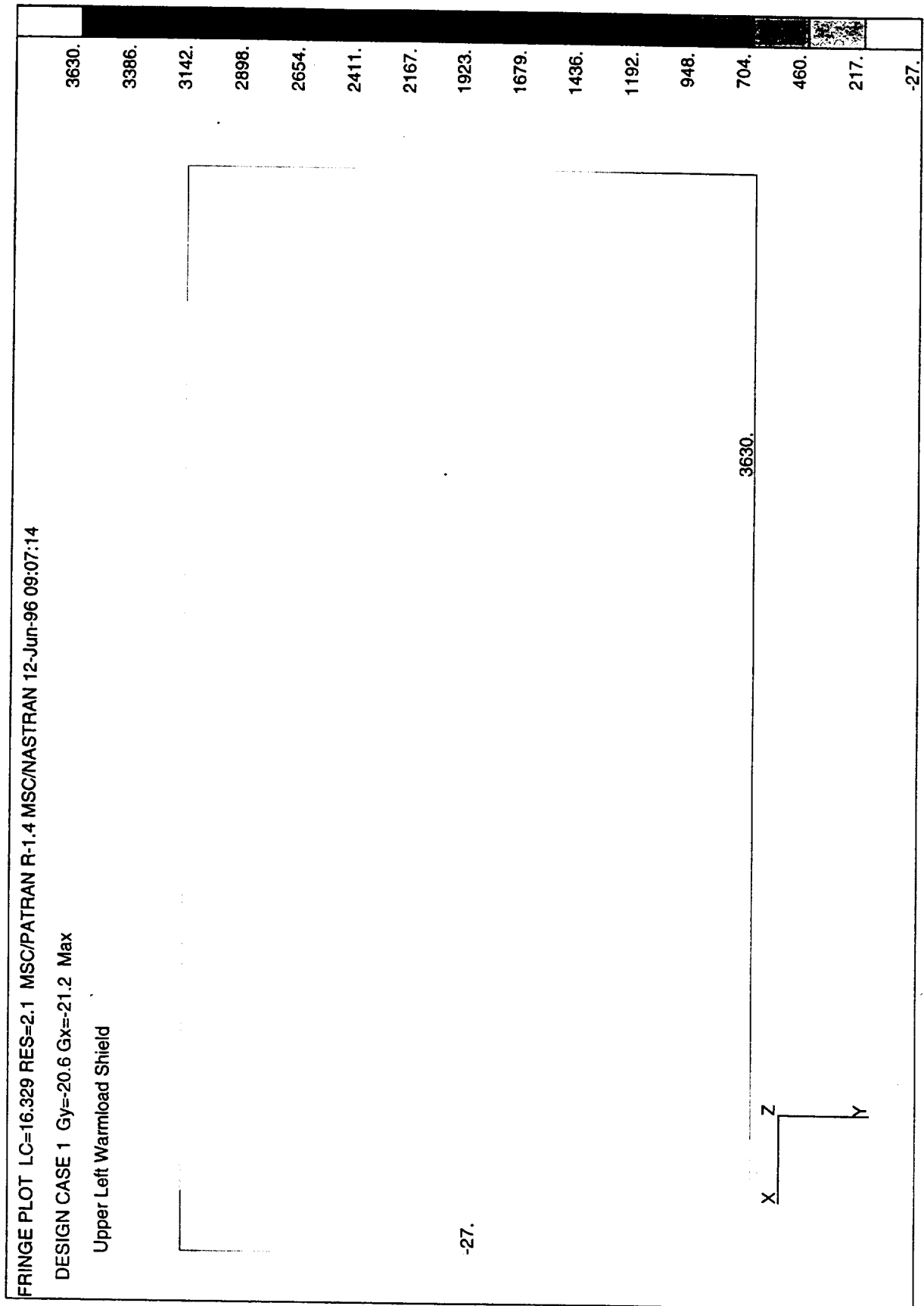
Static Subcase

METSAT AMSU-A1

Upper Left

Warmload Shield





Time: 11:11:17
Date: 06/12/96

Contour
Node Scalar1

Color Index

0	0.363E+03
9	0.328E+03
8	0.294E+03
7	0.260E+03
6	0.225E+03
5	0.191E+03
4	0.156E+03
3	0.122E+03
2	0.873E+02
1	0.529E+02

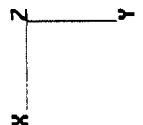
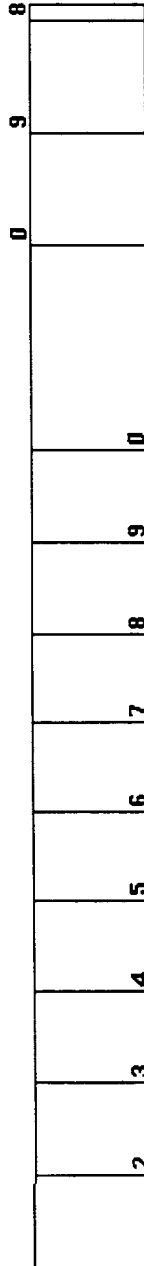
Min = 5.285617E+01
Max = 3.972899E+02
Min ID= 1998
Max ID= 2690

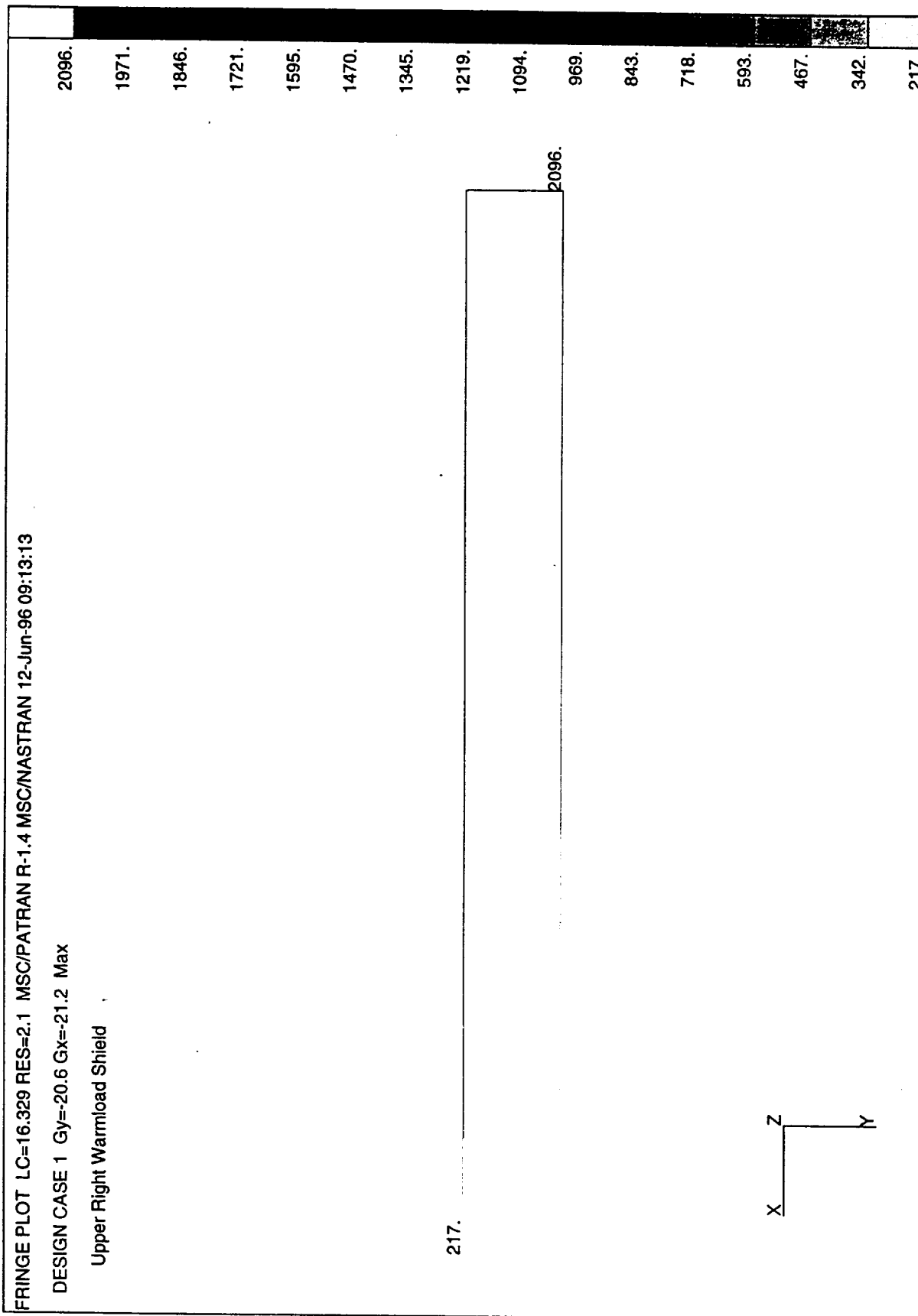
Contour_2:
Stress Tensor

Von Mises
At 22

Design_Gy=-20.6_Gx=-21.2
Static Subcase

METSAT AMSU-A1
Upper Right
Warmload Shield





Time: 08:33:31
Date: 06/14/96

Contour
Node Scalar1

Color Index

—	0	0.280E+04
—	9	0.249E+04
—	8	0.218E+04
—	7	0.187E+04
—	6	0.156E+04
—	5	0.125E+04
—	4	0.938E+03
—	3	0.628E+03
—	2	0.318E+03
—	1	0.818E+01

Min = 8.178007E+00

Max = 3.106662E+03

Min ID= 4831

Max ID= 3733

Contour_7:

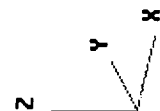
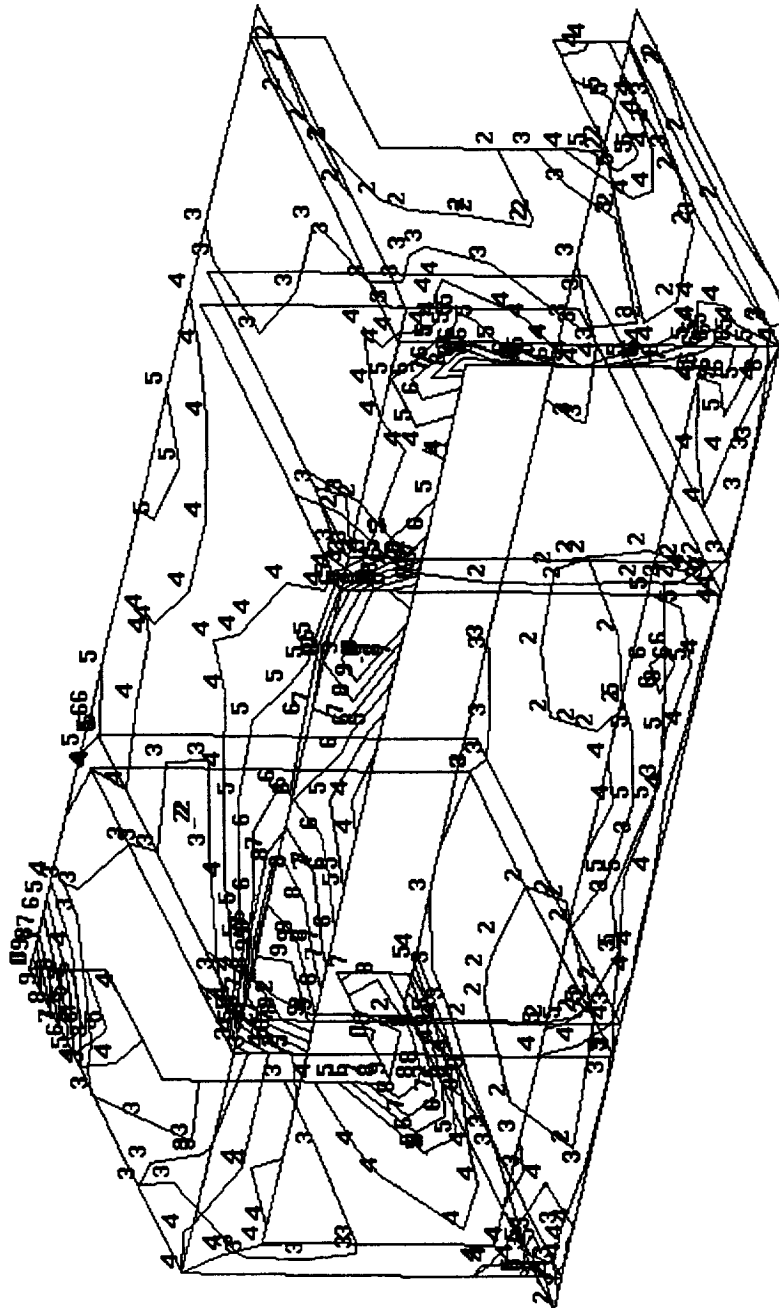
Stress Tensor

Von Mises
At Z1

Design_Gy=-20.6_Gz=19.6
Static Subcase

METSAT AMSU-A1

Lower Card Cage



Time: 08:56:38
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.896E+03
9	0.798E+03
8	0.699E+03
7	0.601E+03
6	0.503E+03
5	0.405E+03
4	0.306E+03
3	0.208E+03
2	0.110E+03
1	0.116E+02

Min = 1.164704E+01

Max = 9.940120E+02

Min ID= 4728

Max ID= 4702

Contour_7:

Stress Tensor

Von Mises

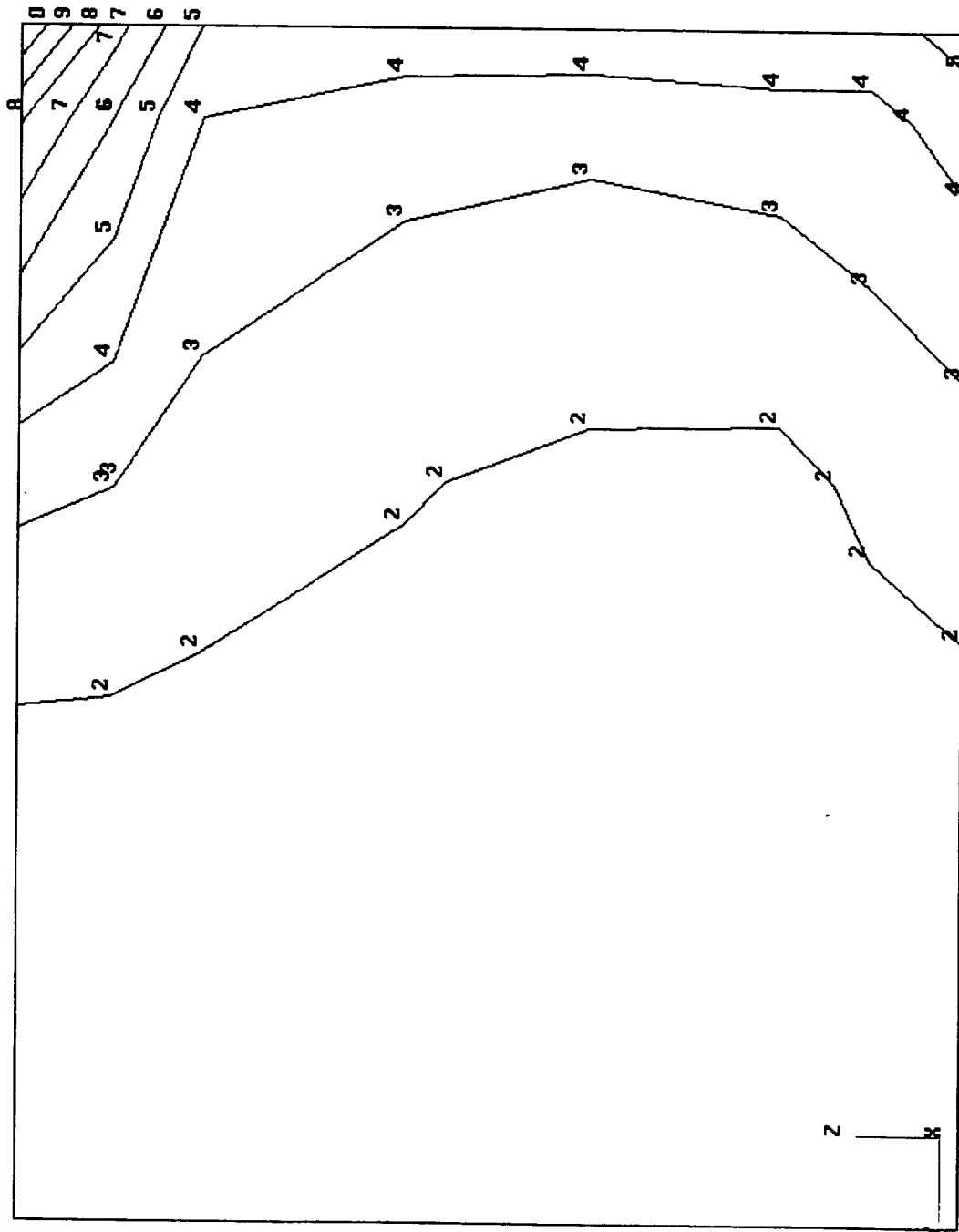
At Z1

Design_Gy=-20.6_Gz=19.6

Static Subcase

METSAT AMSU-A1

Lower Card



Time: 09:09:01
Date: 06/14/96

Contour
Node Scalar1

Color Index

—	0	0.289E+04
—	9	0.257E+04
—	8	0.225E+04
—	7	0.193E+04
—	6	0.161E+04
—	5	0.129E+04
—	4	0.966E+03
—	3	0.645E+03
—	2	0.324E+03
—	1	0.269E+01

Min = 2.694405E+00

Max = 3.213089E+03

Min ID= 11834

Max ID= 3382

Contour_8:

Stress Tensor

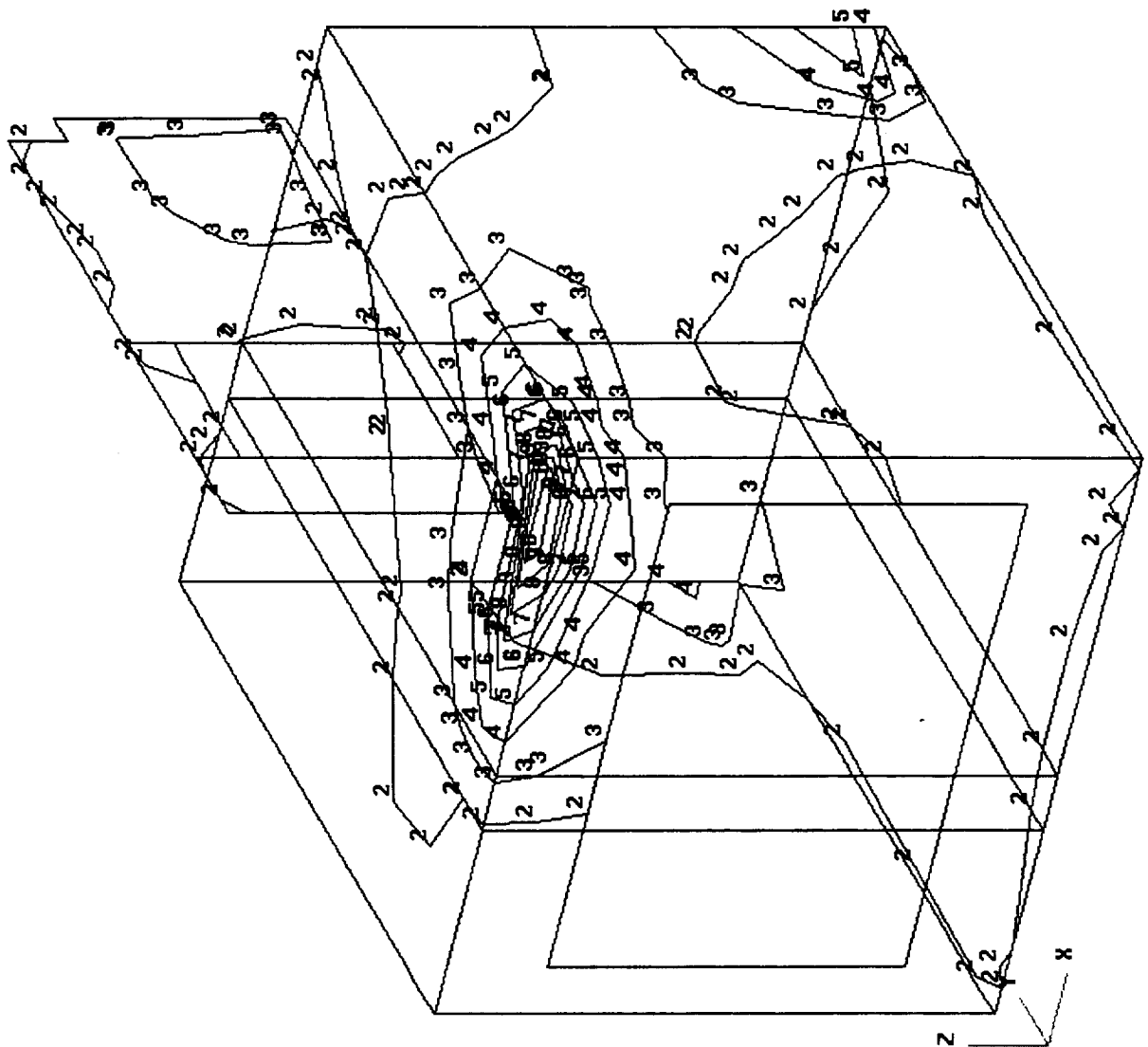
Von Mises

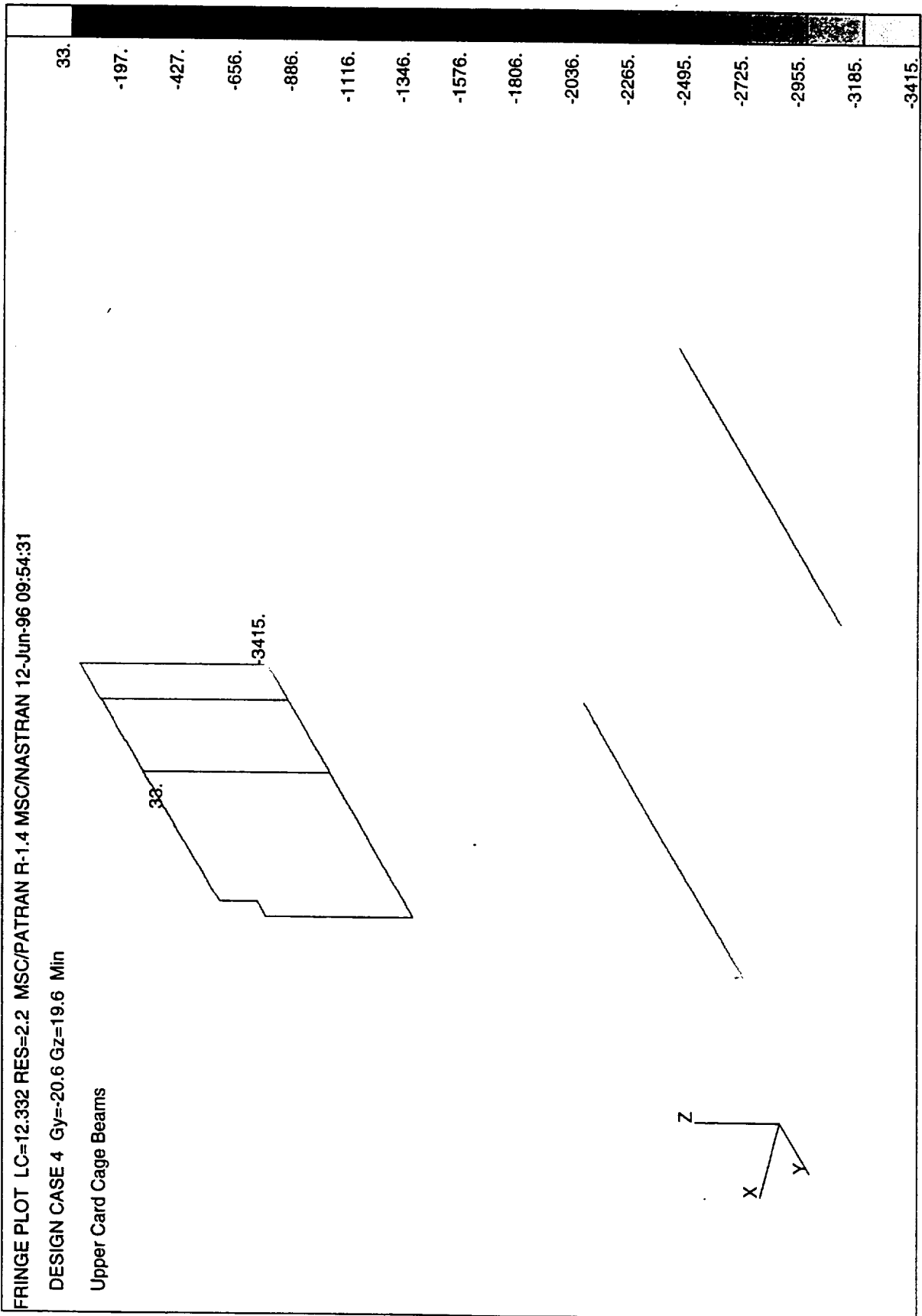
At Z2

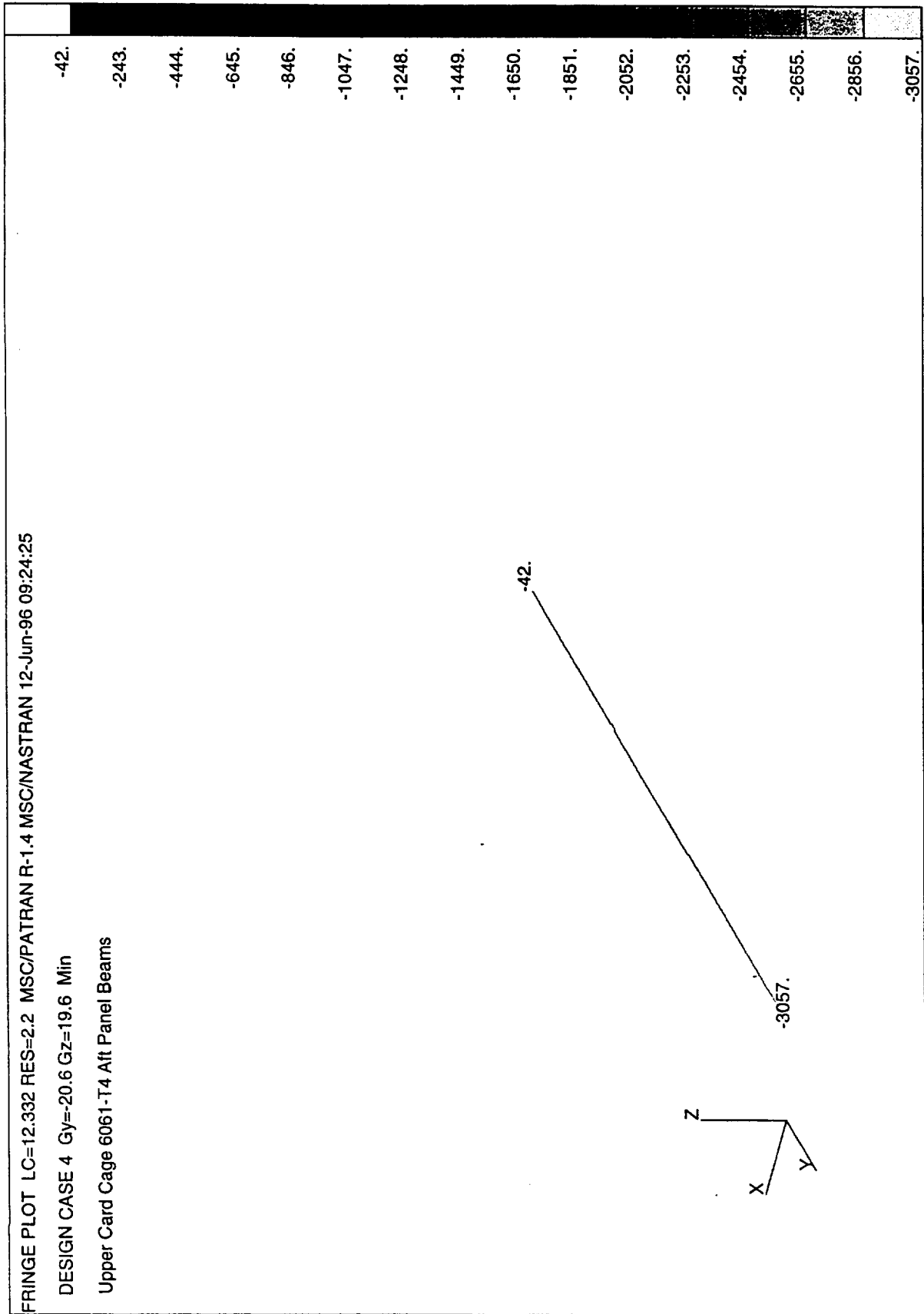
Design_6y=-20.6_gz=19.6

Static Subcase

METSAT AMSU-A1
Upper Card Cage







Time: 09:17:21
Date: 06/14/96

Contour
Node Scalar1

Color Index

—	0	0.144E+04
—	9	0.130E+04
—	8	0.115E+04
—	7	0.101E+04
—	6	0.861E+03
—	5	0.716E+03
—	4	0.572E+03
—	3	0.427E+03
—	2	0.282E+03
—	1	0.137E+03

Min = 1.370855E+02

Max = 1.585423E+03

Min ID= 1474

Max ID= 3556

Contour_8:

Stress Tensor

Von Mises

At Z2

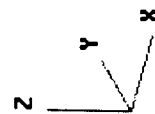
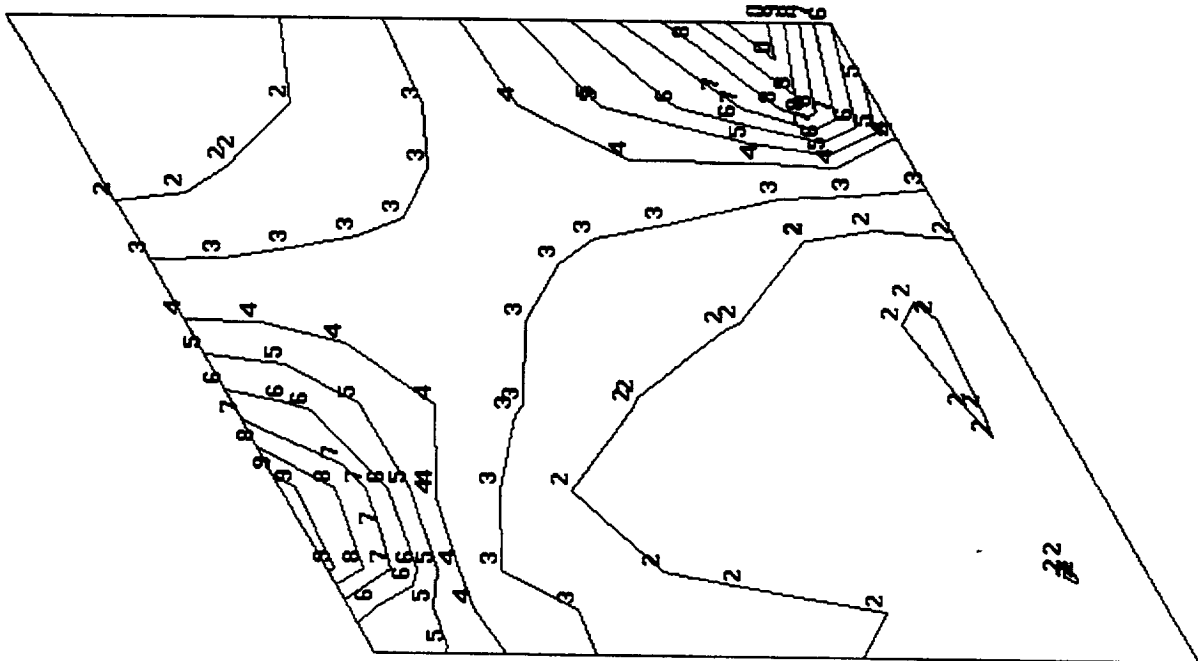
Design_Gy=-20.6_Gz=19.6

Static Subcase

METSAT AMSU-A1

Upper Card Cage

6061-T4 Panel



Time: 09:00:34
Date: 06/14/96

Contour
Node Scalar1

Color Index

— 0 0.488E+03
— 9 0.446E+03
— 8 0.405E+03
— 7 0.363E+03
— 6 0.322E+03
— 5 0.281E+03
— 4 0.239E+03
— 3 0.198E+03
— 2 0.156E+03
— 1 0.115E+03

Min = 1.151174E+02

Max = 5.289375E+02

Min ID= 11292

Max ID= 11334

Contour_2:

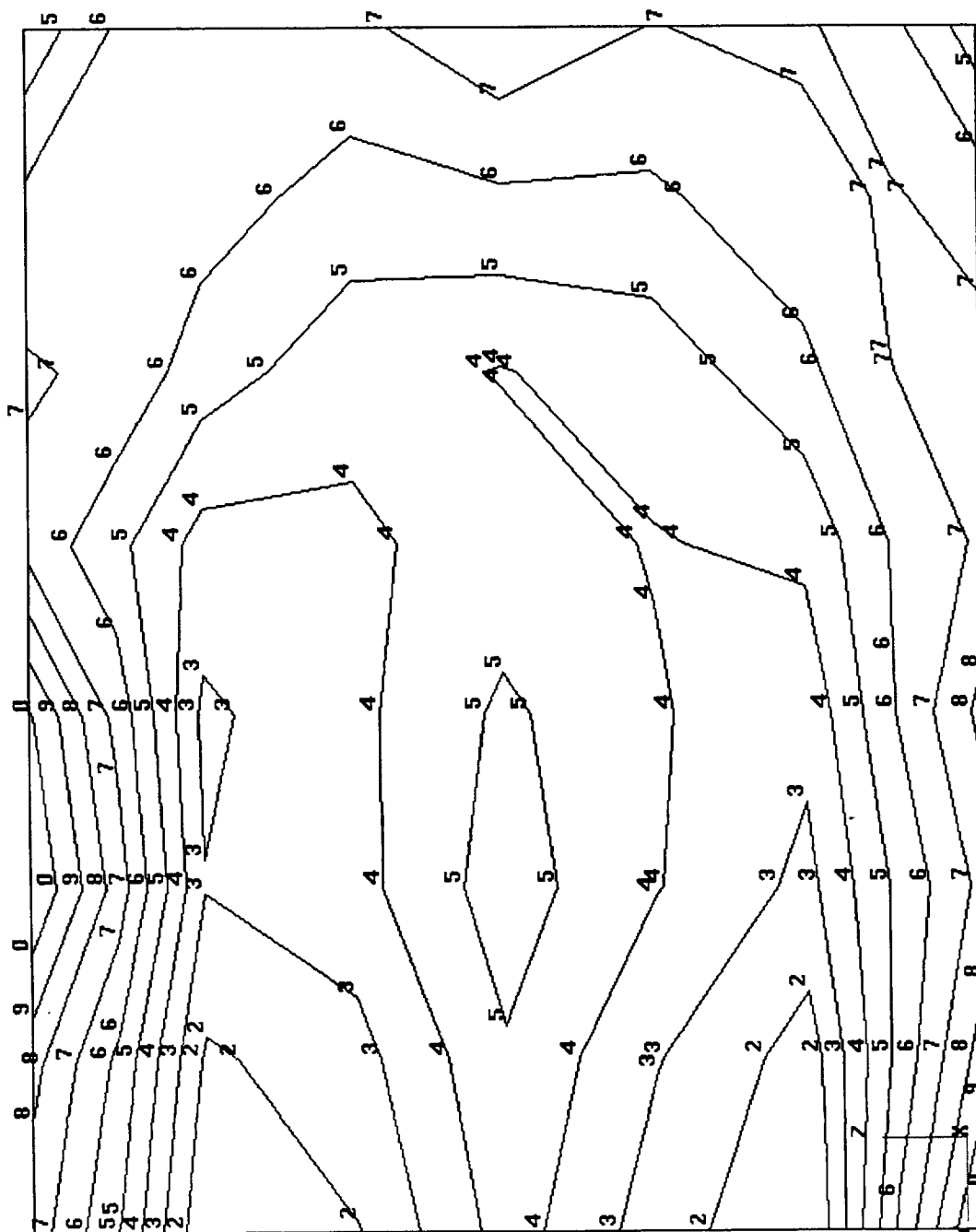
Stress Tensor

Von Mises
At 22

Design_Gy=-20.6_Gx=-21.2
Static Subcase

METSAT AMSU-A1

Upper Card



Time: 09:21:29
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.179E+04
9	0.160E+04
8	0.140E+04
7	0.121E+04
6	0.102E+04
5	0.823E+03
4	0.630E+03
3	0.437E+03
2	0.243E+03
1	0.498E+02

Min = 4.976259E+01

Max = 1.984076E+03

Min ID= 5171

Max ID= 3060

Contour_6:

Stress Tensor

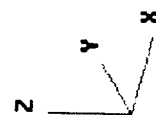
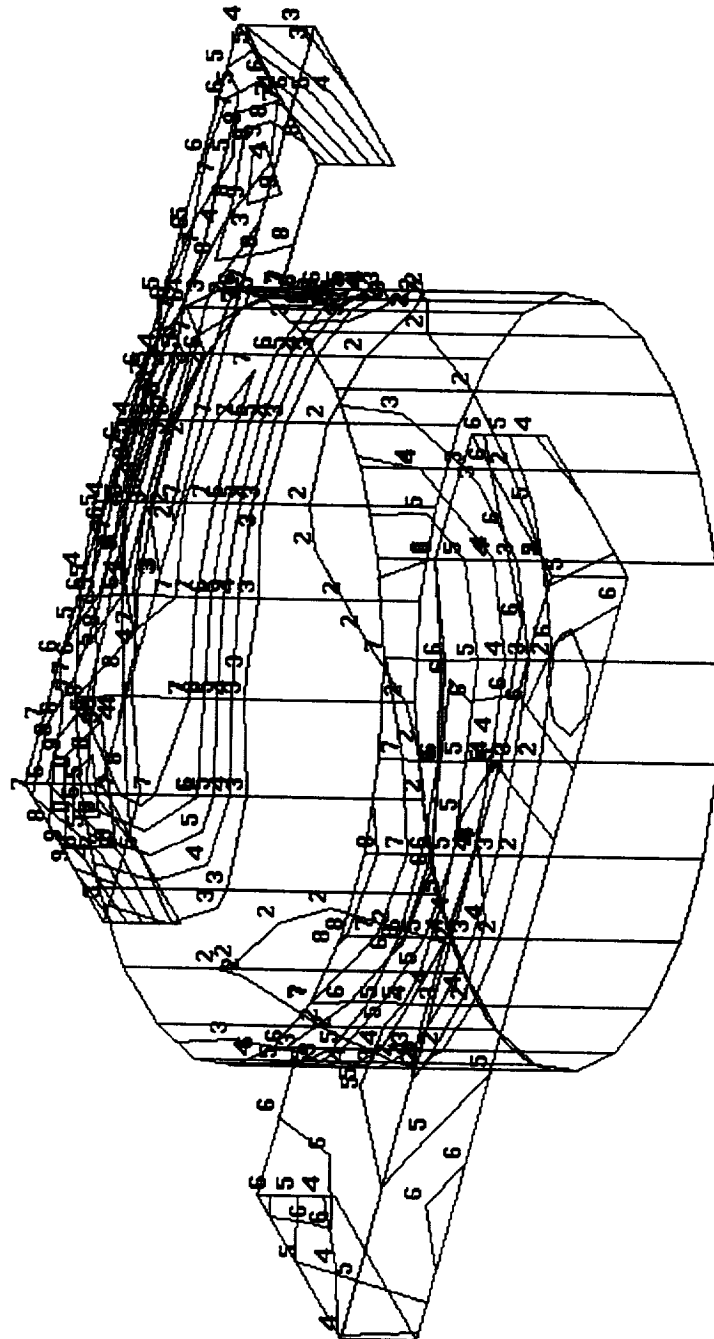
Von Mises

At 22

Design_Gy=-20.6_Gz=-19.6

Static Subcase

METSAT AMSU-A1
Lower Warmload



Time: 09:24:59
Date: 06/14/96

Contour
Node Scalar1

Color Index

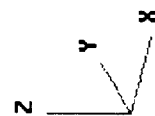
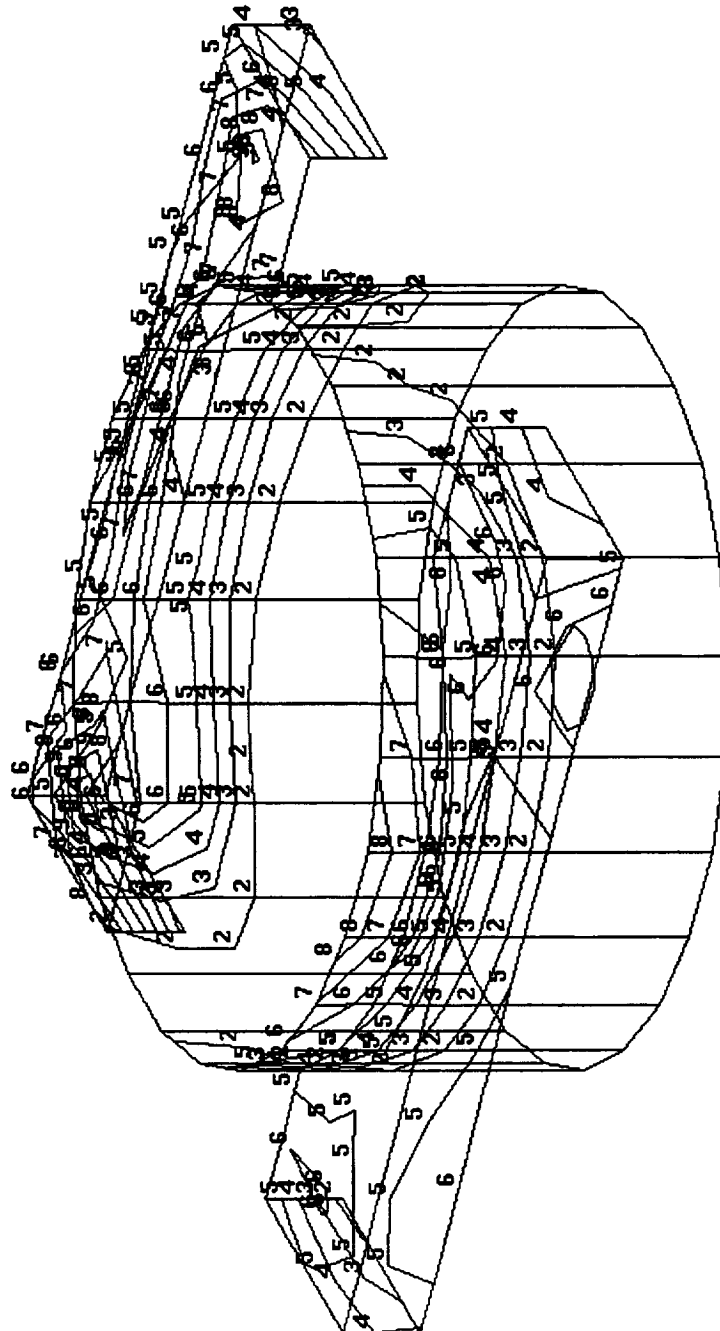
0	0.172E+04
9	0.153E+04
8	0.134E+04
7	0.115E+04
6	0.967E+03
5	0.780E+03
4	0.592E+03
3	0.405E+03
2	0.217E+03
1	0.300E+02

Min = 3.002866E+01
Max = 1.904372E+03
Min ID= 5405
Max ID= 2961
Contour_6:
Stress Tensor

Von Mises
At 22

Design_Gy=-20.6_Gz=-19.6
Static Subcase

METSAT AMSU-A1
Upper Warmload



Time: 09:28:14
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.837E+03
9	0.745E+03
8	0.654E+03
7	0.562E+03
6	0.470E+03
5	0.378E+03
4	0.286E+03
3	0.195E+03
2	0.103E+03
1	0.112E+02

Min = 1.116715E+01

Max = 9.287907E+02

Min ID= 4004

Max ID= 4029

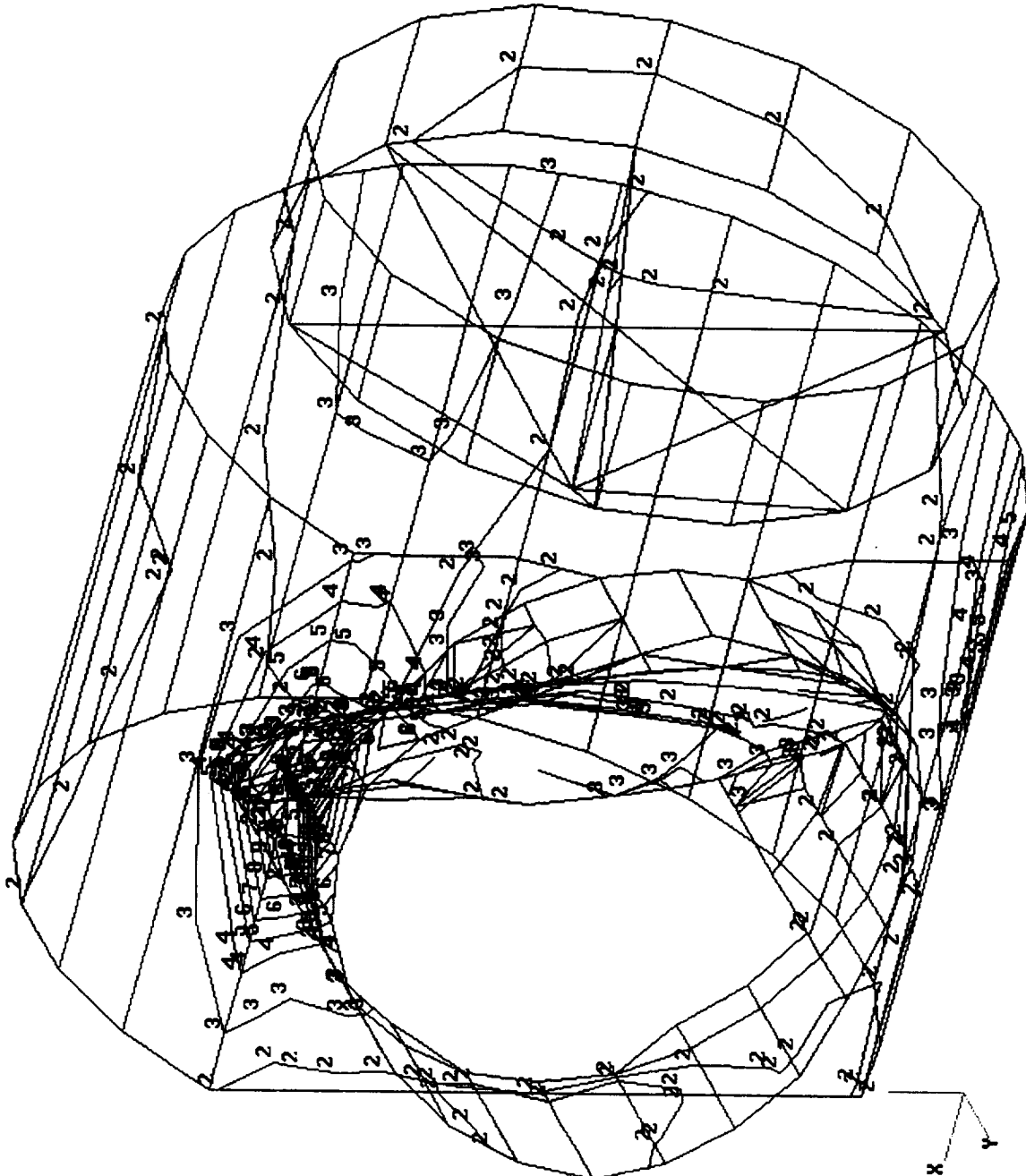
Contour_5:

Stress Tensor

Von Mises
At 21

Design_Gy=-20.6_Gz=-19.6
Static Subcase

METSAT AMSU-A1
Lower Reflector



Time: 09:34:49
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.289E+04
9	0.257E+04
8	0.225E+04
7	0.193E+04
6	0.161E+04
5	0.129E+04
4	0.975E+03
3	0.656E+03
2	0.336E+03
1	0.170E+02

Min = 1.698457E+01

Max = 3.210751E+03

Min ID= 5646

Max ID= 6338

Contour_4:

Stress Tensor

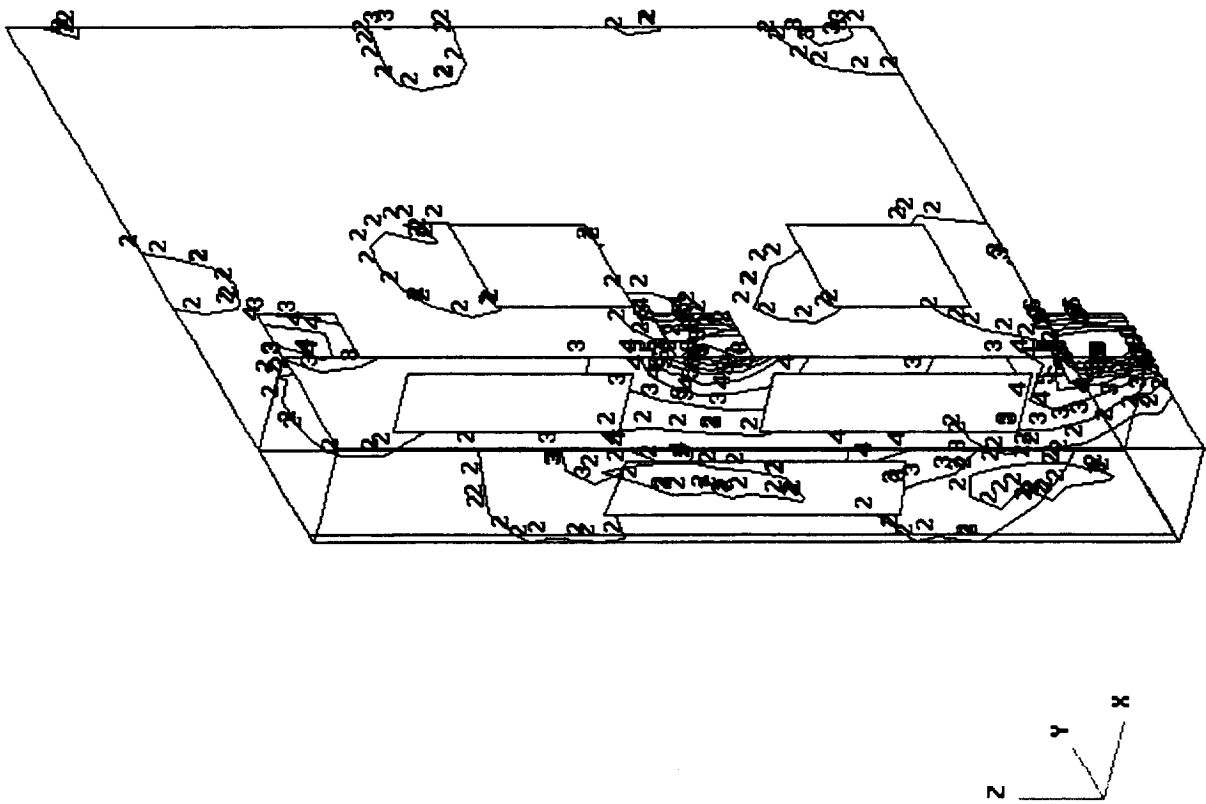
Von Mises

At Z2

Design_Gy=-20.6_Gx=21.2

Static Subcase

METSAT AMSU-A1
Power/Control Mon



Time: 08:17:09
Date: 06/14/96

Contour
Node Scalar1

Color Index

—	0	0.803E+03
—	9	0.715E+03
—	8	0.627E+03
—	7	0.540E+03
—	6	0.452E+03
—	5	0.364E+03
—	4	0.276E+03
—	3	0.188E+03
—	2	0.100E+03
—	1	0.121E+02

Min = 1.209474E+01

Max = 8.911633E+02

Min ID= 5601

Max ID= 6307

Contour_2:

Stress Tensor

Von Mises

At Z2

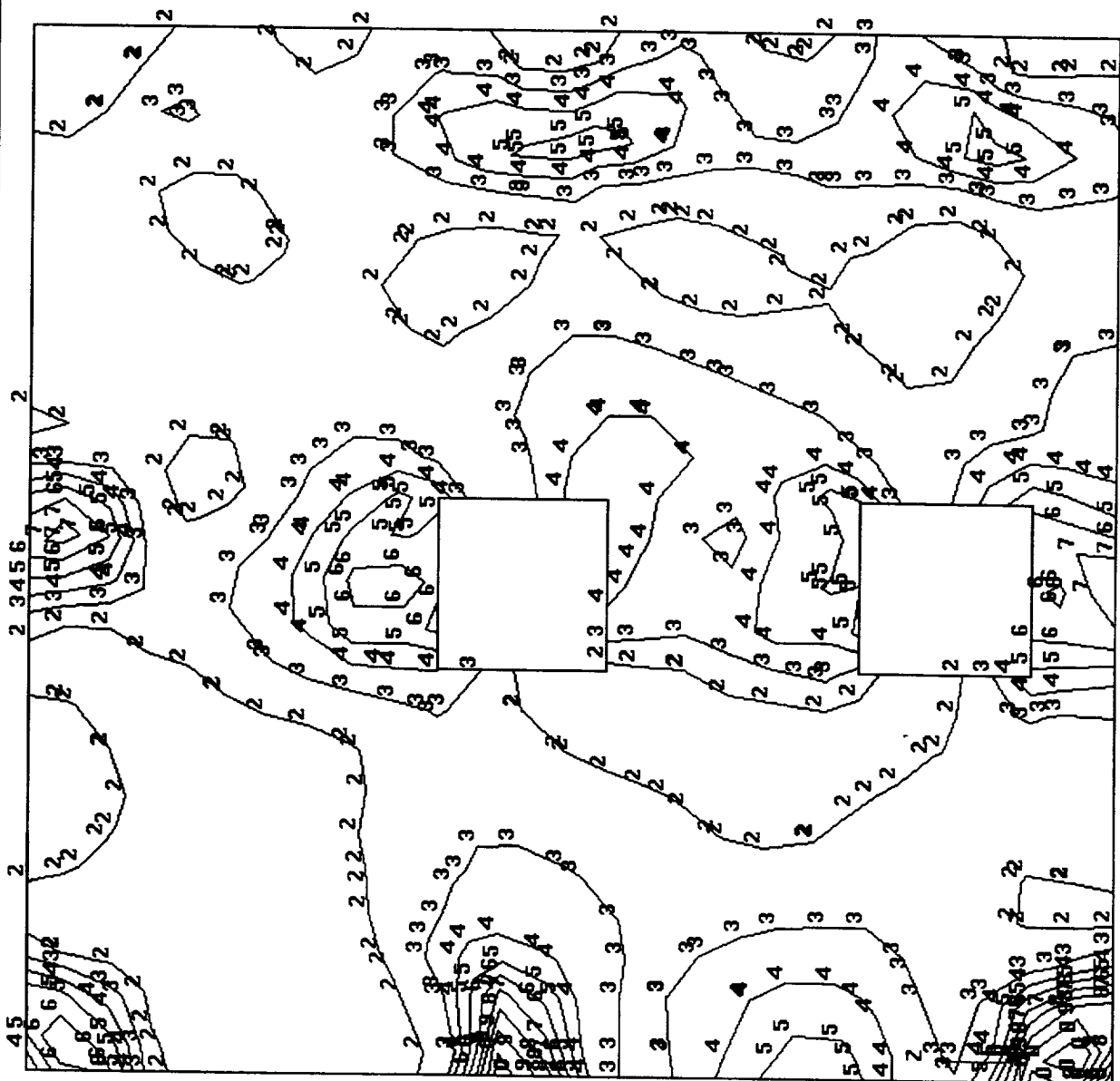
Design_Gy=-20.6_Gx=-21.2

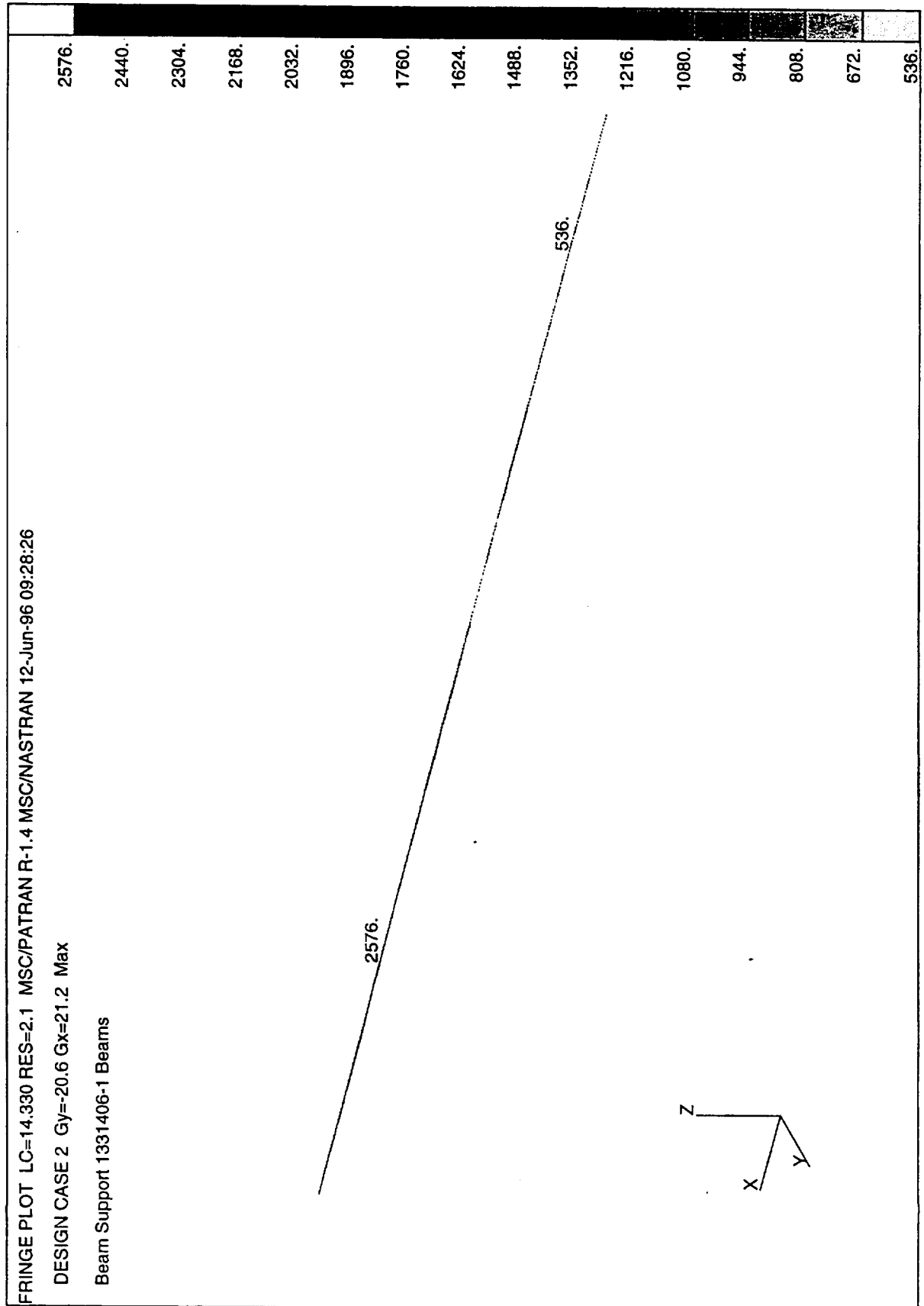
Static Subcase

METSAT AMSU-A1

Power/Control

Monitor PWB





Time: 09:39:02
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.237E+04
9	0.211E+04
8	0.185E+04
7	0.159E+04
6	0.134E+04
5	0.108E+04
4	0.823E+03
3	0.566E+03
2	0.309E+03
1	0.522E+02

Min = 5.216672E+01

Max = 2.622498E+03

Min ID= 10973

Max ID= 10899

Contour_4:

Stress Tensor

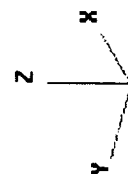
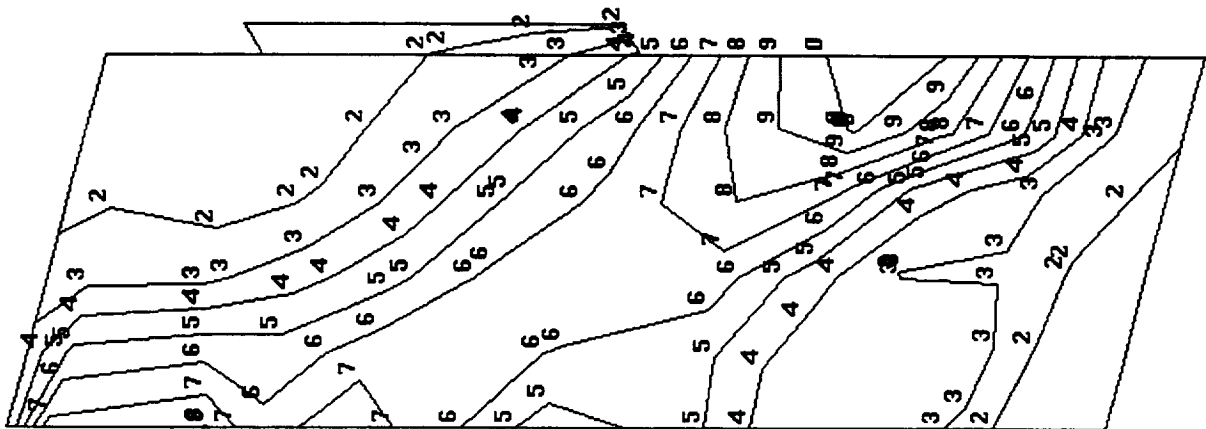
Von Mises

At Z2

Design_0y=-20.6_0x=21.2

Static Subcase

METSAT AMSU-A1
Radiator Panel



Time: 09:43:28
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.966E+04
9	0.858E+04
8	0.751E+04
7	0.644E+04
6	0.537E+04
5	0.429E+04
4	0.322E+04
3	0.215E+04
2	0.108E+04
1	0.649E+01

Min = 6.487691E+00

Max = 1.072748E+04

Min ID= 7936

Max ID= 9642

Contour_8:

Stress Tensor

Von Mises

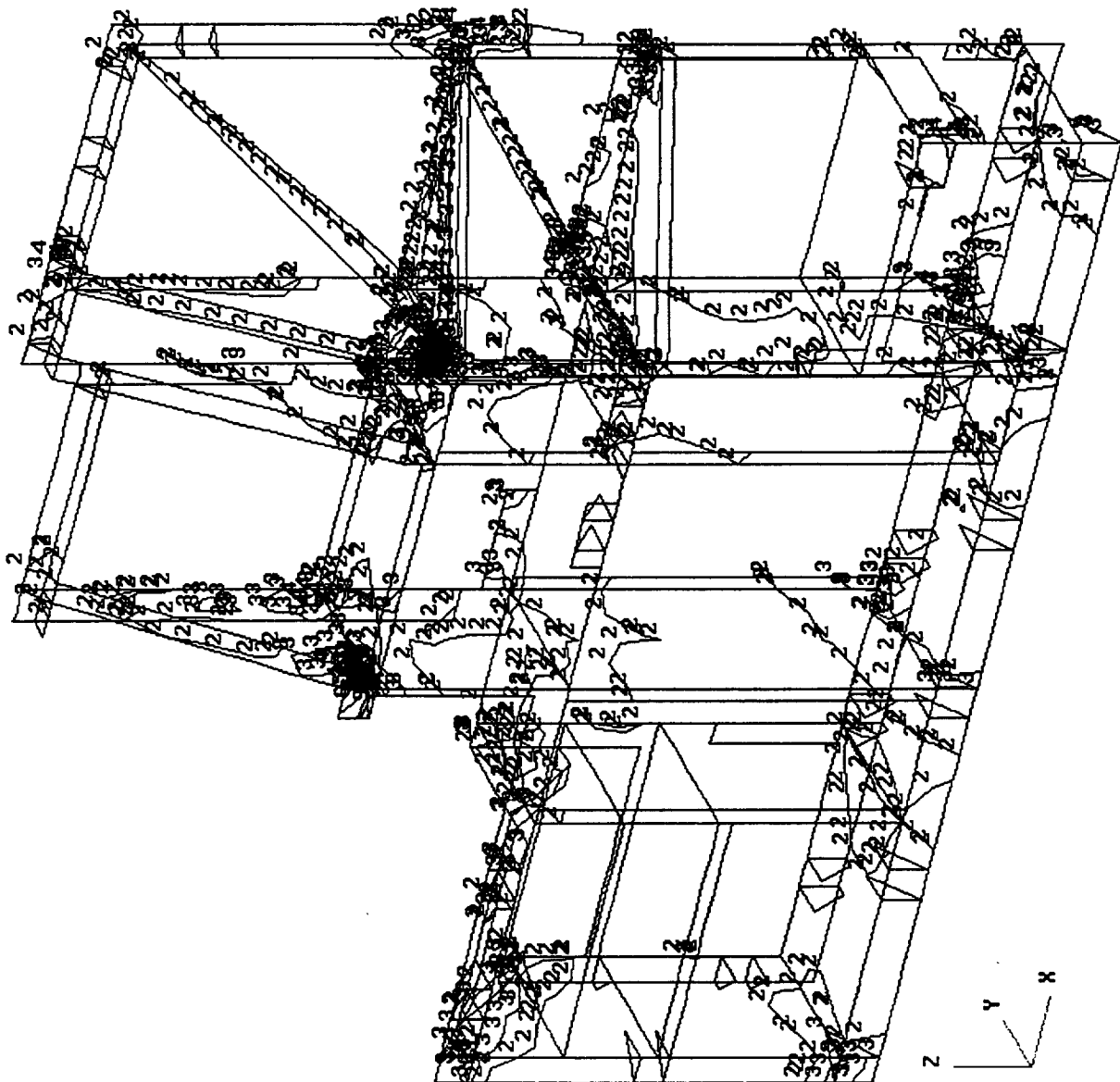
At 22

Design_Gy=-20.6_Gz=19.6

Static Subcase

METSAT AMSU-A1

Sidemount



Time: 09:45:48
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.966E+04
9	0.858E+04
8	0.751E+04
7	0.644E+04
6	0.537E+04
5	0.429E+04
4	0.322E+04
3	0.215E+04
2	0.108E+04
1	0.649E+01

Min = 6.487691E+00

Max = 1.072748E+04

Min ID= 7936

Max ID= 9642

Contour_8:

Stress Tensor

Von Mises

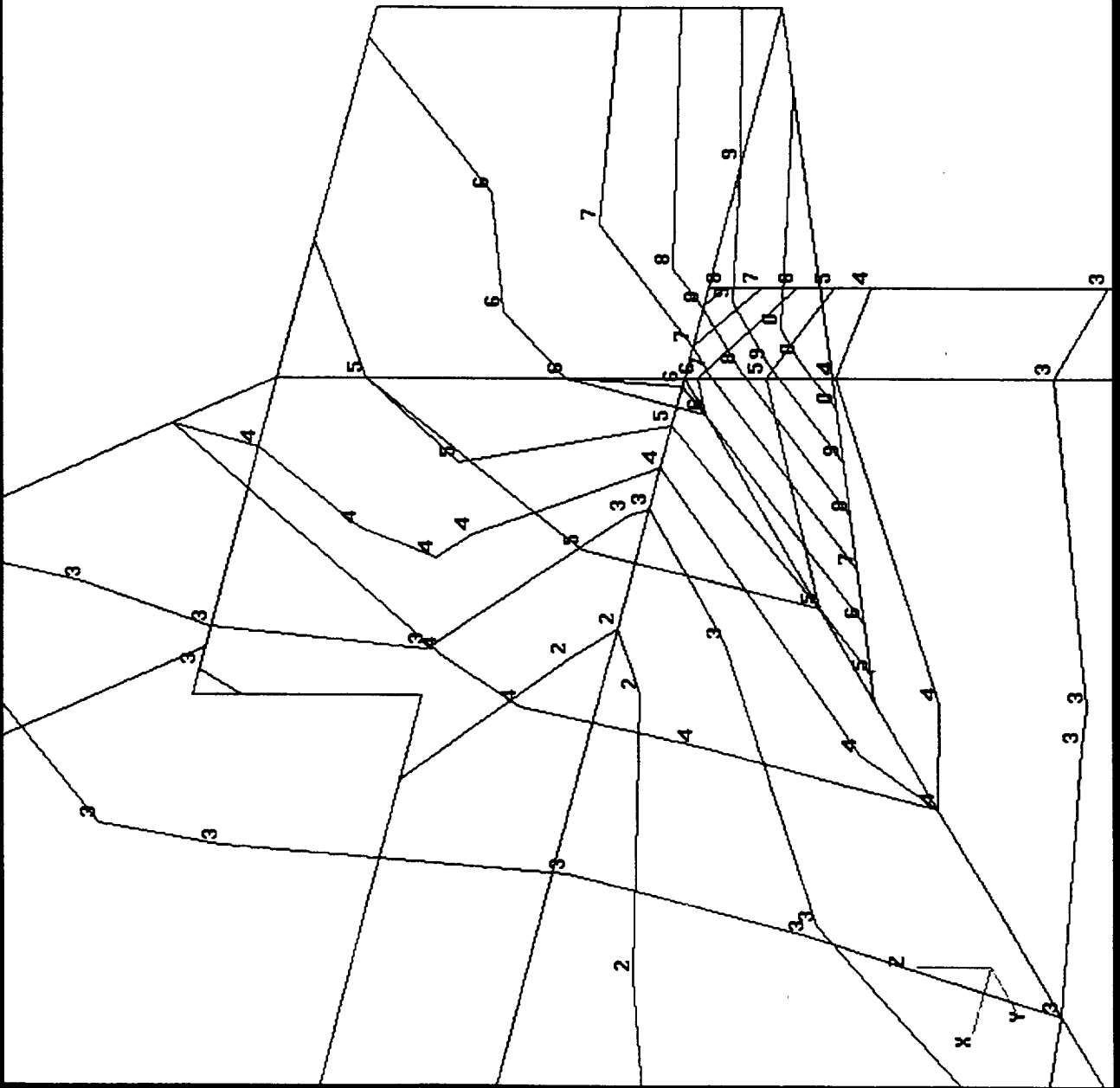
At Z2

Design_Gy=-20.6_Gz=19.6

Static Subcase

METSAT AMSU-A1

Sidemount



Time: 09:43:28
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.966E+04
9	0.858E+04
8	0.751E+04
7	0.644E+04
6	0.537E+04
5	0.429E+04
4	0.322E+04
3	0.215E+04
2	0.108E+04
1	0.649E+01

Min = 6.487691E+00

Max = 1.072748E+04

Min ID= 7936

Max ID= 9642

Contour_8:

Stress Tensor

Von Mises

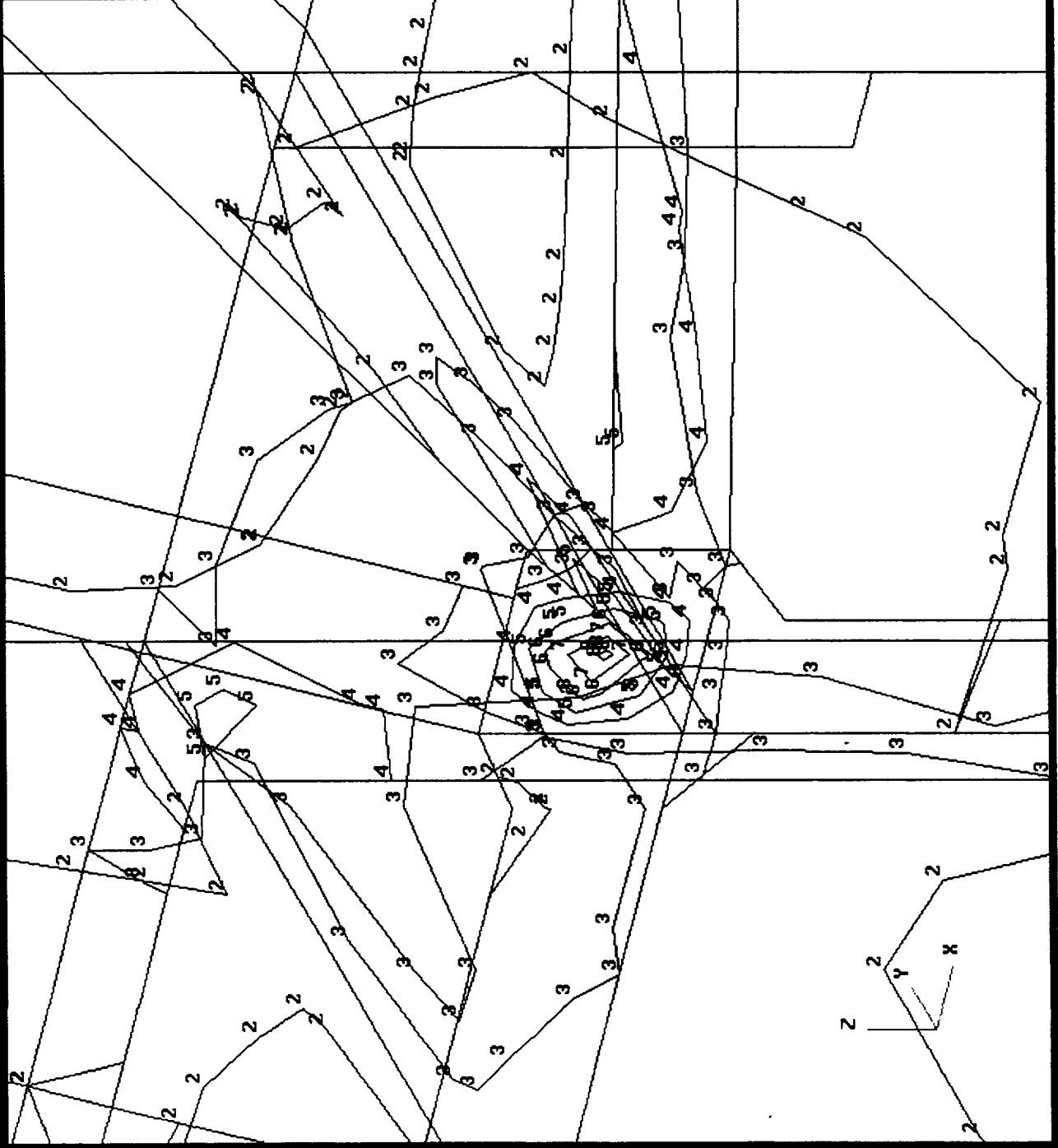
At Z2

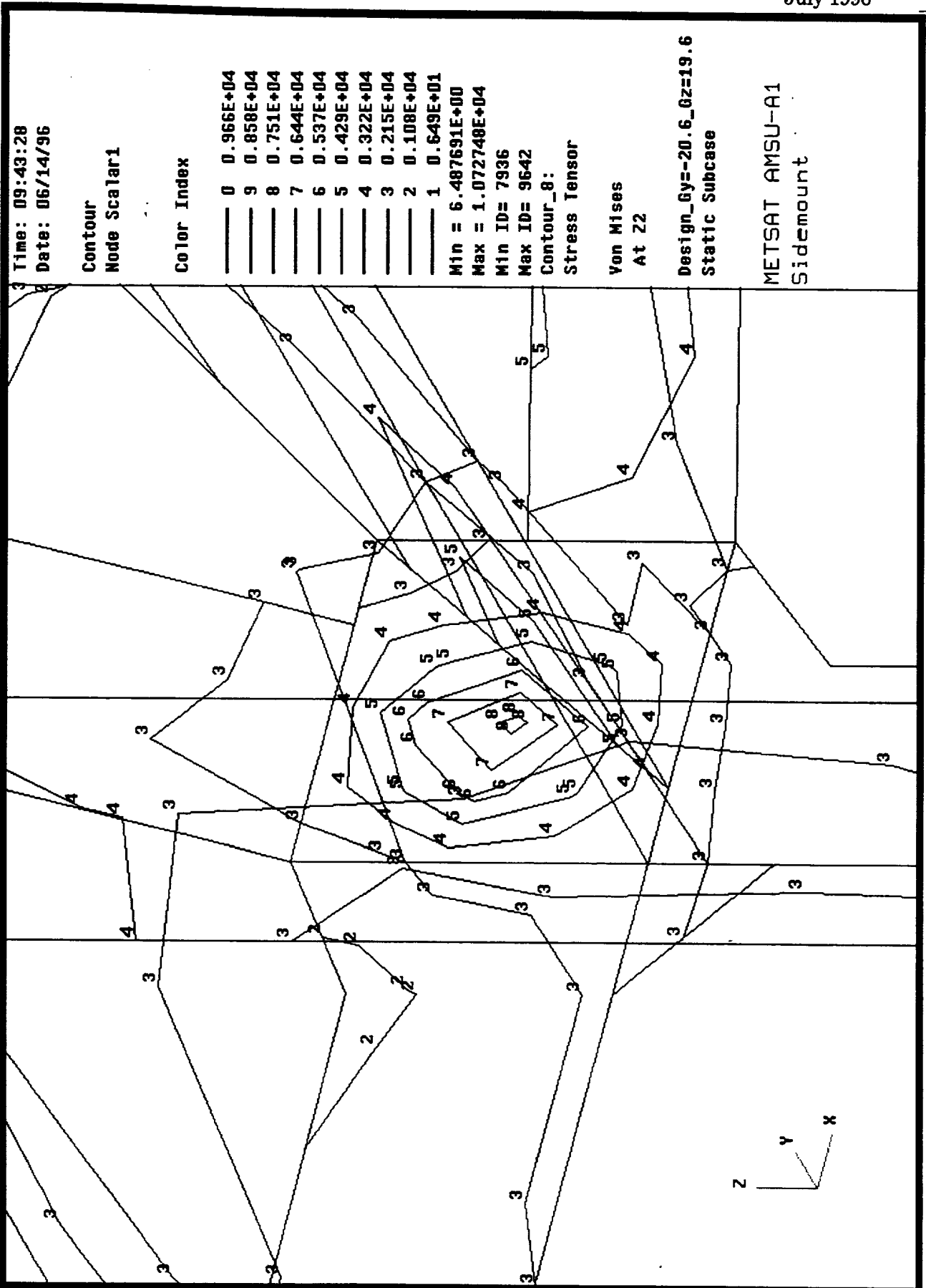
Design_0y=-20.6_Gz=19.6

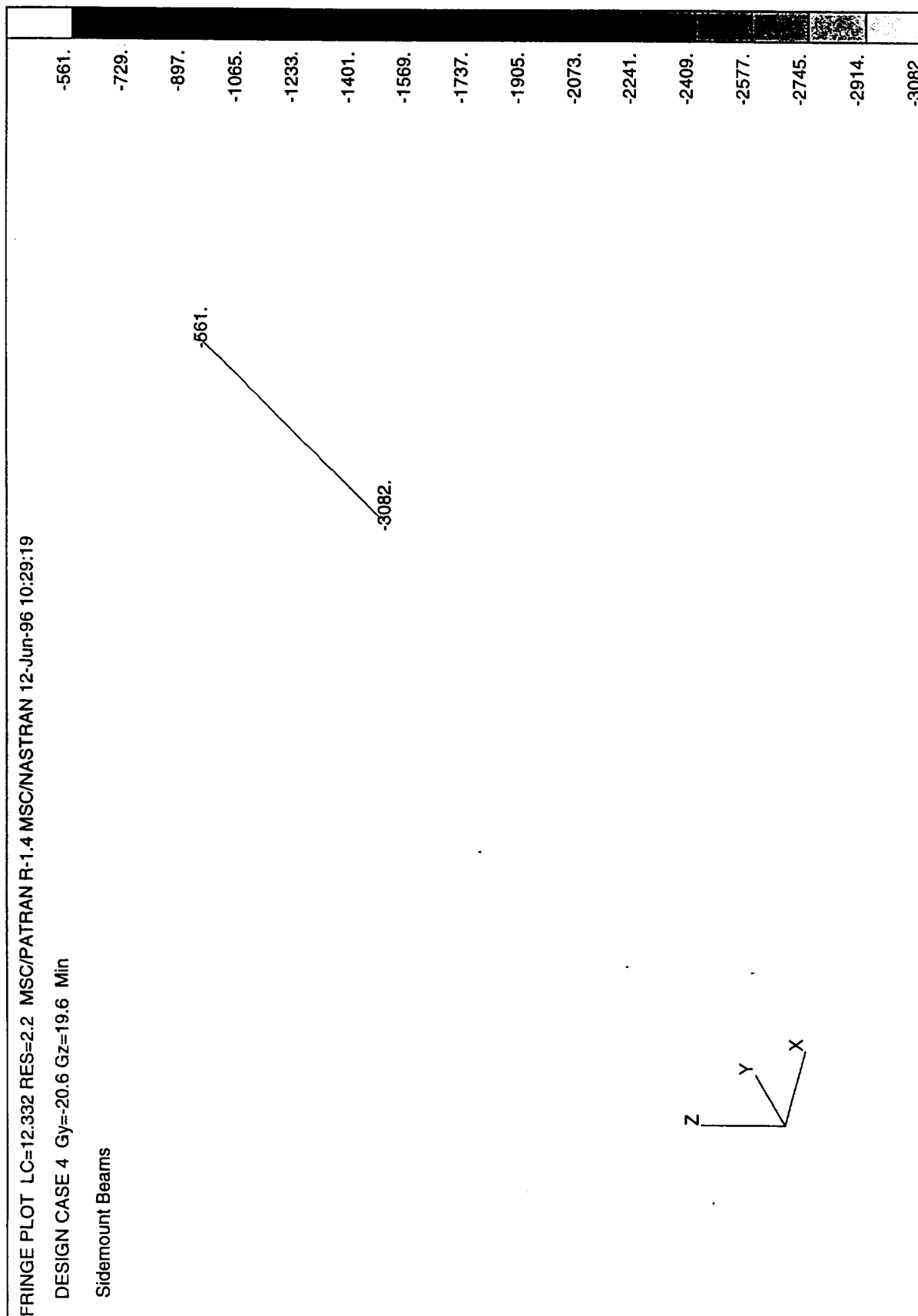
Static Subcase

METSAT AMSU-A1

Sidemount







Time: 09:28:52
Date: 06/14/96

Contour
Node Scalar1

Color Index

0	0.837E+03
9	0.745E+03
8	0.653E+03
7	0.562E+03
6	0.470E+03
5	0.378E+03
4	0.286E+03
3	0.195E+03
2	0.103E+03
1	0.112E+02

Min = 1.117864E+01

Max = 9.287308E+02

Min ID= 4318

Max ID= 4343

Contour_5:

Stress Tensor

Von Mises

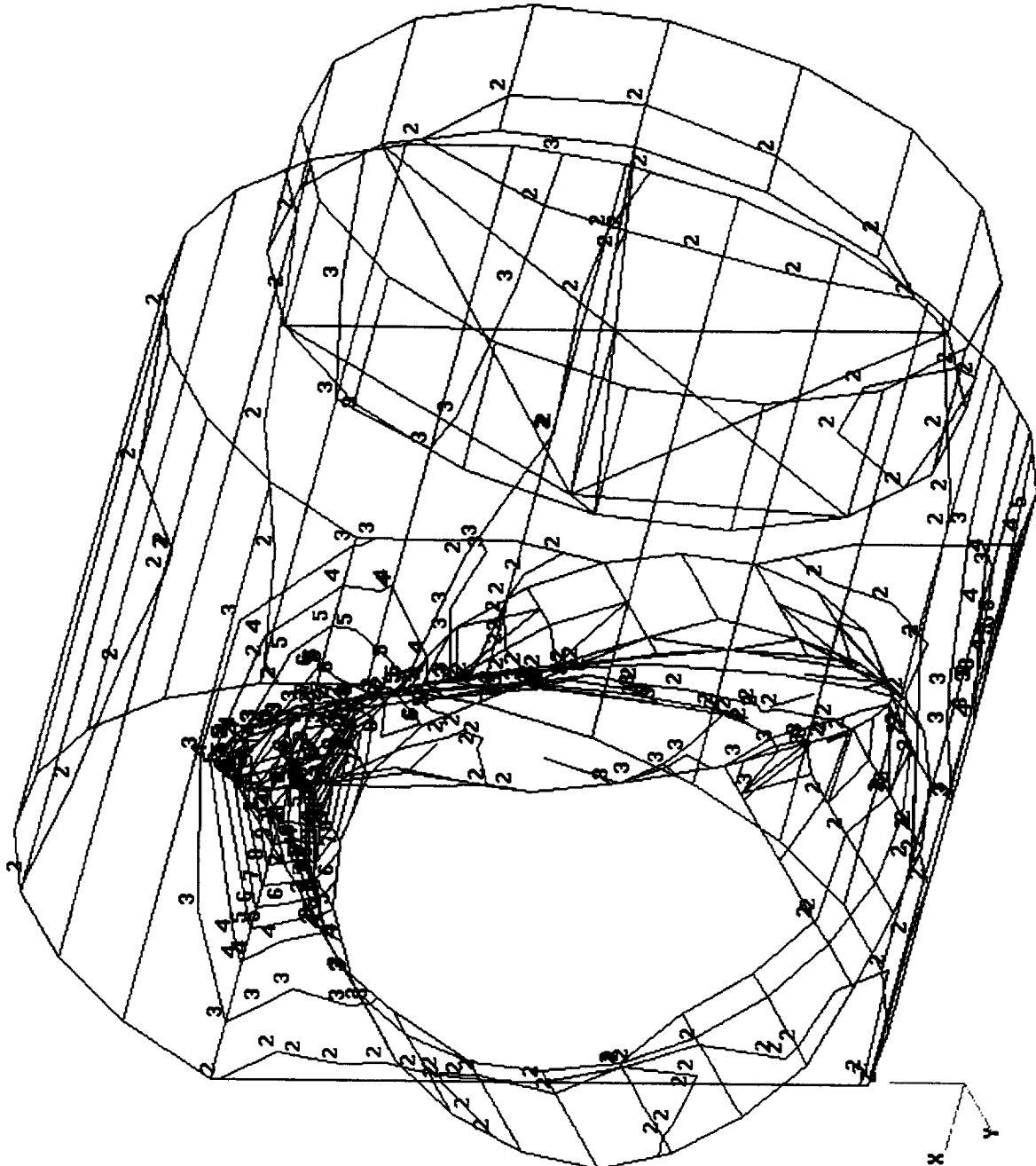
At Z1

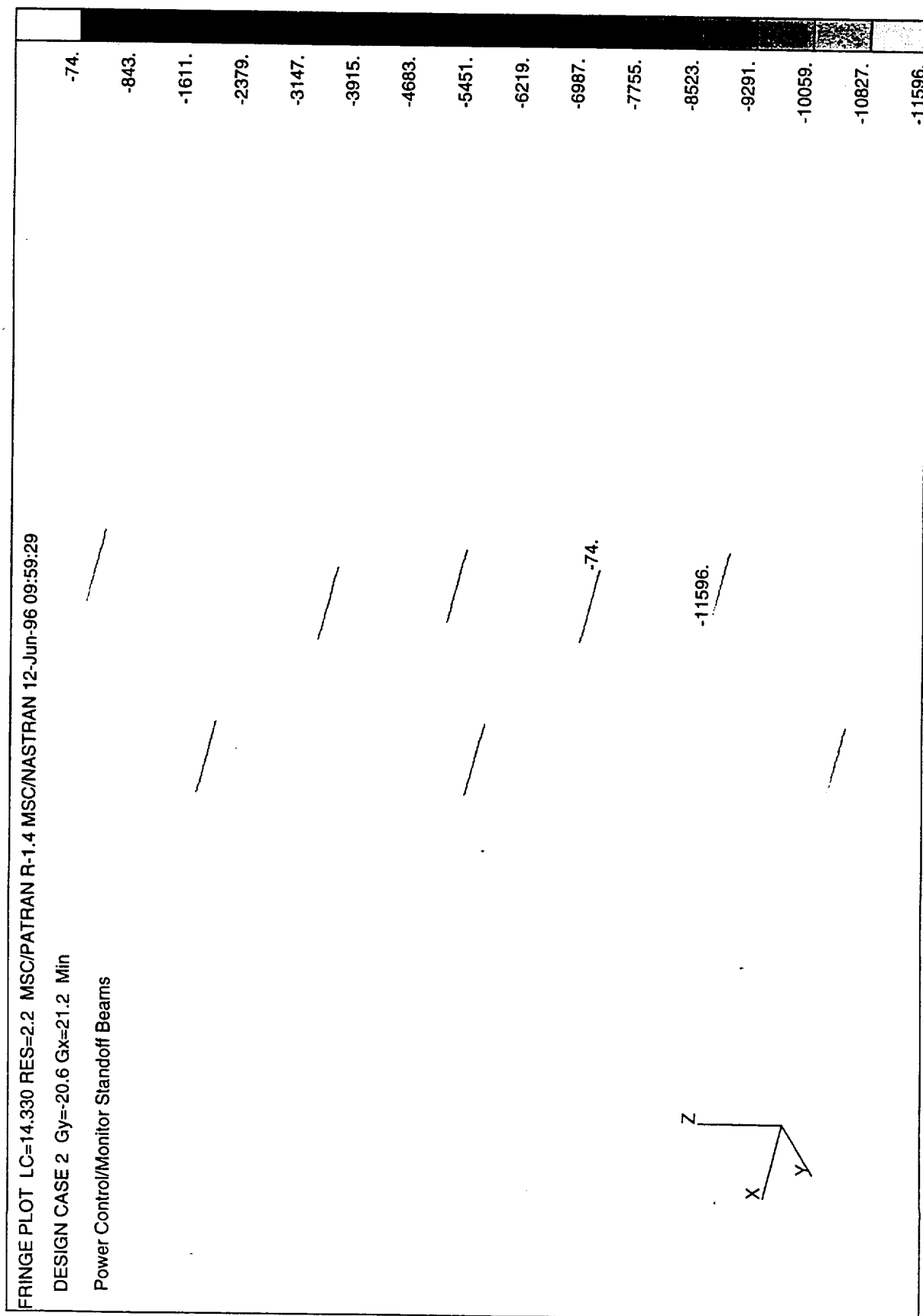
Design_Gy=-20.6_Gz=-19.6

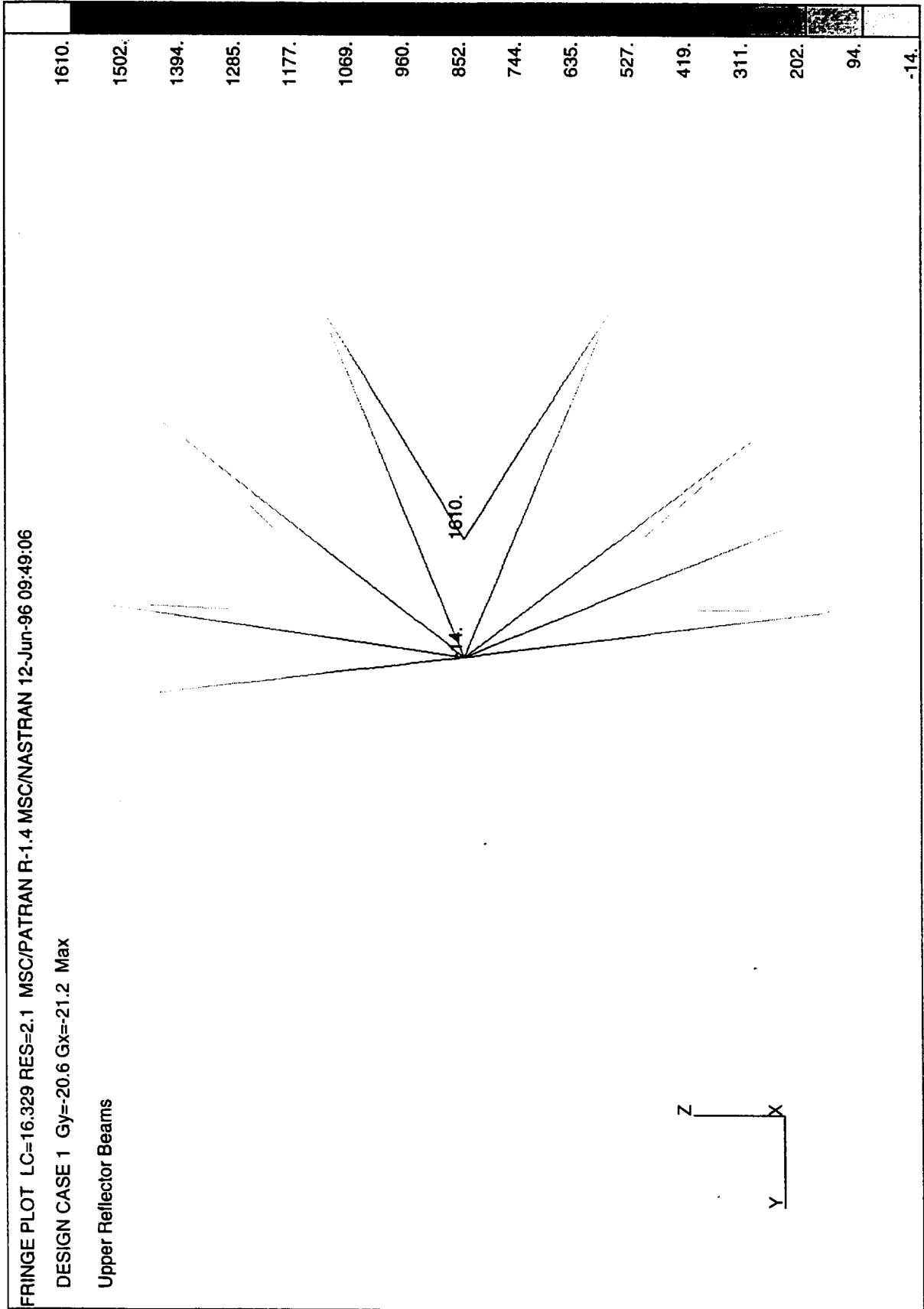
Static Subcase

METSAT AMSU-A1

Upper Reflector







Appendix D
RANDOM VIBRATION STRESSES
(7 PERCENT CRITICAL DAMPING $Q=7$)

RANDOM VIBRATION SUMMARY - METSAT QUALIFICATION LEVEL (8.8 GRMS) JUNE 1996													
LOWER BASEPLATE													
QUAD ELEMENT													
6061-T6 FTY=35000 PSI													
ELEMENT ID.	STRESS ID.	X-LOAD			Y-LOAD			Z-LOAD					
		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS			
179 Z1	3	286	319	957	349	406	1219	1266	1282	3846			
	5	133			140			305					
	7	78			124			125					
179 Z2	10	286	320	961	332	376	1127	1244	1263	3790			
	12	145			108			214					
	14	77			109			143					
197 Z1	3	431	448	1345	497	520	1561	1841	1866	5598			
	5	201			265			802					
	7	66			77			164					
197 Z2	10	423	438	1314	475	516	1548	1783	1785	5355			
	12	242			207			771					
	14	54			113			41					
3523 Z1	3	262	289	868	244	289	866	774	787	2361			
	5	136			135			182					
	7	65			82			87					
3523 Z2	10	276	295	885	266	311	934	860	882	2646			
	12	105			122			207					
	14	60			93			121					
3522 Z1	3	270	455	1364	190	460	1379	666	1310	3931			
	5	394			370			1207					
	7	106			156			258					
3522 Z2	10	252	485	1454	321	537	1610	823	1461	4382			
	12	437			455			1389					
	14	105			133			214					

215 Z1	3	318	327	982	349	383	1148	1387	1413	4238
	5	115			145			449		
	7	45			89			157		
215 Z2	10	329	336	1008	360	394	1182	1403	1412	4237
	12	145			119			434		
	14	38			96			96		
3524 Z1	3	184	209	626	164	239	718	469	534	1601
	5	93			65			150		
	7	54			115			158		
3524 Z2	10	185	206	617	156	255	765	467	520	1561
	12	79			88			169		
	14	51			128			137		

[illegible]

35	2	353	388	1164	994	1045	3134	2952	3027	9081
	3	346	381	1143	996	1047	3142	2927	3002	9006
	4	346	381	1143	996	1047	3142	2927	3002	9006
	5	353	388	1164	994	1045	3134	2952	3027	9081
	6	35			51			75		
	12	344	379	1138	1004	1055	3166	3109	3185	9554
36	13	357	392	1175	992	1043	3128	3104	3179	9536
	14	357	392	1175	992	1043	3128	3104	3179	9536
	15	344	379	1138	1004	1055	3166	3109	3185	9554
	2	468	516	1549	1459	1562	4685	2759	2825	8474
	3	437	486	1457	1453	1556	4667	2747	2813	8440
	4	437	486	1457	1453	1556	4667	2747	2813	8440
87	5	468	516	1549	1459	1562	4685	2759	2825	8474
	6	48			103			66		
	12	256	304	912	859	962	2886	1441	1507	4520
	13	273	321	963	845	948	2843	1469	1535	4605
	14	273	321	963	845	948	2843	1469	1535	4605
	15	256	304	912	859	962	2886	1441	1507	4520
88	2	449	473	1419	803	828	2485	1918	1934	5802
	3	448	472	1417	798	822	2467	1895	1911	5733
	4	448	472	1417	798	822	2467	1895	1911	5733
	5	449	473	1419	803	828	2485	1918	1934	5802
	6	24			25			16		
	12	401	425	1275	877	901	2704	1826	1841	5524
88	13	401	426	1277	857	882	2647	1809	1825	5474
	14	401	426	1277	857	882	2647	1809	1825	5474
	15	401	425	1275	877	901	2704	1826	1841	5524
	2	427	461	1382	924	967	2901	1907	1957	5870
	3	427	461	1384	905	948	2845	1890	1940	5820
	4	427	461	1384	905	948	2845	1890	1940	5820
88	5	427	461	1382	924	967	2901	1907	1957	5870
	6	34			43			50		
	12	646	679	2038	1315	1358	4073	2839	2889	8668
	13	651	684	2053	1504	1547	4642	2993	3043	9129
	14	651	684	2053	1504	1547	4642	2993	3043	9129
	15	646	679	2038	1315	1358	4073	2839	2889	8668

89	2	553	584	1752	1199	1226	3677	2439	2513	7538
	3	550	581	1743	1116	1142	3426	2362	2435	7304
	4	550	581	1743	1116	1142	3426	2362	2435	7304
	5	553	584	1752	1199	1226	3677	2439	2513	7538
	6	31			26			73		
	12	157	189	566	124	150	450	331	404	1211
	13	160	191	574	187	213	640	361	434	1301
	14	160	191	574	187	213	640	361	434	1301
	15	157	189	566	124	150	450	331	404	1211
93	2	131	200	599	289	321	963	927	955	2866
	3	130	199	598	285	317	952	934	962	2885
	4	130	199	598	285	317	952	934	962	2885
	5	131	200	599	289	321	963	927	955	2866
	6	69			32			28		
	12	237	306	918	393	426	1277	1340	1368	4103
	13	231	301	902	403	436	1307	1325	1353	4058
	14	231	301	902	403	436	1307	1325	1353	4058
	15	237	306	918	393	426	1277	1340	1368	4103

LOWER MOTOR MT PANEL													
QUAD ELEMENT													
7075-T6 FTY=66000 PSI													
ELEMENT ID.	STRESS ID.	X-LOAD			Y-LOAD			Z-LOAD					
		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS			
2481 Z1	3	845	2075	6224	176	517	1550	198	613	1838			
	5	2061			479			587					
	7	131			113			102					
2481 Z2	10	697	1845	5536	195	559	1678	230	623	1868			
	12	1798			476			575					
	14	233			174			137					
2501 Z1	3	2180	2246	6738	700	708	2124	780	791	2373			
	5	660			229			246					
	7	323			62			79					
2501 Z2	10	2227	2283	6848	717	725	2175	794	801	2404			
	12	717			216			234					
	14	296			63			67					
2505 Z1	3	1937	1964	5892	1333	1341	4023	569	614	1843			
	5	864			364			251					
	7	172			87			128					
2505 Z2	10	1790	2010	6031	435	520	1560	607	675	2024			
	12	916			273			200					
	14	490			145			179					
2512 Z1	3	2176	2215	6646	1318	1325	3976	707	731	2192			
	5	996			255			252					
	7	218			89			106					
2512 Z2	10	1753	2006	6018	467	491	1474	670	725	2174			
	12	570			194			186					
	14	603			84			171					
2503 Z1	3	3178	3194	9581	916	918	2753	1018	1020	3061			
	5	973			256			291					
	7	186			29			38					

2503 Z2	10	3104	3131	9394	880	883	2650	995	999	2997
	12	1074			315			345		
	14	237			40			50		
2496 Z1	3	2703	2765	8295	861	881	2644	936	954	2863
	5	1219			381			420		
	7	308			100			100		
2496 Z2	10	2949	3009	9027	892	930	2789	1013	1058	3173
	12	1456			485			540		
	14	306			130			152		

LOWER MOTOR MT PANEL													
BAR ELEMENT													
7075-T6 FTY=66000 PSI													
ELEMENT ID.	STRESS ID.	X-LOAD			Y-LOAD			Z-LOAD					
		RMS	PRIN	STRESS	RMS	PRIN	STRESS	RMS	PRIN	STRESS	RMS	PRIN	STRESS
		COMP	STRESS	3 SIG	COMP	STRESS	3 SIG	COMP	STRESS	3 SIG	COMP	STRESS	3 SIG
2546	2	4333	4857	14572	680	841	2522	923	1064	3192			
	3	4507	5031	15093	677	838	2515	949	1090	3271			
	4	4507	5031	15093	677	838	2515	949	1090	3271			
	5	4333	4857	14572	680	841	2522	923	1064	3192			
	6	524			161			141					
	12	1788	2313	6938	247	408	1223	390	531	1594			
2565	13	1898	2423	7268	273	434	1302	422	563	1690			
	14	1898	2423	7268	273	434	1302	422	563	1690			
	15	1788	2313	6938	247	408	1223	390	531	1594			
	2	1645	2922	8765	244	418	1254	380	645	1936			
	3	1624	2901	8702	243	417	1251	371	637	1910			
	4	1624	2901	8702	243	417	1251	371	637	1910			
2603	5	1645	2922	8765	244	418	1254	380	645	1936			
	6	1277			174			266					
	12	3524	4801	14404	502	676	2028	755	1020	3061			
	13	3624	4901	14702	521	695	2085	779	1044	3132			
	14	3624	4901	14702	521	695	2085	779	1044	3132			
	15	3524	4801	14404	502	676	2028	755	1020	3061			
	2	281	975	2926	190	284	853	74	237	712			
	3	456	1150	3450	178	273	818	92	256	769			
	4	2462	3157	9470	446	540	1620	625	789	2368			
	5	2406	3101	9302	456	550	1650	620	784	2352			
	6	694			94			164					
	12	143	837	2511	64	158	474	47	211	632			
	13	199	893	2679	120	214	643	74	237	712			
	14	474	1169	3506	391	485	1455	203	367	1100			
	15	429	1124	3371	362	456	1368	185	349	1047			

2602	2	155	862	2585	62	154	462	31	199	596
	3	254	960	2879	69	160	481	46	214	642
	4	1248	1954	5861	654	746	2237	198	366	1098
	5	1213	1919	5758	653	744	2233	192	360	1079
	6	706			92			168		
	12	384	1091	3272	122	213	640	126	294	883
	13	151	857	2571	63	155	464	66	234	703
	14	1933	2639	7917	717	808	2425	462	630	1889
	15	2000	2706	8119	733	824	2472	487	655	1966
2564	2	765	1754	5262	151	299	897	206	417	1250
	3	789	1778	5335	141	289	866	198	409	1228
	4	789	1778	5335	141	289	866	198	409	1228
	5	765	1754	5262	151	299	897	206	417	1250
	6	989			148			211		
	12	2618	3607	10822	377	526	1577	564	775	2324
	13	2592	3581	10744	371	519	1557	554	765	2294
	14	2592	3581	10744	371	519	1557	554	765	2294
	15	2618	3607	10822	377	526	1577	564	775	2324

UPPER MOTOR MT PANEL													
QUAD ELEMENT													
7075-T6 FTY=66000 PSI													
ELEMENT ID.	STRESS ID.	X-LOAD			Y-LOAD			Z-LOAD					
		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS			
2331 Z1	3	673	754	2262	327	362	1086	158	180	539			
	5	452			123			90					
	7	156			91			44					
2331 Z2	10	728	1010	3029	351	386	1159	171	211	634			
	12	122			44			46					
	14	500			110			82					
2332 Z1	3	2153	2161	6484	855	868	2604	507	511	1532			
	5	666			324			170					
	7	112			84			35					
2332 Z2	10	2125	2228	6685	831	854	2561	496	510	1531			
	12	1124			433			266					
	14	337			98			59					
2287 Z1	3	48	127	380	116	142	425	115	181	544			
	5	40			38			22					
	7	83			52			103					
2287 Z2	10	27	88	264	67	111	333	72	141	423			
	12	18			23			27					
	14	65			62			89					

UPPER MOTOR MT PANEL													
BAR ELEMENT													
7075-T6 FTY=66000 PSI													
ELEMENT ID.	STRESS ID.	X-LOAD				Y-LOAD				Z-LOAD			
		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	
2383	2	697	1227	3682		207	407	1221		110	254	761	
	3	625	1155	3466		174	374	1123		96	240	719	
	4	625	1155	3466		174	374	1123		96	240	719	
	5	697	1227	3682		207	407	1221		110	254	761	
	6	530				200				144			
	12	2296	2826	8478		574	774	2323		511	655	1966	
2365	13	2253	2783	8349		573	773	2319		497	641	1924	
	14	2253	2783	8349		573	773	2319		497	641	1924	
	15	2296	2826	8478		574	774	2323		511	655	1966	
	2	74	167	502		93	166	498		91	182	546	
	3	415	508	1524		216	290	869		156	247	741	
	4	415	508	1524		216	290	869		156	247	741	
2376	5	74	167	502		93	166	498		91	182	546	
	6	94				73				91			
	12	148	241	723		86	159	478		44	135	405	
	13	323	416	1249		187	260	780		82	174	521	
	14	323	416	1249		187	260	780		82	174	521	
	15	148	241	723		86	159	478		44	135	405	
	2	439	680	2041		185	298	895		188	270	809	
	3	162	403	1210		115	228	685		194	276	829	
	4	162	403	1210		115	228	685		194	276	829	
	5	439	680	2041		185	298	895		188	270	809	
	6	242				114				82			
	12	241	483	1448		108	222	666		117	199	597	
	13	106	348	1044		71	185	555		133	215	645	
	14	106	348	1044		71	185	555		133	215	645	
	15	241	483	1448		108	222	666		117	199	597	

2413	2	197	304	913	140	191	573	120	177	531
	3	242	350	1049	217	267	802	297	354	1061
	4	242	350	1049	217	267	802	297	354	1061
	5	197	304	913	140	191	573	120	177	531
	6	107			51			57		
	12	36	143	429	17	67	202	17	74	221
	13	44	151	453	34	85	254	27	83	250
	14	44	151	453	34	85	254	27	83	250
	15	36	143	429	17	67	202	17	74	221

UPPER MOTOR MT PANEL													
BEAM ELEMENT													
7075-T6 FTY=66000 PSI													
		X-LOAD			Y-LOAD			Z-LOAD					
ELEMENT ID.	STRESS ID.	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS
2444	4	1522		4566	353		1060	266					797
	5	1079		3238	281		843	207					620
	6	1475		4424	356		1069	266					799
	7	1124		3371	274		822	204					613
	104	891		2673	470		1411	231					694
	105	2575		7725	701		2102	487					1460
	106	819		2458	435		1306	212					636
2397	107	2633		7899	730		2190	503					1509
	4	1237		3712	409		1226	352					1057
	5	777		2330	355		1066	273					820
	6	1098		3295	333		999	298					893
	7	1176		3528	349		1048	312					937
	104	302		907	117		351	98					294
	105	518		1553	155		464	109					327
2396	106	1294		3881	314		943	260					780
	107	1161		3482	288		865	242					726
	4	1615		4846	610		1829	422					1265
	5	2139		6417	591		1773	530					1591
	6	1774		5321	573		1718	421					1262
	7	1695		5086	598		1793	408					1223
	104	654		1961	261		782	230					690
	105	300		899	318		954	181					543
	106	1491		4472	466		1398	289					867
	107	1403		4209	414		1243	266					799

2395	4	869	2607	236	709	164	493
	5	1506	4518	348	1045	267	802
	6	569	1706	410	1230	236	707
	7	558	1673	420	1261	248	745
	104	1656	4967	632	1896	477	1430
	105	981	2944	413	1239	319	956
	106	627	1880	510	1530	326	978
	107	728	2184	550	1650	350	1051

[illegible]

1225 Z2	10	387	658	1975	466	596	1787	132	421	1263
	12	396			170			65		
	14	267			235			321		
1226 Z1	3	167	200	601	218	321	962	61	367	1102
	5	75			101			145		
	7	64			150			261		
1226 Z2	10	239	264	791	347	426	1277	92	410	1230
	12	99			157			61		
	14	63			145			333		
1240 Z1	3	246	333	1000	298	343	1030	121	217	652
	5	125			108			72		
	7	135			103			119		
1240 Z2	10	254	354	1061	306	367	1100	126	210	630
	12	136			122			63		
	14	147			122			111		
1334 Z1	3	52	177	530	65	208	623	62	179	538
	5	40			66			53		
	7	131			142			122		
1334 Z2	10	63	204	613	65	109	328	49	184	552
	12	78			49			73		
	14	134			52			123		

[illegible]

1183	2	468	505	1516	249	299	896	206	267	802
	3	163	201	603	55	104	311	79	140	421
	4	163	201	603	55	104	311	79	140	421
	5	468	505	1516	249	299	896	206	267	802
	6	37			49			61		
	12	447	484	1452	255	304	911	198	260	780
	13	204	241	723	71	120	360	70	132	395
	14	204	241	723	71	120	360	70	132	395
	15	447	484	1452	255	304	911	198	260	780
1161	2	239	255	764	619	640	1919	358	367	1102
	3	407	423	1268	215	236	707	180	189	567
	4	407	423	1268	215	236	707	180	189	567
	5	239	255	764	619	640	1919	358	367	1102
	6	16			21			9		
	12	108	124	371	227	248	743	71	80	240
	13	110	126	377	44	65	196	29	38	113
	14	110	126	377	44	65	196	29	38	113
	15	108	124	371	227	248	743	71	80	240
1216	2	504	544	1631	195	260	781	416	583	1750
	3	137	176	529	137	202	607	430	597	1791
	4	137	176	529	137	202	607	430	597	1791
	5	504	544	1631	195	260	781	416	583	1750
	6	40			65			167		
	12	449	488	1465	200	266	797	371	538	1615
	13	229	268	805	267	333	998	570	737	2211
	14	229	268	805	267	333	998	570	737	2211
	15	449	488	1465	200	266	797	371	538	1615
1363	2	37	150	451	73	218	655	63	174	522
	3	43	157	470	40	186	557	36	148	443
	4	43	157	470	40	186	557	36	148	443
	5	37	150	451	73	218	655	63	174	522
	6	113			145	291	872	111		
	12	43	157	470	79	224	673	67	178	535
	13	68	181	544	49	194	583	37	148	444
	14	68	181	544	49	194	583	37	148	444
	15	43	157	470	79	224	673	67	178	535

1369	2	132	178	533	139	245	735	92	169	506
	3	189	235	706	152	258	775	113	190	569
	4	189	235	706	152	258	775	113	190	569
	5	132	178	533	139	245	735	92	169	506
	6	46			106			77		
	12	91	137	412	68	174	522	48	125	375
	13	162	208	624	130	237	710	105	182	546
	14	162	208	624	130	237	710	105	182	546
	15	91	137	412	68	174	522	48	125	375

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334 Z2	10	57	398	1195	59	333	998	108	400	1201
	12	292			249			122		
	14	190			151			285		
344 Z1	3	116	308	924	173	274	823	461	517	1551
	5	126			77			139		
	7	187			142			146		
344 Z2	10	114	332	995	122	316	947	390	726	2177
	12	159			166			287		
	14	194			170			384		
352 Z1	3	134	363	1090	63	276	829	150	460	1381
	5	227			149			77		
	7	177			164			345		
352 Z2	10	118	345	1034	68	275	825	140	532	1596
	12	194			159			260		
	14	185			155			327		

[illegible]

313	2	250	284	851	128	154	463	178	191	572
	3	277	311	933	443	469	1407	492	505	1516
	4	277	311	933	443	469	1407	492	505	1516
	5	250	284	851	128	154	463	178	191	572
	6	34			26			13		
	12	560	594	1781	178	205	614	243	256	767
	13	411	445	1335	279	305	916	171	184	552
	14	411	445	1335	279	305	916	171	184	552
	15	560	594	1781	178	205	614	243	256	767
305	2	309	324	973	246	262	786	468	520	1561
	3	266	281	844	207	222	667	259	311	933
	4	266	281	844	207	222	667	259	311	933
	5	309	324	973	246	262	786	468	520	1561
	6	15			16			52		
	12	63	78	233	40	56	168	35	87	260
	13	196	212	635	172	188	563	189	241	723
	14	196	212	635	172	188	563	189	241	723
	15	63	78	233	40	56	168	35	87	260

[illegible]

[illegible]

657	2	43	105	314	101	223	668	112	162	486
	3	61	124	371	83	205	614	32	82	247
	4	61	124	371	83	205	614	32	82	247
	5	43	105	314	101	223	668	112	162	486
	6	62			122			50		
	12	150	212	637	429	551	1653	214	264	792
658	13	151	213	639	428	550	1649	340	391	1172
	14	151	213	639	428	550	1649	340	391	1172
	15	150	212	637	429	551	1653	214	264	792
	2	87	154	461	316	440	1319	310	353	1060
	3	107	173	520	287	411	1232	394	437	1311
	4	107	173	520	287	411	1232	394	437	1311
659	5	87	154	461	316	440	1319	310	353	1060
	6	67			123			43		
	12	342	409	1226	557	681	2043	205	249	746
	13	371	438	1313	554	678	2033	253	296	889
	14	371	438	1313	554	678	2033	253	296	889
	15	342	409	1226	557	681	2043	205	249	746
670	2	383	431	1294	560	640	1920	209	252	755
	3	310	359	1076	535	615	1845	322	365	1096
	4	310	359	1076	535	615	1845	322	365	1096
	5	383	431	1294	560	640	1920	209	252	755
	6	49			80			43		
	12	334	383	1148	552	632	1896	199	242	725
670	13	257	306	917	545	625	1876	533	576	1729
	14	257	306	917	545	625	1876	533	576	1729
	15	334	383	1148	552	632	1896	199	242	725
	2	102	166	498	62	158	473	90	259	778
	3	143	208	623	153	249	748	307	476	1429
	4	143	208	623	153	249	748	307	476	1429
670	5	102	166	498	62	158	473	90	259	778
	6	64			96			169		
	12	75	140	419	43	139	417	96	265	796
	13	122	187	560	141	237	711	282	451	1353
	14	122	187	560	141	237	711	282	451	1353
	15	75	140	419	43	139	417	96	265	796

UPPER FRONT PANEL											
QUAD ELEMENT											
7075-T6 FTY=66000 PSI											
ELEMENT ID.	STRESS ID.	X-LOAD					Y-LOAD				
		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS			RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS
1532 Z1	3	108	221	664	126		315	944	91	232	695
	5	35			49				91		
	7	145			224				141		
1532 Z2	10	125	239	718	108		322	965	184	283	850
	12	49			44				49		
	14	147			244				153		
1540 Z1	3	255	357	1072	256		436	1309	171	394	1181
	5	195			247				216		
	7	129			185				199		
1540 Z2	10	277	384	1152	227		549	1648	100	434	1301
	12	269			474				343		
	14	111			156				174		
1538 Z1	3	269	390	1171	197		308	923	141	373	1119
	5	73			158				184		
	7	196			129				209		
1538 Z2	10	209	297	891	259		388	1163	243	434	1302
	12	72			157				158		
	14	141			172				230		
1539 Z1	3	171	394	1181	170		386	1157	112	377	1132
	5	172			260				205		
	7	222			164				214		
1539 Z2	10	185	404	1212	141		514	1543	78	440	1319
	12	253			415				262		
	14	182			193				253		
1550 Z1	3	140	309	927	107		310	930	52	332	997
	5	180			255				167		
	7	147			106				215		

1550 Z2	10	136	341	1024	130	413	1240	78	416	1247
	12	309			331			222		
	14	82			153			256		
1551 Z1	3	134	540	1620	110	568	1704	76	412	1236
	5	494			528			266		
	7	136			135			221		
1551 Z2	10	201	732	2196	159	673	2018	68	508	1525
	12	714			634			351		
	14	99			140			264		
1544 Z1	3	88	236	708	79	389	1166	147	347	1042
	5	68			161			205		
	7	158			265			169		
1544 Z2	10	100	243	730	119	389	1168	212	337	1011
	12	50			101			75		
	14	166			279			181		

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1397	2	1184	1717	5152	723	1222	3665	706	1227	3682
	3	1165	1699	5096	777	1276	3828	677	1199	3596
	4	1165	1699	5096	777	1276	3828	677	1199	3596
	5	1184	1717	5152	723	1222	3665	706	1227	3682
	6	533			498			522		
	12	204	737	2211	318	816	2449	217	739	2216
1346	13	213	746	2239	312	810	2431	240	762	2286
	14	213	746	2239	312	810	2431	240	762	2286
	15	204	737	2211	318	816	2449	217	739	2216
	2	478	493	1478	324	396	1189	268	417	1252
	3	394	409	1227	146	218	653	168	317	951
	4	394	409	1227	146	218	653	168	317	951
1350	5	478	493	1478	324	396	1189	268	417	1252
	6	15			72			150		
	12	233	248	743	252	324	972	201	351	1053
	13	110	125	376	87	159	478	135	284	853
	14	110	125	376	87	159	478	135	284	853
	15	233	248	743	252	324	972	201	351	1053
1350	2	133	179	536	101	146	437	98	148	443
	3	92	138	414	163	208	623	112	162	486
	4	92	138	414	163	208	623	112	162	486
	5	133	179	536	101	146	437	98	148	443
	6	46			45			50		
	12	33	78	235	34	79	237	21	71	214
	13	75	121	363	164	209	626	125	175	525
	14	75	121	363	164	209	626	125	175	525
	15	33	78	235	34	79	237	21	71	214

UPPER AFT PANEL									
QUAD ELEMENT									
2024-T851 FTY=58000 PSI									

6897 Z2	10	512	1441	4323	317	520	1560	263	385	1155
	12	1080			371			315		
	14	579			174			93		
6898 Z1	3	1362	2139	6416	409	759	2278	523	668	2003
	5	1318			682			479		
	7	798			164			166		
6898 Z2	10	1189	2044	6132	429	737	2211	513	665	1995
	12	1238			632			472		
	14	830			180			171		
7334 Z1	3	2138	3826	11477	726	898	2695	422	531	1593
	5	3634			620			397		
	7	568			219			121		
7334 Z2	10	2016	3716	11149	711	835	2506	440	519	1557
	12	3562			619			387		
	14	512			164			102		
7335 Z1	3	3260	4486	13459	454	646	1938	544	652	1956
	5	3915			554			443		
	7	837			133			150		
7335 Z2	10	3061	4288	12865	404	598	1795	468	582	1747
	12	3835			543			428		
	14	746			104			133		

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1680	2	743	848	2545	284	310	930	396	445	1335
	3	599	705	2115	320	346	1038	340	390	1169
	4	599	705	2115	320	346	1038	340	390	1169
	5	743	848	2545	284	310	930	396	445	1335
	6	106			26			49		
	12	541	647	1941	191	217	651	302	351	1054
1681	13	466	572	1715	222	248	745	247	296	889
	14	466	572	1715	222	248	745	247	296	889
	15	541	647	1941	191	217	651	302	351	1054
	2	282	436	1307	71	127	380	64	161	484
	3	287	440	1321	157	212	637	178	275	826
	4	287	440	1321	157	212	637	178	275	826
7262	5	282	436	1307	71	127	380	64	161	484
	6	154			56			97		
	12	128	282	846	80	135	406	87	185	554
	13	88	241	724	77	132	397	71	168	505
	14	88	241	724	77	132	397	71	168	505
	15	128	282	846	80	135	406	87	185	554
	2	522	1040	3121	172	282	845	106	178	535
	3	527	1045	3136	168	278	835	102	175	526
	4	527	1045	3136	168	278	835	102	175	526
	5	522	1040	3121	172	282	845	106	178	535
	6	519			110			73		
	12	739	1258	3773	96	206	619	88	161	484
	13	685	1203	3610	91	202	605	77	150	451
	14	685	1203	3610	91	202	605	77	150	451
	15	739	1258	3773	96	206	619	88	161	484

[illegible]

1047 Z2	10		349	769	2306		696	1466	4399	374	2319	6957
	12		394				993			1840		
	14		397				603			965		
1115, Z1	3		174	417	1252		271	635	1905	959	1943	5829
	5		201				154			206		
	7		229				418			1307		
1115 Z2	10		193	491	1474		203	520	1560	905	1844	5533
	12		154				196			331		
	14		317				321			1193		
1116 Z1	3		164	422	1266		254	709	2127	841	2293	6879
	5		100				188			529		
	7		288				487			1601		
1116 Z2	10		172	475	1424		173	628	1883	781	2209	6628
	12		129				171			501		
	14		323				456			1562		
1135 Z1	3		168	357	1070		247	624	1873	795	1932	5796
	5		100				191			564		
	7		220				404			1247		
1135 Z2	10		161	433	1300		213	546	1639	790	1834	5502
	12		134				173			539		
	14		286				353			1163		
1113 Z1	3		146	341	1024		204	467	1402	190	1039	3116
	5		49				92			70		
	7		239				315			907		
1113 Z2	10		312	483	1450		281	541	1623	212	987	2962
	12		148				110			79		
	14		240				335			839		

[illegible]

[illegible]

1989	10	538	777	2332	929	1189	3567	1261	1476	4429
1989	12	341			467			434		
1989	14	323			433			473		
1801	3	217	247	742	228	258	773	351	395	1184
1801	5	43			27			46		
1801	7	78			82			124		
1801	10	212	234	703	186	211	632	352	392	1175
1801	12	27			30			40		
1801	14	68			67			118		
1802	3	142	355	1064	124	320	961	247	576	1728
1802	5	86			64			73		
1802	7	239			225			407		
1802	10	153	340	1021	165	363	1089	243	549	1647
1802	12	39			43			37		
1802	14	237			252			396		
1815	3	133	546	1638	145	522	1565	191	783	2348
1815	5	202			132			140		
1815	7	377			383			617		
1815	10	126	474	1421	90	454	1361	188	752	2255
1815	12	106			130			95		
1815	14	357			343			609		
2090	3	217	647	1942	314	776	2327	372	916	2747
2090	5	152			192			191		
2090	7	462			519			628		
2090	10	255	647	1940	203	580	1740	436	1051	3152
2090	12	137			140			223		
2090	14	447			407			713		
2089	3	131	667	2002	215	733	2198	275	963	2888
2089	5	196			199			254		
2089	7	503			525			698		
2089	10	149	661	1982	132	609	1828	286	988	2964
2089	12	181			169			263		
2089	14	495			458			714		
2099	3	214	647	1941	324	768	2304	309	863	2590
2099	5	94			92			143		
2099	7	490			548			632		
2099	10	230	642	1926	191	576	1728	357	978	2933

2099	12	92				93				153		
2099	14	477				431				716		
2023	3	43		1167	3500	50		1253	3759	70	1418	4253
2023	5	1151				1243				1382		
2023	7	133				112				218		
2023	10	40		1227	3681	40		1152	3456	68	1446	4338
2023	12	1215				1135				1417		
2023	14	118				139				201		
1983	3	137		867	2600	131		956	2869	238	1178	3535
1983	5	828				923				1073		
1983	7	169				165				314		
1983	10	123		927	2782	128		815	2446	237	1262	3786
1983	12	898				776				1161		
1983	14	154				164				321		

[illegible]

LOWER RIGHT FRONT SUPPORT PANEL										
QUAD ELEMENT										
6061-T6 FTY=35000 PSI										
		X-LOAD		Y-LOAD		Z-LOAD				
ELEMENT ID.	STRESS ID.	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS
2205	3	131	242	725	94	219	658	67	145	434
2205	5	141			184			86		
2205	7	106			66			68		
2205	10	88	222	665	120	218	653	60	151	453
2205	12	141			164			98		
2205	14	103			72			70		

LOWER RIGHT FRONT SUPPORT PANEL														
BAR ELEMENT														
6061-T6 FTY=35000 PSI														
ELEMENT ID.	STRESS ID.	X-LOAD			Y-LOAD			Z-LOAD						
		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	
2222	2	123	250	750	170	205	614	147	234	703				
2222	3	105	233	698	234	269	807	180	267	800				
2222	4	105	233	698	234	269	807	180	267	800				
2222	5	123	250	750	170	205	614	147	234	703				
2222	6	127			34			87						
2222	12	111	239	716	285	319	958	199	286	858				
2222	13	138	265	794	310	345	1035	266	353	1060				
2222	14	138	265	794	310	345	1035	266	353	1060				
2222	15	111	239	716	285	319	958	199	286	858				
2219	2	351	421	1263	106	149	447	92	133	400				
2219	3	571	640	1921	140	183	549	164	205	614				
2219	4	571	640	1921	140	183	549	164	205	614				
2219	5	351	421	1263	106	149	447	92	133	400				
2219	6	69			43			41						
2219	12	852	921	2763	202	245	734	239	280	841				
2219	13	1229	1299	3896	303	346	1038	362	403	1209				
2219	14	1229	1299	3896	303	346	1038	362	403	1209				
2219	15	852	921	2763	202	245	734	239	280	841				
2223	2	200	424	1271	40	94	282	36	106	319				
2223	3	411	635	1904	81	135	404	87	156	469				
2223	4	411	635	1904	81	135	404	87	156	469				
2223	5	200	424	1271	40	94	282	36	106	319				
2223	6	223			54			70						
2223	12	232	455	1365	69	122	367	52	122	365				
2223	13	302	525	1575	84	137	412	72	142	426				
2223	14	302	525	1575	84	137	412	72	142	426				
2223	15	232	455	1365	69	122	367	52	122	365				

2212	2	205	361	1083	165	261	783	121	229	686
2212	3	115	271	812	110	206	618	132	239	718
2212	4	115	271	812	110	206	618	132	239	718
2212	5	205	361	1083	165	261	783	121	229	686
2212	6	155			96			108		
2212	12	27	182	546	24	121	362	20	127	382
2212	13	19	174	523	21	117	351	21	128	384
2212	14	19	174	523	21	117	351	21	128	384
2212	15	27	182	546	24	121	362	20	127	382

LOWER RIGHT PANEL													
QUAD ELEMENT													
6061-T6 FTY=35000 PSI													
ELEMENT ID.	STRESS ID.	X-LOAD			Y-LOAD			Z-LOAD					
		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS
529	3	264	486	1458	1192	1405	4214	1076	1189	3567			
529	5	139			799			480					
529	7	278			359			283					
529	10	203	369	1106	1165	1368	4103	575	641	1923			
529	12	115			796			354					
529	14	205			340			138					
465	3	53	88	263	84	135	404	53	160	481			
465	5	43			84			44					
465	7	39			51			112					
465	10	34	69	208	88	149	448	55	127	380			
465	12	51			79			53					
465	14	25			66			73					
606	3	263	345	1034	175	298	895	116	361	1082			
606	5	66			58			33					
606	7	151			172			283					
606	10	214	286	859	218	304	913	268	426	1278			
606	12	94			76			84					
606	14	118			141			233					
599	3	62	463	1389	105	307	920	73	313	939			
599	5	408			170			150					
599	7	149			166			198					
599	10	100	342	1027	121	235	704	147	287	861			
599	12	326			143			129					
599	14	63			102			149					
589	3	79	299	896	250	695	2084	334	413	1238			
589	5	282			671			172					
589	7	60			104			138					

589	10	210	270	811	286	694	2082	119	520	1560
589	12	129			657			510		
589	14	93			123			64		
588	3	47	210	629	84	411	1234	44	290	869
588	5	142			231			107		
588	7	105			243			212		
588	10	105	192	575	93	374	1122	71	351	1054
588	12	43			211			76		
588	14	113			214			278		
598	3	157	725	2176	176	530	1590	168	420	1259
598	5	664			386			282		
598	7	186			226			186		
598	10	242	329	987	186	373	1119	183	375	1126
598	12	291			265			151		
598	14	58			142			208		
530	3	185	439	1316	675	829	2486	894	952	2857
530	5	105			359			309		
530	7	291			269			194		
530	10	198	449	1346	808	937	2810	332	563	1688
530	12	109			374			302		
530	14	292			269			245		
580	3	174	345	1034	657	1286	3859	180	341	1023
580	5	248			1157			291		
580	7	129			285			90		
580	10	172	321	962	765	1300	3901	236	425	1274
580	12	239			1180			301		
580	14	110			253			153		

[illegible]

434	2	316	435	1304	157	283	849	125	233	699
434	3	201	319	958	100	226	679	77	185	555
434	4	201	319	958	100	226	679	77	185	555
434	5	316	435	1304	157	283	849	125	233	699
434	6	119			126			108		
434	12	258	377	1131	138	264	792	109	217	652
434	13	220	338	1015	115	241	724	78	186	557
434	14	220	338	1015	115	241	724	78	186	557
434	15	258	377	1131	138	264	792	109	217	652

[illegible]

464	4	34	101	39	117	92	277
464	5	25	76	39	116	99	296
464	6	32	97	88	264	223	670
464	7	32	96	88	263	223	668
464	104	88	265	242	726	242	727
464	105	98	295	248	745	279	836
464	106	238	715	594	1781	653	1960
464	107	236	709	594	1782	650	1949

[illegible]

UPPER RIGHT FRONT SUPPORT PANEL														
BAR ELEMENT														
6061-T6 FTY=35000 PSI														

2265	2	67	107	321	37	91	273	60	109	328
2265	3	75	115	346	70	125	375	115	164	492
2265	4	75	115	346	70	125	375	115	164	492
2265	5	67	107	321	37	91	273	60	109	328
2265	6	40			55			49		
2265	12	100	140	420	103	158	473	222	271	813
2265	13	84	124	373	167	221	663	182	231	692
2265	14	84	124	373	167	221	663	182	231	692
2265	15	100	140	420	103	158	473	222	271	813
2261	2	190	224	672	119	165	496	121	152	456
2261	3	122	156	468	70	116	349	93	124	371
2261	4	122	156	468	70	116	349	93	124	371
2261	5	190	224	672	119	165	496	121	152	456
2261	6	34			46			31		
2261	12	128	162	486	241	287	862	146	177	530
2261	13	130	164	493	142	189	566	129	160	479
2261	14	130	164	493	142	189	566	129	160	479
2261	15	128	162	486	241	287	862	146	177	530
2266	2	729	1110	3329	161	278	834	129	208	623
2266	3	185	566	1699	47	165	494	33	112	335
2266	4	185	566	1699	47	165	494	33	112	335
2266	5	729	1110	3329	161	278	834	129	208	623
2266	6	381			117			79		
2266	12	346	727	2180	129	246	739	64	142	427
2266	13	183	564	1693	86	204	611	36	115	344
2266	14	183	564	1693	86	204	611	36	115	344
2266	15	346	727	2180	129	246	739	64	142	427

[illegible]

1575	10	110	218	654	200	281	844	110	236	708
1575	12	80			158			163		
1575	14	122			100			96		
1615	3	372	757	2272	456	1048	3145	350	692	2076
1615	5	640			1001			643		
1615	7	213			167			130		
1615	10	311	689	2067	485	1077	3231	306	677	2032
1615	12	618			1015			648		
1615	14	164			192			104		

UPPER RIGHT PANEL														
BAR ELEMENT														
6061-T6 FTY=35000 PSI														
ELEMENT ID.	STRESS ID.	X-LOAD			Y-LOAD			Z-LOAD						
		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	
1571	2	199	255	766	233	328	984	253	411	1234				
1571	3	79	135	406	117	213	638	82	240	721				
1571	4	79	135	406	117	213	638	82	240	721				
1571	5	199	255	766	233	328	984	253	411	1234				
1571	6	56			95			158						
1571	12	196	252	755	219	314	943	244	402	1207				
1571	13	85	141	423	114	209	628	93	251	754				
1571	14	85	141	423	114	209	628	93	251	754				
1571	15	196	252	755	219	314	943	244	402	1207				

[illegible]

10139	10	148	162	486	41	57	170	436	463	1389
10139	12	38			20			93		
10139	14	41			24			100		
10140	3	99	115	344	86	90	269	391	416	1247
10140	5	26			13			63		
10140	7	38			16			93		
10140	10	93	112	337	80	84	253	383	412	1236
10140	12	25			13			62		
10140	14	42			17			101		

[illegible]

2194	4	25	75	32	97	35	106
2194	5	75	225	114	341	138	414
2194	6	66	198	30	90	91	273
2194	7	76	228	109	328	143	430
2194	104	25	75	32	97	35	106
2194	105	39	118	22	67	33	99
2194	106	60	179	86	258	68	205
2194	107	41	123	28	85	34	101

LEFT PANEL													
QUAD ELEMENT													
6061-T6 FTY=35000 PSI													
ELEMENT ID.	STRESS ID.	X-LOAD			Y-LOAD			Z-LOAD					
		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS
2982	3	84	329	988		536	1608				111	214	641
2982	5	266									188		
2982	7	125									52		
2982	10	55	275	826		483	1449				167	268	803
2982	12	210									237		
2982	14	120									55		
3041	3	530	570	1711		952	2856				570	586	1759
3041	5	195									239		
3041	7	123									75		
3041	10	669	704	2112		834	2502				383	403	1208
3041	12	285									176		
3041	14	120									66		
5765	3	172	397	1191		826	2477				267	513	1538
5765	5	385									495		
5765	7	52									65		
5765	10	252	520	1561		752	2257				155	299	896
5765	12	449									278		
5765	14	139									55		
2809	3	44	134	403		598	1794				75	266	797
2809	5	65									176		
2809	7	79									130		
2809	10	46	151	454		552	1655				162	604	1811
2809	12	88									531		
2809	14	81									178		
2810	3	38	117	350		657	1970				93	278	835
2810	5	69									216		
2810	7	61									107		

2810	10	46	152	455	174	589	1767	216	660	1981
2810	12	111			577			644		
2810	14	65			72			86		
5724	3	635	668	2004	816	833	2499	360	377	1130
5724	5	250			254			146		
5724	7	117			98			63		
5724	10	572	610	1829	755	770	2311	251	271	813
5724	12	209			261			107		
5724	14	123			87			57		
3050	3	429	790	2369	450	669	2006	193	283	848
3050	5	595			420			203		
3050	7	265			233			84		
3050	10	390	543	1629	414	625	1875	150	225	675
3050	12	401			424			148		
3050	14	147			206			76		
5725	3	231	602	1806	116	481	1444	71	234	703
5725	5	558			449			202		
5725	7	128			108			73		
5725	10	240	519	1557	168	450	1349	83	196	589
5725	12	477			421			165		
5725	14	108			90			59		
5029	3	109	372	1117	53	304	913	53	230	690
5029	5	56			221			176		
5029	7	289			145			97		
5029	10	92	346	1039	69	236	709	50	133	400
5029	12	54			163			85		
5029	14	273			110			63		
5030	3	56	268	803	39	240	721	64	151	452
5030	5	84			101			96		
5030	7	197			167			69		
5030	10	48	247	740	33	178	534	63	145	436
5030	12	65			90			80		
5030	14	190			113			73		
3052	3	44	203	610	147	251	753	96	158	473
3052	5	31			93			116		
3052	7	165			128			51		

3052	10	146	295	886	209	254	762	237	258	773
3052	12	80			54			31		
3052	14	179			95			68		
3030	3	201	286	858	514	523	1570	235	249	746
3030	5	99			124			101		
3030	7	126			61			44		
3030	10	264	341	1024	549	568	1703	335	341	1022
3030	12	119			125			53		
3030	14	131			91			41		
3042	3	115	270	809	408	449	1348	172	191	572
3042	5	76			140			111		
3042	7	173			114			39		
3042	10	199	337	1010	451	473	1418	295	311	933
3042	12	103			122			45		
3042	14	179			87			65		
2815	3	96	290	870	337	927	2780	109	256	769
2815	5	248			917			201		
2815	7	90			78			91		
2815	10	77	278	835	227	762	2286	271	759	2278
2815	12	237			740			746		
2815	14	92			109			81		
2816	3	68	254	761	240	739	2217	80	261	783
2816	5	204			725			212		
2816	7	96			84			94		
2816	10	60	285	856	165	657	1971	204	752	2257
2816	12	242			629			725		
2816	14	98			118			123		
2883	3	260	365	1096	161	262	785	80	204	613
2883	5	54			153			54		
2883	7	182			105			137		
2883	10	311	410	1231	143	339	1017	89	204	611
2883	12	69			212			64		
2883	14	184			158			127		
2884	3	261	386	1159	201	269	808	74	189	567
2884	5	85			57			37		
2884	7	194			120			132		
2884	10	320	436	1309	150	252	756	95	208	625

2884	12	87					39			35		
2884	14	201					147			140		
5008	3	183	280	840			147	734		56	182	547
5008	5	39					112			36		
5008	7	153					114			136		
5008	10	225	323	970			120	1064		71	181	544
5008	12	51					158			42		
5008	14	164					215			124		

[illegible]

[illegible]

2652	2	949	1143	3428		240	514	1542	351	445	1335
2652	3	1120	1314	3941		459	734	2201	397	492	1476
2652	4	1120	1314	3941		459	734	2201	397	492	1476
2652	5	949	1143	3428		240	514	1542	351	445	1335
2652	6	194				274			94		
2652	12	747	941	2822		149	423	1269	247	341	1024
2652	13	525	719	2157		277	551	1653	205	299	898
2652	14	525	719	2157		277	551	1653	205	299	898
2652	15	747	941	2822		149	423	1269	247	341	1024
2646	2	69	268	803		97	181	544	134	187	562
2646	3	113	312	937		115	199	596	131	184	553
2646	4	113	312	937		115	199	596	131	184	553
2646	5	69	268	803		97	181	544	134	187	562
2646	6	199				84			53		
2646	12	155	354	1061		108	192	575	40	93	278
2646	13	171	370	1109		135	219	658	51	104	312
2646	14	171	370	1109		135	219	658	51	104	312
2646	15	155	354	1061		108	192	575	40	93	278
2651	2	294	462	1387		410	704	2111	340	672	2016
2651	3	445	613	1839		390	684	2051	381	713	2139
2651	4	445	613	1839		390	684	2051	381	713	2139
2651	5	294	462	1387		410	704	2111	340	672	2016
2651	6	168				293			332		
2651	12	419	587	1760		590	884	2651	637	969	2907
2651	13	298	466	1397		299	593	1778	964	1296	3888
2651	14	298	466	1397		299	593	1778	964	1296	3888
2651	15	419	587	1760		590	884	2651	637	969	2907

LOWER RIGHT WARMLOAD SHIELD										
QUAD ELEMENT										
6061-T6 FTY=35000 PSI										
					X-LOAD		Y-LOAD		Z-LOAD	
ELEMENT ID.	STRESS ID.	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS
2619	3	399	527	1580	173	204	612	183	215	645
2619	5	90			40			41		
2619	7	236			72			75		
2619	10	441	558	1673	176	202	605	179	218	653
2619	12	108			33			40		
2619	14	228			65			83		
2618	3	261	290	869	174	180	540	122	136	408
2618	5	109			80			57		
2618	7	71			25			33		
2618	10	286	312	936	122	137	410	122	137	411
2618	12	77			79			58		
2618	14	77			29			34		
2614	3	226	288	863	106	124	373	82	107	322
2614	5	78			39			39		
2614	7	114			39			41		
2614	10	227	292	875	92	120	359	94	129	387
2614	12	64			43			55		
2614	14	121			47			51		
2613	3	153	222	666	86	115	344	63	112	335
2613	5	33			38			49		
2613	7	114			47			55		
2613	10	150	192	577	85	104	311	76	108	323
2613	12	27			28			38		
2613	14	83			38			47		

[illegible]

UPPER LEFT WARMLOAD SHIELD										
QUAD ELEMENT										
6061-T6 FTY=35000 PSI										
</										

2682	3	217	441	1322	205	315	946	153	254	763
2682	5	151			234			201		
2682	7	254			95			73		
2682	10	193	421	1262	167	264	792	176	218	653
2682	12	122			183			161		
2682	14	261			89			49		

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2687	2	857	1364	4093	479	919	2757	832	1128	3384
2687	3	902	1409	4228	463	903	2708	798	1094	3282
2687	4	902	1409	4228	463	903	2708	798	1094	3282
2687	5	857	1364	4093	479	919	2757	832	1128	3384
2687	6	507			440			296		
2687	12	387	894	2683	311	750	2251	418	714	2142
2687	13	410	917	2750	316	755	2265	426	722	2165
2687	14	410	917	2750	316	755	2265	426	722	2165
2687	15	387	894	2683	311	750	2251	418	714	2142
2692	2	508	617	1852	670	729	2188	753	813	2439
2692	3	426	536	1607	537	597	1792	482	542	1625
2692	4	426	536	1607	537	597	1792	482	542	1625
2692	5	508	617	1852	670	729	2188	753	813	2439
2692	6	110			60			60		
2692	12	155	265	794	118	178	533	109	169	506
2692	13	147	257	770	138	198	594	112	172	515
2692	14	147	257	770	138	198	594	112	172	515
2692	15	155	265	794	118	178	533	109	169	506

[illegible]

[illegible]

LOWER CARD CAGE ASSEMBLY										
QUAD ELEMENT										
6061-T6 FTY=35000 PSI										
		X-LOAD				Y-LOAD		Z-LOAD		
ELEMENT ID.	STRESS ID.	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS
3710	3	100	177	532	505	1134	3402	294	358	1074
3710	5	128			1016			180		
3710	7	62			272			107		
3710	10	155	192	576	558	942	2826	519	621	1863
3710	12	108			812			169		
3710	14	55			224			214		
3760	3	690	739	2217	857	974	2921	1235	1393	4178
3760	5	311			552			296		
3760	7	145			222			415		
3760	10	690	716	2148	864	918	2754	1288	1346	4039
3760	12	245			298			689		
3760	14	111			183			196		
3689	3	177	363	1088	322	532	1595	441	633	1900
3689	5	254			327			136		
3689	7	142			207			309		
3689	10	211	395	1185	282	479	1438	515	793	2378
3689	12	204			155			403		
3689	14	187			253			329		
3599	3	230	514	1542	797	1849	5546	231	429	1287
3599	5	103			149			90		
3599	7	341			1337			259		
3599	10	311	637	1912	750	1793	5378	424	672	2016
3599	12	122			194			107		
3599	14	410			1291			374		
3711	3	105	217	650	822	1439	4316	259	360	1079
3711	5	183			1415			270		
3711	7	61			120			95		

3711	10	183	218	654	876	1138	3413	714	753	2260
3711	12	156			1088			214		
3711	14	47			114			146		
3686	3	611	698	2093	998	1057	3171	1472	1543	4630
3686	5	217			221			236		
3686	7	203			221			306		
3686	10	595	634	1901	962	1033	3098	1551	1647	4942
3686	12	141			215			384		
3686	14	137			240			349		
3585	3	315	476	1429	547	799	2397	350	2392	7177
3585	5	454			768			2372		
3585	7	60			89			205		
3585	10	290	520	1561	493	892	2675	444	2447	7342
3585	12	491			850			2427		
3585	14	82			129			201		
3593	3	623	659	1976	1564	1701	5102	362	406	1217
3593	5	104			392			116		
3593	7	139			423			112		
3593	10	745	767	2301	1696	1824	5471	693	722	2167
3593	12	59			255			59		
3593	14	125			449			140		
3594	3	153	381	1142	636	2163	6489	464	627	1882
3594	5	75			237			67		
3594	7	264			1715			302		
3594	10	205	418	1255	868	2243	6730	716	939	2817
3594	12	67			166			66		
3594	14	274			1690			441		
3685	3	235	287	860	313	356	1068	489	572	1716
3685	5	52			41			29		
3685	7	110			117			212		
3685	10	197	244	732	344	403	1209	521	586	1759
3685	12	54			29			47		
3685	14	95			149			187		
3595	3	404	538	1613	961	2181	6544	1034	1116	3348
3595	5	129			415			93		
3595	7	234			1468			290		

3595	10	274	472	1417	849	2109	6326	796	952	2856
3595	12	134			409			120		
3595	14	259			1463			360		
3547	3	100	317	950	361	2193	6579	321	556	1668
3547	5	39			170			62		
3547	7	245			1925			341		
3547	10	78	331	994	289	2138	6413	191	631	1892
3547	12	39			141			60		
3547	14	272			1921			501		
3571	3	699	1222	3665	1174	2043	6130	801	1794	5381
3571	5	200			398			910		
3571	7	731			1196			936		
3571	10	660	1190	3569	1081	2020	6060	932	1761	5284
3571	12	268			417			994		
3571	14	698			1227			798		

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UPPER CARD CAGE ASSEMBLY									
QUAD ELEMENT									
6061-T6 FTY=35000 PSI									
ELEMENT ID.	STRESS ID.	X-LOAD			Y-LOAD			Z-LOAD	
		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS
3250	3	938	1261	3784		1156	3469	1114	1505
3250	5	1079						1302	
3250	7	243						281	
3250	10	1058	1412	4237		1294	3882	1250	1692
3250	12	1231						1477	
3250	14	253						308	
3259	3	608	968	2903		902	2705	716	1161
3259	5	874						1047	
3259	7	183						225	
3259	10	770	1074	3223		996	2989	942	1280
3259	12	1011						1213	
3259	14	139						152	
3251	3	228	1732	5195		1596	4788	274	2066
3251	5	1438						1719	
3251	7	664						788	
3251	10	60	1640	4919		1513	4538	76	1959
3251	12	1378						1648	
3251	14	643						765	
3260	3	186	1418	4253		1301	3904	230	1695
3260	5	1271						1520	
3260	7	425						506	
3260	10	40	1337	4012		1224	3673	49	1601
3260	12	1214						1454	
3260	14	400						479	
3249	3	961	988	2964		907	2721	1113	1150
3249	5	645						753	
3249	7	96						122	

3249	10	986	998	2994	896	910	2731	1139	1157	3470
3249	12	559			512			644		
3249	14	72			77			96		
3258	3	461	573	1719	446	543	1630	538	676	2028
3258	5	478			451			565		
3258	7	103			95			124		
3258	10	531	594	1782	496	543	1630	651	735	2205
3258	12	393			362			466		
3258	14	112			93			150		
3475	3	98	410	1230	77	364	1093	62	353	1060
3475	5	381			312			283		
3475	7	95			123			143		
3475	10	100	438	1313	101	468	1405	83	449	1346
3475	12	399			439			391		
3475	14	114			104			145		
3476	3	217	793	2380	177	647	1940	125	568	1704
3476	5	772			613			488		
3476	7	110			126			189		
3476	10	217	806	2418	211	781	2343	151	649	1947
3476	12	781			761			579		
3476	14	123			107			187		

UPPER CARD CAGE ASSEMBLY									
BAR ELEMENT									
6061-T6 FTY=35000 PSI									
					X-LOAD				
ELEMENT ID.	STRESS ID.	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS
3229	2	1905	1933	5800	1749	1776	5328	2244	2285
3229	3	2090	2119	6356	1928	1954	5863	2510	2551
3229	4	2090	2119	6356	1928	1954	5863	2510	2551
3229	5	1905	1933	5800	1749	1776	5328	2244	2285
3229	6	28			26			41	
3229	12	98	127	380	101	127	382	190	231
3229	13	135	164	491	128	155	464	160	201
3229	14	135	164	491	128	155	464	160	201
3229	15	98	127	380	101	127	382	190	231
3215	2	112	185	556	114	186	557	143	233
3215	3	176	249	748	189	260	780	234	323
3215	4	176	249	748	189	260	780	234	323
3215	5	112	185	556	114	186	557	143	233
3215	6	73			71			90	
3215	12	596	669	2008	562	633	1899	704	794
3215	13	486	560	1679	459	530	1590	580	669
3215	14	486	560	1679	459	530	1590	580	669
3215	15	596	669	2008	562	633	1899	704	794
3208	2	1347	1449	4347	2037	2235	6706	1280	1419
3208	3	1669	1771	5312	2259	2457	7371	1479	1618
3208	4	1669	1771	5312	2259	2457	7371	1479	1618
3208	5	1347	1449	4347	2037	2235	6706	1280	1419
3208	6	102			199			139	
3208	12	544	646	1938	809	1007	3022	571	709
3208	13	712	814	2441	585	784	2352	558	697
3208	14	712	814	2441	585	784	2352	558	697
3208	15	544	646	1938	809	1007	3022	571	709

3216	2	460	482	1447	619	651	1953	408	436	1309
3216	3	374	397	1190	675	707	2120	435	463	1389
3216	4	374	397	1190	675	707	2120	435	463	1389
3216	5	460	482	1447	619	651	1953	408	436	1309
3216	6	22			32			28		
3216	12	39	61	183	39	71	212	20	48	144
3216	13	50	72	217	42	74	222	36	64	191
3216	14	50	72	217	42	74	222	36	64	191
3216	15	39	61	183	39	71	212	20	48	144

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3396	10	690	751	2254	941	1025	3075	504	597	1791
3396	12	431			587			282		
3396	14	140			192			171		
3387	3	58	292	875	58	342	1026	56	227	680
3387	5	259			307			206		
3387	7	88			99			60		
3387	10	82	272	816	78	344	1032	104	187	562
3387	12	241			310			149		
3387	14	77			95			57		

UPPER CARD CAGE ASSEMBLY										
BAR ELEMENT										
6061-T4 FTY=16000 PSI										
		X-LOAD			Y-LOAD			Z-LOAD		
ELEMENT ID.	STRESS ID.	RMS COMP	RMS PRIN	RMS 3 SIG	RMS COMP	RMS PRIN	RMS 3 SIG	RMS COMP	RMS PRIN	RMS 3 SIG
		STRESS	STRESS	STRESS	STRESS	STRESS	STRESS	STRESS	STRESS	STRESS
3482	2	663	673	2020	712	724	2171	1114	1138	3414
3482	3	722	733	2200	754	766	2297	1251	1275	3826
3482	4	722	733	2200	754	766	2297	1251	1275	3826
3482	5	663	673	2020	712	724	2171	1114	1138	3414
3482	6	11			11			24		
3482	12	77	88	265	61	72	216	112	136	409
3482	13	74	85	255	62	74	221	53	77	231
3482	14	74	85	255	62	74	221	53	77	231
3482	15	77	88	265	61	72	216	112	136	409

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10693	3	54	382	1145	15	108	323	15	104	313
10693	5	377			106			102		
10693	7	38			13			13		
10693	10	54	354	1061	15	102	305	14	121	363
10693	12	348			99			120		
10693	14	40			14			13		
10292	3	53	332	997	14	103	310	17	79	237
10292	5	327			102			77		
10292	7	40			11			10		
10292	10	53	359	1077	14	110	330	16	82	245
10292	12	354			109			80		
10292	14	38			11			11		

LOWER WARMLOAD									
QUAD ELEMENT									
6061-T6 FTY=35000 PSI									
ELEMENT ID.	STRESS ID.	X-LOAD			Y-LOAD			Z-LOAD	
		RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS
2789	3	668	1496	4489	822	1965	5896	470	1053
2789	5	1444			1945			1026	
2789	7	209			152			127	
2789	10	916	1542	4626	1579	2150	6451	686	1102
2789	12	1489			2103			1072	
2789	14	182			164			113	
2778	3	469	675	2026	873	949	2848	474	543
2778	5	407			505			286	
2778	7	235			184			134	
2778	10	1165	1214	3642	1457	1490	4471	913	935
2778	12	442			551			310	
2778	14	194			177			115	
2777	3	409	647	1940	724	1032	3095	392	555
2777	5	373			479			251	
2777	7	255			412			223	
2777	10	1006	1056	3168	1199	1265	3794	769	793
2777	12	387			526			266	
2777	14	183			220			114	
2788	3	639	1354	4063	742	1849	5547	481	1050
2788	5	1293			1733			975	
2788	7	209			358			206	
2788	10	817	1353	4059	1427	1974	5923	630	1030
2788	12	1311			1899			1002	
2788	14	150			203			107	
2771	3	224	671	2014	132	587	1761	39	390
2771	5	194			204			103	
2771	7	462			417			317	

2771	10	464	817	2452	353	584	1751	150	442	1325
2771	12	163			197			98		
2771	14	481			299			317		
2772	3	858	1080	3241	647	838	2513	506	681	2042
2772	5	279			192			227		
2772	7	421			351			282		
2772	10	818	1075	3225	554	708	2124	465	630	1890
2772	12	299			210			186		
2772	14	446			277			270		
2760	3	1423	1523	4570	953	1091	3273	996	1091	3273
2760	5	580			398			354		
2760	7	307			309			265		
2760	10	1464	1603	4808	939	1067	3200	953	1062	3185
2760	12	569			421			364		
2760	14	379			287			276		
2761	3	1320	1567	4700	982	1154	3463	1058	1212	3636
2761	5	404			300			329		
2761	7	536			383			369		
2761	10	1238	1600	4800	973	1139	3416	1069	1239	3717
2761	12	402			285			301		
2761	14	658			376			400		
2782	3	186	749	2247	101	750	2250	- 100	494	1481
2782	5	135			184			112		
2782	7	588			606			388		
2782	10	315	809	2426	312	786	2357	255	582	1747
2782	12	123			156			94		
2782	14	582			546			400		
2783	3	1118	1335	4006	1169	1347	4042	813	926	2779
2783	5	419			384			305		
2783	7	447			415			266		
2783	10	1243	1424	4271	1192	1363	4088	836	963	2889
2783	12	430			413			311		
2783	14	424			402			287		
2793	3	1550	1614	4843	1512	1566	4699	1088	1158	3473
2793	5	607			652			388		
2793	7	255			224			231		

2793	10	1563	1680	5040		1396	1487	4461	1018	1115	3346
2793	12	615				661			386		
2793	14	353				274			266		
2794	3	1844	2107	6321		1833	2021	6064	1408	1570	4711
2794	5	605				564			440		
2794	7	629				523			429		
2794	10	1767	2135	6405		1834	2060	6180	1413	1606	4817
2794	12	591				571			436		
2794	14	754				581			475		
2767	3	567	708	2124		696	757	2272	275	365	1094
2767	5	376				318			217		
2767	7	216				164			115		
2767	10	1116	1162	3487		965	1002	3006	634	663	1989
2767	12	401				352			233		
2767	14	188				155			111		
2768	3	568	693	2078		622	827	2480	250	371	1112
2768	5	328				308			176		
2768	7	213				326			153		
2768	10	975	1018	3053		824	868	2603	547	565	1696
2768	12	349				349			189		
2768	14	169				151			83		
5091	3	826	982	2945		1606	1832	5497	675	791	2373
5091	5	975				1831			788		
5091	7	33				20			20		
5091	10	378	1002	3005		827	1878	5635	351	810	2429
5091	12	993				1874			805		
5091	14	75				69			45		
5092	3	844	1118	3353		1405	1762	5287	638	823	2469
5092	5	1050				1647			769		
5092	7	136				203			100		
5092	10	273	1051	3153		641	1666	4999	288	775	2325
5092	12	1045				1659			768		
5092	14	68				88			57		
5132	3	786	999	2998		1091	1318	3953	452	573	1718
5132	5	952				1229			542		
5132	7	101				142			61		

5132	10	289	962	2887	485	1246	3739	163	539	1616
5132	12	951			1241			535		
5132	14	85			65			39		
5133	3	811	983	2948	1240	1401	4203	488	576	1729
5133	5	974			1398			575		
5133	7	39			22			12		
5133	10	321	999	2998	639	1437	4312	206	592	1775
5133	12	992			1433			587		
5133	14	68			57			41		

[illegible]

2730	10	277	408	1223	258	468	1403	157	378	1134
2730	12	91			140			111		
2730	14	204			262			243		
2744	3	206	623	1868	360	979	2936	268	687	2062
2744	5	540			851			596		
2744	7	186			281			195		
2744	10	394	629	1886	775	1064	3192	515	711	2134
2744	12	572			952			651		
2744	14	115			180			109		
2745	3	219	670	2009	418	985	2954	303	692	2076
2745	5	584			955			682		
2745	7	196			129			63		
2745	10	445	695	2084	897	1136	3409	607	781	2344
2745	12	619			1047			742		
2745	14	138			146			83		
2733	3	236	377	1130	325	569	1706	255	415	1246
2733	5	139			234			167		
2733	7	183			285			199		
2733	10	275	338	1014	578	639	1917	471	499	1497
2733	12	156			265			186		
2733	14	107			151			94		
2734	3	213	388	1164	433	508	1523	349	370	1111
2734	5	154			258			194		
2734	7	202			137			61		
2734	10	353	425	1276	753	791	2374	634	645	1936
2734	12	164			283			214		
2734	14	137			140			71		
2707	3	986	1013	3039	2577	2625	7876	1400	1449	4348
2707	5	325			834			457		
2707	7	137			294			223		
2707	10	597	721	2163	2279	2345	7034	1282	1326	3977
2707	12	228			810			464		
2707	14	247			318			194		
2708	3	397	477	1431	859	986	2959	498	589	1768
2708	5	282			498			231		
2708	7	125			250			181		

2708	10	765	831	2492	1030	1170	3511	619	699	2097
2708	12	212			410			211		
2708	14	202			326			197		
2718	3	211	315	944	802	848	2543	438	488	1463
2718	5	75			251			137		
2718	7	157			165			131		
2718	10	337	410	1230	1277	1355	4064	585	622	1867
2718	12	75			355			186		
2718	14	156			279			128		
2719	3	89	262	785	120	527	1582	36	325	976
2719	5	61			98			57		
2719	7	186			418			279		
2719	10	171	324	971	191	638	1913	104	331	993
2719	12	69			115			78		
2719	14	197			483			240		
5710	3	75	131	392	35	165	496	13	54	161
5710	5	74			74			26		
5710	7	56			109			34		
5710	10	74	189	567	34	143	429	12	69	207
5710	12	137			64			45		
5710	14	77			93			37		
2711	3	327	615	1846	716	1583	4748	237	768	2303
2711	5	524			1549			746		
2711	7	162			172			106		
2711	10	760	835	2505	890	1584	4752	541	807	2422
2711	12	644			1551			797		
2711	14	120			150			52		
2712	3	362	697	2092	739	1658	4973	219	803	2410
2712	5	353			1645			793		
2712	7	340			108			77		
2712	10	635	707	2120	945	1717	5151	566	853	2558
2712	12	413			1691			840		
2712	14	145			141			61		
2722	3	212	314	942	225	540	1621	59	263	788
2722	5	45			423			183		
2722	7	166			192			127		
2722	10	185	268	805	892	942	2826	326	344	1032

2722	12	74				440				204		
2722	14	127				158				50		
2723	3	159	466	1398		333		539	1618	93	254	761
2723	5	68				474				215		
2723	7	350				117				79		
2723	10	187	280	840		1139		1165	3496	403	419	1257
2723	12	65				523				232		
2723	14	141				131				55		

[illegible]

3156	4	60	179	67		200	44	131
3156	5	116	348	73		220	36	108
3156	6	71	213	77		232	57	172
3156	7	117	352	75		224	36	109
3156	104	109	327	135		405	57	171
3156	105	37	110	125		374	44	133
3156	106	108	323	83		250	112	336
3156	107	34	102	99		297	44	132
3157	4	50	151	99		298	78	234
3157	5	124	371	135		406	60	179
3157	6	57	172	134		403	198	594
3157	7	123	370	111		334	49	146
3157	104	104	311	211		632	126	377
3157	105	141	424	261		784	241	722
3157	106	140	420	390		1169	242	726
3157	107	134	401	203		608	262	785
3158	4	46	139	41		124	164	492
3158	5	94	281	139		418	83	250
3158	6	66	198	246		739	275	825
3158	7	96	287	125		375	98	293
3158	104	161	484	257		770	143	430
3158	105	137	412	259		777	229	687
3158	106	290	871	396		1189	169	507
3158	107	107	322	181		543	220	659
3155	4	56	169	53		158	25	76
3155	5	322	965	96		287	116	347
3155	6	52	155	144		433	57	172
3155	7	320	959	101		304	113	340
3155	104	217	651	114		343	57	172
3155	105	91	274	107		320	44	132
3155	106	203	610	55		164	69	208
3155	107	93	279	96		288	41	123

3154	4	301	902	93	279	77	231
3154	5	478	1434	149	448	149	447
3154	6	279	836	154	461	90	269
3154	7	481	1442	155	465	150	450
3154	104	141	422	77	230	89	266
3154	105	526	1578	169	506	208	623
3154	106	122	366	91	272	70	210
3154	107	528	1585	171	514	211	632
3162	4	38	113	59	176	26	79
3162	5	95	285	160	480	147	441
3162	6	77	230	41	122	83	248
3162	7	103	308	165	494	141	423
3162	104	153	460	252	755	166	497
3162	105	88	264	79	236	54	161
3162	106	196	589	363	1090	163	490
3162	107	63	190	20	60	77	232
3153	4	30	91	62	186	53	160
3153	5	249	748	235	706	390	1170
3153	6	211	634	104	311	232	695
3153	7	244	732	233	699	369	1107
3153	104	163	490	123	370	187	561
3153	105	173	519	132	396	315	946
3153	106	168	504	220	660	179	538
3153	107	172	516	134	403	318	954

[illegible]

4532	10	1012	2791	8372		442	1197	3592	416	1150	3451
4532	12	2779				1191			1147		
4532	14	146				69			54		
4533	3	950	2876	8628		411	1225	3676	411	1195	3584
4533	5	2830				1197			1169		
4533	7	299				151			141		
4533	10	1063	2815	8445		434	1197	3592	444	1166	3497
4533	12	2811				1195			1164		
4533	14	88				44			34		
4514	3	607	1908	5723		347	1005	3015	309	907	2720
4514	5	1894				1001			903		
4514	7	135				55			48		
4514	10	637	1885	5656		371	1030	3091	333	923	2768
4514	12	1863				1004			901		
4514	14	166				131			113		
4515	3	1939	1945	5836		1037	1043	3129	930	932	2797
4515	5	841				385			352		
4515	7	79				63			37		
4515	10	1914	1926	5778		1002	1007	3020	902	905	2716
4515	12	686				359			312		
4515	14	120				53			45		
4398	3	155	374	1122		134	273	820	73	190	570
4398	5	274				212			131		
4398	7	148				92			83		
4398	10	206	479	1436		143	329	987	81	252	756
4398	12	280				289			189		
4398	14	232				86			103		
4381	3	315	1016	3048		183	549	1647	130	386	1158
4381	5	952				543			374		
4381	7	211				49			56		
4381	10	381	1004	3012		182	589	1768	133	402	1207
4381	12	957				575			390		
4381	14	172				77			57		

[illegible]

POWER CONTROL/MONITOR - BRACKET												
QUAD ELEMENT												
6061-T6 FTY=35000 PSI												
		X-LOAD			Y-LOAD			Z-LOAD				
ELEMENT ID.	STRESS ID.	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS
6495	3	160	2082	6247	117	473	1419	91	468	1403		
6495	5	1980			437			421				
6495	7	443			113			132				
6495	10	225	2075	6225	87	427	1282	78	406	1218		
6495	12	1990			410			396				
6495	14	397			76			55				
6496	3	1147	2661	7982	127	421	1264	141	491	1472		
6496	5	2164			383			429				
6496	7	867			106			147				
6496	10	1110	2683	8050	133	444	1333	148	458	1374		
6496	12	2174			410			417				
6496	14	895			103			112				
10383	3	268	1000	2999	146	252	756	185	340	1021		
10383	5	901			221			244				
10383	7	269			57			122				
10383	10	208	1106	3319	137	290	871	231	445	1334		
10383	12	876			236			280				
10383	14	455			91			188				
10351	3	172	828	2483	73	210	630	102	296	887		
10351	5	731			198			237				
10351	7	251			41			106				
10351	10	166	659	1978	49	246	738	57	357	1071		
10351	12	531			211			255				
10351	14	251			84			175				
6497	3	923	966	2898	108	128	384	115	126	377		
6497	5	397			50			39				
6497	7	156			40			31				

6497	10	892	930	2789		110	139	416	117	137	411
6497	12	396				49			40		
6497	14	142				50			44		
6498	3	297	847	2541		98	131	392	78	122	367
6498	5	479				53			49		
6498	7	450				50			57		
6498	10	323	878	2634		96	139	417	89	143	428
6498	12	479				53			50		
6498	14	470				61			70		
10401	3	364	840	2520		219	239	716	494	502	1506
10401	5	683				112			212		
10401	7	273				50			49		
10401	10	737	880	2641		140	177	531	396	414	1242
10401	12	655				116			193		
10401	14	180				48			63		
10403	3	378	1216	3647		151	249	748	338	494	1481
10403	5	1167				192			299		
10403	7	203				75			174		
10403	10	604	1312	3936		98	276	827	248	552	1655
10403	12	1225				194			319		
10403	14	248				121			266		
10395	3	76	488	1464		20	114	343	22	184	552
10395	5	481				108			160		
10395	7	55				24			62		
10395	10	104	395	1184		18	186	557	18	392	1176
10395	12	359				184			386		
10395	14	101				16			47		
10394	3	151	502	1505		61	84	253	113	157	472
10394	5	323				52			69		
10394	7	251				27			63		
10394	10	133	451	1352		77	104	311	172	202	605
10394	12	405				56			67		
10394	14	120				35			63		

6492	3	260	1933	5800	77	384	1153	98	712	2135
6492	5	1887			368			641		
6492	7	279			70			208		
6492	10	111	1744	5233	76	369	1107	80	560	1681
6492	12	1728			362			544		
6492	14	164			45			88		

POWER CONTROL/MONITOR - PWB													
QUAD ELEMENT													
REINFORCED PLASTIC FTY=24000 PSI													
X-LOAD													
ELEMENT ID.	STRESS ID.	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS
5861	10	44	1053	3160	76	554	1663	51	411	1233			
5861	12	1044			542			404					
5861	14	97			75			50					
5861	3	81	597	1790	70	513	1539	51	375	1124			
5861	5	593			508			371					
5861	7	47			44			33					
5875	10	134	509	1528	69	254	761	68	220	661			
5875	12	380			189			166					
5875	14	220			109			91					
5875	3	104	404	1213	54	226	677	54	188	563			
5875	5	377			179			152					
5875	7	90			90			69					
5874	10	52	805	2414	91	497	1491	79	374	1121			
5874	12	788			482			356					
5874	14	112			78			72					
5874	3	57	582	1746	80	470	1410	73	347	1042			
5874	5	573			465			338					
5874	7	68			42			50					
5987	10	544	1063	3189	215	493	1479	163	367	1102			
5987	12	913			466			348					
5987	14	280			87			63					
5987	3	281	591	1773	152	443	1328	119	328	985			
5987	5	584			437			323					
5987	7	46			42			33					
5866	10	25	452	1356	18	322	966	14	241	724			
5866	12	444			317			237					
5866	14	57			40			32					

5866	3	27	566	1698		18	338	1015	14	250	749
5866	5	560				334			246		
5866	7	56				38			32		
5862	10	80	456	1368		35	183	550	27	140	419
5862	12	439				172			132		
5862	14	79				42			30		
5862	3	64	186	557		27	149	447	21	116	347
5862	5	148				133			105		
5862	7	67				43			32		
5860	10	47	183	550		48	81	244	36	61	184
5860	12	60				31			25		
5860	14	130				41			30		
5860	3	125	208	624		54	89	266	42	63	189
5860	5	141				53			38		
5860	7	75				35			23		
5986	10	493	598	1793		212	283	848	156	210	629
5986	12	134				82			62		
5986	14	220				119			89		
5986	3	178	294	882		109	205	616	90	159	478
5986	5	66				83			59		
5986	7	163				109			83		
5867	10	41	523	1568		29	372	1117	21	277	830
5867	12	521				372			276		
5867	14	25				11			14		
5867	3	40	628	1883		29	391	1172	21	286	857
5867	5	626				390			285		
5867	7	27				11			13		

POWER CONTROL/MONITOR - STANDOFFS									
BAR ELEMENT									
CRES 1/A HARD FTY=44000 PSI									
		X-LOAD				Y-LOAD			

6484	2	215	230	689	141	150	451	82	89	267
6484	3	215	230	689	141	150	451	82	89	267
6484	4	134	149	446	156	166	497	105	112	335
6484	5	134	149	446	156	166	497	105	112	335
6484	6	15			9			7		
6484	12	3979	3993	11980	985	995	2984	1179	1186	3559
6484	13	3979	3993	11980	985	995	2984	1179	1186	3559
6484	14	2152	2166	6499	1948	1957	5871	2343	2350	7051
6484	15	2152	2166	6499	1948	1957	5871	2343	2350	7051
6486	2	414	446	1337	280	297	890	206	220	660
6486	3	414	446	1337	280	297	890	206	220	660
6486	4	352	384	1151	187	203	610	160	174	521
6486	5	352	384	1151	187	203	610	160	174	521
6486	6	32			17			14		
6486	12	2236	2268	6803	939	956	2867	1067	1081	3244
6486	13	2236	2268	6803	939	956	2867	1067	1081	3244
6486	14	4119	4151	12453	2338	2355	7064	1820	1834	5502
6486	15	4119	4151	12453	2338	2355	7064	1820	1834	5502
6479	2	507	535	1606	306	321	963	228	239	718
6479	3	507	535	1606	306	321	963	228	239	718
6479	4	121	149	448	148	163	490	105	117	350
6479	5	121	149	448	148	163	490	105	117	350
6479	6	29			15			12		
6479	12	7043	7071	21214	1470	1485	4456	981	993	2979
6479	13	7043	7071	21214	1470	1485	4456	981	993	2979
6479	14	2295	2323	6970	1733	1748	5244	1365	1377	4130
6479	15	2295	2323	6970	1733	1748	5244	1365	1377	4130
6485	2	202	216	647	96	103	310	74	78	235
6485	3	202	216	647	96	103	310	74	78	235
6485	4	150	164	491	130	137	412	103	107	322
6485	5	150	164	491	130	137	412	103	107	322
6485	6	14			7			5		
6485	12	2776	2790	8369	974	981	2943	1320	1325	3974
6485	13	2776	2790	8369	974	981	2943	1320	1325	3974
6485	14	2431	2445	7335	2230	2237	6712	2255	2260	6780
6485	15	2431	2445	7335	2230	2237	6712	2255	2260	6780

6480	2	134	180	541	36	47	140	35	44	132
6480	3	134	180	541	36	47	140	35	44	132
6480	4	95	142	425	75	85	255	67	76	229
6480	5	95	142	425	75	85	255	67	76	229
6480	6	46			10			9		
6480	12	7190	7236	21709	1197	1208	3623	999	1007	3022
6480	13	7190	7236	21709	1197	1208	3623	999	1007	3022
6480	14	5498	5545	16634	1116	1126	3378	1319	1328	3984
6480	15	5498	5545	16634	1116	1126	3378	1319	1328	3984

RADIATOR PANEL																					
QUAD ELEMENT																					
6061-T6 FTY=35000 PSI																					
				X-LOAD								Y-LOAD								Z-LOAD	
ELEMENT ID.	STRESS ID.	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	
10065	3	200	617	1852	101	335	1004	54	439	1318											
10065	5	309			191			264						260							
10065	7	359			183																
10065	10	193	587	1762	132	322	967	79	449	1348											
10065	12	356			147			275													
10065	14	302			183			254													
10066	3	140	492	1476	76	257	771	33	330	990											
10066	5	147			70			123													
10066	7	349			184			248													
10066	10	148	490	1469	123	281	842	48	354	1062											
10066	12	212			56			133													
10066	14	308			188			260													
10070	3	156	547	1641	80	337	1011	54	371	1114											
10070	5	252			229			206													
10070	7	339			167			229													
10070	10	169	526	1578	68	298	895	35	360	1081											
10070	12	229			172			185													
10070	14	325			171			239													
10071	3	201	589	1767	90	339	1017	90	429	1287											
10071	5	201			169			222													
10071	7	388			206			265													
10071	10	217	554	1663	64	307	921	56	400	1199											
10071	12	185			142			196													
10071	14	353			200			265													
10113	3	118	667	2002	106	486	1457	48	556	1669											
10113	5	641			429			550													
10113	7	121			147			55													
10113	10	91	827	2480	100	503	1510	47	544	1631											
10113	12	782			438			534													

10113	14	182						162				70		
10058	3	91	679	2037				93	460	1380		37	488	1464
10058	5	625						386				477		
10058	7	178						165				69		
10058	10	76	870	2611				93	486	1457		33	536	1607
10058	12	837						415				524		
10058	14	163						167				75		
10097	3	1239	1557	4670				147	188	565		292	376	1128
10097	5	271						89				121		
10097	7	640						65				147		
10097	10	604	836	2509				61	145	436		165	248	745
10097	12	414						100				161		
10097	14	313						62				86		
10092	3	1972	2024	6071				243	271	812		497	567	1702
10092	5	767						116				242		
10092	7	254						65				151		
10092	10	1513	1562	4686				154	223	669		336	468	1403
10092	12	660						81				186		
10092	14	210						99				193		
10108	3	74	611	1832				101	365	1095		38	580	1739
10108	5	559						314				550		
10108	7	166						116				127		
10108	10	83	602	1806				109	360	1079		36	451	1352
10108	12	524						290				420		
10108	14	201						133				113		
10062	3	241	362	1086				199	243	728		61	344	1033
10062	5	282						175				317		
10062	7	98						54				88		
10062	10	170	375	1126				160	276	829		64	423	1269
10062	12	255						186				306		
10062	14	157						102				205		
10067	3	167	523	1570				145	344	1031		38	176	527
10067	5	307						210				89		
10067	7	278						163				110		
10067	10	108	450	1349				109	331	993		35	229	686
10067	12	269						211				119		
10067	14	248						163				146		

SIDEMOUNT													
QUAD ELEMENT													
6061-T6 FTY=35000 PSI													
		X-LOAD				Y-LOAD				Z-LOAD			
ELEMENT ID.	STRESS ID.	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS	RMS COMP STRESS	RMS PRIN STRESS	RMS 3 SIG STRESS
9376	10	62	765	2294	88	1143	3429	170	2412	7237			
9376	12	756			1125			2360					
9376	14	77			139			343					
9376	3	108	590	1771	161	938	2815	378	2142	6425			
9376	5	589			937			2141					
9376	7	25			35			20					
9377	10	257	1007	3022	191	1464	4392	417	3230	9691			
9377	12	633			967			2166					
9377	14	530			796			1731					
9377	3	69	610	1830	47	942	2826	186	2085	6256			
9377	5	364			577			1222					
9377	7	365			571			1280					
9378	10	79	862	2585	140	1234	3703	269	2650	7951			
9378	12	726			988			2060					
9378	14	326			540			1185					
9378	3	140	666	1997	249	972	2916	585	2114	6343			
9378	5	520			676			1514					
9378	7	277			463			958					
9335	10	281	482	1446	414	687	2062	879	1516	4548			
9335	12	108			204			335					
9335	14	274			363			867					
9335	3	145	344	1032	210	557	1670	470	1222	3666			
9335	5	184			233			602					
9335	7	179			335			683					
9337	10	267	454	1361	416	751	2253	913	1645	4936			
9337	12	256			397			905					
9337	14	192			344			737					

9337	3	298	568	1705	426	808	2423	927	1785	5355
9337	5	269			427			923		
9337	7	285			381			860		
8523	10	937	1053	3160	1226	1495	4486	1970	2412	7236
8523	12	762			1454			2227		
8523	14	184			105			286		
8523	3	429	822	2466	1537	1787	5360	2706	3036	9108
8523	5	546			1262			2314		
8523	7	329			361			488		
8524	10	564	971	2913	937	1466	4398	1658	2418	7254
8524	12	871			1229			2045		
8524	14	202			354			533		
8524	3	628	862	2587	1414	1619	4858	2229	2785	8355
8524	5	492			1578			2742		
8524	7	294			92			154		
10778	10	328	919	2756	1344	1616	4849	2367	2600	7800
10778	12	606			1498			2356		
10778	14	429			180			239		
10778	3	1194	1340	4021	1545	1698	5095	2493	2794	8383
10778	5	758			1295			2335		
10778	7	292			249			372		
10779	10	189	330	991	417	497	1491	779	860	2579
10779	12	120			213			265		
10779	14	172			150			219		
10779	3	433	445	1335	906	919	2758	1467	1490	4471
10779	5	58			160			367		
10779	7	67			100			163		
7509	10	645	771	2313	338	458	1373	722	851	2552
7509	12	230			122			249		
7509	14	262			200			278		
7509	3	342	881	2644	212	502	1506	351	943	2829
7509	5	273			153			300		
7509	7	573			318			617		
7510	10	396	769	2307	240	510	1530	428	776	2328
7510	12	183			87			227		
7510	14	468			338			437		

7510	3	545	1292	3875		302	735	2206		572	1424	4273
7510	5	142				130				114		
7510	7	926				512				1056		
7511	10	1002	1007	3022		578	596	1788		1066	1069	3208
7511	12	831				504				896		
7511	14	30				40				23		
7511	3	2188	2240	6719		1293	1333	3999		2361	2415	7246
7511	5	225				127				257		
7511	7	324				219				344		
7512	10	82	546	1637		42	364	1093		82	564	1691
7512	12	168				192				149		
7512	14	419				236				447		
7512	3	1280	1781	5343		761	1023	3070		1393	1963	5889
7512	5	385				180				475		
7512	7	837				471				921		
7877	10	424	1223	3670		303	987	2960		169	471	1413
7877	12	464				404				166		
7877	14	779				631				303		
7877	3	360	1476	4428		222	1151	3453		190	669	2007
7877	5	381				294				140		
7877	7	1105				892				503		
7977	10	399	432	1296		116	208	625		109	161	484
7977	12	197				191				115		
7977	14	88				40				49		
7977	3	79	988	2964		67	781	2342		47	420	1259
7977	5	741				567				318		
7977	7	474				390				194		
7406	10	2099	2172	6516		1675	1725	5175		2080	2159	6478
7406	12	137				127				99		
7406	14	384				282				404		
7406	3	2104	2198	6595		1658	1732	5197		1968	2022	6066
7406	5	108				105				121		
7406	7	443				349				321		
7407	10	1931	2265	6794		1556	1821	5462		1633	1974	5921
7407	12	695				476				570		
7407	14	724				596				691		

7407	3		1797	2185	6554		1400	1712	5136	1742	2181	6544
7407	5		694				433			725		
7407	7		760				632			801		
7408	10		618	2784	8353		438	2040	6120	517	2671	8013
7408	12		2640				1934			2519		
7408	14		560				412			571		
7408	3		499	2686	8058		323	1961	5882	275	2198	6595
7408	5		2598				1900			2170		
7408	7		439				316			232		
7409	10		3492	3540	10621		2579	2633	7900	3311	3327	9981
7409	12		3111				2329			2825		
7409	14		143				129			90		
7409	3		3522	3612	10836		2609	2667	8001	3123	3411	10233
7409	5		3224				2419			3124		
7409	7		187				120			287		
7410	10		2129	3300	9900		1667	2603	7810	2015	2853	8558
7410	12		3264				2589			2827		
7410	14		204				115			147		
7410	3		2199	3396	10188		1710	2657	7972	2135	3557	10670
7410	5		3343				2634			3501		
7410	7		252				150			281		
8996	10		505	1163	3489		958	1124	3372	1005	1182	3547
8996	12		467				733			881		
8996	14		677				255			231		
8996	3		227	1143	3429		692	999	2996	710	857	2572
8996	5		514				340			207		
8996	7		759				449			310		
8997	10		307	1521	4562		833	1744	5231	840	1691	5073
8997	12		684				1394			1490		
8997	14		1008				565			414		
8997	3		278	1461	4383		1085	1772	5315	1088	1702	5107
8997	5		240				705			618		
8997	7		1202				856			816		
9253	10		66	254	763		180	417	1252	185	490	1471
9253	12		126				325			350		
9253	14		156				148			207		

9253	3	170	442	1326	568	701	2104	631	788	2364
9253	5	122			387			615		
9253	7	295			204			165		
9257	10	167	357	1070	511	536	1608	503	564	1693
9257	12	244			72			182		
9257	14	146			106			154		
9257	3	259	456	1369	935	953	2858	940	965	2895
9257	5	122			573			712		
9257	7	257			82			80		
7845	10	1046	1332	3997	599	880	2639	870	967	2902
7845	12	181			271			254		
7845	14	574			413			263		
7845	3	514	559	1676	575	602	1807	308	331	992
7845	5	107			134			207		
7845	7	142			112			53		
7846	10	433	1441	4323	569	1371	4114	433	1658	4974
7846	12	950			541			1465		
7846	14	703			816			487		
7846	3	653	1203	3609	490	1000	2999	275	458	1375
7846	5	128			105			208		
7846	7	769			675			214		
7653	10	662	698	2093	686	719	2158	843	894	2683
7653	12	572			605			728		
7653	14	67			62			92		
7653	3	1875	1936	5809	1967	2031	6094	2379	2458	7374
7653	5	86			92			96		
7653	7	337			353			431		
7654	10	111	477	1430	108	495	1484	155	611	1832
7654	12	125			126			158		
7654	14	359			378			454		
7654	3	1215	1643	4930	1277	1723	5169	1536	2075	6226
7654	5	187			188			201		
7654	7	790			827			1005		
7655	10	297	1256	3767	288	1316	3948	364	1568	4704
7655	12	845			892			1053		
7655	14	628			660			787		

7655	3	746	1316	3947	773	1365	4094	960	1707	5121
7655	5	416			429			552		
7655	7	716			744			929		
7788	10	137	972	2915	149	998	2995	180	1255	3765
7788	12	963			988			1244		
7788	14	83			93			108		
7788	3	154	1186	3559	151	1243	3729	201	1494	4482
7788	5	1035			1092			1295		
7788	7	396			406			507		
7789	10	50	906	2719	19	933	2798	52	1143	3430
7789	12	176			187			214		
7789	14	791			826			1007		
7789	3	526	2042	6125	533	2132	6395	668	2601	7802
7789	5	1361			1420			1741		
7789	7	1016			1067			1289		
7790	10	155	531	1594	165	561	1682	183	673	2018
7790	12	42			53			59		
7790	14	429			448			548		
7790	3	1254	1627	4882	1305	1694	5082	1608	2078	6234
7790	5	208			186			270		
7790	7	727			766			922		
9694	10	238	579	1738	573	1030	3089	230	413	1240
9694	12	367			465			223		
9694	14	269			508			187		
9694	3	532	661	1984	798	1120	3359	352	501	1504
9694	5	126			409			264		
9694	7	263			478			189		
9695	10	173	346	1037	345	756	2268	135	353	1058
9695	12	329			674			192		
9695	14	53			184			187		
9695	3	140	757	2270	242	1189	3566	104	583	1749
9695	5	689			1125			530		
9695	7	204			245			160		
7785	10	142	976	2927	135	1014	3043	166	1241	3722
7785	12	282			269			365		
7785	14	760			809			970		

7785	3	138	570	1709	137	586	1759	169	713	2138
7785	5	213			244			263		
7785	7	392			392			495		
7513	10	162	1477	4430	143	932	2795	130	1520	4560
7513	12	1086			673			1123		
7513	14	716			451			742		
7513	3	461	1139	3417	281	643	1929	536	1298	3895
7513	5	522			263			617		
7513	7	647			371			720		

[illegible]

3988	3	453	1350	4051		200	534	1602	115	339	1018
3988	5	1289				523			326		
3988	7	235				61			55		
4005	10	266	551	1653		115	241	724	72	194	583
4005	12	383				204			143		
4005	14	219				69			80		
4005	3	171	490	1469		69	219	656	62	143	430
4005	5	368				166			111		
4005	7	197				89			51		
3865	10	500	878	2635		236	347	1041	132	312	936
3865	12	720				238			281		
3865	14	245				110			75		
3865	3	352	674	2021		165	236	708	128	266	799
3865	5	655				217			260		
3865	7	78				37			29		
3868	10	433	662	1985		187	280	841	125	269	806
3868	12	656				273			266		
3868	14	36				26			20		
3868	3	574	736	2208		250	308	925	189	287	861
3868	5	720				293			284		
3868	7	51				30			16		
3989	10	106	379	1137		67	270	810	64	257	770
3989	12	328				221			211		
3989	14	118				100			93		
3989	3	81	274	822		37	180	541	44	152	457
3989	5	185				114			111		
3989	7	131				97			67		
3972	10	87	535	1606		71	340	1019	60	320	960
3972	12	470				313			309		
3972	14	170				84			54		
3972	3	198	554	1663		124	335	1006	127	289	867
3972	5	389				275			268		
3972	7	242				113			59		
4139	10	961	2481	7442		576	1480	4441	444	1207	3621
4139	12	2470				1478			1203		
4139	14	125				42			53		

4139	3	1050	2574	7721	603	1538	4615	460	1256	3768
4139	5	2515			1510			1235		
4139	7	298			162			128		
4140	10	879	2441	7324	526	1462	4386	450	1213	3639
4140	12	2433			1459			1212		
4140	14	116			57			34		
4140	3	776	2493	7479	476	1502	4506	422	1245	3735
4140	5	2456			1470			1216		
4140	7	254			181			154		
3821	10	956	968	2904	588	591	1772	497	498	1494
3821	12	653			379			273		
3821	14	60			22			19		
3821	3	871	879	2637	536	539	1618	473	474	1423
3821	5	502			279			184		
3821	7	55			28			19		
3822	10	864	872	2615	499	500	1501	413	416	1249
3822	12	327			126			138		
3822	14	65			24			27		
3822	3	939	999	2996	548	565	1694	448	464	1391
3822	5	384			160			119		
3822	7	192			83			73		
* ELEMENTS 3865, 3868, 3821, 3822 2024-T4 FTY=40000 PSI										

[illegible]

Appendix E
RANDOM VIBRATION ACCELERATIONS & DISPLACEMENTS
(7 PERCENT CRITICAL DAMPING $Q=7$)

METSAT RANDOM VIBRATION ACCELERATIONS - TRANSMISSIBILITIES OF SELECTED ELEMENTS - X-LOAD												
COMPONENT	GRID	LOAD DIR	X-RESPONSE		Y-RESPONSE		Z-RESPONSE					
			RMS	GRMS	RMS	GRMS	RMS	GRMS				
									Q	Q	Q	
LARGE MASS	11911	X	3414	8.84	1.00	0	0.00	0	0.00	0.00		
	1001	X	3442	8.92	1.01	2801	7.26	1010	2.62	0.30		
	1172	X	2983	7.72	0.87	2921	7.57	4189	10.85	1.23		
	1019	X	3602	9.33	1.06	3086	7.99	1467	3.80	0.43		
	1205	X	2917	7.56	0.85	3485	9.03	2021	5.23	0.59		
	1124	X	2999	7.77	0.88	1342	3.48	3360	8.70	0.98		
LOWER MOTOR MOUNT	1001	X	3442	8.92	1.01	2801	7.26	1010	2.62	0.30		
	1172	X	2983	7.72	0.87	2921	7.57	4189	10.85	1.23		
	2663	X	3398	8.80	1.00	2640	6.84	4182	10.83	1.22		
	2875	X	3643	9.44	1.07	2322	6.01	1003	2.60	0.29		
	2855	X	8937	23.15	2.62	3251	8.42	3537	9.16	1.04		
	2861	X	5403	14.00	1.58	3199	8.28	2126	5.51	0.62		
	2888	X	2720	7.05	0.80	3093	8.01	6464	16.74	1.89		
	2892	X	2583	6.69	0.76	1866	4.83	4802	12.44	1.41		
	2896	X	2723	7.05	0.80	2633	6.82	5989	15.51	1.75		
	2884	X	2758	7.14	0.81	3599	9.32	6594	17.08	1.93		
UPPER MOTOR MOUNT	1578	X	3222	8.34	0.94	1685	4.36	804	2.08	0.24		
	1569	X	3093	8.01	0.91	1675	4.34	2274	5.89	0.67		
	2710	X	9192	23.81	2.69	2278	5.90	2669	6.91	0.78		
	2779	X	10854	28.11	3.18	2492	6.45	2350	6.09	0.69		
	2783	X	3425	8.87	1.00	2101	5.44	853	2.21	0.25		
	2764	X	3454	8.95	1.01	2075	5.37	1451	3.76	0.42		
	2696	X	2114	5.48	0.62	2414	6.25	8620	22.33	2.52		
	2700	X	2090	5.41	0.61	1595	4.13	5194	13.45	1.52		
	2704	X	2063	5.34	0.60	1908	4.94	5771	14.95	1.69		
	2692	X	2126	5.51	0.62	2586	6.70	7177	18.59	2.10		

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER FRONT PANEL	1455	X	3184	8.25	0.93	2311	5.99	0.68	1146	2.97	0.34
	1608	X	3125	8.09	0.92	2069	5.36	0.61	873	2.26	0.26
	1191	X	2965	7.68	0.87	1933	5.01	0.57	1253	3.25	0.37
	1005	X	3442	8.92	1.01	1696	4.39	0.50	768	1.99	0.23
	1979	X	928	2.40	0.27	2086	5.40	0.61	11441	29.63	3.35
	1970	X	2271	5.88	0.67	1053	2.73	0.31	4502	11.66	1.32
	1973	X	869	2.25	0.25	2151	5.57	0.63	7279	18.85	2.13
	1976	X	1916	4.96	0.56	751	1.94	0.22	4479	11.60	1.31
	1885	X	7343	19.02	2.15	1951	5.05	0.57	797	2.06	0.23
	1961	X	2994	7.75	0.88	2088	5.41	0.61	1236	3.20	0.36
LOWER AFT PANEL	1963	X	3446	8.93	1.01	2313	5.99	0.68	1221	3.16	0.36
	1315	X	3265	8.46	0.96	3661	9.48	1.07	2340	6.06	0.69
	1215	X	3166	8.20	0.93	3552	9.20	1.04	1493	3.87	0.44
	1019	X	3602	9.33	1.06	3086	7.99	0.90	1467	3.80	0.43
	1205	X	2917	7.56	0.85	3485	9.03	1.02	2021	5.23	0.59
	1260	X	3933	10.19	1.15	3584	9.28	1.05	1876	4.86	0.55
	1455	X	3184	8.25	0.93	2311	5.99	0.68	1146	2.97	0.34
	1608	X	3125	8.09	0.92	2069	5.36	0.61	873	2.26	0.26
	1468	X	3166	8.20	0.93	3590	9.30	1.05	1575	4.08	0.46
	1458	X	3265	8.46	0.96	3856	9.99	1.13	2535	6.57	0.74
UPPER FRONT PANEL	1375	X	3216	8.33	0.94	946	2.45	0.28	1282	3.32	0.38
	1528	X	3290	8.52	0.96	943	2.44	0.28	436	1.13	0.13
	1993	X	5359	13.88	1.57	2270	5.88	0.66	1525	3.95	0.45
	2124	X	3226	8.36	0.95	2338	6.06	0.68	382	0.99	0.11
	2047	X	968	2.51	0.28	1856	4.81	0.54	5068	13.13	1.48
	2037	X	2350	6.09	0.69	1116	2.89	0.33	5324	13.79	1.56
	2050	X	997	2.58	0.29	2236	5.79	0.65	6189	16.03	1.81
	2060	X	2065	5.35	0.60	818	2.12	0.24	4048	10.49	1.19
	2078	X	10729	27.79	3.14	1091	2.82	0.32	640	1.66	0.19

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
UPPER AFT PANEL	1468	X	3166	8.20	0.93	3590	9.30	1.05	1575	4.08	0.46
	1458	X	3265	8.46	0.96	3856	9.99	1.13	2535	6.57	0.74
	2210	X	5362	13.89	1.57	2930	7.59	0.86	2753	7.13	0.81
	2240	X	3158	8.18	0.92	2949	7.64	0.86	1469	3.80	0.43
	2218	X	3120	8.08	0.91	3850	9.97	1.13	2282	5.91	0.67
	2223	X	3589	9.30	1.05	3761	9.74	1.10	1916	4.96	0.56
	2219	X	5584	14.46	1.64	3647	9.45	1.07	2320	6.01	0.68
	2224	X	6362	16.48	1.86	3593	9.31	1.05	1931	5.00	0.57
LOWER SHELF	1826	X	4032	10.44	1.18	2429	6.29	0.71	1480	3.83	0.43
	1828	X	4059	10.51	1.19	2021	5.24	0.59	1705	4.42	0.50
	1777	X	3778	9.78	1.11	2413	6.25	0.71	2757	7.14	0.81
	1779	X	3777	9.78	1.11	2019	5.23	0.59	2346	6.08	0.69
	1744	X	3539	9.17	1.04	2984	7.73	0.87	2302	5.96	0.67
	1767	X	3582	9.28	1.05	1324	3.43	0.39	2016	5.22	0.59
	1671	X	3258	8.44	0.95	1469	3.81	0.43	1582	4.10	0.46
	1641	X	3169	8.21	0.93	1894	4.91	0.55	1380	3.57	0.40
	1703	X	3370	8.73	0.99	1284	3.33	0.38	2372	6.14	0.69
	1699	X	3322	8.60	0.97	1697	4.40	0.50	1790	4.64	0.52
	1647	X	3286	8.51	0.96	2559	6.63	0.75	2859	7.41	0.84
	1631	X	3266	8.46	0.96	1995	5.17	0.58	1897	4.91	0.56
	1842	X	4092	10.60	1.20	3230	8.36	0.95	1273	3.30	0.37
	1776	X	3739	9.68	1.10	2643	6.85	0.77	2814	7.29	0.82
	1800	X	3798	9.84	1.11	1575	4.08	0.46	1803	4.67	0.53
	1827	X	4115	10.66	1.21	2270	5.88	0.66	1537	3.98	0.45
	1732	X	3481	9.02	1.02	1962	5.08	0.57	2907	7.53	0.85
	1718	X	3386	8.77	0.99	1499	3.88	0.44	1767	4.58	0.52
	1783	X	3671	9.51	1.08	1363	3.53	0.40	1698	4.40	0.50
	1853	X	4054	10.50	1.19	1298	3.36	0.38	661	1.71	0.19
	1858	X	4084	10.58	1.20	1828	4.74	0.54	937	2.43	0.27
	1805	X	3718	9.63	1.09	1416	3.67	0.41	2595	6.72	0.76
	1625	X	3308	8.57	0.97	3551	9.20	1.04	1902	4.93	0.56
	1694	X	3355	8.69	0.98	2883	7.47	0.84	2792	7.23	0.82

COMPONENT	GRID	LOAD	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
UPPER SHELF		DIR									
	2572	X	3521	9.12	1.03	2709	7.02	0.79	829	2.15	0.24
	2577	X	3515	9.10	1.03	2037	5.28	0.60	416	1.08	0.12
	2547	X	3245	8.40	0.95	2915	7.55	0.85	2887	7.48	0.85
	2551	X	3235	8.38	0.95	2260	5.85	0.66	1588	4.11	0.47
	2520	X	3135	8.12	0.92	2712	7.03	0.79	3086	7.99	0.90
	2525	X	3129	8.10	0.92	2042	5.29	0.60	1156	2.99	0.34
	2469	X	3085	7.99	0.90	2044	5.30	0.60	2194	5.68	0.64
	2464	X	3077	7.97	0.90	2721	7.05	0.80	3244	8.40	0.95
	2489	X	3067	7.94	0.90	3205	8.30	0.94	4427	11.47	1.30
	2420	X	3170	8.21	0.93	3571	9.25	1.05	4644	12.03	1.36
	2426	X	3172	8.22	0.93	2887	7.48	0.85	2987	7.74	0.87
	2355	X	3549	9.19	1.04	3448	8.93	1.01	3092	8.01	0.91
	2360	X	3506	9.08	1.03	2892	7.49	0.85	2355	6.10	0.69
LOWER RIGHT FRONT SUP	2300	X	3959	10.25	1.16	3293	8.53	0.96	2352	6.09	0.69
	2323	X	3893	10.08	1.14	2276	5.90	0.67	2228	5.77	0.65
	2318	X	3887	10.07	1.14	2733	7.08	0.80	2233	5.78	0.65
	2275	X	4200	10.88	1.23	3212	8.32	0.94	2182	5.65	0.64
	2241	X	4600	11.91	1.35	2737	7.09	0.80	2039	5.28	0.60
	2442	X	3118	8.08	0.91	2570	6.66	0.75	3016	7.81	0.88
	1176	X	2953	7.65	0.86	1925	4.99	0.56	1217	3.15	0.36
	1963	X	3446	8.93	1.01	2313	5.99	0.68	1221	3.16	0.36
	1172	X	2983	7.72	0.87	2921	7.57	0.86	4189	10.85	1.23
LOWER RIGHT PANEL	2663	X	3398	8.80	1.00	2640	6.84	0.77	4182	10.83	1.22
	2661	X	3036	7.86	0.89	10225	26.48	2.99	2487	6.44	0.73
	1379	X	3120	8.08	0.91	1845	4.78	0.54	1214	3.15	0.36
	1191	X	2965	7.68	0.87	1933	5.01	0.57	1253	3.25	0.37
	1205	X	2917	7.56	0.85	3485	9.03	1.02	2021	5.23	0.59
	1315	X	3265	8.46	0.96	3661	9.48	1.07	2340	6.06	0.69
	1455	X	3184	8.25	0.93	2311	5.99	0.68	1146	2.97	0.34

COMPONENT	GRID	LOAD	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
UPPER RIGHT FRONT SUP		DIR									
	1998	X	4226	10.95	1.24	1840	4.76	0.54	1278	3.31	0.37
	1569	X	3093	8.01	0.91	1675	4.34	0.49	2274	5.89	0.67
	2691	X	4279	11.08	1.25	1923	4.98	0.56	2532	6.56	0.74
	1519	X	3103	8.04	0.91	953	2.47	0.28	1283	3.32	0.38
	2679	X	3586	9.29	1.05	4827	12.50	1.41	1439	3.73	0.42
UPPER RIGHT PANEL											
	1457	X	3265	8.46	0.96	3772	9.77	1.10	2451	6.35	0.72
	2154	X	3983	10.32	1.17	4146	10.74	1.21	2040	5.28	0.60
	2130	X	3953	10.24	1.16	6378	16.52	1.87	1524	3.95	0.45
	1345	X	3250	8.42	0.95	2311	5.99	0.68	1887	4.89	0.55
	1365	X	3221	8.34	0.94	1053	2.73	0.31	1385	3.59	0.41
	2210	X	5362	13.89	1.57	2930	7.59	0.86	2753	7.13	0.81
	1993	X	5359	13.88	1.57	2270	5.88	0.66	1525	3.95	0.45
	1375	X	3216	8.33	0.94	946	2.45	0.28	1282	3.32	0.38
TOP PANEL											
	11025	X	4470	11.58	1.31	2408	6.24	0.71	4146	10.74	1.21
	11038	X	4498	11.65	1.32	3023	7.83	0.89	1403	3.63	0.41
	11175	X	3315	8.59	0.97	3465	8.97	1.01	1392	3.61	0.41
	11170	X	13686	35.45	4.01	3438	8.90	1.01	1712	4.43	0.50
	2627	X	3132	8.11	0.92	1414	3.66	0.41	4491	11.63	1.32
	11128	X	9914	25.68	2.90	3424	8.87	1.00	1485	3.85	0.43
	2210	X	5362	13.89	1.57	2930	7.59	0.86	2753	7.13	0.81
	1993	X	5359	13.88	1.57	2270	5.88	0.66	1525	3.95	0.45
LEFT PANEL											
	3243	X	3029	7.85	0.89	7622	19.74	2.23	870	2.25	0.25
	3251	X	3084	7.99	0.90	4148	10.74	1.22	557	1.44	0.16
	1001	X	3442	8.92	1.01	2801	7.26	0.82	1010	2.62	0.30
	1019	X	3602	9.33	1.06	3086	7.99	0.90	1467	3.80	0.43
	1608	X	3125	8.09	0.92	2069	5.36	0.61	873	2.26	0.26
	2124	X	3226	8.36	0.95	2338	6.06	0.68	382	0.99	0.11
	2240	X	3158	8.18	0.92	2949	7.64	0.86	1469	3.80	0.43

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER LEFT SHIELD	1878	X	3630	9.40	1.06	1836	4.76	0.54	798	2.07	0.23
	2829	X	3558	9.22	1.04	2214	5.73	0.65	991	2.57	0.29
	1840	X	3861	10.00	1.13	1918	4.97	0.56	794	2.06	0.23
	2831	X	3701	9.59	1.08	2466	6.39	0.72	1682	4.36	0.49
	2913	X	3599	9.32	1.05	1634	4.23	0.48	2741	7.10	0.80
LOWER RIGHT SHIELD	1938	X	3429	8.88	1.00	2200	5.70	0.64	1070	2.77	0.31
	2835	X	3586	9.29	1.05	2640	6.84	0.77	3511	9.09	1.03
	2901	X	3419	8.86	1.00	2459	6.37	0.72	2546	6.60	0.75
	1963	X	3446	8.93	1.01	2313	5.99	0.68	1221	3.16	0.36
	2663	X	3398	8.80	1.00	2640	6.84	0.77	4182	10.83	1.22
UPPER LEFT SHIELD	2119	X	3403	8.81	1.00	1917	4.97	0.56	404	1.05	0.12
	2083	X	3084	7.99	0.90	2032	5.26	0.60	606	1.57	0.18
	2763	X	3251	8.42	0.95	1963	5.09	0.58	850	2.20	0.25
	2760	X	3276	8.48	0.96	2023	5.24	0.59	1466	3.80	0.43
	2933	X	3157	8.18	0.92	1591	4.12	0.47	3256	8.43	0.95
UPPER RIGHT SHIELD	2010	X	4154	10.76	1.22	1835	4.75	0.54	1239	3.21	0.36
	2918	X	4378	11.34	1.28	1926	4.99	0.56	2472	6.40	0.72
	1998	X	4226	10.95	1.24	1840	4.76	0.54	1278	3.31	0.37
	2691	X	4279	11.08	1.25	1923	4.98	0.56	2532	6.56	0.74
LOWER CARD CAGE	3673	X	4905	12.70	1.44	3065	7.94	0.90	2411	6.24	0.71
	3662	X	4459	11.55	1.31	5562	14.41	1.63	2008	5.20	0.59
	3671	X	5927	15.35	1.74	3047	7.89	0.89	2746	7.11	0.80
	3660	X	5961	15.44	1.75	5474	14.18	1.60	5809	15.05	1.70
	3655	X	2915	7.55	0.85	3196	8.28	0.94	1715	4.44	0.50
	3650	X	3359	8.70	0.98	3091	8.01	0.91	2529	6.55	0.74
	1179	X	2933	7.60	0.86	1582	4.10	0.46	1904	4.93	0.56
	1084	X	3141	8.13	0.92	1439	3.73	0.42	1970	5.10	0.58
	3735	X	5218	13.51	1.53	6652	17.23	1.95	1792	4.64	0.52
	3730	X	4920	12.74	1.44	3366	8.72	0.99	1635	4.23	0.48
	3764	X	4520	11.71	1.32	6264	16.22	1.83	1797	4.65	0.53

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER CARD CAGE	3711	X	3663	9.49	1.07	7289	18.88	2.14	2605	6.75	0.76
	3708	X	3560	9.22	1.04	5099	13.21	1.49	2070	5.36	0.61
	3753	X	3454	8.95	1.01	2133	5.53	0.62	1895	4.91	0.56
LOWER CARD CAGE CARDS	4845	X	6395	16.56	1.87	2910	7.54	0.85	3731	9.66	1.09
	4822	X	4624	11.98	1.35	3655	9.47	1.07	1686	4.37	0.49
	4826	X	5710	14.79	1.67	4999	12.95	1.46	10601	27.46	3.11
	4813	X	2885	7.47	0.85	5447	14.11	1.60	10733	27.80	3.14
	4817	X	3476	9.00	1.02	1910	4.95	0.56	1657	4.29	0.49
	4777	X	3461	8.96	1.01	1905	4.93	0.56	1733	4.49	0.51
	4725	X	6562	17.00	1.92	2725	7.06	0.80	1594	4.13	0.47
	4765	X	6627	17.17	1.94	2916	7.55	0.85	1564	4.05	0.46
UPPER CARD CAGE	1469	X	3212	8.32	0.94	2958	7.66	0.87	1769	4.58	0.52
	1509	X	3181	8.24	0.93	1055	2.73	0.31	1476	3.82	0.43
	1514	X	3071	7.95	0.90	1050	2.72	0.31	5229	13.54	1.53
	1474	X	3188	8.26	0.93	2962	7.67	0.87	5024	13.01	1.47
	3372	X	4095	10.61	1.20	4535	11.75	1.33	1893	4.90	0.55
	3381	X	4673	12.10	1.37	4546	11.77	1.33	5026	13.02	1.47
	3404	X	4766	12.34	1.40	4054	10.50	1.19	5318	13.77	1.56
	3395	X	4228	10.95	1.24	4006	10.38	1.17	1690	4.38	0.50
	2249	X	4504	11.67	1.32	3011	7.80	0.88	2112	5.47	0.62
	2448	X	3096	8.02	0.91	2996	7.76	0.88	3517	9.11	1.03
	3369	X	4329	11.21	1.27	4144	10.73	1.21	3242	8.40	0.95
UPPER CARD CAGE CARDS	11836	X	4202	10.88	1.23	3802	9.85	1.11	4366	11.31	1.28
	11251	X	4635	12.01	1.36	3725	9.65	1.09	5465	14.16	1.60
	11829	X	3189	8.26	0.93	2498	6.47	0.73	4425	11.46	1.30
	11203	X	3224	8.35	0.94	1753	4.54	0.51	5641	14.61	1.65
	11879	X	11250	29.14	3.30	2910	7.54	0.85	4791	12.41	1.40
	11312	X	11105	28.76	3.25	2835	7.34	0.83	4746	12.29	1.39

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER WARMLOAD	1945	X	3892	10.08	1.14	2096	5.43	0.61	1097	2.84	0.32
	1929	X	7299	18.91	2.14	2052	5.32	0.60	1001	2.59	0.29
	2812	X	9865	25.55	2.89	2557	6.62	0.75	2227	5.77	0.65
	2810	X	5315	13.77	1.56	2542	6.58	0.74	1690	4.38	0.49
	1892	X	7814	20.24	2.29	1884	4.88	0.55	786	2.03	0.23
	1908	X	5500	14.24	1.61	1901	4.92	0.56	843	2.18	0.25
	2814	X	3563	9.23	1.04	2665	6.90	0.78	3195	8.28	0.94
	2815	X	4205	10.89	1.23	2644	6.85	0.77	3691	9.56	1.08
UPPER WARMLOAD	3006	X	6896	17.86	2.02	5899	15.28	1.73	6896	17.86	2.02
	3036	X	4152	10.75	1.22	6203	16.07	1.82	7775	20.14	2.28
	2009	X	4193	10.86	1.23	1539	3.99	0.45	1273	3.30	0.37
	2033	X	4491	11.63	1.32	1536	3.98	0.45	1129	2.92	0.33
	2080	X	4737	12.27	1.39	1449	3.75	0.42	646	1.67	0.19
	2056	X	4812	12.46	1.41	1485	3.85	0.43	801	2.07	0.23
	2939	X	4224	10.94	1.24	1780	4.61	0.52	1498	3.88	0.44
	2941	X	4744	12.29	1.39	2315	6.00	0.68	2928	7.58	0.86
LOWER REFLECTOR	2738	X	4086	10.58	1.20	1832	4.74	0.54	2450	6.35	0.72
	2739	X	4784	12.39	1.40	1835	4.75	0.54	2274	5.89	0.67
	2976	X	4632	12.00	1.36	4143	10.73	1.21	2908	7.53	0.85
	2945	X	4228	10.95	1.24	3601	9.33	1.05	1647	4.27	0.48
	4662	X	5087	13.18	1.49	2777	7.19	0.81	3346	8.67	0.98
	4085	X	3413	8.84	1.00	2769	7.17	0.81	2746	7.11	0.80
	4087	X	3633	9.41	1.06	3534	9.15	1.04	3035	7.86	0.89
	4176	X	6002	15.54	1.76	12456	32.26	3.65	11457	29.67	3.36
POWER CNTRL/MON BKT	4096	X	23713	61.42	6.95	19403	50.25	5.68	22987	59.54	6.73
	5591	X	6492	16.81	1.90	3010	7.80	0.88	1771	4.59	0.52
	5600	X	7444	19.28	2.18	3286	8.51	0.96	1771	4.59	0.52
	5590	X	5388	13.95	1.58	3505	9.08	1.03	1780	4.61	0.52
	11803	X	6581	17.04	1.93	26606	68.91	7.79	5519	14.29	1.62

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
POWER CNTRL/MON PWB	5878	X	6654	17.23	1.95	5422	14.04	1.59	3592	9.30	1.05
	6204	X	10018	25.95	2.93	5183	13.43	1.52	4072	10.55	1.19
PWR CNTRL/MON STANDOFF	6247	X	3569	9.24	1.05	5535	14.34	1.62	4451	11.53	1.30
	6015	X	7632	19.77	2.24	5642	14.61	1.65	3441	8.91	1.01
	5932	X	3159	8.18	0.93	5738	14.86	1.68	3485	9.03	1.02
	5999	X	6719	17.40	1.97	4902	12.70	1.44	3636	9.42	1.06
RADIATOR PANEL	10963	X	13685	35.45	4.01	3459	8.96	1.01	2276	5.89	0.67
	10968	X	3313	8.58	0.97	3349	8.67	0.98	1471	3.81	0.43
	10974	X	7713	19.98	2.26	1475	3.82	0.43	1356	3.51	0.40
	10969	X	8409	21.78	2.46	1464	3.79	0.43	1210	3.13	0.35
	10917	X	6153	15.94	1.80	1761	4.56	0.52	1667	4.32	0.49
SIDEMOUNT	8313	X	3414	8.84	1.00	0	0.00	0.00	0	0.00	0.00
	8229	X	3318	8.59	0.97	163	0.42	0.05	233	0.60	0.07
	7796	X	3424	8.87	1.00	252	0.65	0.07	59	0.15	0.02
	7840	X	3414	8.84	1.00	0	0.00	0.00	0	0.00	0.00
	8178	X	2373	6.15	0.70	2073	5.37	0.61	2580	6.68	0.76
	9338	X	3591	9.30	1.05	1482	3.84	0.43	1093	2.83	0.32
	9576	X	3414	8.84	1.00	0	0.00	0.00	0	0.00	0.00
	7508	X	3403	8.81	1.00	1195	3.09	0.35	36	0.09	0.01
	7240	X	3414	8.84	1.00	0	0.00	0.00	0	0.00	0.00
	9642	X	3182	8.24	0.93	284	0.73	0.08	132	0.34	0.04
UPPER REFLECTOR	4629	X	7623	19.74	2.23	3197	8.28	0.94	3683	9.54	1.08
	4399	X	4607	11.93	1.35	2148	5.56	0.63	2134	5.53	0.63
	4401	X	4997	12.94	1.46	2462	6.38	0.72	3005	7.78	0.88
	4490	X	8004	20.73	2.34	9369	24.27	2.74	10662	27.62	3.12
	4410	X	24553	63.59	7.19	18831	48.77	5.52	27398	70.96	8.03

METSAT RANDOM VIBRATION ACCELERATIONS - TRANSMISSIBILITIES OF SELECTED ELEMENTS - Y-LOAD													
COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE				
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q		
LARGE MASS	11911	Y	0	0.00	0.00	3414	8.84	1.00	0	0.00	0.00		
LOWER BASEPLATE	1001	Y	497	1.29	0.15	5483	14.20	1.61	735	1.90	0.22		
	1172	Y	2008	5.20	0.59	5671	14.69	1.66	4215	10.92	1.23		
	1019	Y	606	1.57	0.18	4068	10.54	1.19	1325	3.43	0.39		
	1205	Y	1689	4.38	0.49	4377	11.34	1.28	2849	7.38	0.83		
	1124	Y	1409	3.65	0.41	4587	11.88	1.34	5346	13.85	1.57		
LOWER MOTOR MOUNT	1001	Y	497	1.29	0.15	5483	14.20	1.61	735	1.90	0.22		
	1172	Y	2008	5.20	0.59	5671	14.69	1.66	4215	10.92	1.23		
	2663	Y	1743	4.51	0.51	6207	16.08	1.82	4260	11.03	1.25		
	2875	Y	575	1.49	0.17	5917	15.33	1.73	742	1.92	0.22		
	2855	Y	3240	8.39	0.95	7532	19.51	2.21	3708	9.60	1.09		
	2861	Y	1839	4.76	0.54	7494	19.41	2.19	2182	5.65	0.64		
	2888	Y	2818	7.30	0.83	7329	18.98	2.15	1563	4.05	0.46		
	2892	Y	6314	16.35	1.85	1719	4.45	0.50	2919	7.56	0.85		
	2896	Y	2735	7.09	0.80	6006	15.56	1.76	1397	3.62	0.41		
	2884	Y	6649	17.22	1.95	3850	9.97	1.13	3072	7.96	0.90		
UPPER MOTOR MOUNT	1578	Y	643	1.67	0.19	4688	12.14	1.37	930	2.41	0.27		
	1569	Y	1806	4.68	0.53	4719	12.22	1.38	3155	8.17	0.92		
	2710	Y	3817	9.89	1.12	3887	10.07	1.14	3373	8.74	0.99		
	2779	Y	3778	9.79	1.11	3798	9.84	1.11	2889	7.48	0.85		
	2783	Y	1464	3.79	0.43	3591	9.30	1.05	911	2.36	0.27		
	2764	Y	1370	3.55	0.40	3892	10.08	1.14	1690	4.38	0.49		
	2696	Y	2546	6.59	0.75	3826	9.91	1.12	2893	7.49	0.85		
	2700	Y	3909	10.12	1.14	1835	4.75	0.54	2755	7.14	0.81		
	2704	Y	2507	6.49	0.73	4185	10.84	1.23	3122	8.09	0.91		
	2692	Y	3974	10.29	1.16	3241	8.39	0.95	3126	8.10	0.92		

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER FRONT PANEL	1455	Y	1853	4.80	0.54	4747	12.29	1.39	3053	7.91	0.89
	1608	Y	868	2.25	0.25	4357	11.28	1.28	1501	3.89	0.44
	1191	Y	2123	5.50	0.62	5419	14.03	1.59	2775	7.19	0.81
	1005	Y	492	1.27	0.14	5110	13.24	1.50	1435	3.72	0.42
	1979	Y	2332	6.04	0.68	5335	13.82	1.56	3511	9.09	1.03
	1970	Y	5615	14.54	1.64	2357	6.11	0.69	2027	5.25	0.59
	1973	Y	1945	5.04	0.57	4851	12.56	1.42	2920	7.56	0.86
	1976	Y	4852	12.57	1.42	1983	5.14	0.58	1218	3.15	0.36
LOWER AFT PANEL	1885	Y	2109	5.46	0.62	4998	12.95	1.46	1599	4.14	0.47
	1961	Y	1443	3.74	0.42	5559	14.40	1.63	2792	7.23	0.82
	1963	Y	1781	4.61	0.52	5710	14.79	1.67	2819	7.30	0.83
	1315	Y	2131	5.52	0.62	3942	10.21	1.15	3138	8.13	0.92
	1215	Y	470	1.22	0.14	3856	9.99	1.13	1244	3.22	0.36
	1019	Y	606	1.57	0.18	4068	10.54	1.19	1325	3.43	0.39
UPPER BASEPLATE	1205	Y	1689	4.38	0.49	4377	11.34	1.28	2849	7.38	0.83
	1260	Y	1391	3.60	0.41	4350	11.27	1.27	2128	5.51	0.62
	1455	Y	1853	4.80	0.54	4747	12.29	1.39	3053	7.91	0.89
	1608	Y	868	2.25	0.25	4357	11.28	1.28	1501	3.89	0.44
	1468	Y	470	1.22	0.14	3795	9.83	1.11	1267	3.28	0.37
UPPER FRONT PANEL	1458	Y	2131	5.52	0.62	3929	10.18	1.15	3196	8.28	0.94
	1375	Y	2020	5.23	0.59	4427	11.47	1.30	3449	8.93	1.01
	1528	Y	476	1.23	0.14	4248	11.00	1.24	901	2.33	0.26
	1993	Y	3098	8.02	0.91	3655	9.47	1.07	3559	9.22	1.04
	2124	Y	1516	3.93	0.44	3714	9.62	1.09	897	2.32	0.26
	2047	Y	2480	6.42	0.73	4099	10.62	1.20	3049	7.90	0.89
	2037	Y	4202	10.88	1.23	2957	7.66	0.87	3900	10.10	1.14
	2050	Y	2490	6.45	0.73	3885	10.06	1.14	7528	19.50	2.21
	2060	Y	4105	10.63	1.20	2036	5.27	0.60	3763	9.75	1.10
	2078	Y	9674	25.06	2.83	3977	10.30	1.16	1633	4.23	0.48

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
UPPER AFT PANEL	1468	Y	470	1.22	0.14	3795	9.83	1.11	1267	3.28	0.37
	1458	Y	2131	5.52	0.62	3929	10.18	1.15	3196	8.28	0.94
	2210	Y	3114	8.07	0.91	3619	9.37	1.06	3384	8.76	0.99
	2240	Y	1500	3.88	0.44	3576	9.26	1.05	1265	3.28	0.37
	2218	Y	1511	3.91	0.44	3882	10.05	1.14	2605	6.75	0.76
	2223	Y	1150	2.98	0.34	3831	9.92	1.12	1960	5.08	0.57
	2219	Y	1719	4.45	0.50	3573	9.25	1.05	2633	6.82	0.77
	2224	Y	1025	2.66	0.30	3546	9.18	1.04	1963	5.08	0.57
LOWER SHELF	1826	Y	707	1.83	0.21	5060	13.11	1.48	2438	6.31	0.71
	1828	Y	716	1.86	0.21	5246	13.59	1.54	3973	10.29	1.16
	1777	Y	905	2.34	0.26	5055	13.09	1.48	4421	11.45	1.30
	1779	Y	906	2.35	0.27	5242	13.58	1.54	5524	14.31	1.62
	1744	Y	1188	3.08	0.35	4771	12.36	1.40	2313	5.99	0.68
	1767	Y	1051	2.72	0.31	5611	14.53	1.64	3176	8.23	0.93
	1671	Y	1707	4.42	0.50	5511	14.28	1.61	3329	8.62	0.98
	1641	Y	1824	4.72	0.53	5946	15.40	1.74	3507	9.08	1.03
	1703	Y	1471	3.81	0.43	5743	14.87	1.68	6510	16.86	1.91
	1699	Y	1522	3.94	0.45	5376	13.92	1.57	2050	5.31	0.60
	1647	Y	1779	4.61	0.52	4979	12.90	1.46	4917	12.74	1.44
	1631	Y	1923	4.98	0.56	5234	13.56	1.53	3277	8.49	0.96
	1842	Y	639	1.66	0.19	4618	11.96	1.35	1087	2.82	0.32
	1776	Y	946	2.45	0.28	4944	12.80	1.45	3487	9.03	1.02
	1800	Y	895	2.32	0.26	5463	14.15	1.60	4855	12.58	1.42
	1827	Y	681	1.76	0.20	5138	13.31	1.50	2763	7.16	0.81
	1732	Y	1238	3.21	0.36	5255	13.61	1.54	4914	12.73	1.44
	1718	Y	1384	3.59	0.41	5480	14.19	1.61	2339	6.06	0.69
	1783	Y	974	2.52	0.29	5579	14.45	1.63	3614	9.36	1.06
	1853	Y	745	1.93	0.22	5615	14.54	1.64	1020	2.64	0.30
	1858	Y	959	2.48	0.28	5076	13.15	1.49	1574	4.08	0.46
	1805	Y	867	2.24	0.25	5595	14.49	1.64	1992	5.16	0.58
	1625	Y	2002	5.19	0.59	4539	11.76	1.33	2820	7.30	0.83
	1694	Y	1516	3.93	0.44	4832	12.52	1.42	3693	9.57	1.08

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
UPPER SHELF	2572	Y	1326	3.43	0.39	4588	11.88	1.34	822	2.13	0.24
	2577	Y	1357	3.51	0.40	4295	11.12	1.26	882	2.28	0.26
	2547	Y	1331	3.45	0.39	4645	12.03	1.36	2852	7.39	0.84
	2551	Y	1334	3.45	0.39	4434	11.48	1.30	1645	4.26	0.48
	2520	Y	1383	3.58	0.41	4593	11.90	1.35	3238	8.39	0.95
	2525	Y	1376	3.56	0.40	4300	11.14	1.26	1424	3.69	0.42
	2469	Y	1605	4.16	0.47	4302	11.14	1.26	2144	5.55	0.63
	2464	Y	1631	4.22	0.48	4603	11.92	1.35	3115	8.07	0.91
	2489	Y	1504	3.89	0.44	4711	12.20	1.38	4649	12.04	1.36
	2420	Y	1830	4.74	0.54	4746	12.29	1.39	4685	12.14	1.37
	2426	Y	1828	4.73	0.54	4640	12.02	1.36	2744	7.11	0.80
	2355	Y	2195	5.68	0.64	4724	12.24	1.38	3140	8.13	0.92
	2360	Y	2163	5.60	0.63	4647	12.04	1.36	2365	6.13	0.69
	2300	Y	2465	6.39	0.72	4709	12.20	1.38	2776	7.19	0.81
LOWER RIGHT FRONT SUP	2323	Y	2434	6.30	0.71	4402	11.40	1.29	3373	8.74	0.99
	2318	Y	2423	6.27	0.71	4631	12.00	1.36	2839	7.35	0.83
	2275	Y	2603	6.74	0.76	4701	12.18	1.38	2913	7.54	0.85
	2241	Y	2814	7.29	0.82	4634	12.00	1.36	3430	8.88	1.00
	2442	Y	1734	4.49	0.51	4575	11.85	1.34	2664	6.90	0.78
	1176	Y	2008	5.20	0.59	5409	14.01	1.58	2758	7.14	0.81
	1963	Y	1781	4.61	0.52	5710	14.79	1.67	2819	7.30	0.83
	1172	Y	2008	5.20	0.59	5671	14.69	1.66	4215	10.92	1.23
	2663	Y	1743	4.51	0.51	6207	16.08	1.82	4260	11.03	1.25
	2661	Y	1084	2.81	0.32	10269	26.60	3.01	2314	5.99	0.68
LOWER RIGHT PANEL	1379	Y	1953	5.06	0.57	4879	12.64	1.43	3424	8.87	1.00
	1191	Y	2123	5.50	0.62	5419	14.03	1.59	2775	7.19	0.81
	1205	Y	1689	4.38	0.49	4377	11.34	1.28	2849	7.38	0.83
	1315	Y	2131	5.52	0.62	3942	10.21	1.15	3138	8.13	0.92
	1455	Y	1853	4.80	0.54	4747	12.29	1.39	3053	7.91	0.89

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
UPPER RIGHT FRONT SUP	1998	Y	1610	4.17	0.47	4187	10.84	1.23	3294	8.53	0.96
	1569	Y	1806	4.68	0.53	4719	12.22	1.38	3155	8.17	0.92
	2691	Y	1615	4.18	0.47	4152	10.75	1.22	3264	8.45	0.96
	1519	Y	1877	4.86	0.55	4428	11.47	1.30	3294	8.53	0.96
	2679	Y	1673	4.33	0.49	27282	70.66	7.99	3199	8.29	0.94
UPPER RIGHT PANEL	1457	Y	2131	5.52	0.62	3934	10.19	1.15	3170	8.21	0.93
	2154	Y	2508	6.50	0.73	6105	15.81	1.79	3388	8.78	0.99
	2130	Y	2521	6.53	0.74	10172	26.35	2.98	3463	8.97	1.01
	1345	Y	2113	5.47	0.62	4096	10.61	1.20	3277	8.49	0.96
	1365	Y	2037	5.28	0.60	4363	11.30	1.28	3414	8.84	1.00
	2210	Y	3114	8.07	0.91	3619	9.37	1.06	3384	8.76	0.99
	1993	Y	3098	8.02	0.91	3655	9.47	1.07	3559	9.22	1.04
	1375	Y	2020	5.23	0.59	4427	11.47	1.30	3449	8.93	1.01
TOP PANEL	11025	Y	1697	4.40	0.50	3700	9.58	1.08	3083	7.98	0.90
	11038	Y	1672	4.33	0.49	3590	9.30	1.05	1159	3.00	0.34
	11175	Y	547	1.42	0.16	3903	10.11	1.14	1243	3.22	0.36
	11170	Y	3640	9.43	1.07	3898	10.10	1.14	1098	2.84	0.32
	2627	Y	1937	5.02	0.57	3479	9.01	1.02	4555	11.80	1.33
	11128	Y	3284	8.51	0.96	3461	8.96	1.01	1138	2.95	0.33
	2210	Y	3114	8.07	0.91	3619	9.37	1.06	3384	8.76	0.99
	1993	Y	3098	8.02	0.91	3655	9.47	1.07	3559	9.22	1.04
LEFT PANEL	3243	Y	611	1.58	0.18	16655	43.14	4.88	970	2.51	0.28
	3251	Y	736	1.91	0.22	18379	47.60	5.38	901	2.33	0.26
	1001	Y	497	1.29	0.15	5483	14.20	1.61	735	1.90	0.22
	1019	Y	606	1.57	0.18	4068	10.54	1.19	1325	3.43	0.39
	1608	Y	868	2.25	0.25	4357	11.28	1.28	1501	3.89	0.44
	2124	Y	1516	3.93	0.44	3714	9.62	1.09	897	2.32	0.26
	2240	Y	1500	3.88	0.44	3576	9.26	1.05	1265	3.28	0.37

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER LEFT SHIELD	1878	Y	535	1.39	0.16	4823	12.49	1.41	1416	3.67	0.41
	2829	Y	515	1.33	0.15	5606	14.52	1.64	725	1.88	0.21
	1840	Y	835	2.16	0.24	5036	13.04	1.47	1775	4.60	0.52
	2831	Y	1409	3.65	0.41	6003	15.55	1.76	1518	3.93	0.44
	2913	Y	732	1.90	0.21	5575	14.44	1.63	2095	5.43	0.61
LOWER RIGHT SHIELD	1938	Y	2150	5.57	0.63	5600	14.50	1.64	2552	6.61	0.75
	2835	Y	1753	4.54	0.51	6171	15.98	1.81	3699	9.58	1.08
	2901	Y	1824	4.72	0.53	6112	15.83	1.79	2782	7.21	0.81
	1963	Y	1781	4.61	0.52	5710	14.79	1.67	2819	7.30	0.83
	2663	Y	1743	4.51	0.51	6207	16.08	1.82	4260	11.03	1.25
UPPER LEFT SHIELD	2119	Y	1275	3.30	0.37	4033	10.45	1.18	908	2.35	0.27
	2083	Y	1423	3.69	0.42	4199	10.88	1.23	1610	4.17	0.47
	2763	Y	1204	3.12	0.35	3831	9.92	1.12	910	2.36	0.27
	2760	Y	1449	3.75	0.42	3931	10.18	1.15	1677	4.34	0.49
	2933	Y	1241	3.21	0.36	4054	10.50	1.19	3360	8.70	0.98
UPPER RIGHT SHIELD	2010	Y	1547	4.01	0.45	4183	10.83	1.23	3177	8.23	0.93
	2918	Y	1638	4.24	0.48	4149	10.75	1.22	3156	8.18	0.92
	1998	Y	1610	4.17	0.47	4187	10.84	1.23	3294	8.53	0.96
	2691	Y	1615	4.18	0.47	4152	10.75	1.22	3264	8.45	0.96
LOWER CARD CAGE	3673	Y	1823	4.72	0.53	4514	11.69	1.32	3196	8.28	0.94
	3662	Y	1405	3.64	0.41	4960	12.85	1.45	2819	7.30	0.83
	3671	Y	3061	7.93	0.90	4489	11.63	1.31	3725	9.65	1.09
	3660	Y	3487	9.03	1.02	5001	12.95	1.46	5468	14.16	1.60
	3655	Y	1673	4.33	0.49	4329	11.21	1.27	3111	8.06	0.91
	3650	Y	1070	2.77	0.31	4236	10.97	1.24	3256	8.43	0.95
	1179	Y	2008	5.20	0.59	4916	12.73	1.44	3374	8.74	0.99
	1084	Y	1038	2.69	0.30	4834	12.52	1.42	2725	7.06	0.80
	3735	Y	1576	4.08	0.46	10662	27.62	3.12	2580	6.68	0.76
	3730	Y	1350	3.50	0.40	11536	29.88	3.38	3693	9.57	1.08
	3764	Y	1765	4.57	0.52	10259	26.57	3.01	2685	6.95	0.79

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER CARD CAGE	3711	Y	971	2.51	0.28	12442	32.23	3.64	2905	7.52	0.85
	3708	Y	924	2.39	0.27	12429	32.19	3.64	2577	6.67	0.75
	3753	Y	1178	3.05	0.35	13611	35.25	3.99	2875	7.45	0.84
LOWER CARD CAGE CARDS	4845	Y	2083	5.39	0.61	4470	11.58	1.31	2654	6.87	0.78
	4822	Y	1434	3.71	0.42	4670	12.10	1.37	3259	8.44	0.95
	4826	Y	3231	8.37	0.95	4933	12.78	1.45	5120	13.26	1.50
	4813	Y	1846	4.78	0.54	5137	13.31	1.50	5114	13.24	1.50
	4817	Y	1056	2.74	0.31	4628	11.99	1.36	3110	8.06	0.91
	4777	Y	1058	2.74	0.31	4667	12.09	1.37	3070	7.95	0.90
	4725	Y	1291	3.34	0.38	4403	11.41	1.29	2567	6.65	0.75
	4765	Y	1252	3.24	0.37	4498	11.65	1.32	2631	6.81	0.77
UPPER CARD CAGE	1469	Y	2076	5.38	0.61	4012	10.39	1.18	3077	7.97	0.90
	1509	Y	1993	5.16	0.58	4363	11.30	1.28	3434	8.89	1.01
	1514	Y	1132	2.93	0.33	4299	11.13	1.26	7040	18.23	2.06
	1474	Y	1134	2.94	0.33	4005	10.37	1.17	6604	17.10	1.93
	3372	Y	2594	6.72	0.76	5106	13.23	1.50	3228	8.36	0.95
	3381	Y	2538	6.57	0.74	5119	13.26	1.50	6599	17.09	1.93
	3404	Y	2687	6.96	0.79	6322	16.37	1.85	7143	18.50	2.09
	3395	Y	2622	6.79	0.77	6276	16.25	1.84	3502	9.07	1.03
	2249	Y	2764	7.16	0.81	4679	12.12	1.37	3265	8.46	0.96
	2448	Y	1693	4.38	0.50	4659	12.07	1.36	3439	8.91	1.01
	3369	Y	2459	6.37	0.72	5284	13.69	1.55	3106	8.04	0.91
UPPER CARD CAGE CARDS	11836	Y	2623	6.79	0.77	5436	14.08	1.59	6525	16.90	1.91
	11251	Y	2579	6.68	0.76	5642	14.61	1.65	7410	19.19	2.17
	11829	Y	2024	5.24	0.59	5124	13.27	1.50	6606	17.11	1.93
	11203	Y	1169	3.03	0.34	4256	11.02	1.25	7655	19.83	2.24
	11879	Y	2585	6.70	0.76	5104	13.22	1.50	6902	17.88	2.02
	11312	Y	2586	6.70	0.76	5184	13.43	1.52	6695	17.34	1.96

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER WARMLOAD	1945	Y	3822	9.90	1.12	5565	14.41	1.63	2612	6.77	0.77
	1929	Y	4964	12.86	1.45	5489	14.22	1.61	2494	6.46	0.73
	2812	Y	8406	21.77	2.46	5758	14.91	1.69	2053	5.32	0.60
	2810	Y	3413	8.84	1.00	5675	14.70	1.66	1381	3.58	0.40
	1892	Y	5702	14.77	1.67	5222	13.52	1.53	1856	4.81	0.54
	1908	Y	5154	13.35	1.51	5288	13.70	1.55	2103	5.45	0.62
	2814	Y	8322	21.55	2.44	5963	15.44	1.75	3281	8.50	0.96
	2815	Y	1592	4.12	0.47	6008	15.56	1.76	3807	9.86	1.12
UPPER WARMLOAD	3006	Y	4755	12.31	1.39	12098	31.33	3.54	7094	18.37	2.08
	3036	Y	2886	7.48	0.85	14874	38.53	4.36	10554	27.34	3.09
	2009	Y	1662	4.30	0.49	4189	10.85	1.23	3227	8.36	0.95
	2033	Y	2439	6.32	0.71	4174	10.81	1.22	2865	7.42	0.84
	2080	Y	3909	10.12	1.14	4005	10.37	1.17	1706	4.42	0.50
	2056	Y	6604	17.11	1.93	4069	10.54	1.19	2074	5.37	0.61
	2939	Y	3952	10.24	1.16	4205	10.89	1.23	1726	4.47	0.51
	2941	Y	2458	6.37	0.72	4829	12.51	1.41	6913	17.90	2.02
LOWER REFLECTOR	2738	Y	1764	4.57	0.52	4274	11.07	1.25	3176	8.23	0.93
	2739	Y	2369	6.14	0.69	4270	11.06	1.25	2852	7.39	0.84
	2976	Y	4273	11.07	1.25	9454	24.49	2.77	5282	13.68	1.55
	2945	Y	1719	4.45	0.50	6335	16.41	1.86	8691	22.51	2.55
	4662	Y	2248	5.82	0.66	7374	19.10	2.16	2711	7.02	0.79
	4085	Y	794	2.06	0.23	6605	17.11	1.93	2801	7.26	0.82
	4087	Y	982	2.54	0.29	7510	19.45	2.20	3101	8.03	0.91
	4176	Y	2367	6.13	0.69	9162	23.73	2.68	4225	10.94	1.24
POWER CNTRL/MON BKT	4096	Y	7434	19.26	2.18	12281	31.81	3.60	6833	17.70	2.00
	5591	Y	2996	7.76	0.88	3525	9.13	1.03	1709	4.43	0.50
	5600	Y	2738	7.09	0.80	3449	8.93	1.01	1705	4.42	0.50
	5590	Y	1152	2.98	0.34	3500	9.06	1.03	1711	4.43	0.50
	11803	Y	4997	12.94	1.46	14760	38.23	4.32	1944	5.04	0.57

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
POWER CNTRL/MON PWB	5878	Y	5786	14.99	1.69	4390	11.37	1.29	2771	7.18	0.81
	6204	Y	5990	15.52	1.75	4511	11.68	1.32	3539	9.17	1.04
PWR CNTRL/MON STANDOFF	6247	Y	1999	5.18	0.59	4345	11.25	1.27	3823	9.90	1.12
	6015	Y	1530	3.96	0.45	4626	11.98	1.36	2902	7.52	0.85
	5932	Y	1951	5.05	0.57	4408	11.42	1.29	2833	7.34	0.83
	5999	Y	2269	5.88	0.66	4421	11.45	1.29	2937	7.61	0.86
RADIATOR PANEL	10963	Y	3640	9.43	1.07	3856	9.99	1.13	1516	3.93	0.44
	10968	Y	546	1.41	0.16	3832	9.93	1.12	1236	3.20	0.36
	10974	Y	3636	9.42	1.06	3124	8.09	0.92	1272	3.30	0.37
	10969	Y	4068	10.54	1.19	3117	8.07	0.91	792	2.05	0.23
	10917	Y	7476	19.36	2.19	3022	7.83	0.89	1208	3.13	0.35
SIDEMOUNT	8313	Y	0	0.00	0.00	3414	8.84	1.00	0	0.00	0.00
	8229	Y	240	0.62	0.07	3103	8.04	0.91	220	0.57	0.06
	7796	Y	149	0.39	0.04	3238	8.39	0.95	72	0.19	0.02
	7840	Y	0	0.00	0.00	3414	8.84	1.00	0	0.00	0.00
	8178	Y	3214	8.32	0.94	745	1.93	0.22	1693	4.38	0.50
	9338	Y	1354	3.51	0.40	2606	6.75	0.76	2050	5.31	0.60
	9576	Y	0	0.00	0.00	3414	8.84	1.00	0	0.00	0.00
	7508	Y	35	0.09	0.01	4963	12.85	1.45	43	0.11	0.01
UPPER REFLECTOR	7240	Y	0	0.00	0.00	3414	8.84	1.00	0	0.00	0.00
	9642	Y	328	0.85	0.10	3126	8.10	0.92	273	0.71	0.08
	4629	Y	7092	18.37	2.08	4613	11.95	1.35	2714	7.03	0.79
	4399	Y	2125	5.50	0.62	3974	10.29	1.16	2551	6.61	0.75
	4401	Y	2751	7.13	0.81	4449	11.52	1.30	2911	7.54	0.85
	4490	Y	3645	9.44	1.07	4766	12.35	1.40	4640	12.02	1.36
	4410	Y	11756	30.45	3.44	11646	30.17	3.41	11901	30.83	3.49

METSAT RANDOM VIBRATION ACCELERATIONS - TRANSMISSIBILITIES OF SELECTED ELEMENTS - Z-LOAD													
COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE				
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q		
LARGE MASS	11911	Z	0	0.00	0.00	0	0.00	0.00	3414	8.84	1.00		
LOWER BASEPLATE	1001	Z	346	0.90	0.10	2463	6.38	0.72	3350	8.68	0.98		
	1172	Z	724	1.88	0.21	2571	6.66	0.75	4427	11.47	1.30		
	1019	Z	408	1.06	0.12	2263	5.86	0.66	3417	8.85	1.00		
	1205	Z	848	2.20	0.25	2456	6.36	0.72	3027	7.84	0.89		
	1124	Z	575	1.49	0.17	2298	5.95	0.67	6781	17.56	1.99		
LOWER MOTOR MOUNT	1001	Z	346	0.90	0.10	2463	6.38	0.72	3350	8.68	0.98		
	1172	Z	724	1.88	0.21	2571	6.66	0.75	4427	11.47	1.30		
	2663	Z	2509	6.50	0.73	1010	2.62	0.30	4456	11.54	1.31		
	2875	Z	604	1.57	0.18	1758	4.55	0.51	3328	8.62	0.97		
	2855	Z	2901	7.51	0.85	3556	9.21	1.04	3549	9.19	1.04		
	2861	Z	1912	4.95	0.56	3452	8.94	1.01	2655	6.88	0.78		
	2888	Z	2824	7.32	0.83	3105	8.04	0.91	1674	4.33	0.49		
	2892	Z	1448	3.75	0.42	2810	7.28	0.82	1776	4.60	0.52		
	2896	Z	2806	7.27	0.82	1095	2.84	0.32	2008	5.20	0.59		
	2884	Z	1576	4.08	0.46	3723	9.64	1.09	2792	7.23	0.82		
UPPER MOTOR MOUNT	1578	Z	417	1.08	0.12	1696	4.39	0.50	3529	9.14	1.03		
	1569	Z	911	2.36	0.27	1692	4.38	0.50	2892	7.49	0.85		
	2710	Z	2853	7.39	0.84	3854	9.98	1.13	3033	7.85	0.89		
	2779	Z	2818	7.30	0.83	4515	11.69	1.32	2822	7.31	0.83		
	2783	Z	1025	2.66	0.30	3677	9.52	1.08	3458	8.96	1.01		
	2764	Z	1453	3.76	0.43	3249	8.42	0.95	3023	7.83	0.89		
	2696	Z	2784	7.21	0.82	4254	11.02	1.25	1993	5.16	0.58		
	2700	Z	3268	8.47	0.96	2969	7.69	0.87	1790	4.64	0.52		
	2704	Z	2761	7.15	0.81	2526	6.54	0.74	1705	4.42	0.50		
	2692	Z	3345	8.66	0.98	2952	7.65	0.86	2126	5.51	0.62		

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER FRONT PANEL	1455	Z	973	2.52	0.28	1333	3.45	0.39	3035	7.86	0.89
	1608	Z	483	1.25	0.14	1142	2.96	0.33	3840	9.95	1.12
	1191	Z	707	1.83	0.21	2614	6.77	0.77	3356	8.69	0.98
	1005	Z	333	0.86	0.10	2306	5.97	0.68	3828	9.91	1.12
	1979	Z	3079	7.98	0.90	1698	4.40	0.50	1814	4.70	0.53
	1970	Z	1713	4.44	0.50	3145	8.14	0.92	2052	5.31	0.60
	1973	Z	3169	8.21	0.93	1354	3.51	0.40	2960	7.67	0.87
	1976	Z	1355	3.51	0.40	3356	8.69	0.98	1604	4.15	0.47
	1885	Z	714	1.85	0.21	1621	4.20	0.47	3873	10.03	1.13
	1961	Z	1057	2.74	0.31	2436	6.31	0.71	3306	8.56	0.97
LOWER AFT PANEL	1963	Z	2601	6.74	0.76	1951	5.05	0.57	3268	8.46	0.96
	1315	Z	1265	3.28	0.37	1572	4.07	0.46	3045	7.89	0.89
	1215	Z	386	1.00	0.11	1516	3.93	0.44	3435	8.90	1.01
	1019	Z	408	1.06	0.12	2263	5.86	0.66	3417	8.85	1.00
	1205	Z	848	2.20	0.25	2456	6.36	0.72	3027	7.84	0.89
	1260	Z	773	2.00	0.23	1515	3.92	0.44	2786	7.22	0.82
	1455	Z	973	2.52	0.28	1333	3.45	0.39	3035	7.86	0.89
	1608	Z	483	1.25	0.14	1142	2.96	0.33	3840	9.95	1.12
	1468	Z	386	1.00	0.11	1489	3.86	0.44	3454	8.94	1.01
	1458	Z	1265	3.28	0.37	1633	4.23	0.48	3073	7.96	0.90
UPPER BASEPLATE	1375	Z	1087	2.82	0.32	1919	4.97	0.56	3134	8.12	0.92
	1528	Z	385	1.00	0.11	1828	4.73	0.54	3440	8.91	1.01
	1993	Z	2219	5.75	0.65	3944	10.22	1.16	3270	8.47	0.96
	2124	Z	1362	3.53	0.40	3874	10.04	1.13	3510	9.09	1.03
	2047	Z	2742	7.10	0.80	2889	7.48	0.85	2243	5.81	0.66
	2037	Z	3330	8.62	0.98	2848	7.38	0.83	2961	7.67	0.87
	2050	Z	2765	7.16	0.81	3491	9.04	1.02	6397	16.57	1.87
	2060	Z	3240	8.39	0.95	2784	7.21	0.82	2954	7.65	0.87
	2078	Z	4163	10.78	1.22	1957	5.07	0.57	2995	7.76	0.88
UPPER FRONT PANEL	1375	Z	1087	2.82	0.32	1919	4.97	0.56	3134	8.12	0.92
	1528	Z	385	1.00	0.11	1828	4.73	0.54	3440	8.91	1.01
	1993	Z	2219	5.75	0.65	3944	10.22	1.16	3270	8.47	0.96
	2124	Z	1362	3.53	0.40	3874	10.04	1.13	3510	9.09	1.03
	2047	Z	2742	7.10	0.80	2889	7.48	0.85	2243	5.81	0.66
	2037	Z	3330	8.62	0.98	2848	7.38	0.83	2961	7.67	0.87
	2050	Z	2765	7.16	0.81	3491	9.04	1.02	6397	16.57	1.87
	2060	Z	3240	8.39	0.95	2784	7.21	0.82	2954	7.65	0.87
	2078	Z	4163	10.78	1.22	1957	5.07	0.57	2995	7.76	0.88

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
UPPER AFT PANEL	1468	Z	386	1.00	0.11	1489	3.86	0.44	3454	8.94	1.01
	1458	Z	1265	3.28	0.37	1633	4.23	0.48	3073	7.96	0.90
	2210	Z	2196	5.69	0.64	4652	12.05	1.36	3056	7.91	0.90
	2240	Z	1353	3.50	0.40	4702	12.18	1.38	3232	8.37	0.95
	2218	Z	943	2.44	0.28	1744	4.52	0.51	2798	7.25	0.82
	2223	Z	1149	2.98	0.34	1703	4.41	0.50	2931	7.59	0.86
	2219	Z	1491	3.86	0.44	2894	7.50	0.85	2806	7.27	0.82
	2224	Z	984	2.55	0.29	2850	7.38	0.83	2925	7.57	0.86
LOWER SHELF	1826	Z	619	1.60	0.18	1744	4.52	0.51	4348	11.26	1.27
	1828	Z	619	1.60	0.18	1816	4.70	0.53	5546	14.36	1.62
	1777	Z	598	1.55	0.18	1742	4.51	0.51	5791	15.00	1.70
	1779	Z	597	1.55	0.17	1815	4.70	0.53	6049	15.67	1.77
	1744	Z	655	1.70	0.19	1650	4.27	0.48	5009	12.97	1.47
	1767	Z	619	1.60	0.18	1972	5.11	0.58	5916	15.32	1.73
	1671	Z	886	2.29	0.26	1933	5.01	0.57	5212	13.50	1.53
	1641	Z	932	2.41	0.27	2051	5.31	0.60	3325	8.61	0.97
	1703	Z	768	1.99	0.22	2036	5.27	0.60	6094	15.79	1.79
	1699	Z	792	2.05	0.23	1873	4.85	0.55	7081	18.34	2.07
	1647	Z	932	2.41	0.27	1718	4.45	0.50	5877	15.22	1.72
	1631	Z	1012	2.62	0.30	1817	4.71	0.53	4809	12.46	1.41
	1842	Z	648	1.68	0.19	1606	4.16	0.47	3384	8.77	0.99
	1776	Z	602	1.56	0.18	1704	4.41	0.50	5683	14.72	1.66
	1800	Z	598	1.55	0.18	1910	4.95	0.56	6046	15.66	1.77
	1827	Z	635	1.64	0.19	1774	4.60	0.52	4875	12.63	1.43
	1732	Z	672	1.74	0.20	1821	4.72	0.53	7239	18.75	2.12
	1718	Z	729	1.89	0.21	1915	4.96	0.56	7147	18.51	2.09
	1783	Z	605	1.57	0.18	1959	5.07	0.57	5858	15.17	1.72
	1853	Z	592	1.53	0.17	1967	5.10	0.58	3754	9.72	1.10
	1858	Z	644	1.67	0.19	1531	3.97	0.45	3969	10.28	1.16
	1805	Z	588	1.52	0.17	1919	4.97	0.56	5044	13.06	1.48
	1625	Z	1068	2.77	0.31	1609	4.17	0.47	3184	8.25	0.93
	1694	Z	792	2.05	0.23	1671	4.33	0.49	5111	13.24	1.50

COMPONENT	GRID	LOAD	X-RESPONSE			Y-RESPONSE			Z-RESPONSE			
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q	
UPPER SHELF												
	2572	Z	923	2.39	0.27	3392	8.79	0.99	3128	8.10	0.92	
	2577	Z	916	2.37	0.27	3253	8.43	0.95	3372	8.73	0.99	
	2547	Z	964	2.50	0.28	3409	8.83	1.00	3472	8.99	1.02	
	2551	Z	955	2.47	0.28	3331	8.63	0.98	3655	9.47	1.07	
	2520	Z	999	2.59	0.29	3396	8.79	0.99	4195	10.86	1.23	
	2525	Z	994	2.57	0.29	3257	8.44	0.95	3524	9.13	1.03	
	2469	Z	1151	2.98	0.34	3259	8.44	0.95	4397	11.39	1.29	
	2464	Z	1117	2.89	0.33	3403	8.81	1.00	4884	12.65	1.43	
	2489	Z	1061	2.75	0.31	3412	8.84	1.00	4841	12.54	1.42	
	2420	Z	1184	3.07	0.35	3367	8.72	0.99	6961	18.03	2.04	
	2426	Z	1185	3.07	0.35	3401	8.81	1.00	4450	11.53	1.30	
	2355	Z	1339	3.47	0.39	3366	8.72	0.99	5612	14.54	1.64	
	2360	Z	1323	3.43	0.39	3403	8.81	1.00	3757	9.73	1.10	
2300	Z	1477	3.83	0.43	3381	8.76	0.99	3757	9.73	1.10		
2323	Z	1475	3.82	0.43	3363	8.71	0.99	3516	9.11	1.03		
2318	Z	1455	3.77	0.43	3427	8.88	1.00	3631	9.41	1.06		
2275	Z	1552	4.02	0.45	3389	8.78	0.99	3290	8.52	0.96		
2241	Z	1670	4.33	0.49	3429	8.88	1.00	3179	8.23	0.93		
2442	Z	1156	2.99	0.34	3411	8.84	1.00	5429	14.06	1.59		
LOWER RIGHT FRONT SUP												
	1176	Z	671	1.74	0.20	2604	6.74	0.76	3278	8.49	0.96	
	1963	Z	2601	6.74	0.76	1951	5.05	0.57	3268	8.46	0.96	
	1172	Z	724	1.88	0.21	2571	6.66	0.75	4427	11.47	1.30	
	2663	Z	2509	6.50	0.73	1010	2.62	0.30	4456	11.54	1.31	
	2661	Z	1708	4.42	0.50	5969	15.46	1.75	3408	8.83	1.00	
LOWER RIGHT PANEL												
	1379	Z	944	2.45	0.28	2380	6.16	0.70	3281	8.50	0.96	
	1191	Z	707	1.83	0.21	2614	6.77	0.77	3356	8.69	0.98	
	1205	Z	848	2.20	0.25	2456	6.36	0.72	3027	7.84	0.89	
	1315	Z	1265	3.28	0.37	1572	4.07	0.46	3045	7.89	0.89	
	1455	Z	973	2.52	0.28	1333	3.45	0.39	3035	7.86	0.89	

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
UPPER RIGHT FRONT SUP	1998	Z	976	2.53	0.29	2827	7.32	0.83	2987	7.74	0.87
	1569	Z	911	2.36	0.27	1692	4.38	0.50	2892	7.49	0.85
	2691	Z	986	2.55	0.29	2633	6.82	0.77	2939	7.61	0.86
	1519	Z	982	2.54	0.29	1924	4.98	0.56	2991	7.75	0.88
	2679	Z	882	2.28	0.26	10332	26.76	3.03	2924	7.57	0.86
UPPER RIGHT PANEL	1457	Z	1265	3.28	0.37	1606	4.16	0.47	3061	7.93	0.90
	2154	Z	1439	3.73	0.42	5030	13.03	1.47	3146	8.15	0.92
	2130	Z	1422	3.68	0.42	6546	16.95	1.92	3118	8.08	0.91
	1345	Z	1220	3.16	0.36	1374	3.56	0.40	3106	8.04	0.91
	1365	Z	1116	2.89	0.33	1810	4.69	0.53	3125	8.09	0.92
	2210	Z	2196	5.69	0.64	4652	12.05	1.36	3056	7.91	0.90
	1993	Z	2219	5.75	0.65	3944	10.22	1.16	3270	8.47	0.96
	1375	Z	1087	2.82	0.32	1919	4.97	0.56	3134	8.12	0.92
TOP PANEL	11025	Z	2487	6.44	0.73	3863	10.01	1.13	12735	32.99	3.73
	11038	Z	2460	6.37	0.72	4805	12.45	1.41	4293	11.12	1.26
	11175	Z	450	1.16	0.13	1499	3.88	0.44	3278	8.49	0.96
	11170	Z	3569	9.25	1.05	1521	3.94	0.45	4399	11.39	1.29
	2627	Z	1224	3.17	0.36	4115	10.66	1.21	8791	22.77	2.57
	11128	Z	3055	7.91	0.89	2904	7.52	0.85	4325	11.20	1.27
	2210	Z	2196	5.69	0.64	4652	12.05	1.36	3056	7.91	0.90
	1993	Z	2219	5.75	0.65	3944	10.22	1.16	3270	8.47	0.96
LEFT PANEL	3243	Z	484	1.25	0.14	5761	14.92	1.69	3310	8.57	0.97
	3251	Z	515	1.33	0.15	8080	20.93	2.37	3478	9.01	1.02
	1001	Z	346	0.90	0.10	2463	6.38	0.72	3350	8.68	0.98
	1019	Z	408	1.06	0.12	2263	5.86	0.66	3417	8.85	1.00
	1608	Z	483	1.25	0.14	1142	2.96	0.33	3840	9.95	1.12
	2124	Z	1362	3.53	0.40	3874	10.04	1.13	3510	9.09	1.03
	2240	Z	1353	3.50	0.40	4702	12.18	1.38	3232	8.37	0.95

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER LEFT SHIELD	1878	Z	449	1.16	0.13	1380	3.57	0.40	3823	9.90	1.12
	2829	Z	468	1.21	0.14	1001	2.59	0.29	3316	8.59	0.97
	1840	Z	585	1.52	0.17	1493	3.87	0.44	3608	9.35	1.06
	2831	Z	768	1.99	0.23	1126	2.92	0.33	2860	7.41	0.84
	2913	Z	549	1.42	0.16	913	2.36	0.27	5733	14.85	1.68
LOWER RIGHT SHIELD	1938	Z	2844	7.37	0.83	1881	4.87	0.55	3077	7.97	0.90
	2835	Z	2598	6.73	0.76	1019	2.64	0.30	3596	9.31	1.05
	2901	Z	2613	6.77	0.77	1169	3.03	0.34	4386	11.36	1.28
	1963	Z	2601	6.74	0.76	1951	5.05	0.57	3268	8.46	0.96
	2663	Z	2509	6.50	0.73	1010	2.62	0.30	4456	11.54	1.31
UPPER LEFT SHIELD	2119	Z	840	2.17	0.25	3047	7.89	0.89	3441	8.91	1.01
	2083	Z	1035	2.68	0.30	3190	8.26	0.93	2965	7.68	0.87
	2763	Z	775	2.01	0.23	2998	7.77	0.88	3470	8.99	1.02
	2760	Z	1283	3.32	0.38	3046	7.89	0.89	3023	7.83	0.89
	2933	Z	958	2.48	0.28	2922	7.57	0.86	8697	22.53	2.55
UPPER RIGHT SHIELD	2010	Z	1094	2.83	0.32	2825	7.32	0.83	2933	7.60	0.86
	2918	Z	1016	2.63	0.30	2635	6.82	0.77	2888	7.48	0.85
	1998	Z	976	2.53	0.29	2827	7.32	0.83	2987	7.74	0.87
	2691	Z	986	2.55	0.29	2633	6.82	0.77	2939	7.61	0.86
LOWER CARD CAGE	3673	Z	1688	4.37	0.49	2016	5.22	0.59	4743	12.29	1.39
	3662	Z	1355	3.51	0.40	2797	7.24	0.82	5093	13.19	1.49
	3671	Z	2011	5.21	0.59	1876	4.86	0.55	5122	13.27	1.50
	3660	Z	2248	5.82	0.66	2732	7.08	0.80	5498	14.24	1.61
	3655	Z	825	2.14	0.24	2398	6.21	0.70	3523	9.12	1.03
	3650	Z	572	1.48	0.17	2378	6.16	0.70	4753	12.31	1.39
	1179	Z	731	1.89	0.21	2323	6.02	0.68	3776	9.78	1.11
	1084	Z	470	1.22	0.14	2349	6.09	0.69	5064	13.12	1.48
	3735	Z	1914	4.96	0.56	2974	7.70	0.87	4580	11.86	1.34
	3730	Z	1659	4.30	0.49	3129	8.10	0.92	6780	17.56	1.99
	3764	Z	1294	3.35	0.38	3044	7.88	0.89	5140	13.31	1.51

COMPONENT	GRID	LOAD DIR	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER CARD CAGE	3711	Z	929	2.41	0.27	4514	11.69	1.32	4624	11.98	1.35
	3708	Z	847	2.19	0.25	2633	6.82	0.77	5530	14.32	1.62
	3753	Z	1044	2.70	0.31	4267	11.05	1.25	5089	13.18	1.49
LOWER CARD CAGE CARDS	4845	Z	1922	4.98	0.56	2281	5.91	0.67	5911	15.31	1.73
	4822	Z	1555	4.03	0.46	2169	5.62	0.64	5626	14.57	1.65
	4826	Z	2087	5.41	0.61	2021	5.23	0.59	7617	19.73	2.23
	4813	Z	736	1.91	0.22	3514	9.10	1.03	7651	19.82	2.24
	4817	Z	544	1.41	0.16	2404	6.23	0.70	5485	14.21	1.61
	4777	Z	540	1.40	0.16	2412	6.25	0.71	5351	13.86	1.57
	4725	Z	1633	4.23	0.48	2325	6.02	0.68	5540	14.35	1.62
	4765	Z	2122	5.50	0.62	2315	6.00	0.68	5341	13.83	1.56
UPPER CARD CAGE	1469	Z	1219	3.16	0.36	1361	3.52	0.40	4205	10.89	1.23
	1509	Z	1096	2.84	0.32	1807	4.68	0.53	3386	8.77	0.99
	1514	Z	561	1.45	0.16	1789	4.63	0.52	7031	18.21	2.06
	1474	Z	604	1.56	0.18	1371	3.55	0.40	7033	18.22	2.06
	3372	Z	1798	4.66	0.53	4493	11.64	1.32	5129	13.28	1.50
	3381	Z	5068	13.13	1.48	4467	11.57	1.31	7021	18.18	2.06
	3404	Z	5261	13.63	1.54	6162	15.96	1.80	7145	18.51	2.09
	3395	Z	1875	4.86	0.55	6109	15.82	1.79	3715	9.62	1.09
	2249	Z	1643	4.26	0.48	3408	8.83	1.00	3156	8.17	0.92
	2448	Z	1136	2.94	0.33	3399	8.80	1.00	4576	11.85	1.34
	3369	Z	4571	11.84	1.34	4628	11.99	1.36	4337	11.23	1.27
UPPER CARD CAGE CARDS	11836	Z	1866	4.83	0.55	6154	15.94	1.80	10653	27.59	3.12
	11251	Z	5044	13.07	1.48	5188	13.44	1.52	7711	19.97	2.26
	11829	Z	1260	3.26	0.37	4062	10.52	1.19	10908	28.25	3.19
	11203	Z	568	1.47	0.17	1709	4.43	0.50	7928	20.54	2.32
	11879	Z	2071	5.36	0.61	3994	10.35	1.17	8019	20.77	2.35
	11312	Z	2034	5.27	0.60	4120	10.67	1.21	7556	19.57	2.21

COMPONENT	GRID	LOAD	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
LOWER WARMLOAD	1945	Z	4350	11.27	1.27	2103	5.45	0.62	3097	8.02	0.91
	1929	Z	3608	9.34	1.06	2072	5.37	0.61	3015	7.81	0.88
	2812	Z	2407	6.23	0.70	1411	3.66	0.41	2652	6.87	0.78
	2810	Z	917	2.37	0.27	1374	3.56	0.40	2798	7.25	0.82
	1892	Z	1517	3.93	0.44	1937	5.02	0.57	3567	9.24	1.04
	1908	Z	1936	5.02	0.57	1976	5.12	0.58	3263	8.45	0.96
	2814	Z	6130	15.88	1.80	1473	3.81	0.43	3193	8.27	0.94
	2815	Z	2856	7.40	0.84	1468	3.80	0.43	3768	9.76	1.10
	3006	Z	1195	3.10	0.35	3687	9.55	1.08	4244	10.99	1.24
	3036	Z	3920	10.15	1.15	6292	16.30	1.84	5599	14.50	1.64
UPPER WARMLOAD	2009	Z	955	2.47	0.28	2549	6.60	0.75	2959	7.66	0.87
	2033	Z	1587	4.11	0.46	2537	6.57	0.74	2814	7.29	0.82
	2080	Z	1905	4.93	0.56	2418	6.26	0.71	2907	7.53	0.85
	2056	Z	3615	9.36	1.06	2460	6.37	0.72	2774	7.19	0.81
	2939	Z	1674	4.34	0.49	2248	5.82	0.66	2952	7.65	0.86
	2941	Z	1581	4.10	0.46	2383	6.17	0.70	4103	10.63	1.20
	2738	Z	922	2.39	0.27	2278	5.90	0.67	2893	7.49	0.85
	2739	Z	1334	3.45	0.39	2282	5.91	0.67	2780	7.20	0.81
	2976	Z	1801	4.66	0.53	7101	18.39	2.08	3932	10.18	1.15
	2945	Z	1023	2.65	0.30	3351	8.68	0.98	4466	11.57	1.31
LOWER REFLECTOR	4662	Z	4576	11.85	1.34	1665	4.31	0.49	3001	7.77	0.88
	4085	Z	1241	3.21	0.36	1560	4.04	0.46	2828	7.33	0.83
	4087	Z	1522	3.94	0.45	2107	5.46	0.62	3049	7.90	0.89
	4176	Z	2317	6.00	0.68	5854	15.16	1.71	4398	11.39	1.29
	4096	Z	5120	13.26	1.50	7699	19.94	2.26	5768	14.94	1.69
POWER CNTRL/MON BKT	5591	Z	5231	13.55	1.53	4447	11.52	1.30	3016	7.81	0.88
	5600	Z	7412	19.20	2.17	3712	9.61	1.09	3022	7.83	0.89
	5590	Z	3124	8.09	0.92	2935	7.60	0.86	3054	7.91	0.89
	11803	Z	5649	14.63	1.65	29332	75.97	8.59	7219	18.70	2.11

COMPONENT	GRID	LOAD	X-RESPONSE			Y-RESPONSE			Z-RESPONSE		
			RMS	GRMS	Q	RMS	GRMS	Q	RMS	GRMS	Q
POWER CNTRL/MON PWB	5878	Z	3771	9.77	1.10	6855	17.76	2.01	4727	12.24	1.38
	6204	Z	5190	13.44	1.52	4027	10.43	1.18	4366	11.31	1.28
PWR CNTRL/MON STANDOFF	6247	Z	1496	3.87	0.44	7429	19.24	2.18	4342	11.25	1.27
	6015	Z	1556	4.03	0.46	3427	8.88	1.00	4614	11.95	1.35
	5932	Z	1559	4.04	0.46	7616	19.73	2.23	4521	11.71	1.32
	5999	Z	3960	10.26	1.16	4811	12.46	1.41	4592	11.89	1.35
RADIATOR PANEL	10963	Z	3570	9.25	1.05	1528	3.96	0.45	3791	9.82	1.11
	10968	Z	449	1.16	0.13	1529	3.96	0.45	3389	8.78	0.99
	10974	Z	3113	8.06	0.91	964	2.50	0.28	3402	8.81	1.00
	10969	Z	4784	12.39	1.40	948	2.46	0.28	3333	8.63	0.98
	10917	Z	2363	6.12	0.69	996	2.58	0.29	3423	8.87	1.00
SIDEMOUNT	8313	Z	0	0.00	0.00	0	0.00	0.00	3414	8.84	1.00
	8229	Z	271	0.70	0.08	304	0.79	0.09	3349	8.67	0.98
	7796	Z	76	0.20	0.02	183	0.47	0.05	3388	8.77	0.99
	7840	Z	0	0.00	0.00	0	0.00	0.00	3414	8.84	1.00
	8178	Z	3870	10.02	1.13	2577	6.67	0.75	2011	5.21	0.59
	9338	Z	978	2.53	0.29	2077	5.38	0.61	4061	10.52	1.19
	9576	Z	0	0.00	0.00	0	0.00	0.00	3414	8.84	1.00
	7508	Z	102	0.26	0.03	993	2.57	0.29	3404	8.82	1.00
UPPER REFLECTOR	7240	Z	0	0.00	0.00	0	0.00	0.00	3414	8.84	1.00
	9642	Z	720	1.86	0.21	354	0.92	0.10	3391	8.78	0.99
	4629	Z	3348	8.67	0.98	3251	8.42	0.95	2851	7.39	0.84
	4399	Z	1299	3.36	0.38	3351	8.68	0.98	2805	7.27	0.82
	4401	Z	1792	4.64	0.53	4053	10.50	1.19	3258	8.44	0.95
	4490	Z	2539	6.58	0.74	6234	16.15	1.83	5539	14.35	1.62
	4410	Z	5342	13.84	1.56	6921	17.93	2.03	6221	16.11	1.82

METSAT RANDOM VIBRATION DISPLACEMENTS - RATTLESPACE AT THE LOWER CARD CAGE CIRCUIT CARDS												
COMPONENT	GRID	GRID	ΔX-DEFLECTION			ΔX-DEFLECTION			ΔX-DEFLECTION			
			X-LOAD			Y-LOAD			Z-LOAD			
			GAP	1Σ ΔX	3Σ ΔX	GAP	1Σ ΔX	3Σ ΔX	GAP	1Σ ΔX	3Σ ΔX	
LOWER CARD CAGE CIRCUIT CARDS	4808	4848	0.037	0.00027	0.00080	0.037	0.00173	0.00519				
	4809	4849		0.00031	0.00094		0.00082	0.00246				
	4810	4850		0.00014	0.00043		0.00032	0.00097				
	4811	4851		0.00008	0.00025		0.00028	0.00085				
	4812	4852		0.00007	0.00020		0.00022	0.00067				
	4803	4843		0.00034	0.00101		0.00176	0.00528				
	4804	4844		0.00034	0.00102		0.00080	0.00239				
	4805	4845		0.00015	0.00046		0.00042	0.00126				
	4806	4846		0.00009	0.00026		0.00030	0.00091				
	4807	4847		0.00006	0.00019		0.00022	0.00065				
	4828	4868		0.00117	0.00350		0.00056	0.00169				
	4829	4869		0.00829	0.02488		0.00244	0.00733				
	4830	4870		0.00303	0.00908		0.00103	0.00310				
	4831	4871		0.00443	0.01329		0.00192	0.00575				
	4832	4872		0.00284	0.00852		0.00088	0.00263				
	4823	4863		0.00282	0.00846		0.00098	0.00294				
	4824	4864		0.00720	0.02161		0.00197	0.00590				
4825	4865		0.00134	0.00402		0.00088	0.00265					
4826	4866		0.00161	0.00483		0.00110	0.00331					
4827	4867		0.00178	0.00533		0.00096	0.00287					

METSAT RANDOM VIBRATION DISPLACEMENTS - RATTLESPACE AT THE UPPER CARD CAGE CIRCUIT CARDS													
COMPONENT	GRID	GRID	ΔX-DEFLECTION			ΔX-DEFLECTION			ΔX-DEFLECTION			ΔX-DEFLECTION	
			GAP	1Σ ΔX	3Σ ΔX	GAP	1Σ ΔX	3Σ ΔX	GAP	1Σ ΔX	3Σ ΔX	Z-LOAD	3Σ ΔX
UPPER CARD CAGE CIRCUIT CARDS	11300	11867	0.137	0.00060	0.00180	0.137	0.00018	0.00053	0.137	0.00035	0.00104		
	11301	11868		0.00044	0.00131		0.00012	0.00036		0.00023	0.00068		
	11302	11869		0.00028	0.00083		0.00011	0.00034		0.00016	0.00049		
	11303	11870		0.00015	0.00046		0.00012	0.00035		0.00016	0.00049		
	11304	11871		0.00012	0.00037		0.00010	0.00029		0.00017	0.00052		
	11305	11872		0.00015	0.00044		0.00009	0.00026		0.00016	0.00049		
	11306	11873		0.00012	0.00036		0.00008	0.00024		0.00012	0.00037		
	11307	11874		0.00007	0.00020		0.00008	0.00023		0.00009	0.00027		
	11308	11875		0.00078	0.00235		0.00014	0.00041		0.00037	0.00111		
	11309	11876		0.00060	0.00179		0.00012	0.00037		0.00026	0.00077		
	11310	11877		0.00042	0.00125		0.00014	0.00043		0.00020	0.00060		
	11311	11878		0.00025	0.00075		0.00014	0.00041		0.00018	0.00055		
	11312	11879		0.00015	0.00046		0.00012	0.00036		0.00018	0.00055		
	11313	11880		0.00014	0.00042		0.00012	0.00035		0.00018	0.00053		
	11314	11881		0.00012	0.00036		0.00011	0.00032		0.00014	0.00042		
	11315	11882		0.00010	0.00031		0.00011	0.00034		0.00010	0.00030		
	11316	11883		0.00068	0.00205		0.00020	0.00059		0.00038	0.00113		
	11317	11884		0.00054	0.00161		0.00019	0.00057		0.00029	0.00087		
	11318	11885		0.00040	0.00119		0.00019	0.00056		0.00023	0.00068		
	11319	11886		0.00026	0.00077		0.00015	0.00045		0.00018	0.00053		
	11320	11887		0.00018	0.00053		0.00013	0.00038		0.00017	0.00052		
	11321	11888		0.00016	0.00047		0.00014	0.00042		0.00017	0.00052		
	11322	11889		0.00013	0.00038		0.00012	0.00036		0.00014	0.00043		
	11323	11890		0.00007	0.00021		0.00007	0.00022		0.00008	0.00025		

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7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES) Aerojet 1100 W. Hollyvale Azusa, CA 91702			8. PERFORMING ORGANIZATION REPORT NUMBER CDRL 311B 10805A July 1996	
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